

ME18503 DESIGN OF MACHINE ELEMENTS

OBJECTIVES

- This course will impart the knowledge on various types of stress- selection of materials
- This course will make acquainted design principles on shaft, fits and tolerances.
- This course will familiarize the design principles of springs under dynamic and static conditions.
- This course will enable to check strength of fasteners kind of both rivet and welding.
- This course will facilitate to select and examine the rolling and sliding contact bearings.

OUTCOMES

1. Able to depict the design process, material selection, calculation of stresses under static and variable loading conditions with the effect of stress concentrations.
2. Adept the design of solid, hollow shafts keys and couplings. Also having knowledge of fits and tolerance.
3. Examining the close coil helical springs under variable loading . Acquiring the knowledge of leaf, disc and torsion springs.
4. Proficient in Design of riveted joint and welding joints under eccentric loading.
5. Have design knowledge on sliding and rolling contact bearing. EHD Journal Bearing

TEXT BOOKS:

1. Bhandari,V.B ,”Design of Machine Element”, 3rd edition, TMH Publications.New Delhi,2017.
2. Khurmi R.S., and Gupta J.K., “*Machine Design*”, 14 th Edition,S Chand&Co NewDelhi,2005.
3. Sundararamoorthy, T.V. and Shanmugam, N., “*Machine Design*”, Anuradha Agencies, Chennai,2003.

REFERENCES :

1. Bhandari V.B., “*Design of Machine Elements*”, 4th edition TATA McGraw Hill New delhi,2017.
2. Khurmi R.S., and Gupta J.K., “*Machine Design*”, 14th Edition, S Chand and Co NewDelhi,2005.
3. Lingaiah K., “*Machine Design Data Book*”, 2nd Edition Tata McGraw – Hill.New delhi,2010.
4. Robert.L.Norton,” *Machine Design*”, 5th Edition, Pearson Publisher, New delhi,2018.
5. Sharma,P.C and Agarwal,D.K,“*MachineDesign*”,Agrawal-Kataria and Sons Publication, NewDelhi,2014.
6. Shigley, J.E., Charles, R.M. and Richard, G.B., “*Mechanical Engineering Design*”, 10th ed., McGraw-Hill, New Delhi,2014.
7. Spotts M.F., “*Design of Machine Elements*”, 8th Edition,Pearson Education,NewDelhi, 2019.

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GENERAL DISCUSSION

PRE REQUISITE

Knowledge on

1. Strength of Material
2. Theory of Machines
3. Materials science/Metallurgy

SOME Questions for you

STRESS?

$$\sigma = \text{load/Area } \text{N/mm}^2$$

STRAIN?

$$\varepsilon = \text{change in dimension/Original dimension}$$

no units

TYPES of STRESSES

Tensile stress , compressive stress ,shear stress

Principle stress

Stress acting on principle plane

Principle plane

Plane in which shear is zero

Types of Loads

1. Point load
2. UD load
3. UV load

Others

- 1. Dead load**
- 2. Transverse load**
- 3. Axial load**
- 4. Tangential load**



Name the load

Point load --- transverse load



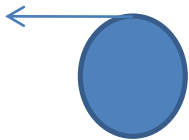
Name the load

Axial load



Name the load

Tangential load



Last session 22- 06- 2020 -discussed

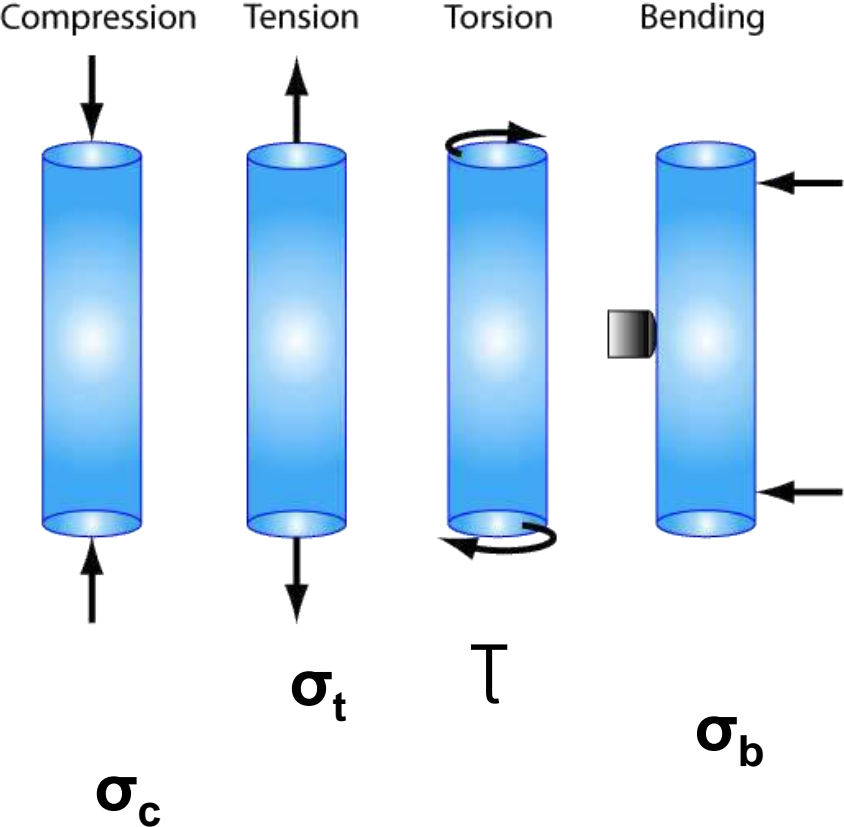
Objectives

Outcomes

types of simple stresses

Types of loads

Diagrams for different type of simple stresses



What is design?

Creating/innovate of an idea

Modify the existing

With the application of scientific principles and mathematical approaches

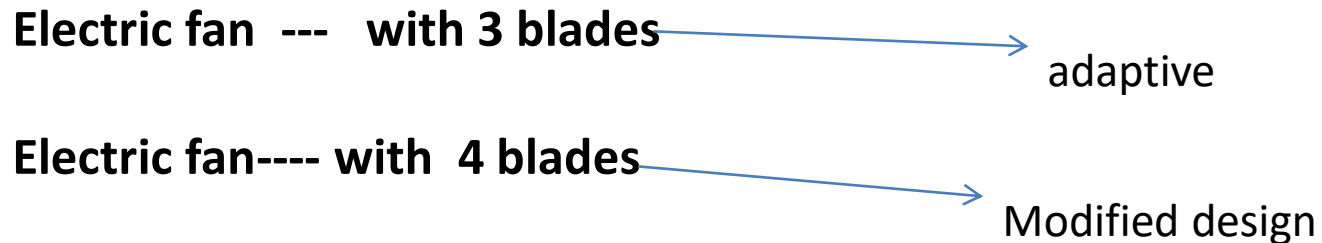
Types of Design

1. Adaptive design

2. Modifying design

3. New design--/ -- Innovation

Examples to understand the types of Design



Discussion –forum for some examples towards the types of design

Assignment to find the other types of design

**Industrial design,
Experimental design,
Aesthetic,
Ergonomics design**

Syllabus

UNIT I

BASICS OF DESIGN PRINCIPLES

UNIT II

DESIGN SHAFT, KEYS AND FITS AND TOLERANCE AND COUPLINGS

UNIT III

DESIGN FOR SPRINGS

UNIT IV

DESIGN FOR RIVETED AND WELDING JOINTS, FASTNERS

UNIT V

DESIGN OF BEARINGS

UNIT I **BASICS OF DESIGN PRINCIPLES**

Design Process-

Types of Stress,

Cyclic stresses ,

Factor of Safety,

Stress concentration factor in tension, bending and torsion,

Theories of failures.

Notch sensitivity,

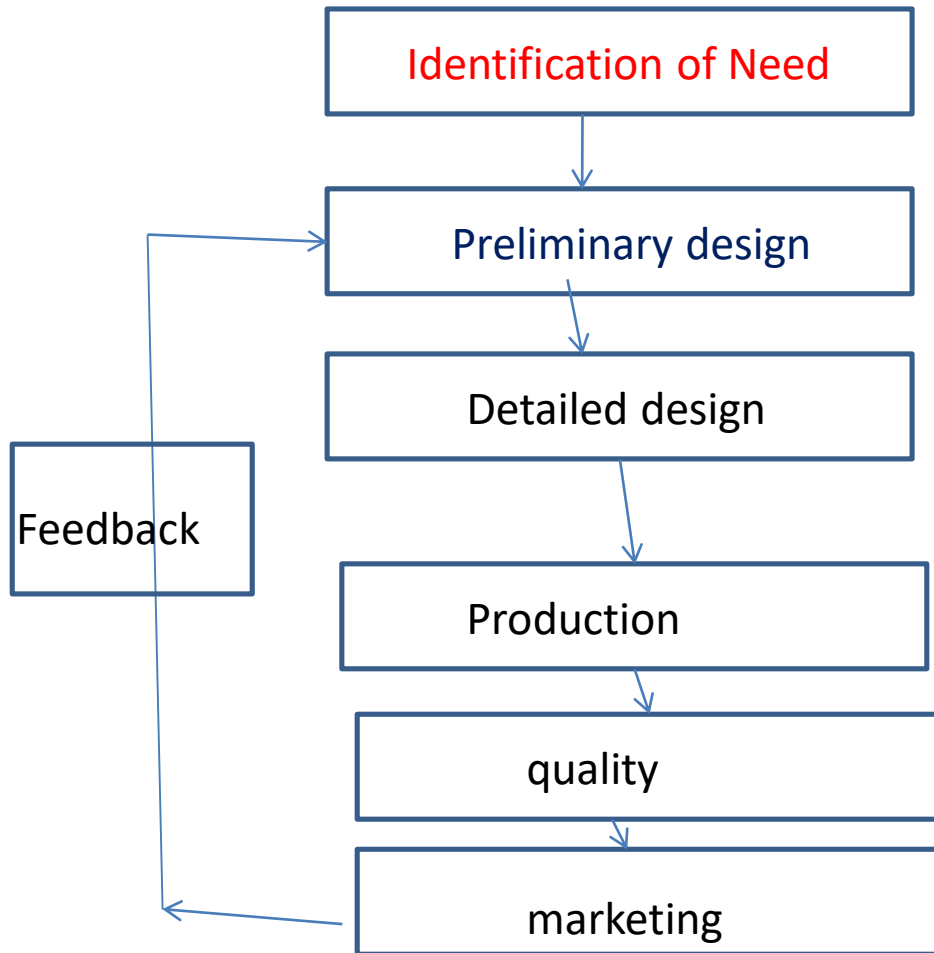
Design for variable and repeated loadings,

Fatigue stress concentration factor,

Endurance diagrams,

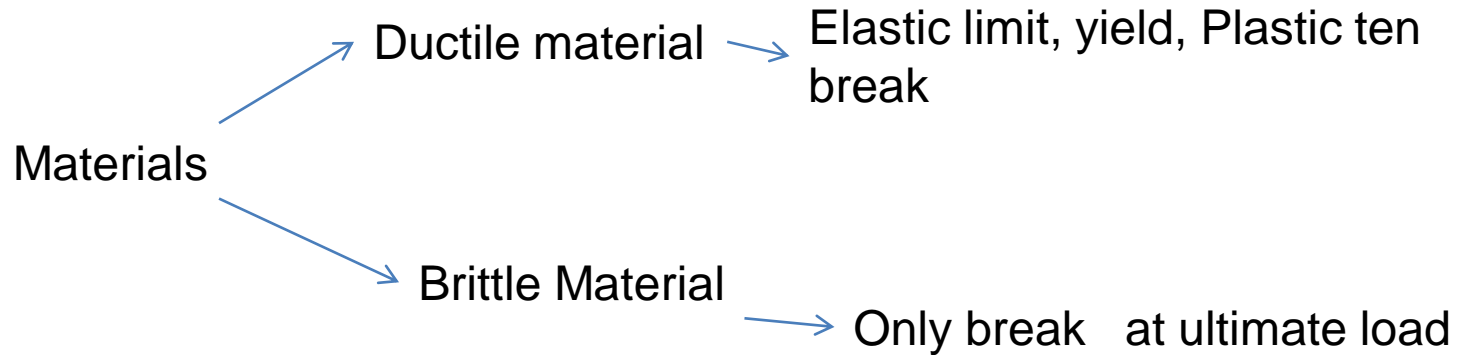
Introduction to fracture mechanics

What is Design process?



It is also known as design cycle

Mechanical properties



Learn some mechanical properties

strength

hardness

Plasticity

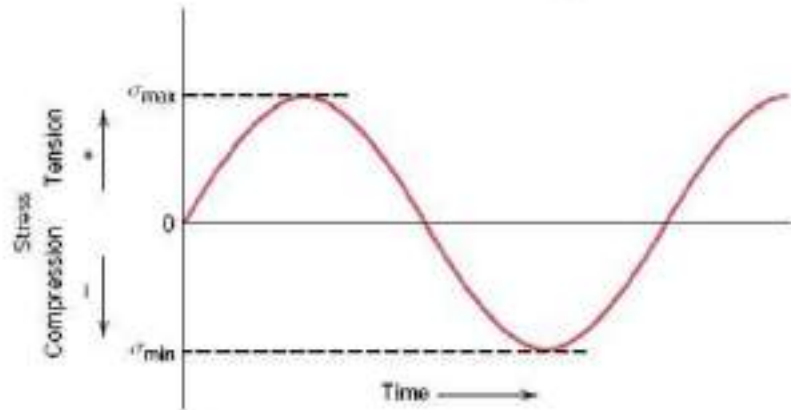
Elasticity

Resilience

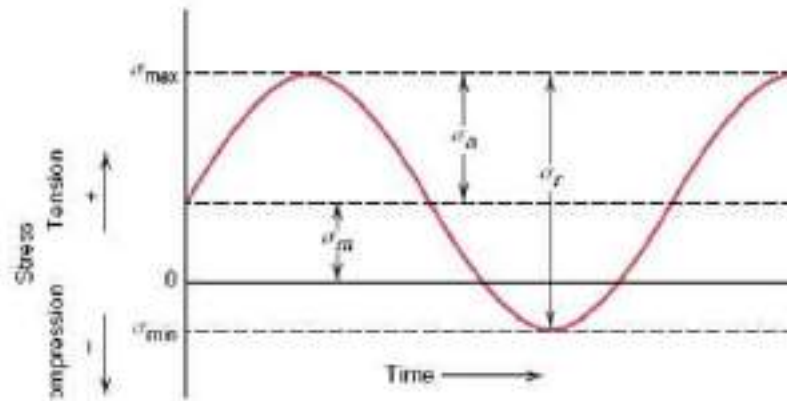
Toughness

Creep

Cyclic stress



Reversed stress cycle

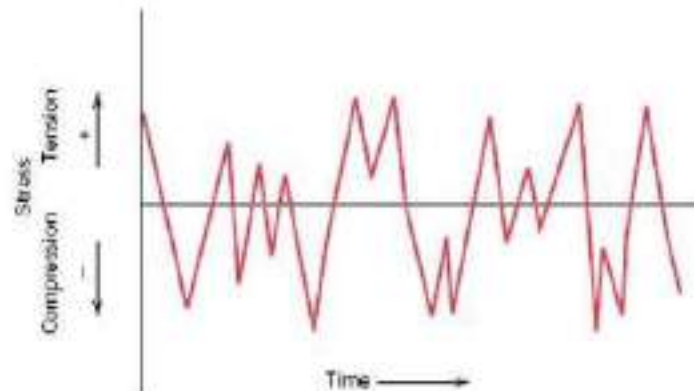


Repeated stress cycle

$$\sigma_m = \frac{\sigma_{max} + \sigma_{min}}{2}$$

$$\sigma_r = \sigma_{max} - \sigma_{min}$$

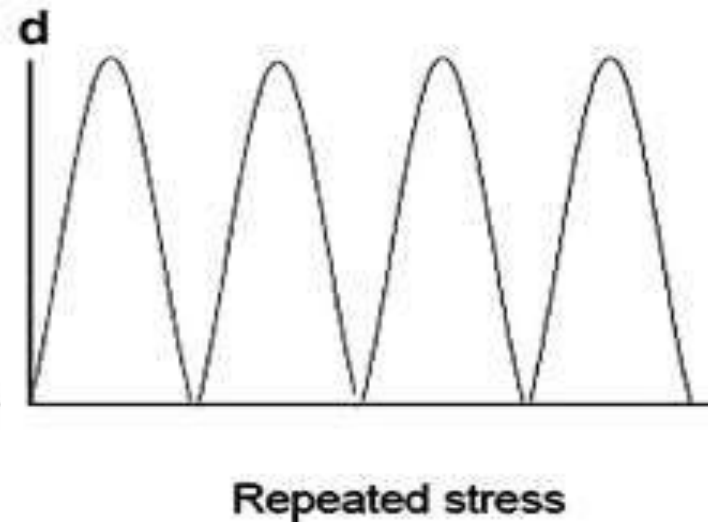
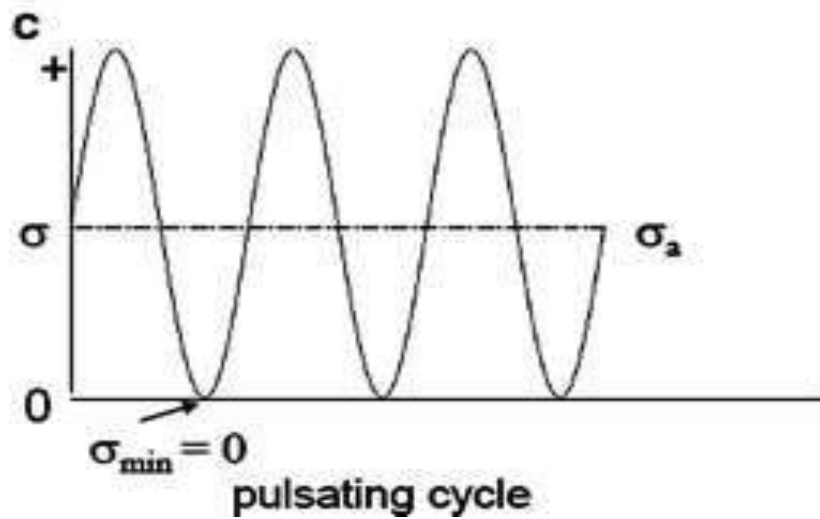
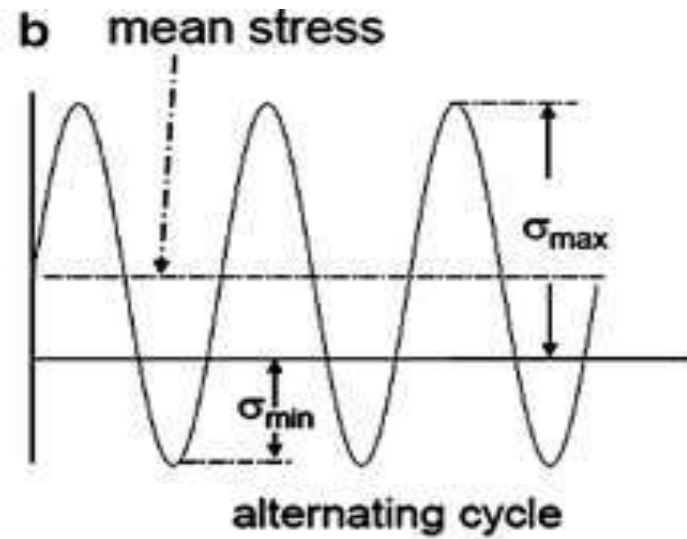
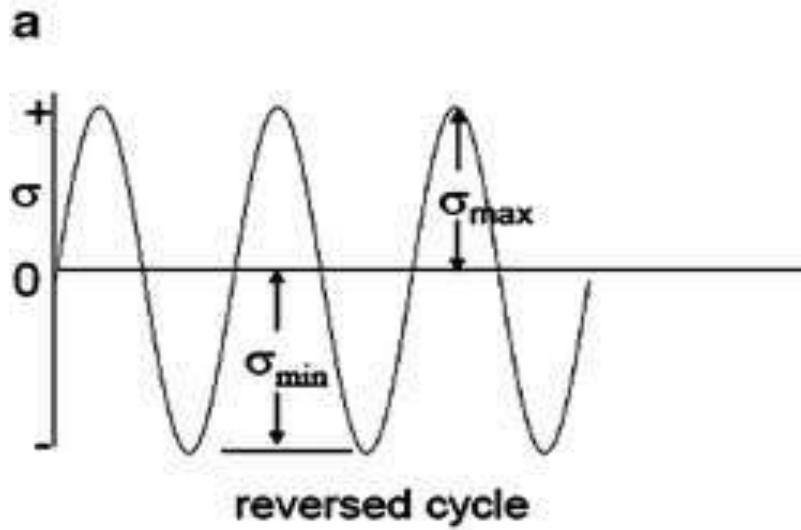
$$\sigma_a = \frac{\sigma_r}{2} = \frac{\sigma_{max} - \sigma_{min}}{2}$$



Random stress cycle



Cyclic stresses



Last session topics

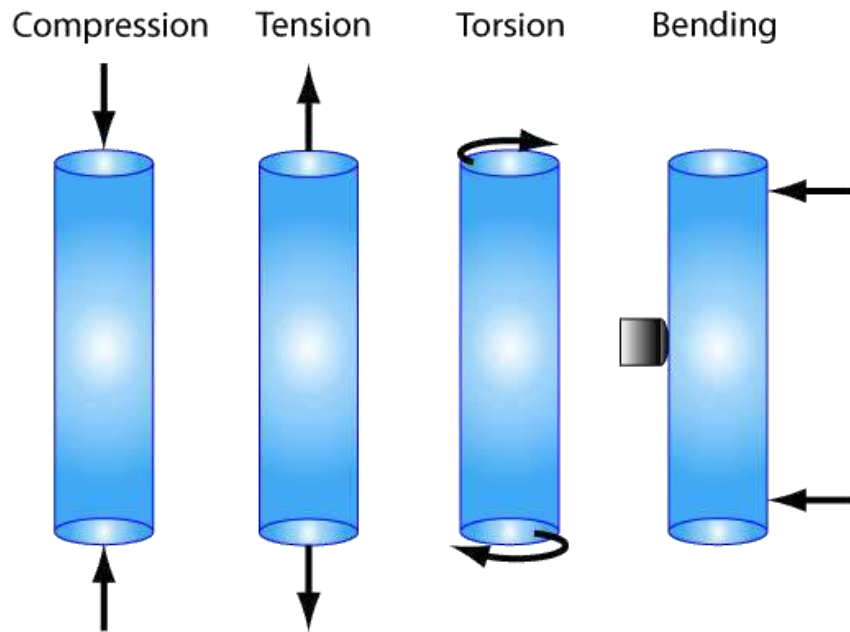
**Types of simple stresses,
cyclic stresses**

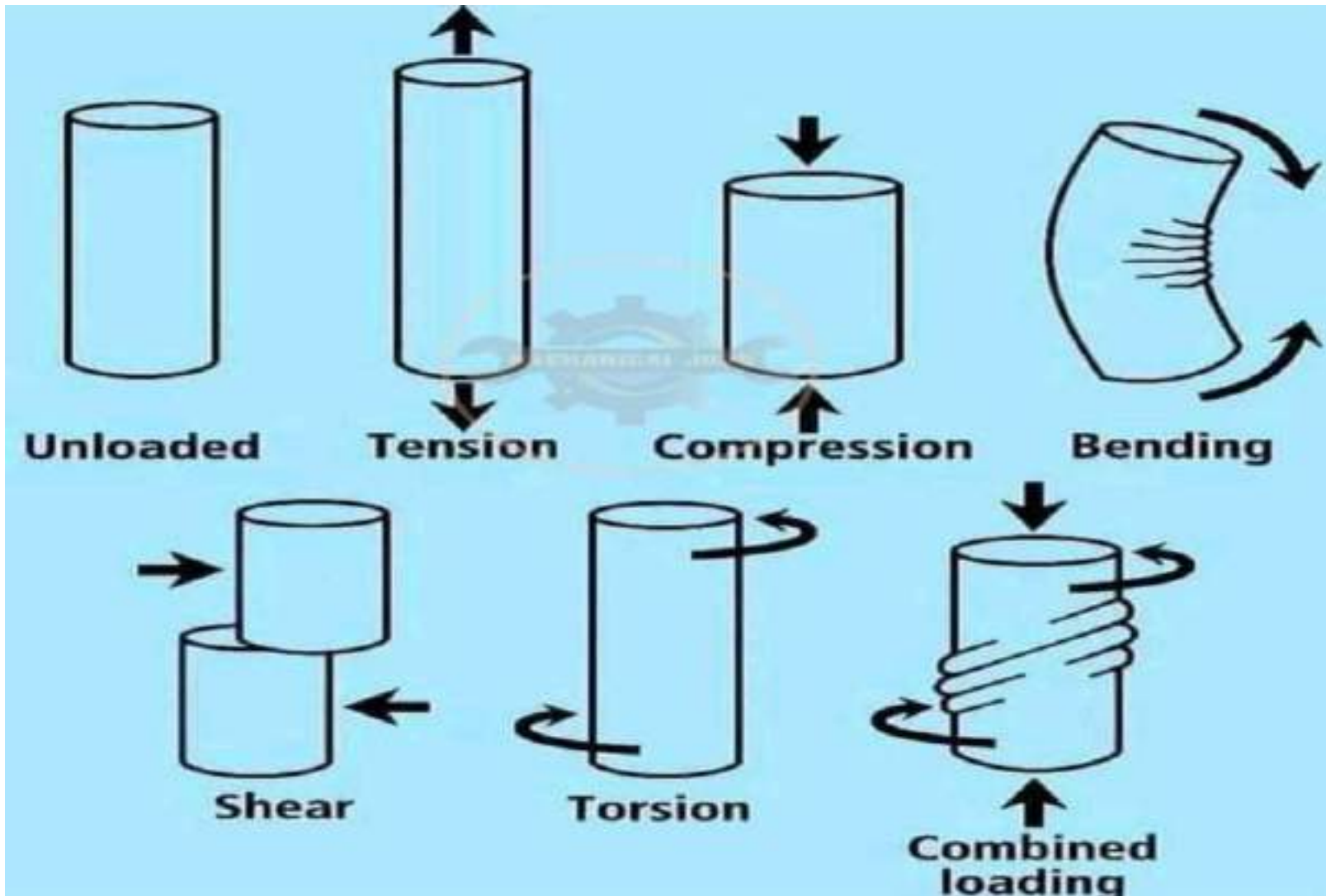
Design processes/cycle

Types of design – adaptive, modified , & new design and others

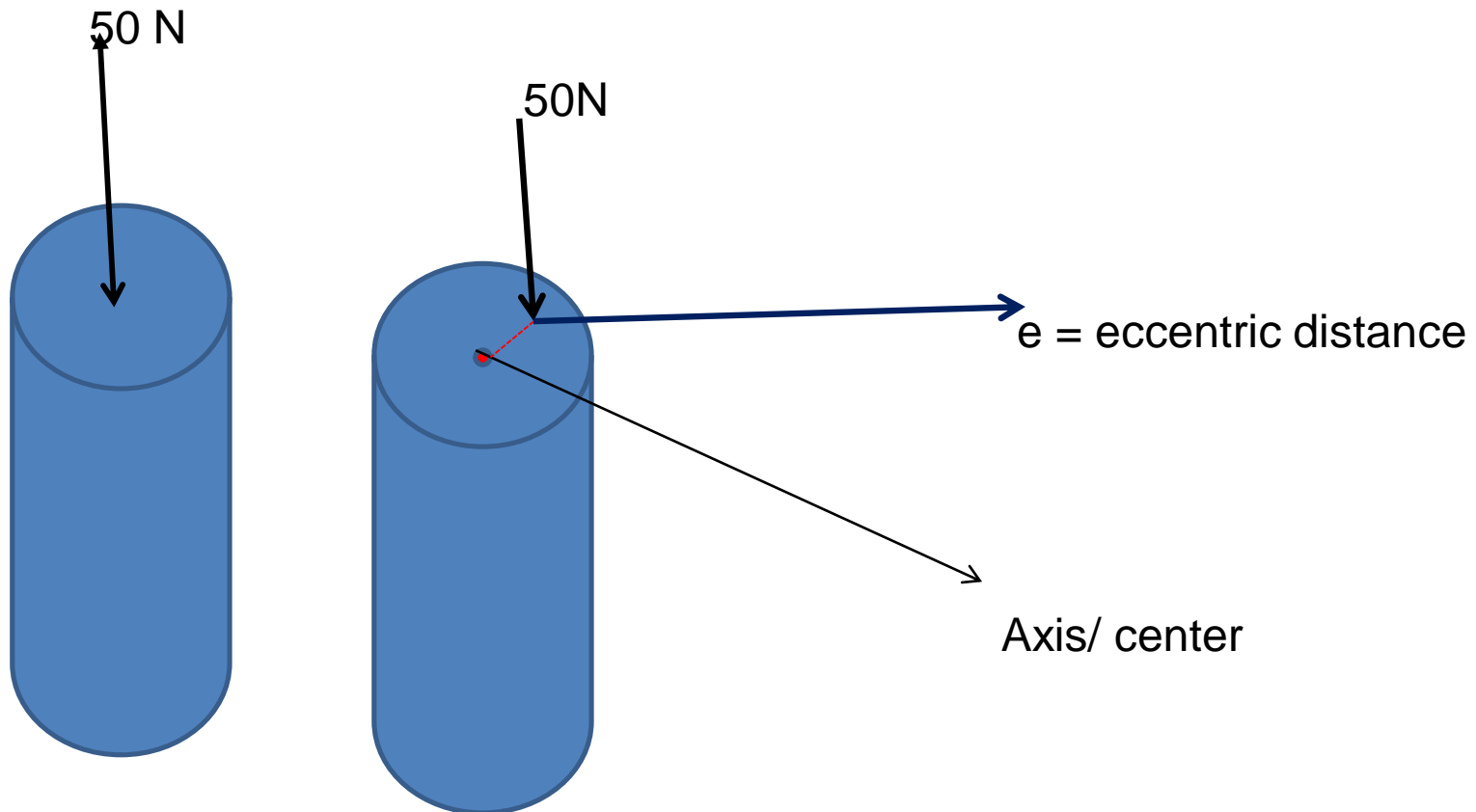
Some important mechanical properties

Failures



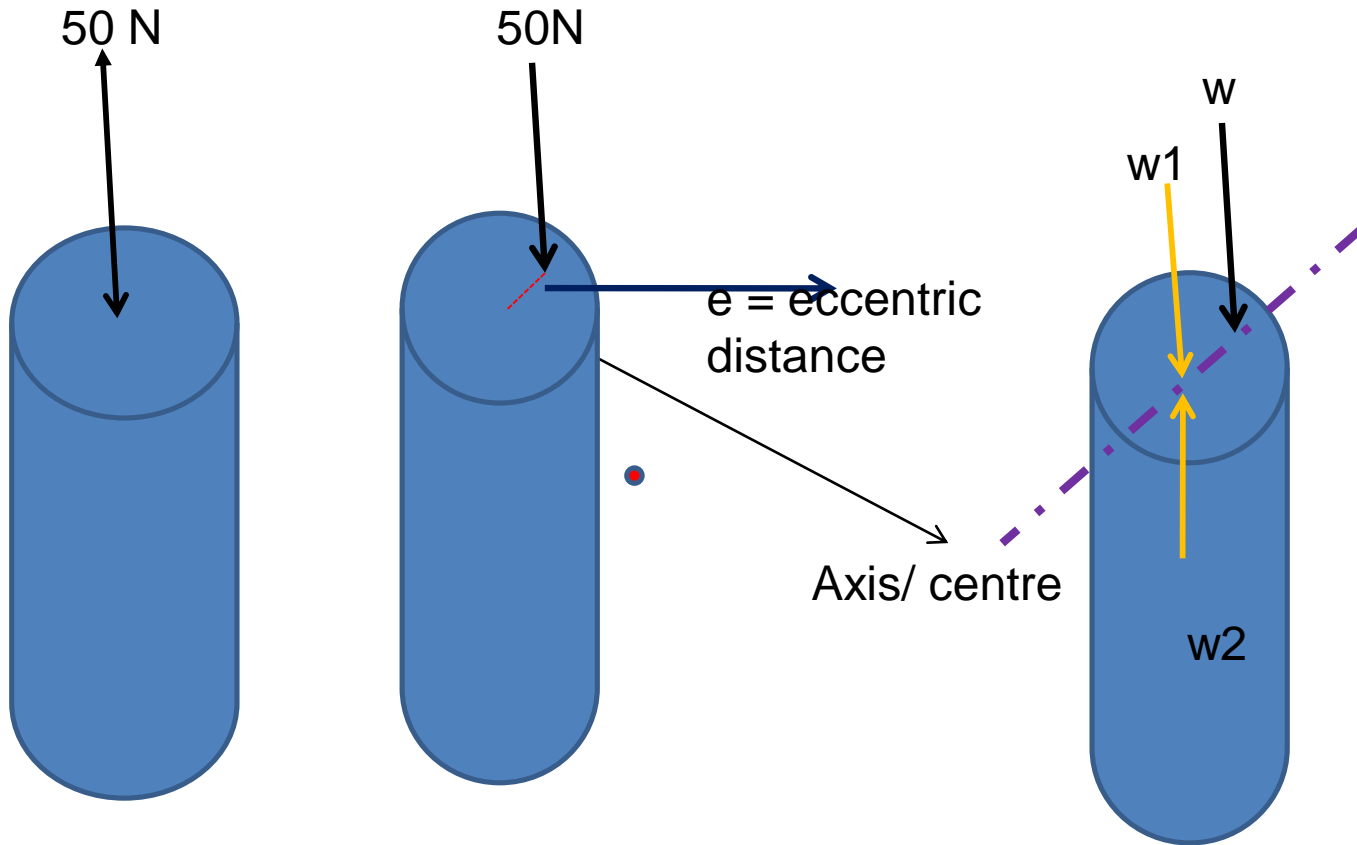


Eccentric load concept



What is the stress?
What is the load?

Eccentric load concept



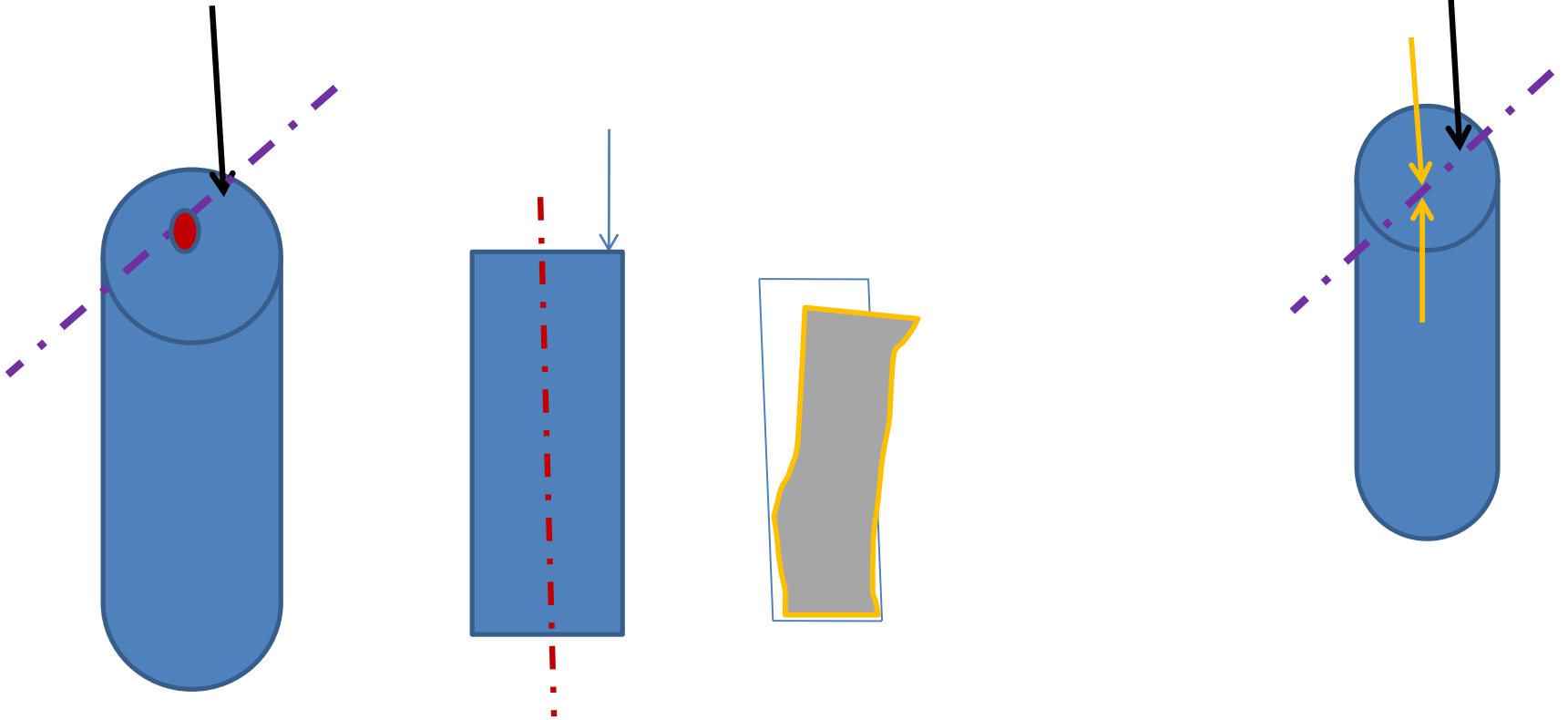
What is the stress?--
compressive or direct stress

What is the load?--- Axial load

W -- original load

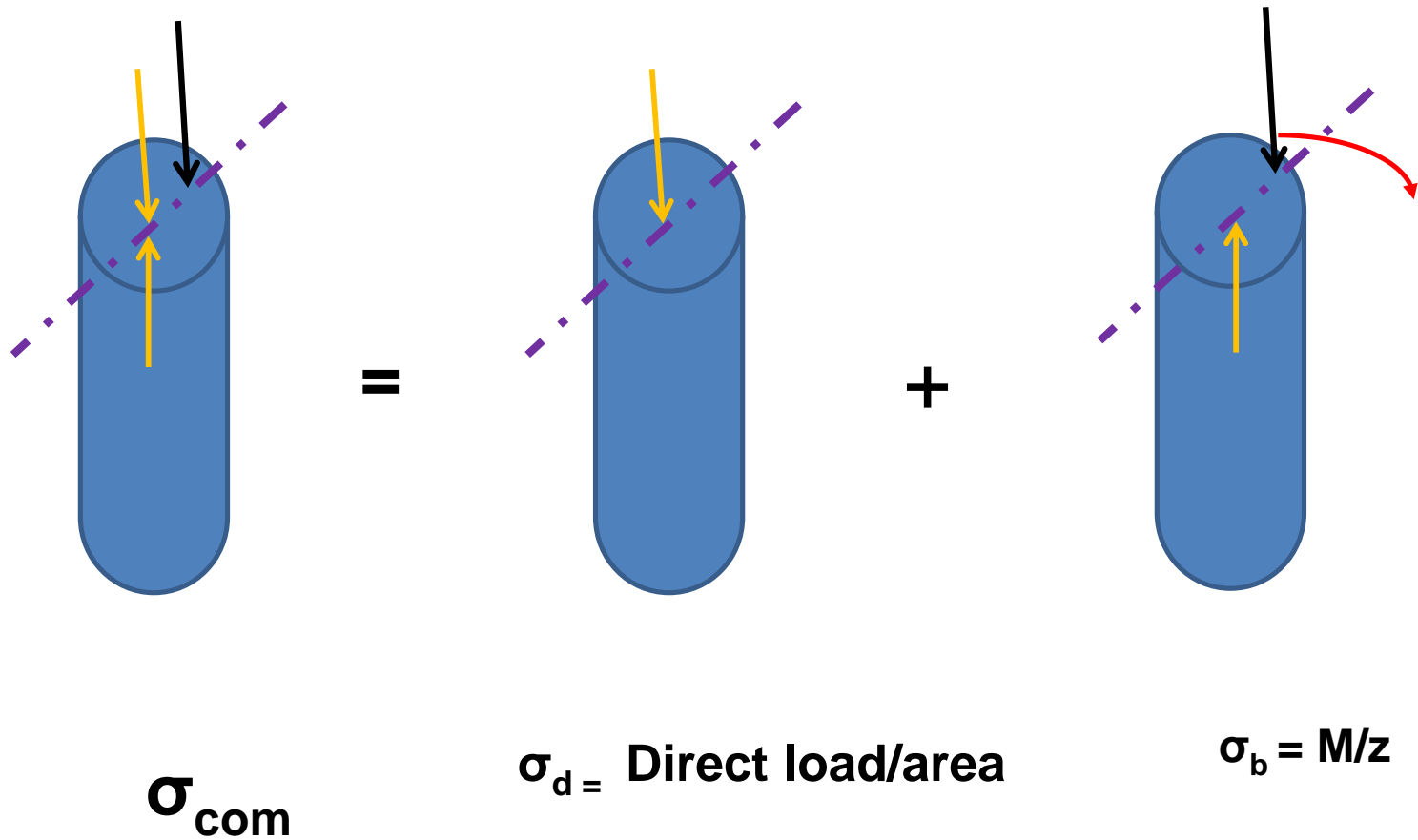
$W1$ - introduced

$W2$ introduced opposite



How the member will fail---???
???? due to which stress

How to get resolution for the effect



Solving the problem

Step1: Introduce imaginary loads equal and opposite at axis(equal to external load value)

Step2: Prepare the equivalent diagram , to have combined stress

Step3: Calculate Direct stress (by axial load)

$$\sigma_d = \text{load/area}$$

Step4: calculate bending Stress (By combination of Original load and introduced load)

$$\sigma_b = M/z$$

where M = moment due to external load= $w \times e$

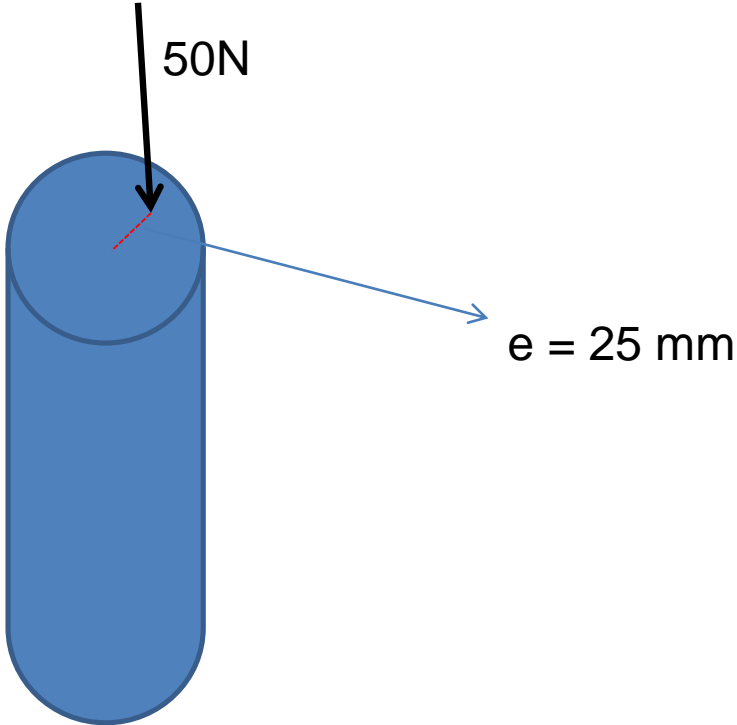
z = section modulus

Step5: Calculate combined stress ‘ σ_{com} ‘

$$\sigma_{com} = \sigma_d \pm \sigma_b$$

Problem :

A circular member of diameter 40 mm is subject to a load of 50 N, eccentrically 25 mm from the axis as shown in figure. Determine the stresses induced in the member.



Data				
diameter	40	mm		
Load	50	N		
e	25	mm		
area				
Pi	3.14			
A	1256	sq.mm		
1	Direct stress			
Stress	0.04	N/sq.mm		
2	Bending			
$\sigma_b =$	M/Z			
M	Load x e			
	1250	N-mm		
z =	I/y			
	$I =$	$\pi * d^4 / 64$		
	$Y =$	$d/2$		

$$I = 125640 \text{ mm}^4$$

$$y = 20 \text{ mm}$$

therefore $z = 6282 \text{ mm}^3$

Now $\sigma_b = 0.198981 \text{ N/sq.mm}$

3 Combined stress

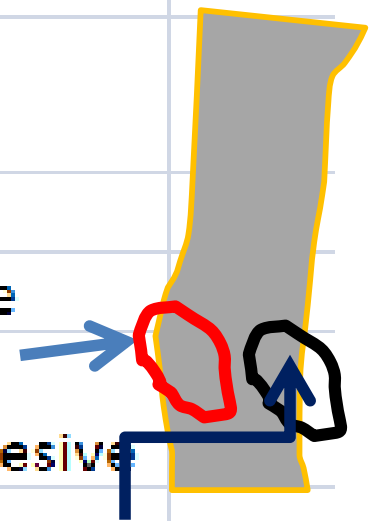
$$\sigma_{com} = \sigma_d + \sigma_{max}$$

or

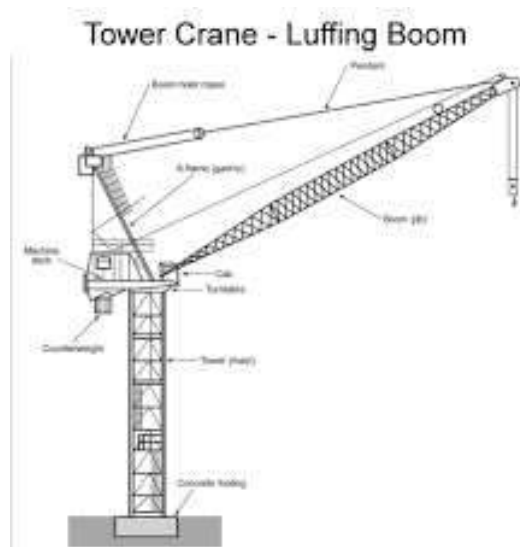
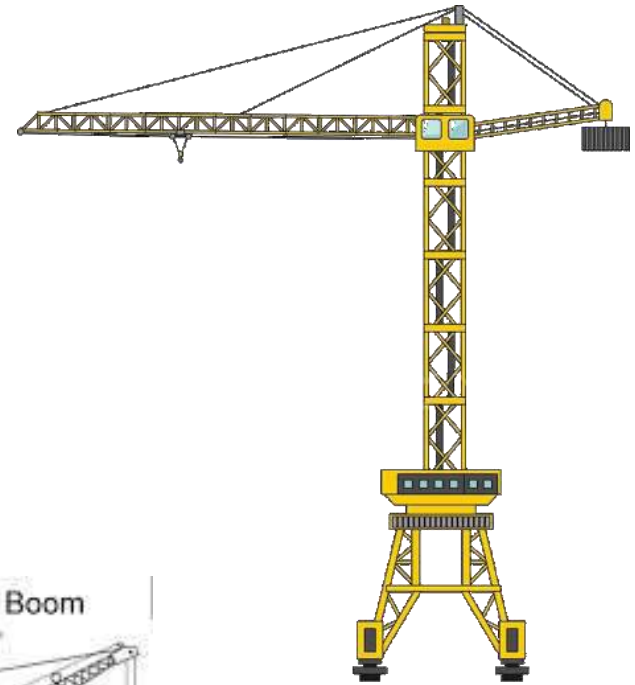
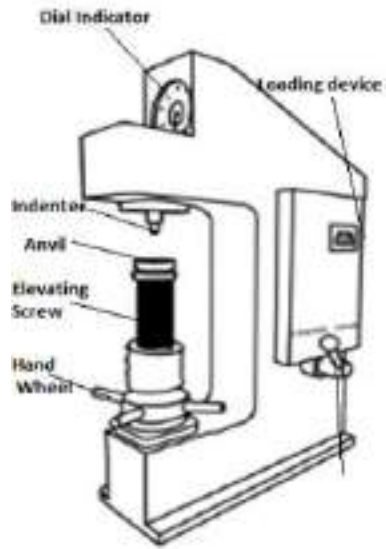
$$\sigma_d - \sigma_{min}$$

0.24 Tensile

-0.16 compressive



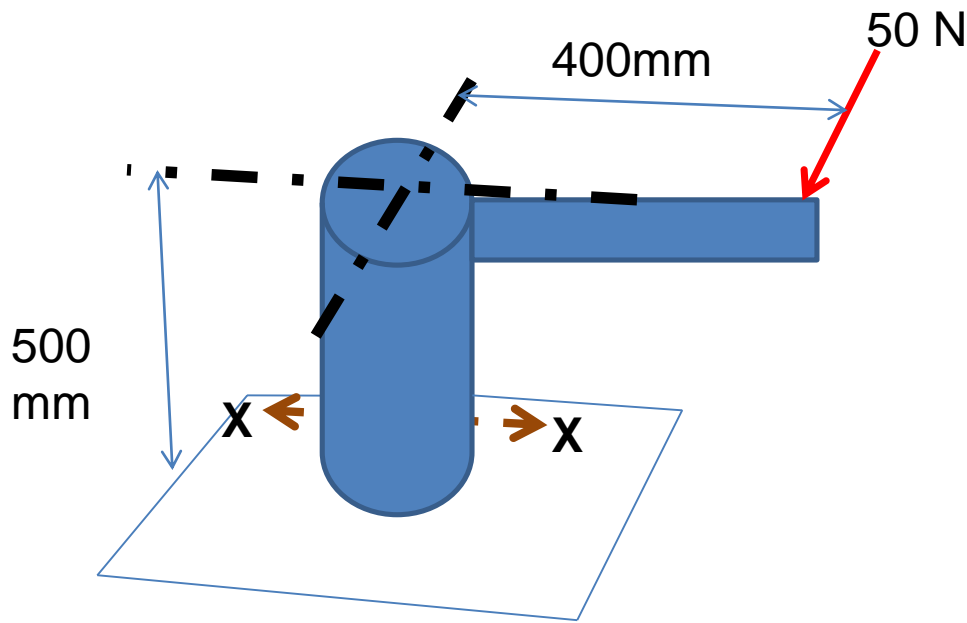
Some applications with the eccentric loading concepts

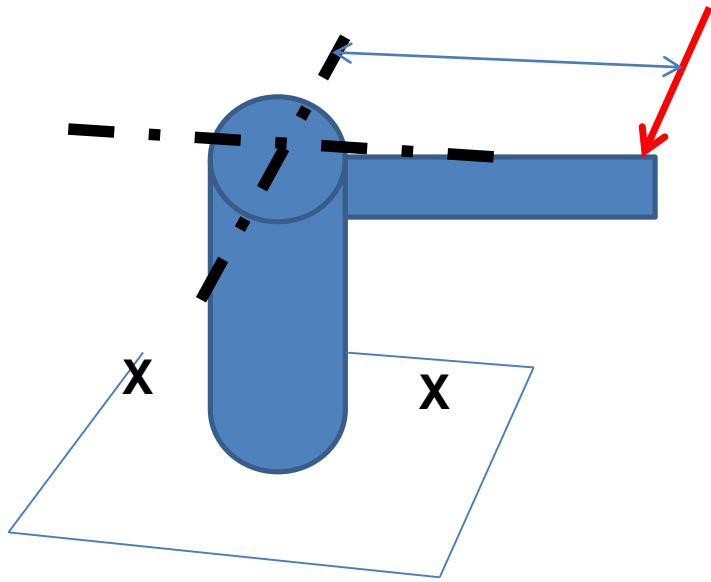


Lifting and Rigging

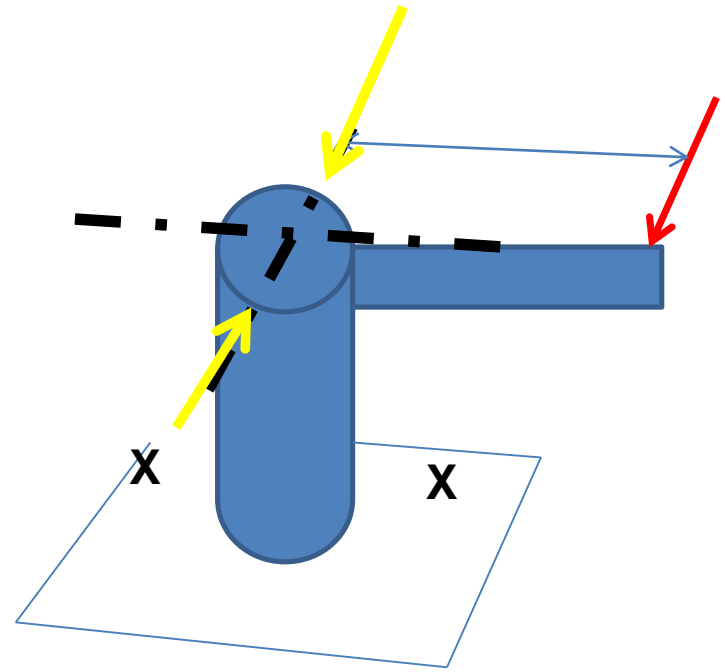
Problem 2

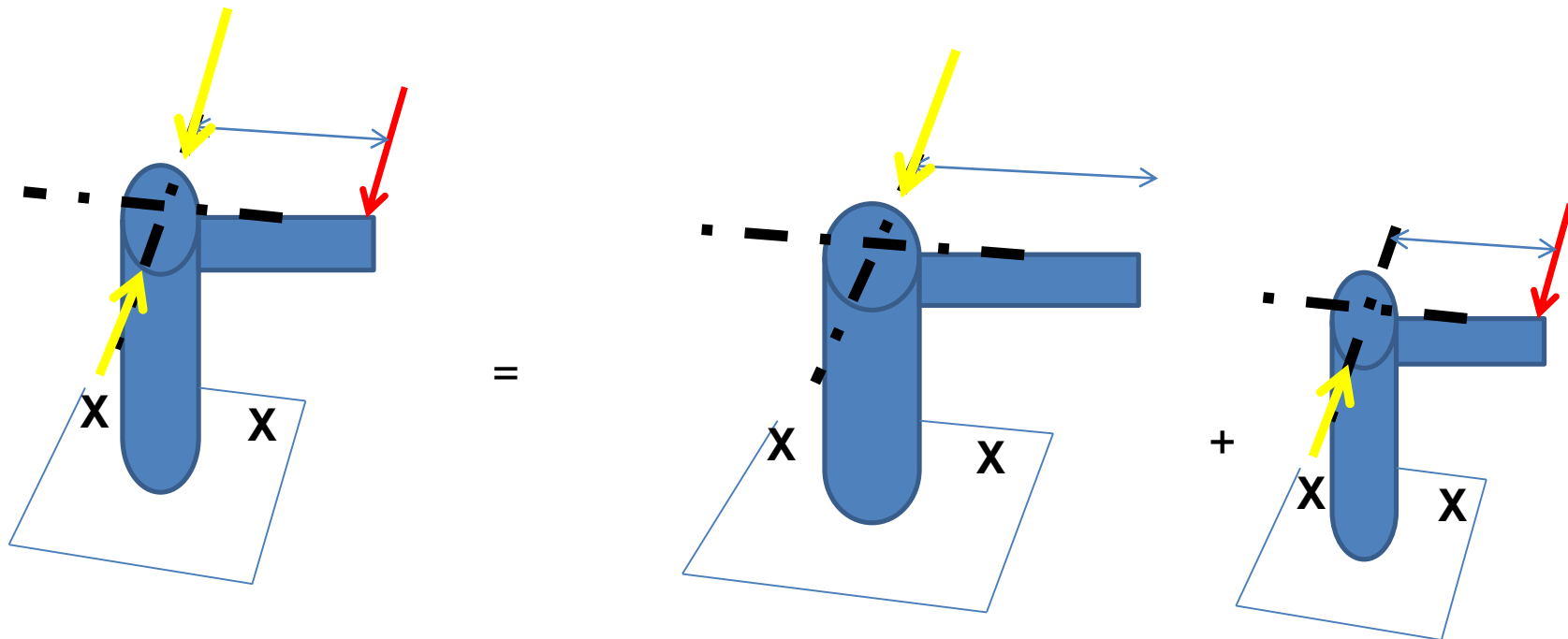
Determine the max and min stresses induced at Section XX, for the member shown in the figure.. Take the diameter of the member is 50 mm

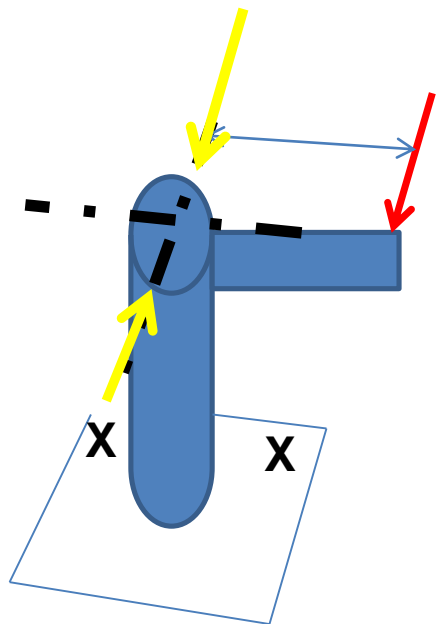




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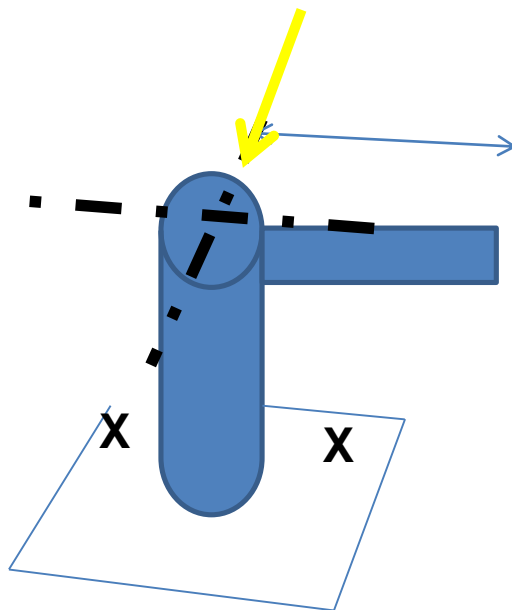




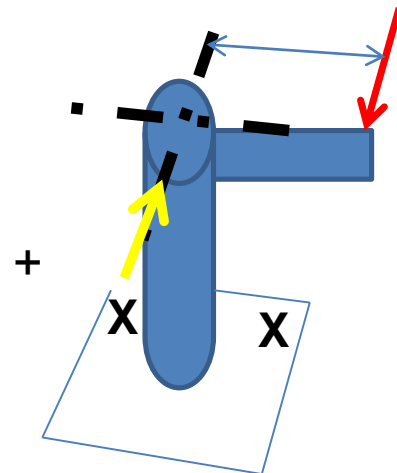


Combined stress

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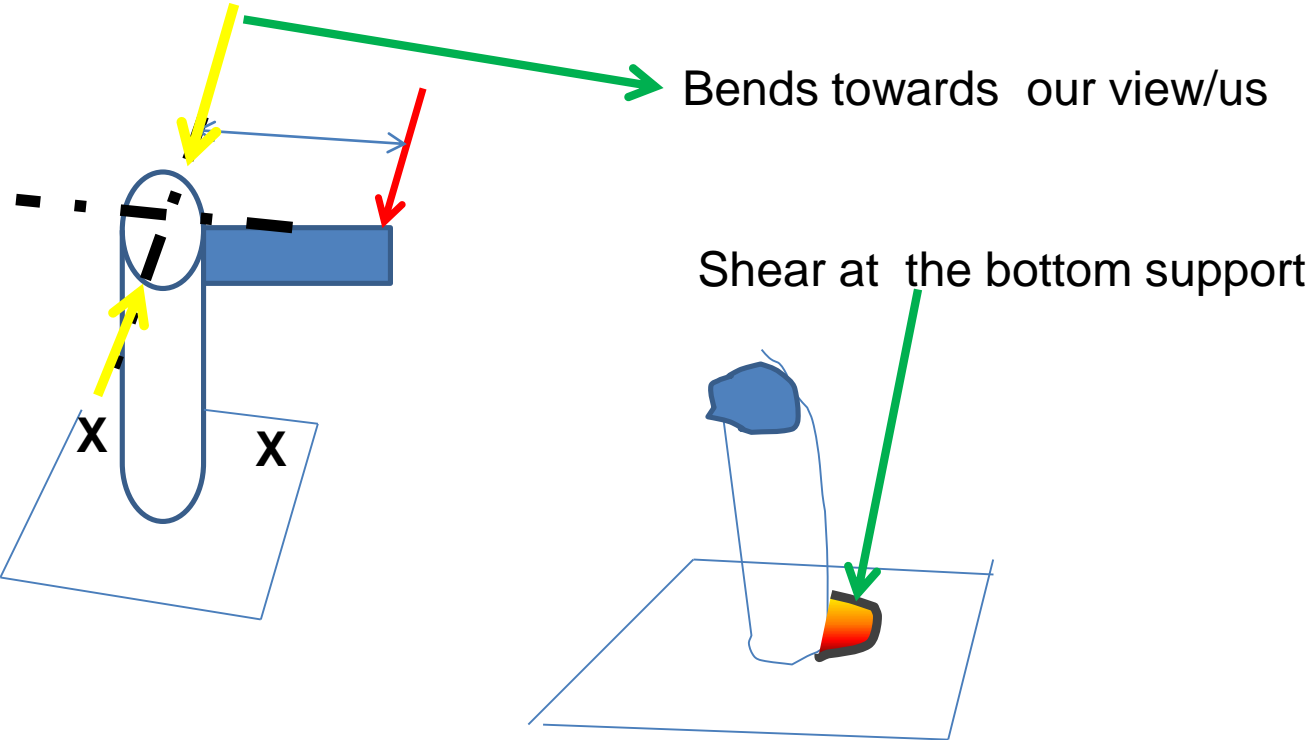


Bending stress



Shear stress

Prediction of failure of the member



Solving the problem

Step1: Introduce imaginary loads equal and opposite at axis(equal to external load value)

Step2: Prepare the equivalent diagram , to have combined stress

Step3: Calculate Direct stress (by axial load)

$$\sigma_d = \text{load/area}$$

Step4: calculate bending Stress (By combination of Original load and introduced load)

$$\sigma_b = M/z$$

where M = moment due to external load= $w \times e$

z = section modulus

Step 5 Calculate Shear stress, $T = \pi/16 \times \tau \times d^3$

Step5: Calculate combined stress ‘ σ_{com} ‘

$$\sigma_{com} = \sigma_d \pm \sigma_b \quad / \quad = \tau \pm \sigma_b$$

Data									
diameter	50 mm								
Load	50 N								
eh	400 mm	(Horizontal to original load)							
ev	500 mm	(Vertical to introduced load from the base)							
area									
Pi	3.141								

		1 Direct stress	
A		Stress	0 N/sq.mm

1 Direct stress	
Stress	0 N/sq.mm

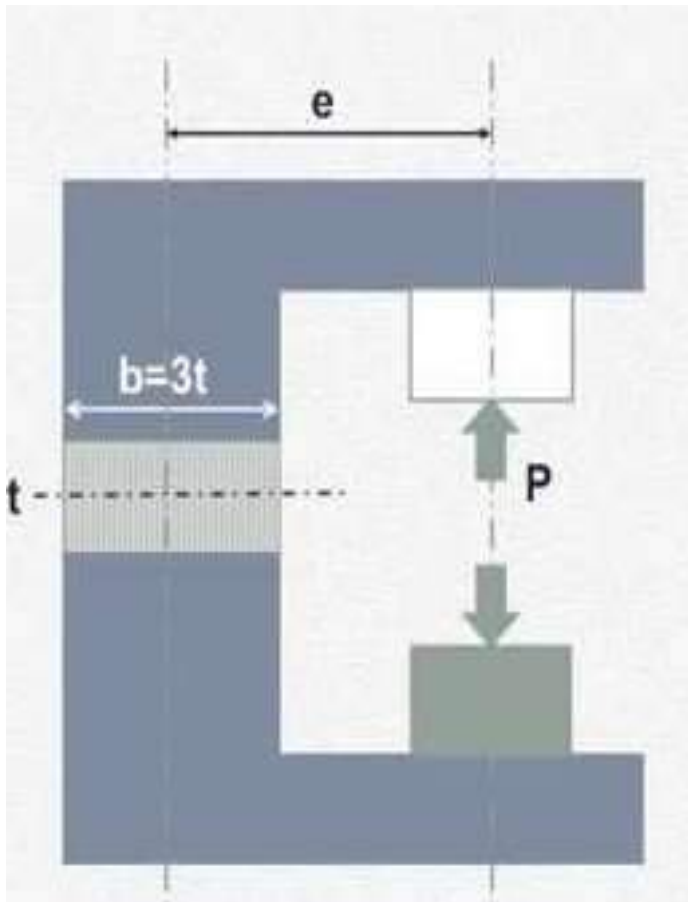
2 Bending					
$\sigma_b =$	M/Z				
M	Load x e				
		25000 N-mm			
z =	I/y				
	$I =$	$\pi * d^4 / 64$			
	$Y =$	$d/2$			
	$I =$	306738 mm ⁴			
	$y =$	25 mm			
therefore	z =	12269.5 mm ³			

Now	$\sigma_b =$	2.03757 N/sq.mm		
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3	Shear stress	$T \times 16 / (\pi \times d^3)$	refer Pg.No 7.1/DDB
	T=	20000 N mm	
		(eccentric distance is vertical distance from base)	
	Shear stress	0.81503 N/sq.mm	

4	Combined stress		
	$\sigma_{com} =$	shear stress + σ_b max	2.85 Tensile
		or	
		σ_b -shear stress min	1.22 Tensile

Problem for solving



Step 1

$$\sigma_d = \frac{P}{A} = \frac{P}{3t^2}$$

Step 2

$$\sigma_b = \frac{M}{Z} = \frac{Pe}{\frac{3}{2}t^3}$$

Step 3:

$$\therefore \sigma_R = \sigma_d + \sigma_b$$

$$\sigma_R = \frac{P}{3t^2} + \frac{Pe}{\frac{3}{2}t^3}$$

$$\therefore \sigma_R = \frac{P}{3t^2} + \frac{2Pe}{3t^3}$$

$$\sigma_R = \frac{Pt + 2Pe}{3t^3}$$

$$\therefore 3t^3 \times \sigma_R = Pt + 2Pe$$

Factor of safety

$$n = \frac{\textit{allowable stress}}{\textit{working stress}}$$

Allowable stress = design stress

Design stress \longrightarrow **$[\sigma]$**

Soft material $[\sigma_y]$, $y = \text{yield}$

Brittle material $[\sigma_u]$, $u = \text{ultimate}$

Directional assignment of stresses under combined load

$$\sigma_d = \sigma_x \quad \text{Direct/ axial load}$$

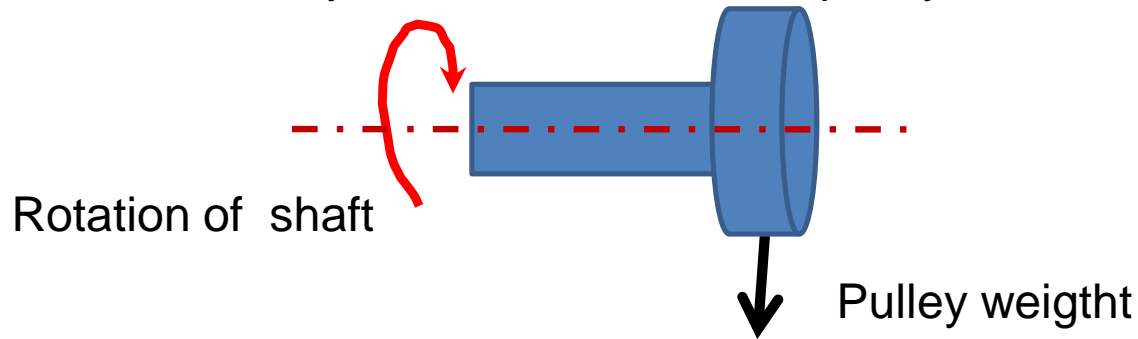
$$\sigma_b = \sigma_y \quad \text{Transverse load}$$

$$\tau_{xy} = \tau_{yx} = \tau \quad \text{Shear stress/load}$$

The directional assignments of stresses will be useful to find the **‘principal stresses’**

- ❑ Principal stresses are nothing but stresses acting on principal plane
- ❑ Quite applicable for finding the stress intensities in the form of
- ❑ **‘ σ_1 ’** maximum and **‘ σ_2 ’** minimum
- ❑ When the members subjected to combined stresses and simple shear stress

Example : belt drive system – shafts carries pulley-



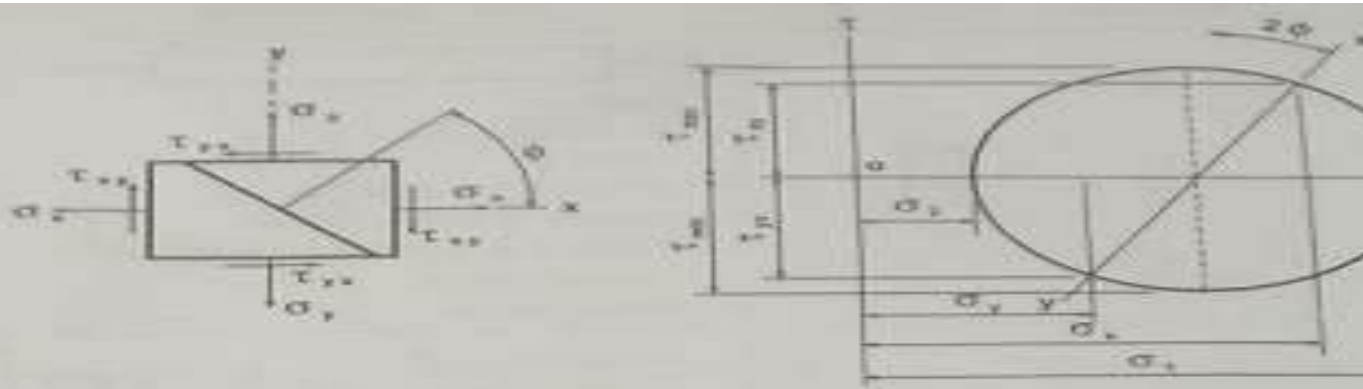
In general, Under strained material all the stress will be mutually perpendicular in three planes xy, xz, yz .

Out of the three **one is max** and the **other is minimum**

Now the combined stress problem is advancing with approach of principal stresses

Now refer the data book page : 7.2 two dimensional stresses

σ_1, σ_2 and τ



Two Dimensional Stresses	
$\sigma_{1,2} = \frac{1}{2} \left[(\sigma_x + \sigma_y) \pm \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2} \right]$	$\sigma_1, \sigma_2, \sigma_3$ Pri Sty
$\tau = \pm \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2}$	$\sigma_x - \sigma_y - \sigma_z$ Str Co $\tau_{xy}, \tau_{yz}, \tau_{zx}$ Sh co σ_y Y σ P
Three Dimensional Stresses	
$\begin{vmatrix} (\sigma_x - \sigma) & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & (\sigma_y - \sigma) & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & (\sigma_z - \sigma) \end{vmatrix} = 0$	

The roots of this cubic equation in σ are the principal stresses

Now The objectives of solving problems

$$n = \frac{\textit{allowable stress}}{\textit{working stress}}$$

Finding dimension of the member

Checking design for safety

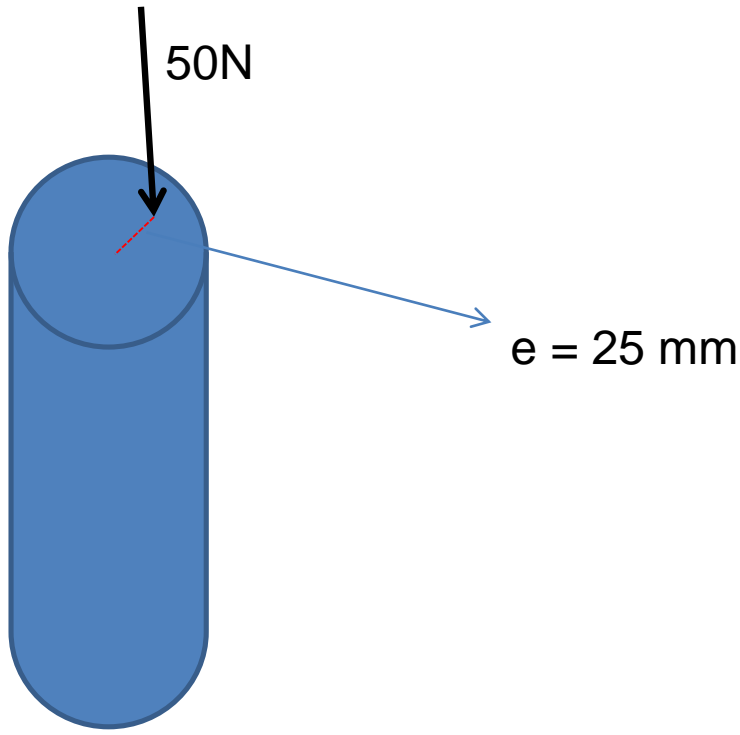
Induced stress < Design Stress

$$\sigma_i < [\sigma],$$

$$\tau < [\tau]$$

Problem 3

A circular member of diameter 40 mm is subject to a load of 50 N, eccentrically 25 mm from the axis as shown in figure. Determine the stresses induced in the member. If the allowable stress for the member material is 2 N/mm^2 Also determine the factor of safety



Solving the problem

Step1: Introduce imaginary loads equal and opposite at axis(equal to external load value)

Step2: Prepare the equivalent diagram , to have combined stress

Step3: Calculate Direct stress (by axial load)

$$\sigma_d = \text{load/area}$$

Step4: calculate bending Stress (By combination of Original load and introduced load)

$$\sigma_b = M/z$$

where $M = \text{moment due to external load} = w \times e$

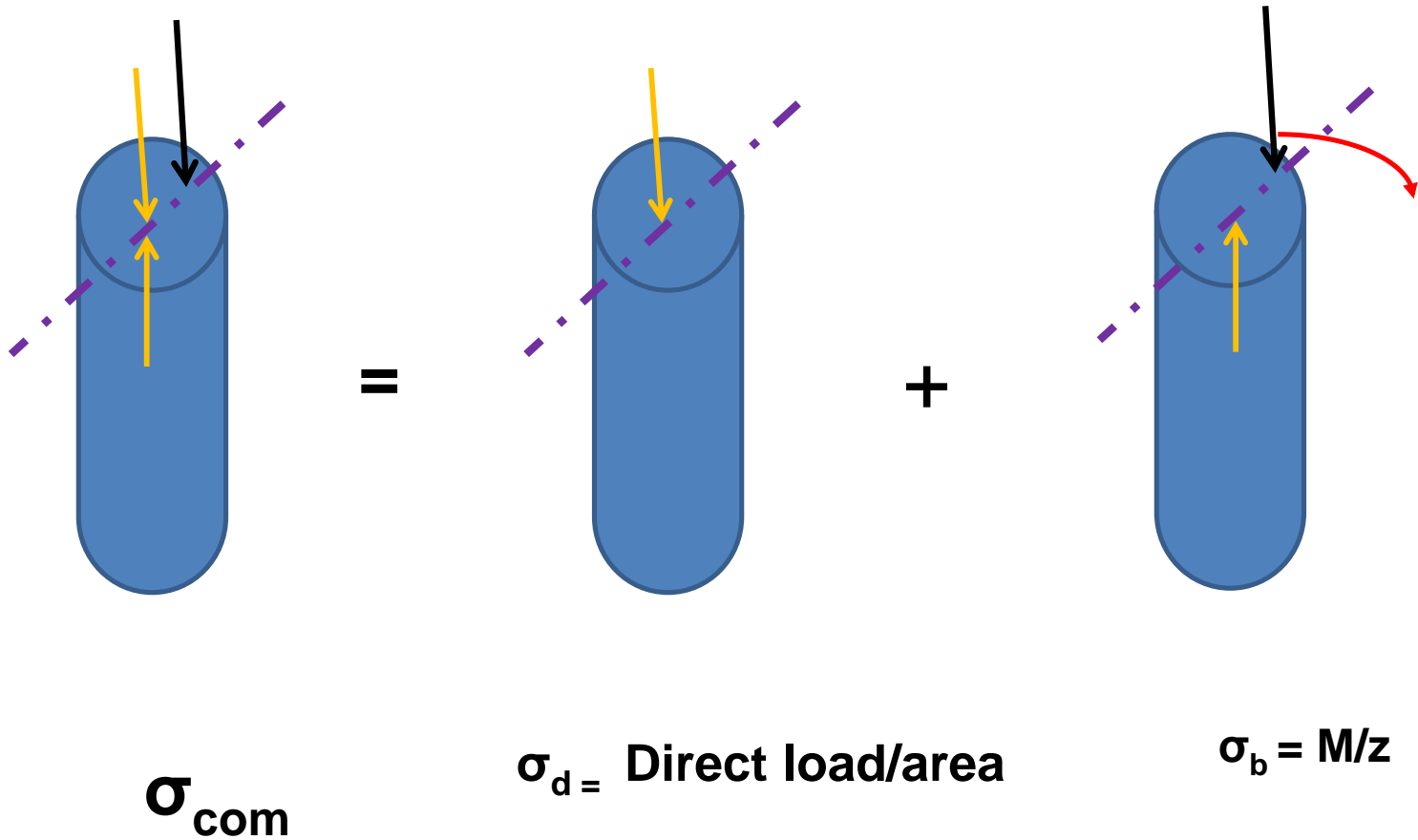
$z = \text{section modulus}$

Step 5 Calculate Shear stress, $T = \pi/16 \times \tau \times d^3$

Step5: Calculate combined stress- ' σ_{com} ' **Principal stresses**

$$\sigma_1 \ \& \ \sigma_2 , \ \tau$$

How to get resolution for the effect



Step3: Calculate Direct stress (by axial load)

$$\sigma_d = \text{load/area}$$

$$\sigma_d = \sigma_x = ???$$

Step4: calculate bending Stress (By combination of Original load and introduced load)

$$\sigma_b = M/z$$

where $M = \text{moment due to external load} = w \times e$

$z = \text{section modulus}$

$$\sigma_b = \sigma_y = ???$$

Step3: Calculate Direct stress (by axial load)

$$\sigma_d = \text{load/area}$$

$$\sigma_d = \sigma_x = 0.04 \text{ N/sq.mm}$$

Step4: calculate bending Stress (By combination of Original load and introduced load)

$$\sigma_b = M/z$$

where $M =$ moment due to external load $= w \times e$

$z =$ section modulus

$$\sigma_b = \sigma_y = 0.198 \text{ N/Sq.mm}$$

~~Step5: Calculate combined stress ' σ_{com} '~~

$$\sigma_{com} = \sigma_d \pm \sigma_b$$

Step6: Calculate σ_1 and σ_2



A photograph of a handwritten equation on a piece of paper. The equation is:
$$\sigma_{1,2} = \frac{1}{2} \left[(\sigma_x + \sigma_y) \pm \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_x^2} \right]$$

In this problem, shear stress is zero, $\tau = 0$

Data		
diameter	40	mm
Load	50	N
e	25	mm
area		
Pi	3.141	
A	1256.4	sq.mm

1	Direct stress	
	Stress	0.0398 N/sq.mm
	σ_x	0.0398

2 Bending

$$\sigma_b = M/Z$$

$$M = \text{Load} \times e$$
$$1250 \text{ N-mm}$$

$$z = I/y$$

$$I = \pi \cdot d^4 / 64$$

$$Y = d/2$$

$$I = 125640 \text{ mm}^4$$

$$y = 20 \text{ mm}$$

$$\text{therefore } z = 6282 \text{ mm}^3$$

$$\text{Now } \sigma_b = 0.1989812 \text{ N/sq.mm}$$

$$\sigma_y = 0.198$$

3 principal stress

$\sigma_1, \sigma_2 = ??$

$\sigma_x + \sigma_y$ 0.2378 N/sq.mm

$\sigma_x - \sigma_y$ -0.1582 N/sq.mm

$(\sigma_x - \sigma_y)^2$ 0.0250

$(\sigma_x + \sigma_y)/2$ 0.1189

$(\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2})/2$ 0.0791

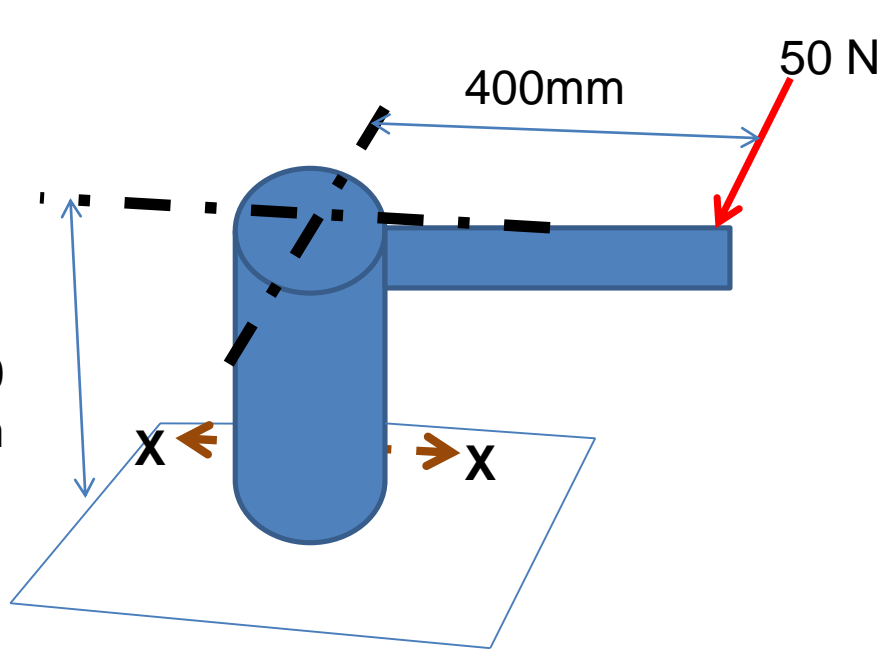
$$\sigma_{1,2} = \frac{1}{2} [(\sigma_x + \sigma_y) \pm \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2}]$$

$$\left(\frac{\sigma_x + \sigma_y}{2}\right) \pm 1/2\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2}$$

$\sigma_1 =$	0.198	N/sq.mm
$\sigma_2 =$	0.0398	N/sq.mm
factor of saftey		
n=	10.1	

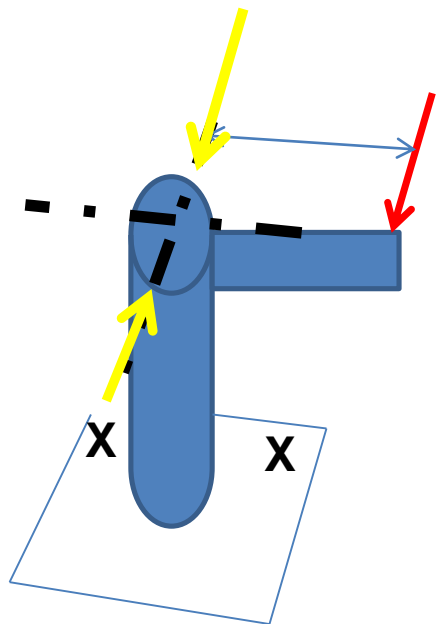
Maximum stress to be taken
As working stress.

Problem 2



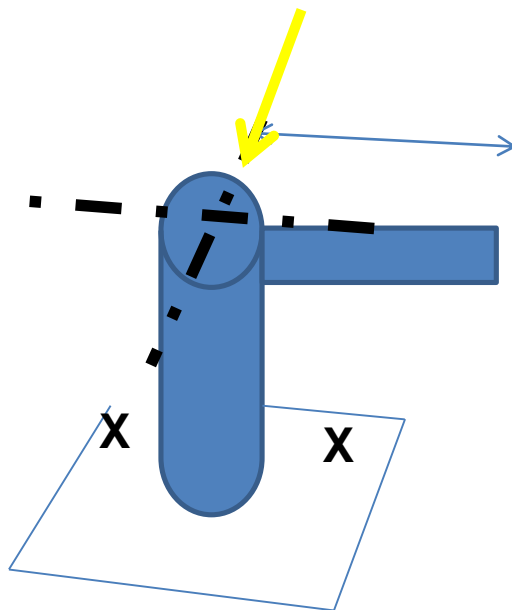
Determine the principal stresses induced at Section XX, for the member shown in the figure.. Take the diameter of the member is 50 mm.

Take allowable stress is 5 N/ sq. Mm
Also find the factor of safety.

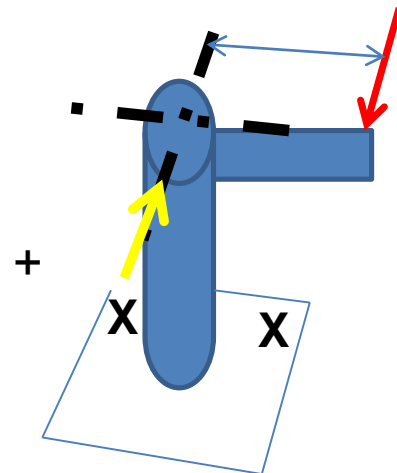


Combined stress

=



Bending stress



Shear stress

Data

diameter

50mm

Load

50N

eh

400mm

(Horizontal to original load)

ev

500mm

(Vertical to introduced load from the base)

area

Pi

3.141

A

1963.125sq.mm

1Direct stress

Stress

0N/sq.mm

σ_x

0

2 Bending

$$\sigma_b = M/Z$$

$$M = \text{Load} \times e = 25000 \text{ N-mm}$$

$$z = I/\gamma$$

$$I = \pi \cdot d^4 / 64$$

$$\gamma = d/2$$

$$I = 306738 \text{ mm}^4$$

$$\gamma = 25 \text{ mm}$$

$$\text{therefore } z = 12269.5 \text{ mm}^3$$

$$\text{Now } \sigma_b = 2.03757 \text{ N/sq.mm}$$

$$\sigma_y = 2.037$$

3	Shear stress		$T \times 16 / (\pi \times d^3)$				refer Pg.No 7.1/1
							DDB
		T=	20000 N mm				
			(eccentric distance is horizontal distar				
	τ Shear stress		0.81503 N/sq.mm				
		τ	0.815				

Principal stress

$\sigma_1, \sigma_2 =$

??

$\sigma_x + \sigma_y$

2.037N/sq.mm

$\sigma_x - \sigma_y$

-2.037N/sq.mm

$(\sigma_x - \sigma_y)^2$

4.1494

$(\sigma_x + \sigma_y)/2$

1.0185

τ

0.815

τ^2

0.664

$(\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2})/2$

1.304

$$\sigma_1 = 2.322941356 \text{ N/sq.mm}$$

$$\sigma_2 = -0.285941356 \text{ N/sq.mm}$$

$$\sigma_i < [\sigma],$$

2.32 < [5] design is safe

factor of safety

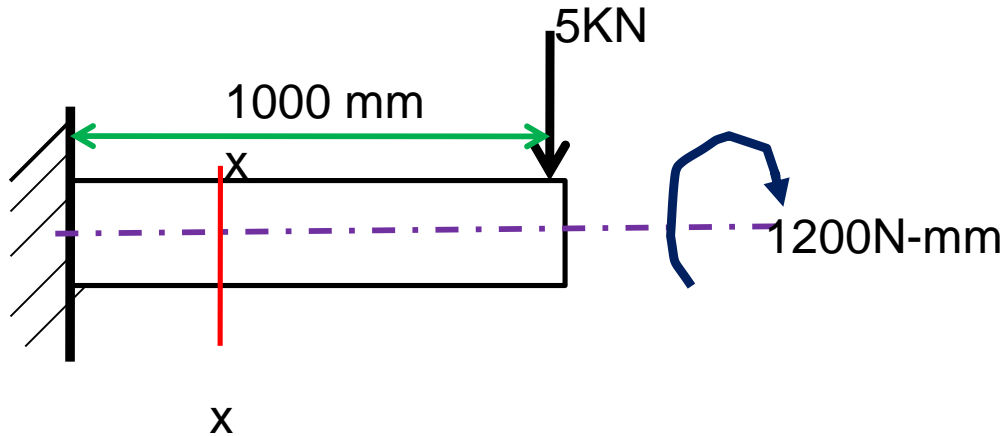
$$n = 2.15$$

AS per shear Stress Theory

$$\tau = \pm \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2}$$

$$\begin{array}{l} \tau \quad + \quad 1.304 \text{ N/sq.mm} \\ \tau \quad - \quad 1.304 \text{ N/sq.mm} \end{array}$$

Problem 3



A cantilever circular member is subject to loads as shown in figure. Determine the diameter of it .If The allowable stress is 90 N/mm^2

STEP1. Identify the loads types and its effect on it

Step2: calculate the individual stresses 2 or 3

Step3 : Take into the Principal stress equations

Note: If the allowable stress given in the problem, then

That is design stress $[\sigma] = [\sigma_1]$,

Find the dimension of the member

ref. pg.No.7.2,/DDB

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

Step1

Calculation of direct stress - $\sigma_d = \sigma_x$

$$\sigma_d = \text{load/area} = \text{NO}$$

Step2

Calculation of bending stress $\sigma_b = \sigma_y$

$$\sigma_b = M/z$$

where $M = \text{moment due to external load} = w \times e$

$z = \text{section modulus}$

Step3: Calculate Shear stress,

$$T = \pi/16 \times \tau \times d^3$$

Step 4 Take into the Principal stress equations

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

$$[\sigma] = [\sigma_1],$$

$$[\sigma_1] = [90] \text{ N/mm}^2$$

Step3

Using principal stress equation, Find diameter of The member

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

σ_1 is to be taken as allowable stress

Data		
diameter	x mm	
bendingLoad	5000 N	
Distance for bending lo	1000 mm	
Turning moment	1200 mm	
area		
Pi	3.141	
[σ1]	90 N/sq.mm	
A	0.78525 d²	sq.mm

1 Direct stress			
Stress	no axial load	N/sq.mm	
σ_x		0	

2 Bending				
$\sigma_b =$	M/Z			
M	Load x e			
		5000000	N-mm	
z =	I/y			
	I =	$\pi * d^4 / 64$		
	Y =	d/2	0.5*d	
	I =	0.049	$* d^4$	mm ⁴

therefore	$z =$	$0.098156 * d^3$	mm^3
Now	$\sigma_b =$	$50939191 * (1/d^3)$	N/sq.mm
	σ_y	$50939191 * (1/d^3)$	

3 Shear stress		$T \times 16 / (\pi \times d \times d \times d)$	
	$T =$	1200 N mm	
τ Shear stress		$6112.703 * (1/d^3)$	N/sq. mm

Principal stress			
$\sigma_1, \sigma_2 =$??		
$\sigma_x + \sigma_y$		50939191.34 $\cdot (1/d^3)$	N/sq.mm
$\sigma_x - \sigma_y$		-50939191.34 $\cdot (1/d^3)$	N/sq.mm
$(\sigma_x - \sigma_y)^2$		2594801214407510 $\cdot (1/d^3)^2$	
$(\sigma_x + \sigma_y)/2$		25469595.67 $\cdot (1/d^3)$	

τ	6112.703	$*(1/d^3)$
τ^2	37365137.487	$*(1/d^3)^2$
$(\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2}/2)$	25469596.404	$*(1/d^3)$

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

Asper principal stress

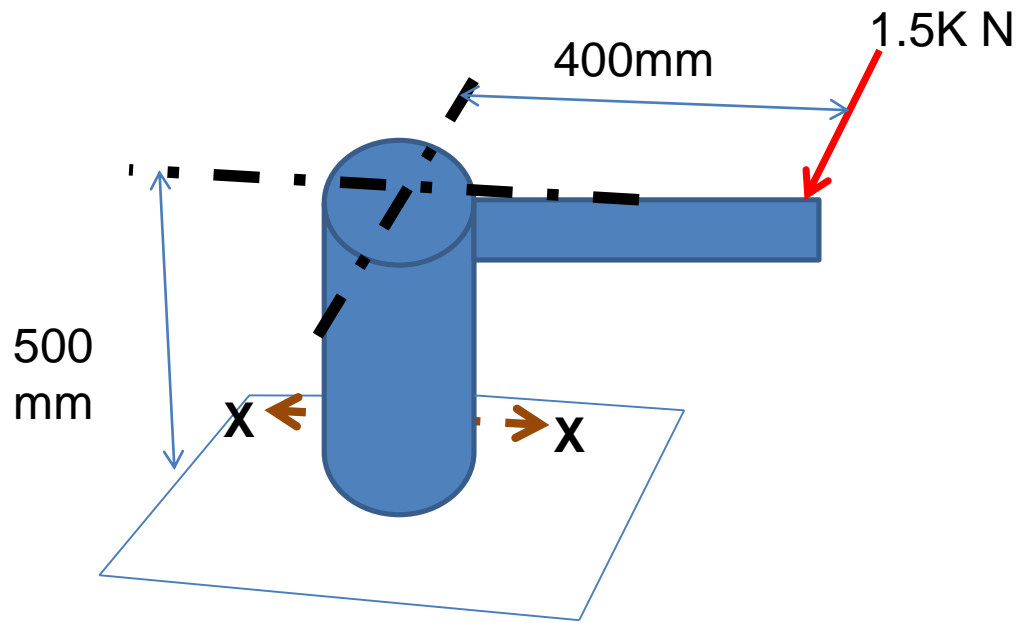
$[90] =$	50939192.074	$*(1/d^3)$
90	50939192.074	$*(1/d^3)$

$$d^3 = 565991.023$$

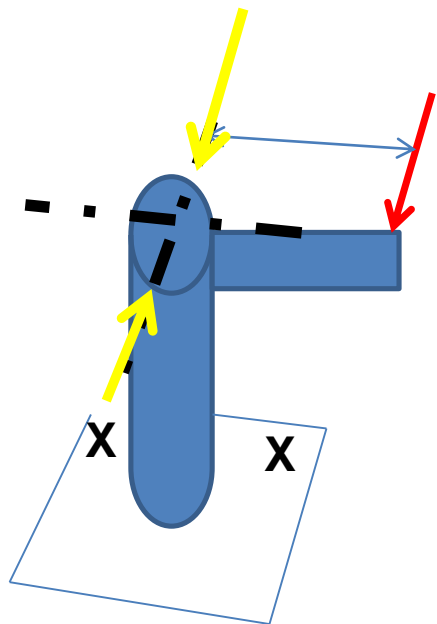
d

82.72 mm

Problem 4

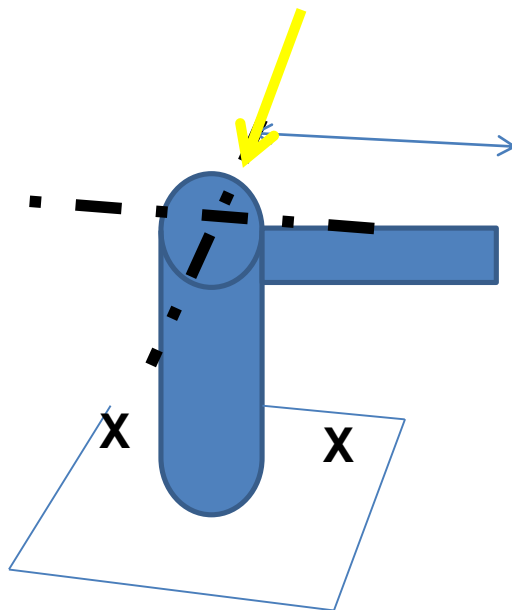


Determine the diameter of the member at Section XX, for the member shown in the figure. Take the allowable stress of the member is 120 N/sq. mm

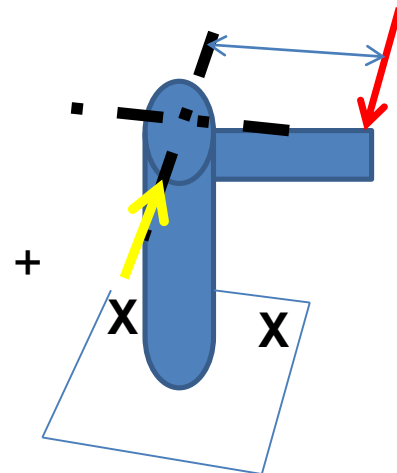


Combined stress

=



Bending stress



+

Shear stress

STEP1. Identify the loads types and its effect on it

Step2: calculate the individual stresses 2 or 3

Step3 : Take into the Principal stress equations

Note: If the allowable stress given in the problem, then

That is design stress $[\sigma] = [\sigma_1]$,

Find the dimension of the member

ref. pg.No.7.2,/DDB

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

Step1

Calculation of direct stress - $\sigma_d = \sigma_x$

$$\sigma_d = \text{load/area} = \text{NO}$$

Step2

Calculation of bending stress $\sigma_b = \sigma_y$

$$\sigma_b = M/z$$

where $M = \text{moment due to external load} = w \times e$

$z = \text{section modulus}$

Step3: Calculate Shear stress,

$$T = \pi/16 \times \tau \times d^3$$

Step 4 Take into the Principal stress equations

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2)}$$

$$[\sigma] = [\sigma_1],$$

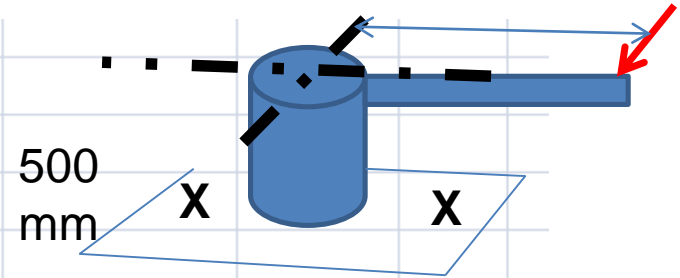
$$[\sigma_1] = [120] \text{ N/mm}^2$$

Data									
diameter		d mm							
Load		1500 N	(1.5 x 1000)						
eh- turning effect		400 mm	(Horizontal to original load)						
ev- bending		500 mm	(Vertical to introduced load from the base)						
area									
Pi		3.141							

1 Direct stress		
Stress		0 N/sq.mm
σ_x		0

Due to no axial load

2 Bending



$\sigma_b =$	M/z			
M	Load x e	750000 N-mm		
$z =$	I/y			
	$I =$	$\pi * d^4 / 64$		
	$Y =$	$d/2$		
	$I =$	0.049 * d^4	mm^4	
	$y =$	0.5 d	mm	
therefore	$z =$	0.0981563	mm^3	
Now	$\sigma_b =$	7640878.7 * $(1/d^3)$	N/sq.mm	
	σ_y	7640878.7 * $(1/d^3)$		

3 Shear stress		$T \times 16 / (\pi \times d^3)$	refer Pg.No 7.1/DDB
	T=	600000 N mm	
		(eccentric distance is horizontal distance from axis)	
τ Shear stress		$\tau = 3056351.5 \times (1/d^3)$	N/sq.mm

4 Principal stress			
$\sigma_1, \sigma_2 =$??		
$\sigma_x + \sigma_y$		7640878.701	N/sq.mm
$\sigma_x - \sigma_y$		-7640878.701	N/sq.mm
$(\sigma_x - \sigma_y)^2$		58383027324169.0000	$\times (1/d^3)^2$
$(\sigma_x + \sigma_y)/2$		3820439.351	$\times (1/d^3)$

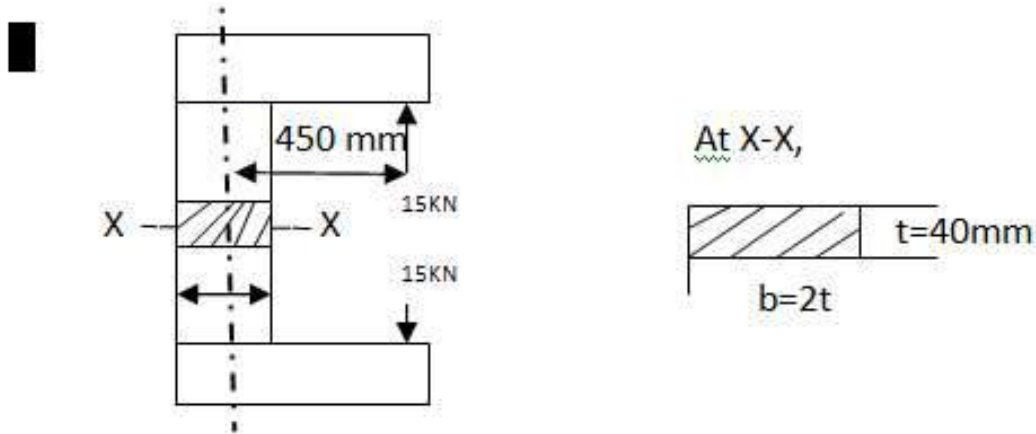
τ		3056351.48 $\cdot (1/d^3)$
τ^2		9341284371867.040 $\cdot (1/d^3)^2$
$(\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2}/2)$		4892549.561 $\cdot (1/d^3)$
Asper principal stress	$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{((\sigma_x - \sigma_y)^2 + 4\tau^2)}$	
	$[90] =$	8712988.911 $\cdot (1/d^3)$
	${}^{120} 90$	8712988.911 $\cdot (1/d^3)$

$d^3 = 96810.9879$

$d = 45.92 \text{ mm}$

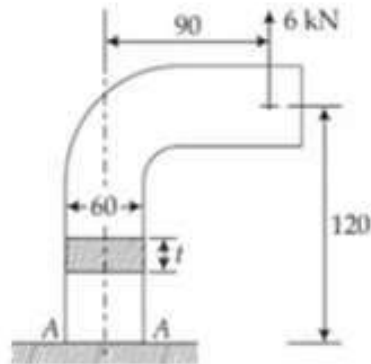
41.72

Tutorial 1 3-7-2020



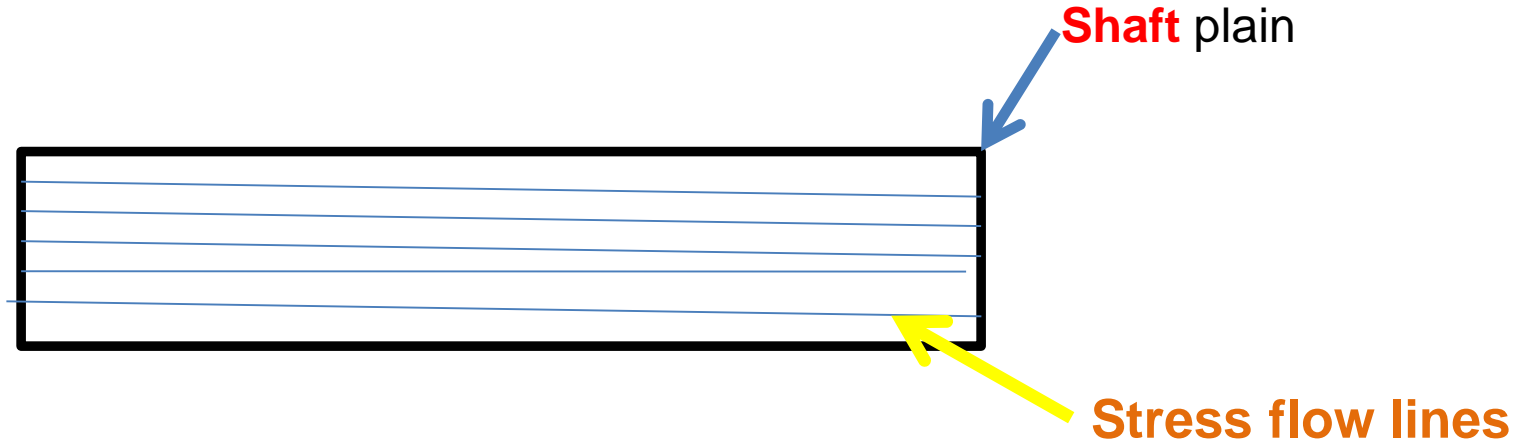
○ Determine the principal stresses in C – clamp at XX section resembles rectangular.

○ Determine the thickness of the steel bracket loaded as shown in fig. taking allowable stress as 90 Mpa



STRESS CONCENTRATION FACTOR

STRESS CONCENTRATION FACTOR- K_t



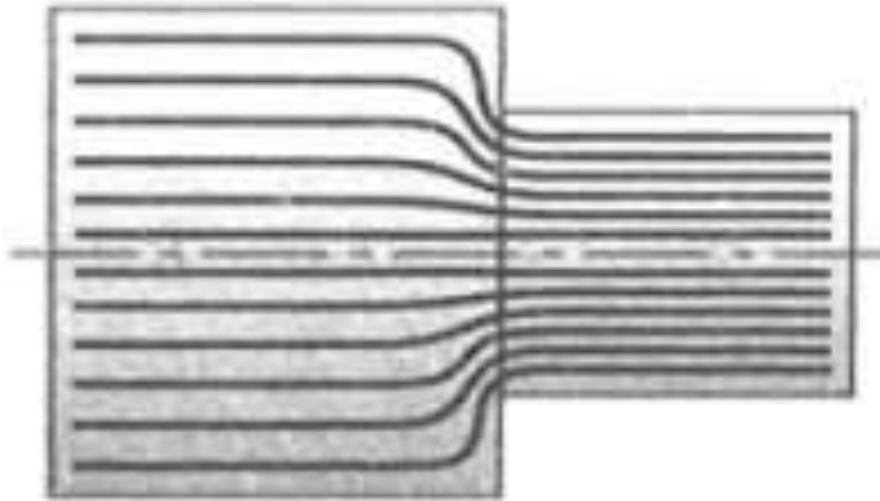
Def:

The stress concentration factor, K_t , is the ratio of the highest stress to a nominal stress of the gross cross-section

$$K_t = \frac{\sigma_{\text{Max}}}{\sigma_{\text{nominal}}}$$

where σ_{Max} is allowable or Design Stress

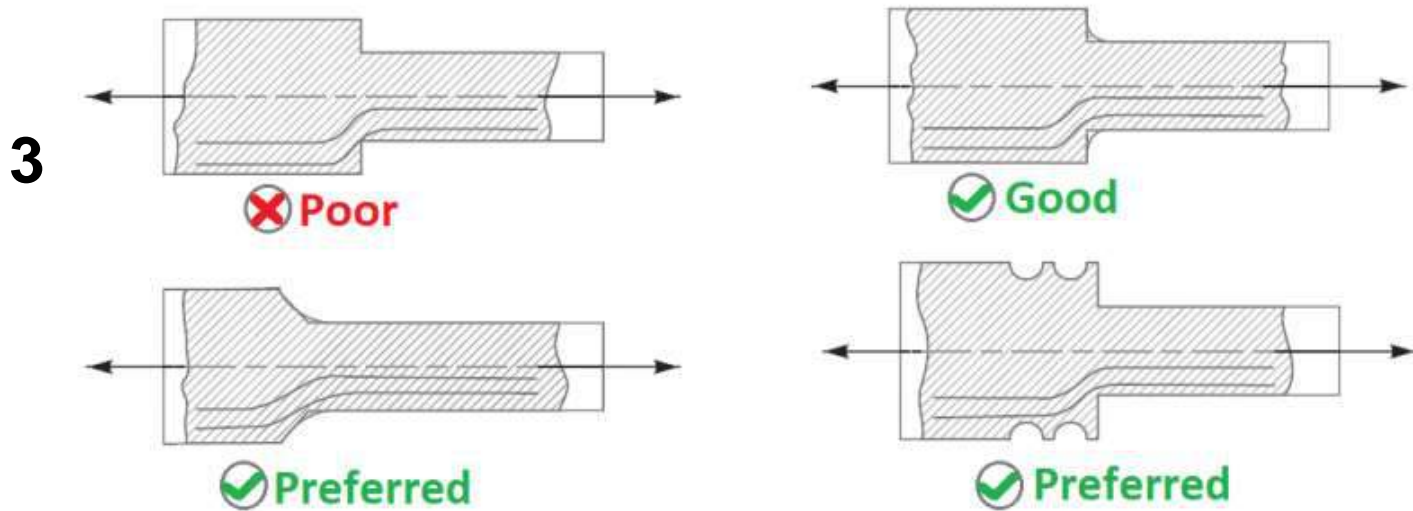
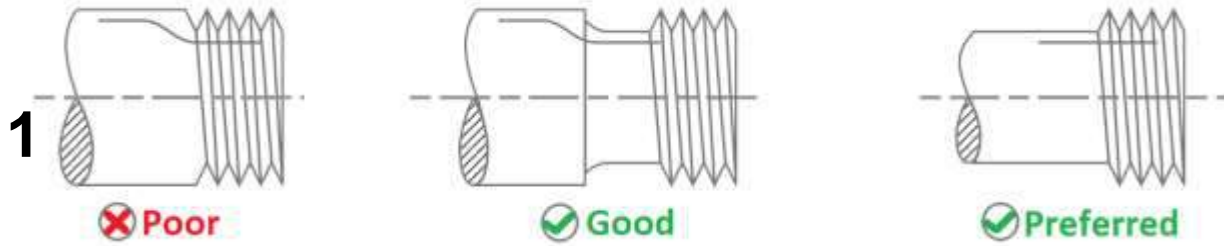
σ_{nominal} is the stress with respect to area of cross section of the member

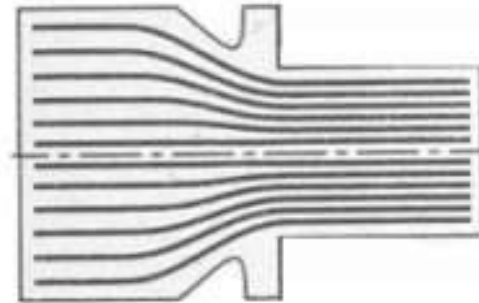
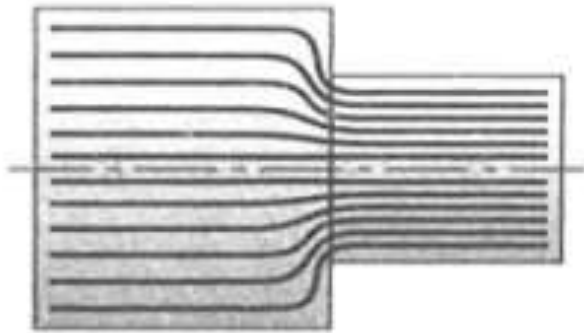


Force flow lines in an abrupt changes in the cross section of the member

Methods of Reducing Stress Concentrations

- ❑ **Avoiding sharp** corners and only **using rounded** corners with maximum radii.
- ❑ **Sanding and polishing** surfaces to remove any notches or defects that occur during forming and processing.
- ❑ **Lowering the stiffness** of straight load-bearing segments.
- ❑ **Placing notches** and threads **in low-stress areas**.



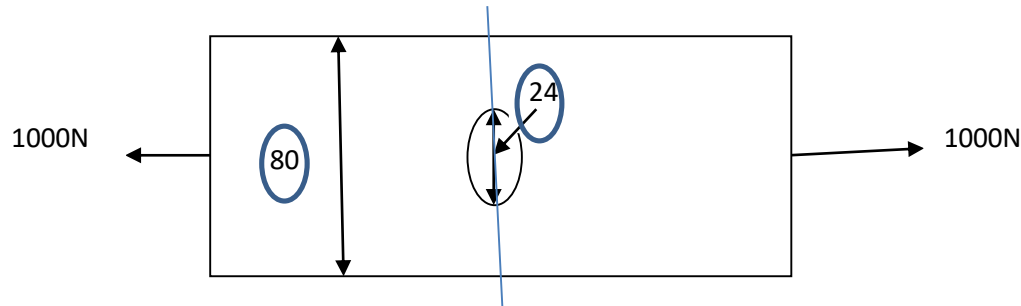


Stress relieving groove

To solve problems

refer design data book pages 7.8 to 7.16 pages
for various types of members and their loading condition.

P5. Find the maximum stress induced in the plate as shown in figure. All sizes are in mm $T=15\text{MM}$



$$K_t = \frac{\sigma_{max}}{\sigma_{nominal/working}}$$

Identify the Notations as per data book Refer pg.No 7.10/DDB

$$W = 80 \text{ mm} \ \& \ a = 24 \text{ mm}$$

Stress concentration Factor

DATA

Load	p	1000 N
Width	W	80 mm
hole dia	a	24 mm
Thuckness		15 mm
Area		(w-a)t
		840 mm ²
σ_{max}		???

$\sigma_{nominal}$

L/A

1.19N/mm²

2Kt ?

To be obtained from DDB 7.10

Ratio a/w

0.3

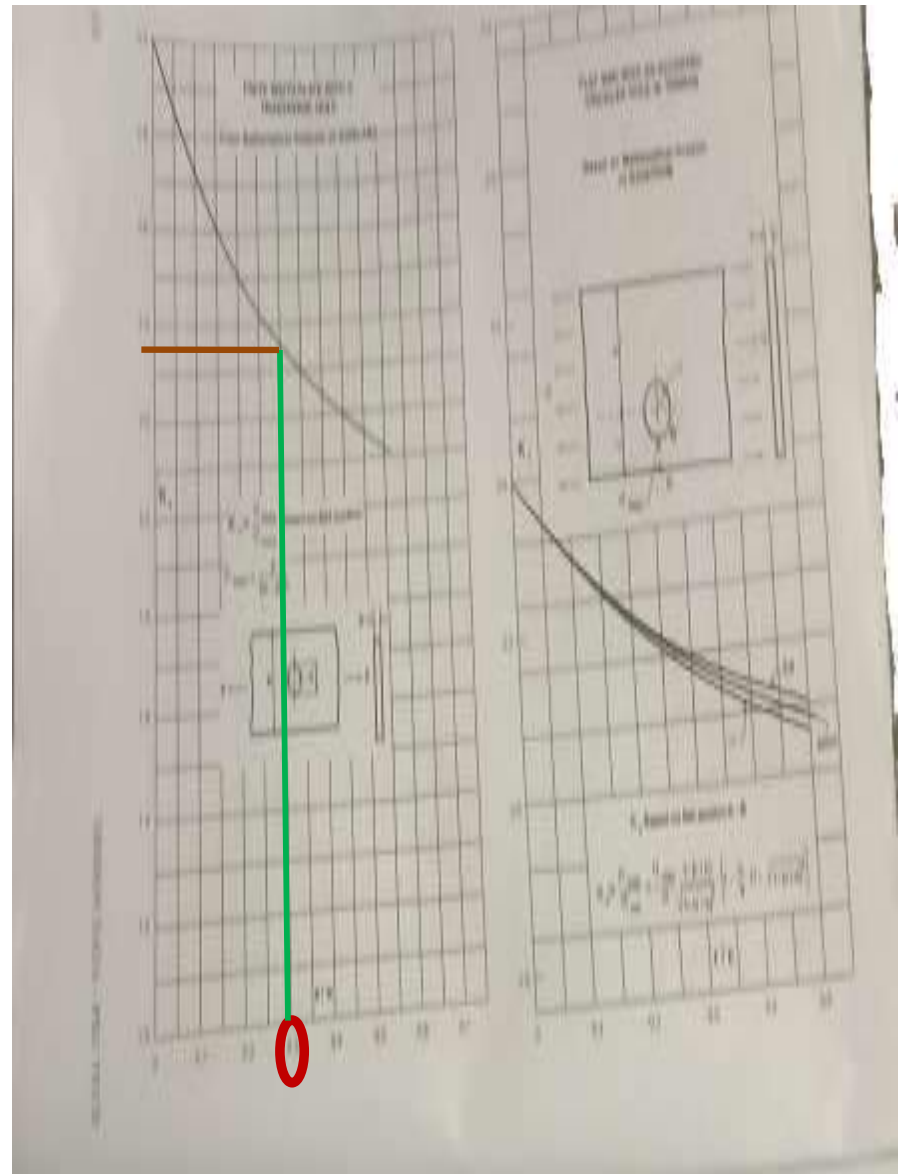
Kt **2.4**

3σ_{max}

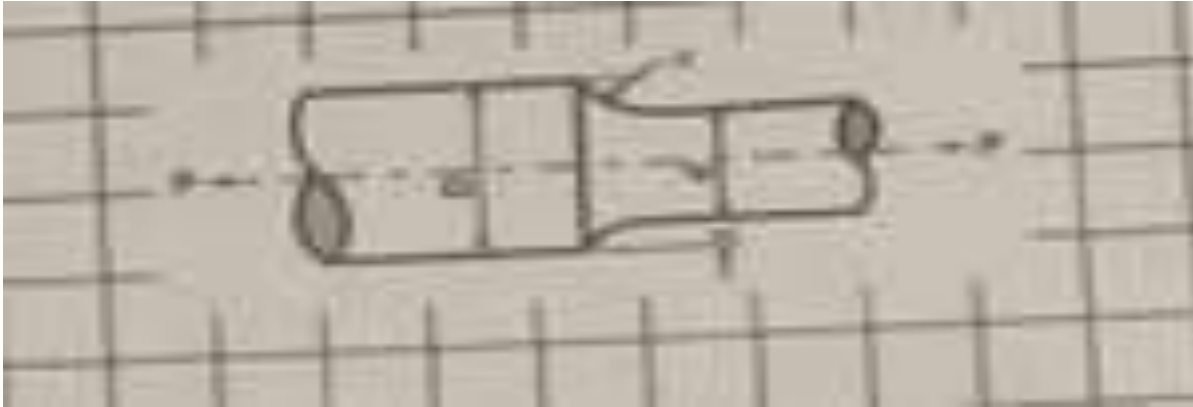
σ_{ma}

$x = Kt \times \sigma_{nom}$

2.8 N/mm²



P6. Find the maximum stress induced in the object as shown in figure. All sizes are in mm



Load is 1KN, Bigger dia is 80 mm and stepped dia is 40 mm. The radius of the fillet is 8 mm

Stress concentration Factor

DATA

Load	p	1000 N
Dia(D)	D	80 mm
smalldia(d)	d	40 mm
radius of fillet r		8 mm

Area

$\pi/4 \cdot (d^2)$ 1256.4 N/mm²

σ_{max} ???

Take small area

$1\sigma_{nom}$

L/A

$0.795924865 \text{ N/mm}^2$

$2K_t$

?

To be obtained from DDB 7.11

Ratio

r/d

0.2

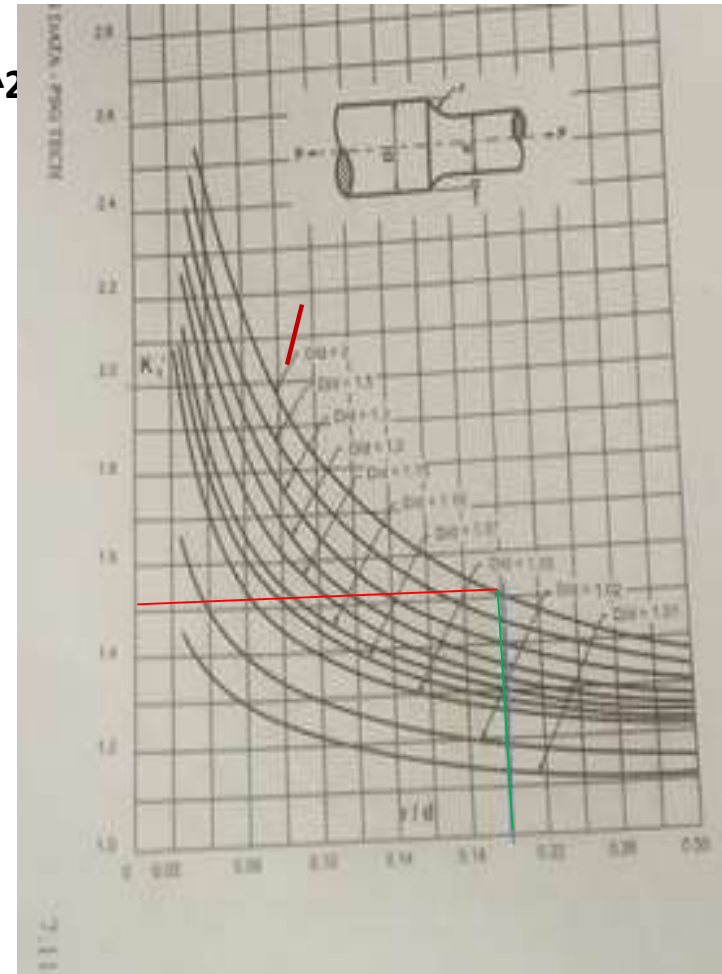
Ratio

D/d

2

K_t

1.5



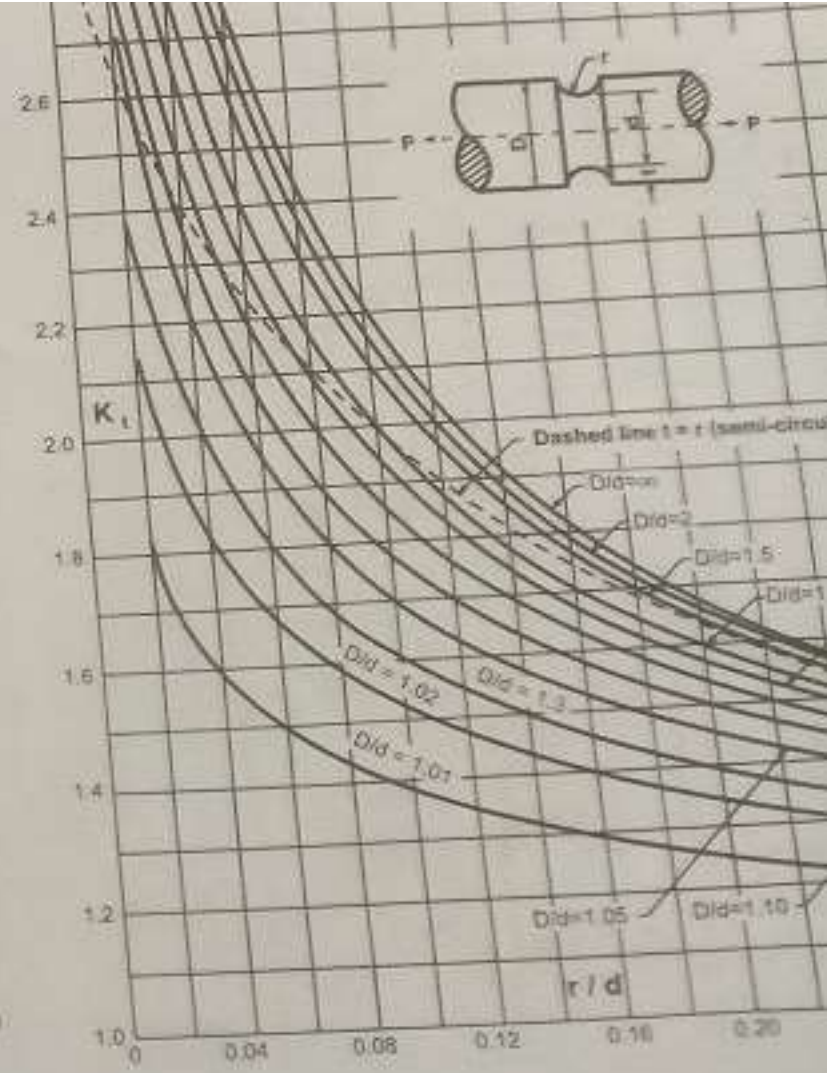
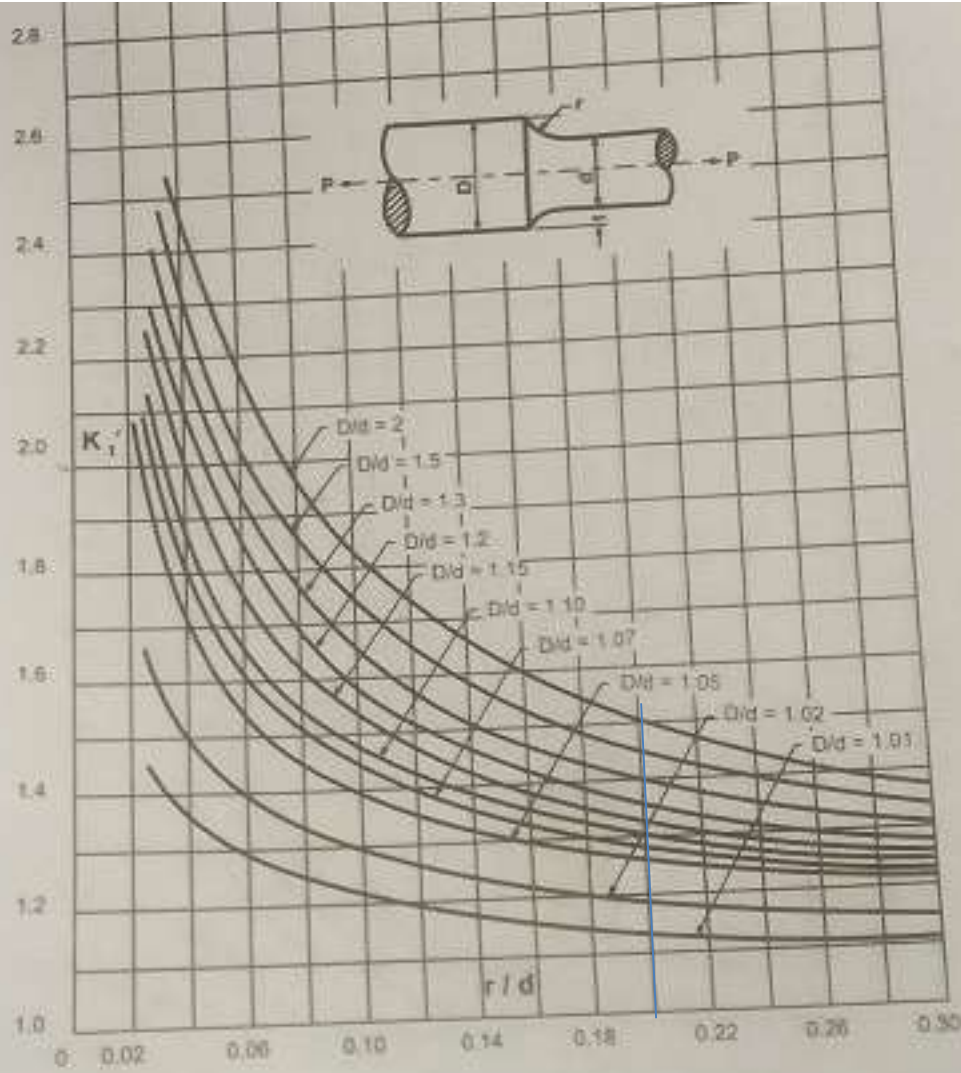
3 σ_{max}

$\sigma_{max} =$

$K_t \times \sigma_{nom}$

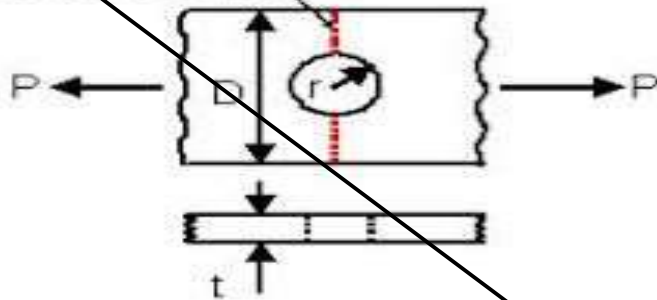
1.194 N/mm²

**Practice by yourself various components
in stress concentration**



1. Determine the stress concentration factor for the plate subjected to load of 9KN As In figure.

Nominal stress area



$$\sigma_{\text{nom}} := \frac{\text{Force}}{\text{Area}}$$

Flat Bar Dimensions

Radius of hole: $r = 10 \cdot \text{mm}$

Width of section: $D = 100 \cdot \text{mm}$

Thickness of plate: $t = 10 \cdot \text{mm}$

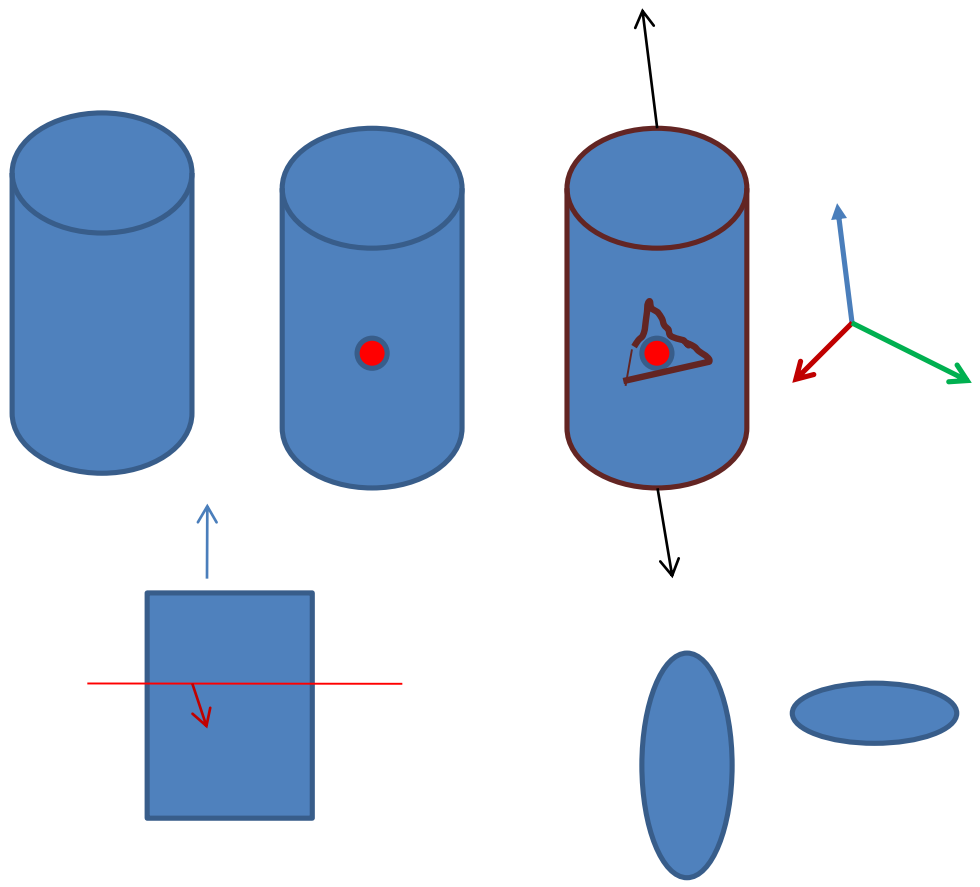
Applied load: $P = 9000 \cdot \text{N}$

Stress concentration factor $k_t := 3.00 - 3.13 \cdot \left(\frac{2 \cdot r}{D}\right) + 3.66 \cdot \left(\frac{2 \cdot r}{D}\right)^2 - 1.53 \cdot \left(\frac{2 \cdot r}{D}\right)^3$

$$k_t = 2.508$$

Nominal stress $\sigma_{\text{nom}} := \frac{P}{t \cdot (D - 2 \cdot r)}$ $\sigma_{\text{nom}} = 11.25 \text{ MPa}$

Peak stress $\sigma_{\text{max}} := k_t \cdot \sigma_{\text{nom}}$ $\sigma_{\text{max}} = 28.217 \text{ MPa}$



THEORIES OF FAILURES

Why ? Need to study

Under combined stress , it is difficult to predict by which stress the failure of member occurred.

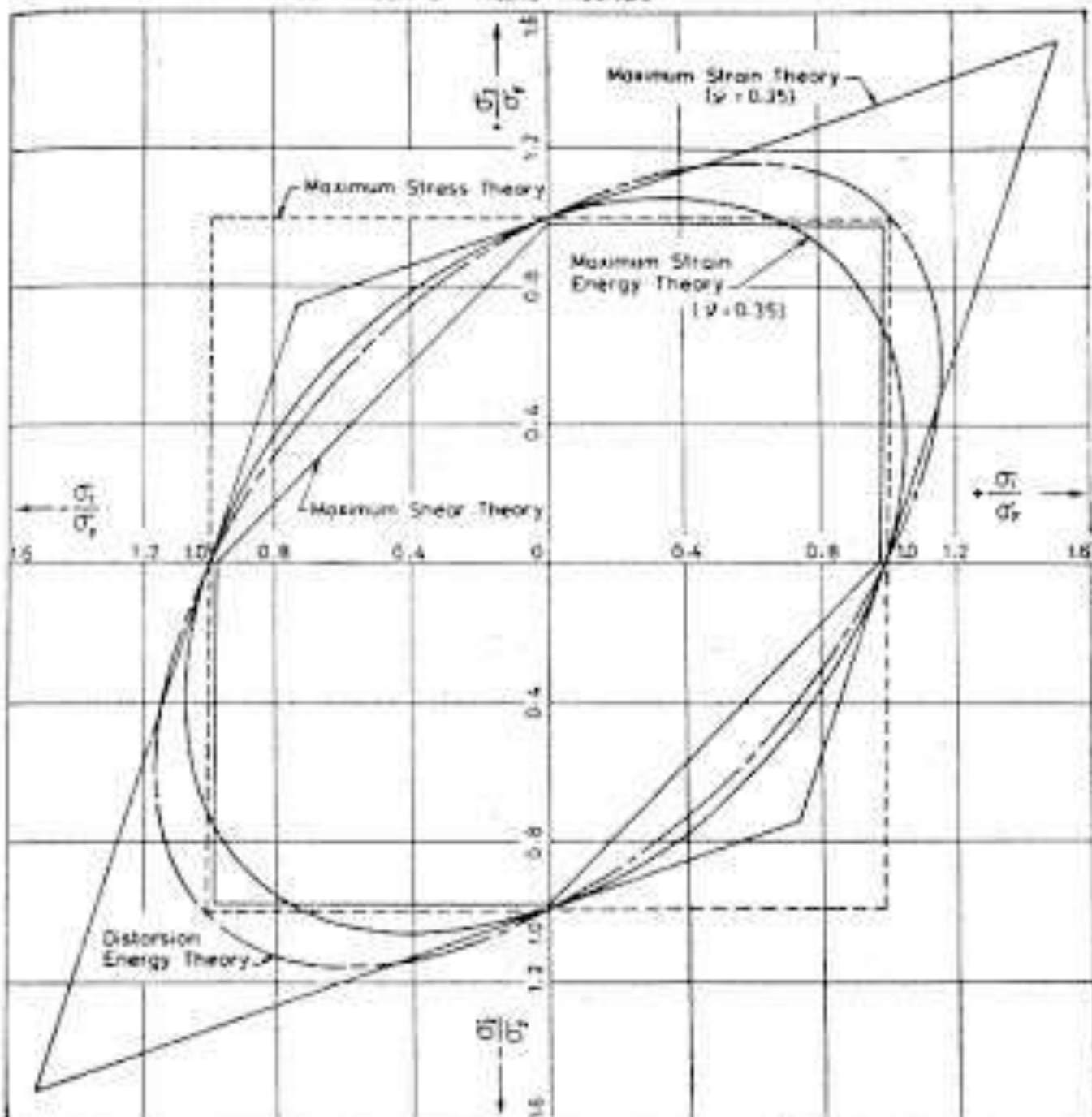
Hence the stresses are accounted in terms of principal stresses

This involves the **assessment of stresses in biaxial or tri axial stresses.**

It is by nature all the engineering components subjected to those systems stresses

TYPES OF THEORIES OF FAILURES

1. Maximum Principal Stress theory (**RANKINE'S THEORY**)—(**σ_{normal}**)
2. Maximum Shear Stress theory (**GUEST AND TRESCA'S THEORY**)--- (**τ_{max}**)
3. Maximum Principal Strain theory (**St. VENANT'S THEORY**)---(**ϵ_{max}**)--- No reliable results- Not followed
4. Total Strain Energy theory (**HAIGH'S THEORY**)
5. Maximum Distortion Energy theory (**VONMISES AND HENCKY'S THEORY**)



How ? The Theories of failure to be defined

Member's σ_y value
under
The biaxial or Tri axial
systems stresses

=

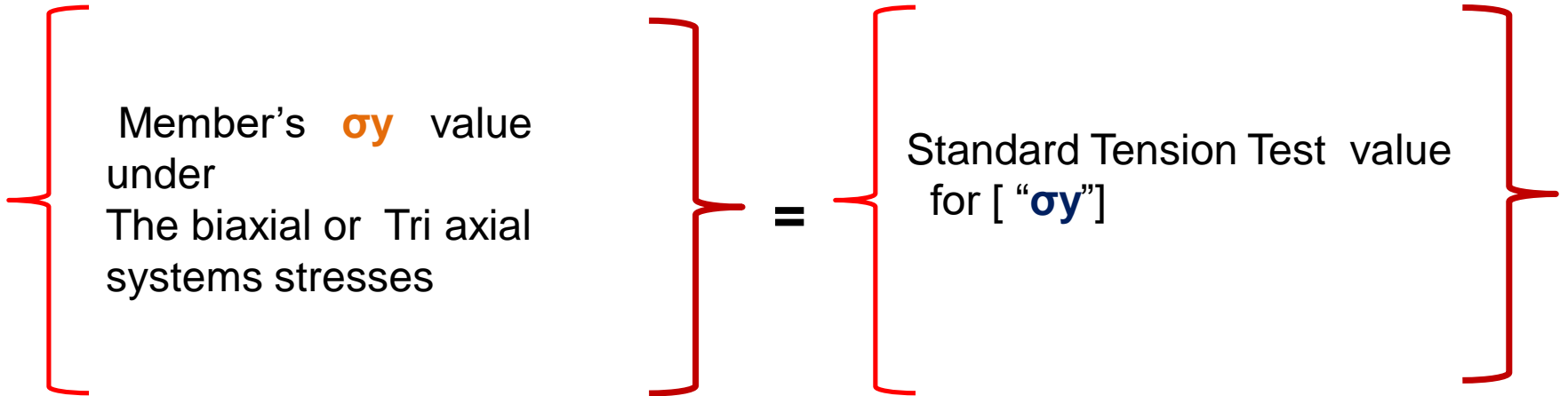
Standard Tension Test value
for [" σ_y "]

The diagram consists of two large red curly braces. The left brace encloses the text 'Member's σ_y value under The biaxial or Tri axial systems stresses'. The right brace encloses the text 'Standard Tension Test value for [" σ_y "]'. An equals sign is positioned between the two braces.

Condition is for same material

1.Max. Normal or principal stress Theory

Def:



Formula

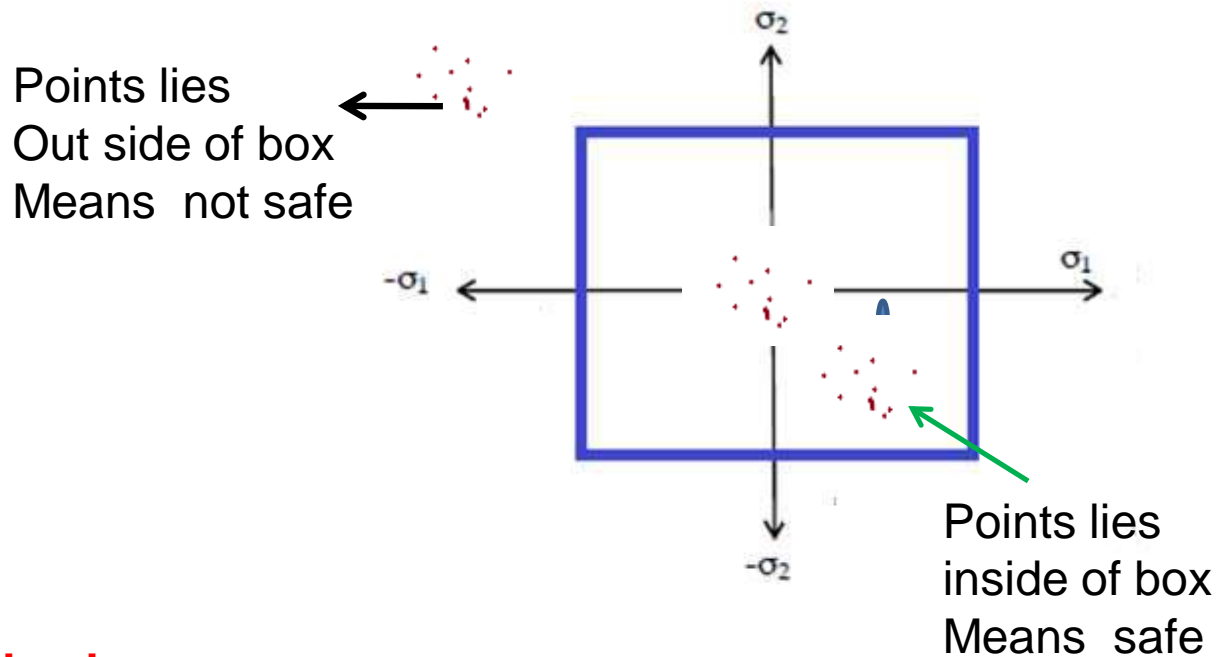
$$\sigma_1 / \sigma_2 / \sigma_3 \leq \sigma_y ,$$

Chose max

$$\sigma_1 / \sigma_2 / \sigma_3 \leq \sigma_y / n$$

Chose max

Graphical study



Design check

' σ_1 / σ_y ' ratio to be calculated and it lies with in square region

Then the design is safe,

if it fall out side then the design is not safe, then rework to be done

2.Max. Shear stress Theory -- τ_{max}

Def:

Member's τ_y value under The biaxial or Tri axial systems stresses } = } Standard Tension Test value for [" τ_y "] }

Formula

$$\frac{(\sigma_1 - \sigma_2)}{(\sigma_2 - \sigma_3)} \text{ / } \frac{(\sigma_3 - \sigma_1)}{(\sigma_2 - \sigma_3)} \leq \sigma_y/2 ,$$

Chose max

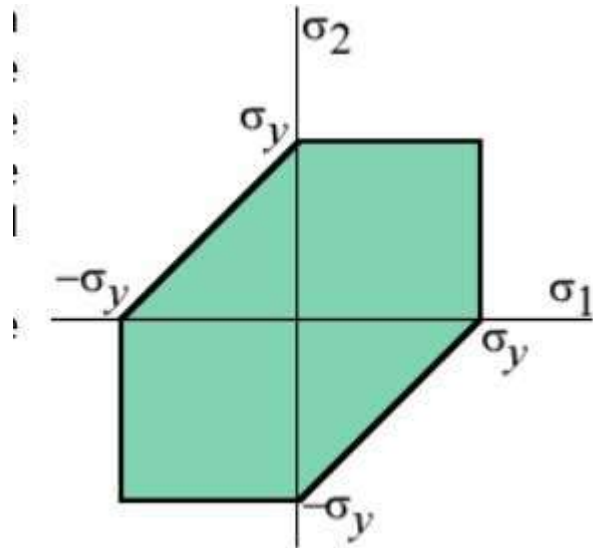
$$\frac{(\sigma_1 - \sigma_2)}{(\sigma_2 - \sigma_3)} \text{ / } \frac{(\sigma_3 - \sigma_1)}{(\sigma_2 - \sigma_3)} \leq \sigma_y/2n$$

Chose max

As per max shear stress theory

$$\tau_{max, simp} = \frac{\sigma_y}{2}$$

Where “ σ_y ” is the material yield value



It is suitable for ductile material and **shaft design** always use it

Design check

“ σ_1 / σ_y ” ratio to be calculated and it lies within shaded/coloured region

Then the design is safe,

if it falls outside then the design is not safe, then rework to be done

3. Max. Strain Theory – ϵ_{max}

Def:

Member's ϵ_y value
under
The biaxial or Tri axial
systems stresses

=

Standard Tension Test value
for [“ ϵ_y ”]

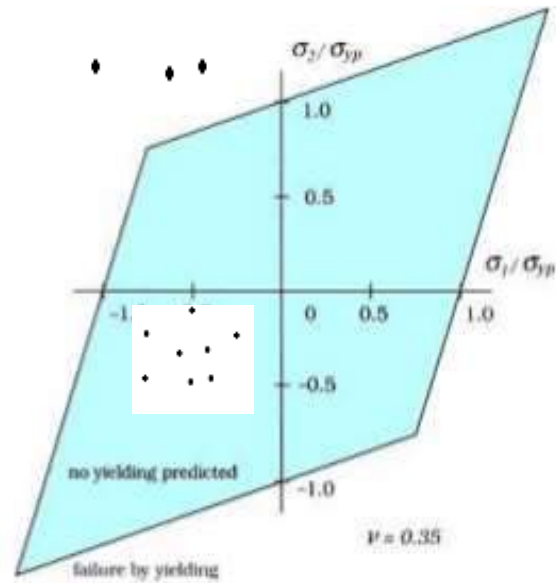
Formula

Maximum strain theory (St. Venant's)

$$\left. \begin{array}{l} \sigma_1 - \nu (\sigma_2 + \sigma_3) \text{ or } \sigma_2 - \nu (\sigma_3 + \sigma_1) \\ \text{or } \sigma_3 - \nu (\sigma_1 + \sigma_2) \text{ whichever is maximum} \end{array} \right\} = \sigma_T$$

As this theory is not providing
Reliable results,

Not being/ recommended used in
the design



4. Max. Strain Energy Theory

Def:

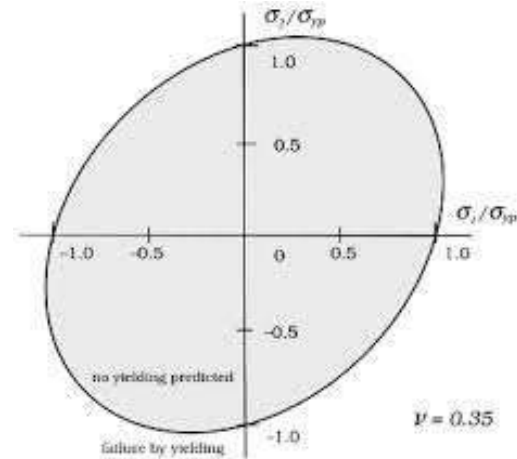
$$\left[\begin{array}{l} \text{Member's energy stored/unit} \\ \text{volume value (U) under} \\ \text{the biaxial or Tri axial} \\ \text{systems stresses} \end{array} \right] = \left[\begin{array}{l} \text{Standard Tension Test value} \\ \text{for limiting strain energy} \\ \text{"U}_y\text{"} \end{array} \right]$$

Formula



Poisson ratio

It holds good results for ductile materials



Design check

' σ_1/σ_y ' ratio to be calculated and it lies within square region

Then the design is safe,
if it falls outside then the design is not safe, then rework to be done

5.Max. Distortion Strain Energy Theory

Def:

Member's energy stored/unit volume value under the biaxial or Tri axial systems stresses (U) = Standard Tension Test value for limiting strain energy [“U y”]

Formula

Maximum strain energy theory

$$\sigma_1^2 + \sigma_2^2 + \sigma_3^2 - 2\nu (\sigma_1\sigma_2 + \sigma_2\sigma_3 + \sigma_3\sigma_1) = \sigma_y^2$$

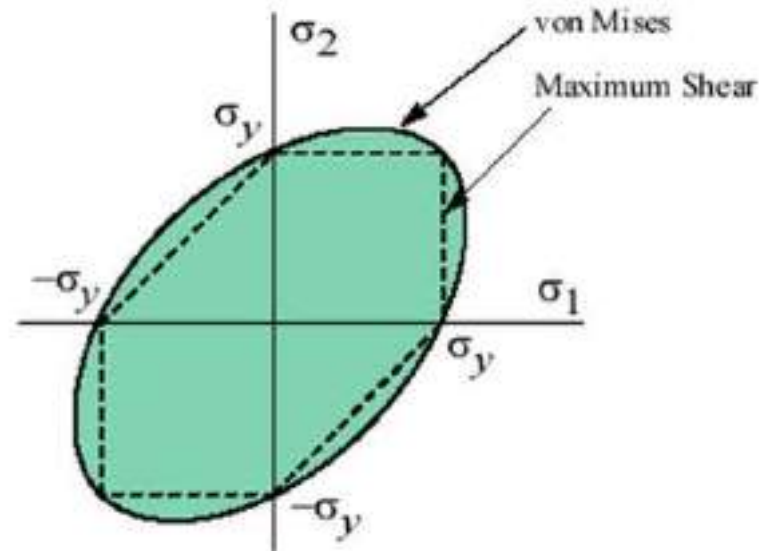
It holds good results for ductile materials

Dist. Strain energy=
strain energy- strain energy by stress

Design check

' σ_1 / σ_y ' ratio to be calculated and it lies within region

Then the design is safe,
if it falls outside then the design is not safe, then
rework to be done



Problems Objectives

- 1. Find the size of the member**
- 2. Find the factor of safety**

A member is having its principal stresses as follows $\sigma_1 = 190$ Mpa, $\sigma_2 = 90$ Mpa and $\sigma_3 = 0$ find the factor of safety for the member using theories of failures. Take material of C45 has 360 Mpa,

$$\sigma_1 = 190$$

$$\sigma_2 = 90$$

$$\sigma_y = 360$$

$$\sigma_1 - \sigma_2 = 100$$

$$\sigma_2 - \sigma_3 = 90$$

$$\sigma_3 - \sigma_1 = -190$$

$$1n = 1.894$$

$$2n = 1.8$$

$3n$

2.27

Max. Strain Theory

$4n^2$

4.021098

Max. Strain energy Theory

n

2.005

$5n^2$

4.782288

Max. Distortion strain energy Theory

n

2.19

STEPS to solve for theories of failure

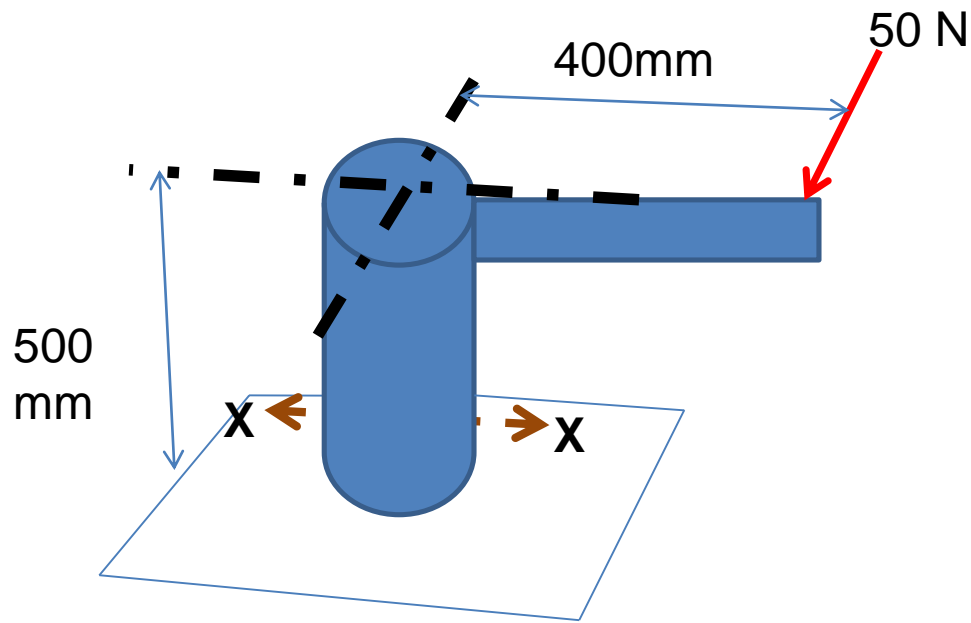
Step1: Identify the stresses available

Step2: Directional assignments the stresses available

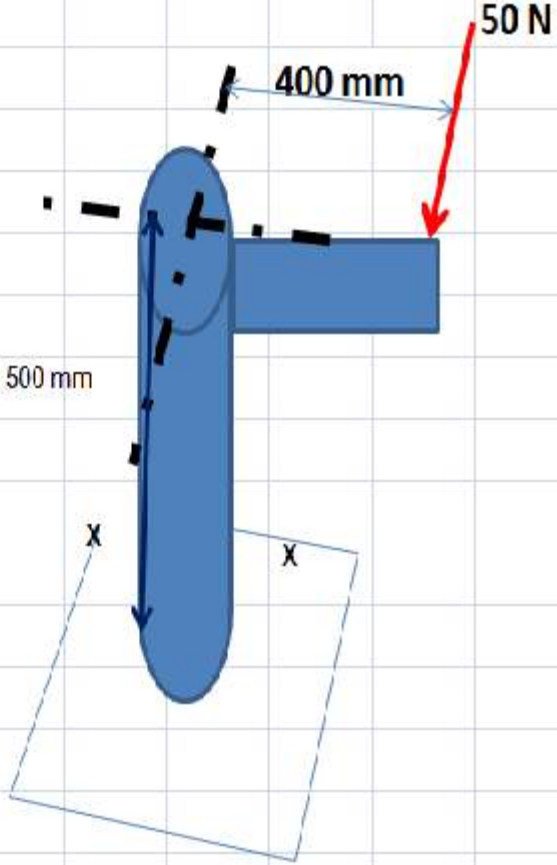
Step3 : Calculate principal stresses

**Step4: Applying theories of failures
& find the objective**

Problem 2



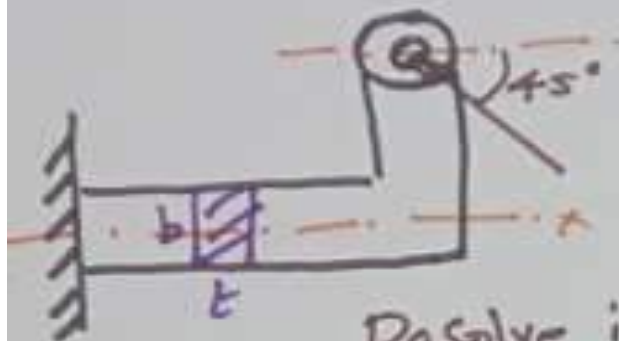
Determine diameter at Section XX, for the member shown in the figure. Find the diameter of the member Using theories of failure. Take factor of safety $n=2.5$,

2			
DATA			
D	???		
twisting moment distance		400 mm	
Bending Moment		500 mm	
n- factor of safety		2.5	
			

First									
calculate σ_1, σ_2	(It should be done bt directional assignment of bending = σ_y, τ , but $\sigma_d = 0$)								
Apply to theories of failure									
1 Caculation of σ_1 & σ_2									
Data									
diameter		x mm							
Load		50 N							
eh		400 mm					(Horizontal to original load)		
ev		500 mm					(Vertical to introduced load from the base)		
area									
Pi		3.141							
A		$0.78525 d^2$					sq.mm		
Direct stress									
1 Stress		0 N/sq.mm							
σ_x		0							

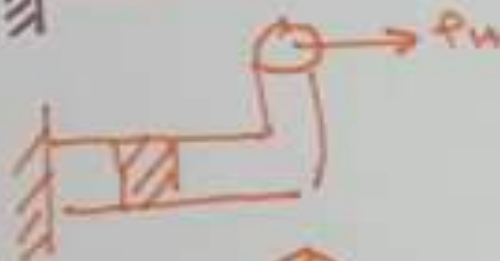
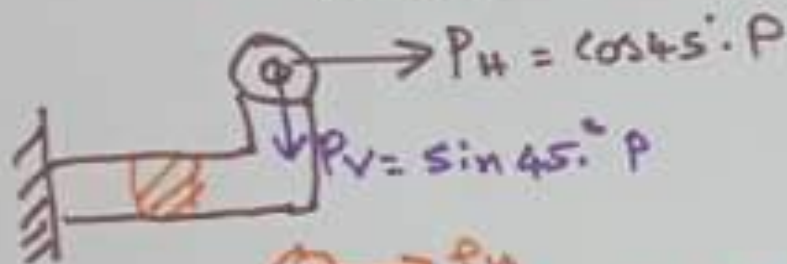
2 Bending					
$\sigma_b =$	M/Z				
M	Load x e				
		25000 N-mm			
$z =$	I/y				
	$I =$	$\pi * d^4 / 64$			
	$y =$	$d/2$			
	$I =$	0.0491 d^4	mm ⁴		
	$y =$	0.5 d	mm		
therefore	$z =$	0.09815625 d^3	mm ³		
Now	$\sigma_b =$	2.547E+05 $*(1/d^3)$	N/sq.mm		
	σ_y	2.547E+05 $*(1/d^3)$			

3 Shear stress		$T \times 16 / (\pi \times d^3)$			refer Pg.No 7.1/DDB
	T=		20000 N mm		
		(eccentric distance is horizontal distance from axis)			
τ Shear stress	τ	101878.3827 $\times (1/d^3)$ N/sq.mm			



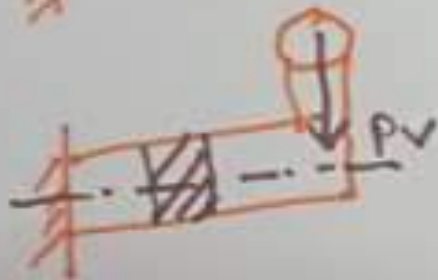
With respect to
x-x
axis.

Resolve it.



only Bending.

$$\sigma_{bH} = \frac{M_H}{Z} \quad (1)$$

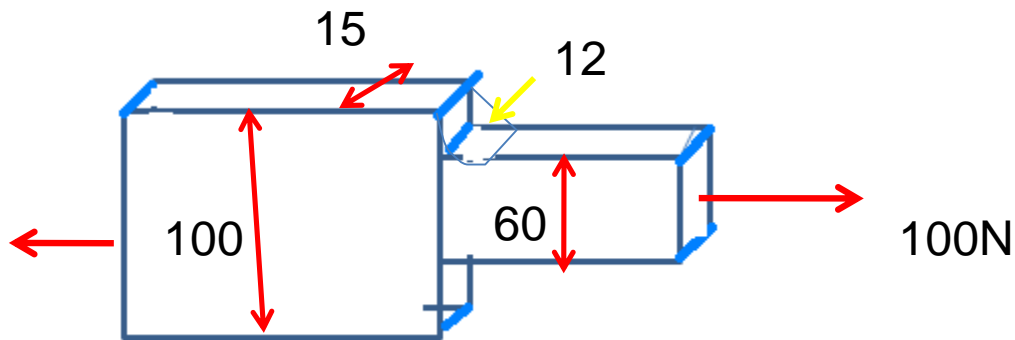
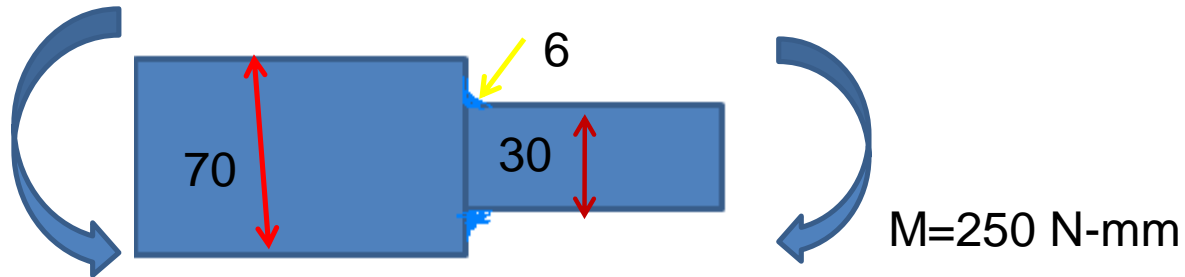


$$\sigma_d = \frac{P_V}{A} \quad (1)$$

$$\Rightarrow \sigma_{bV} = \frac{M_V}{Z} \quad (2)$$

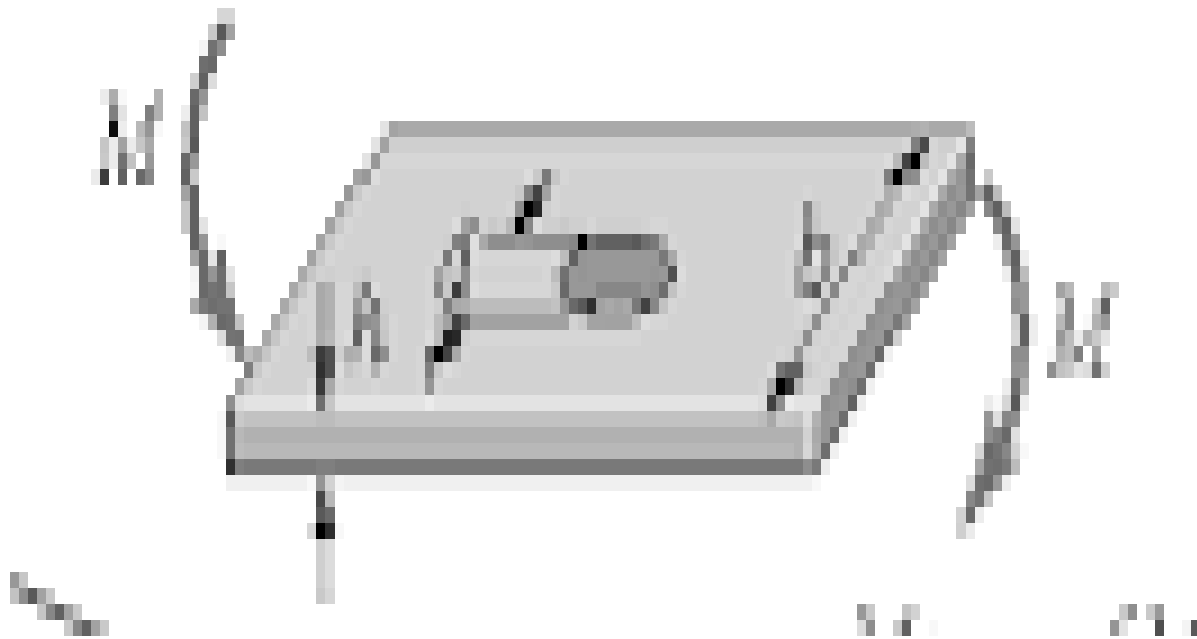
$$\sigma_{com} = \sigma_{bH} + \sigma_{bV} + \sigma_d$$

Tutorial 2 week2 20 marks credit



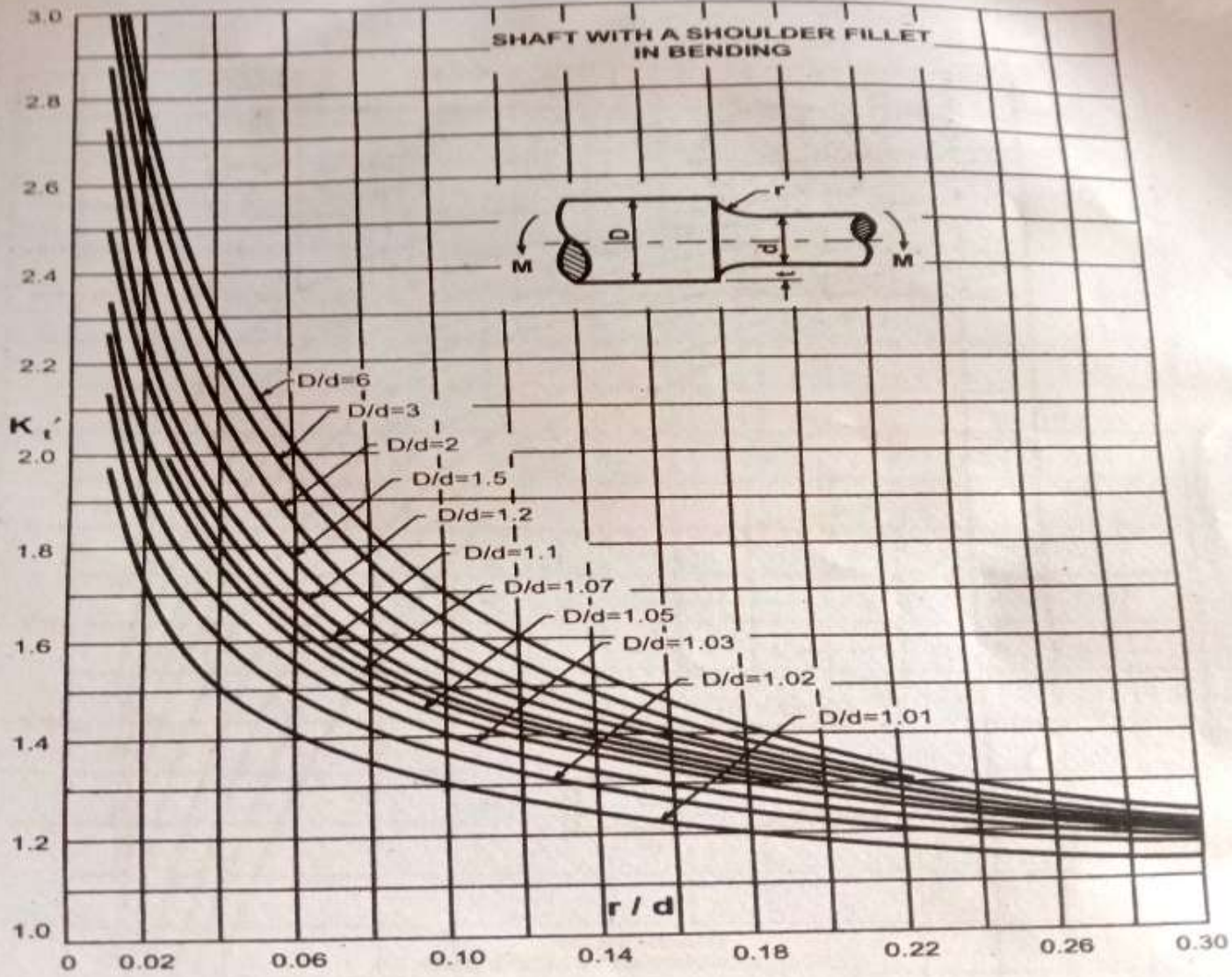
Determine the maximum stress induced in the members, all dimensions are in mm

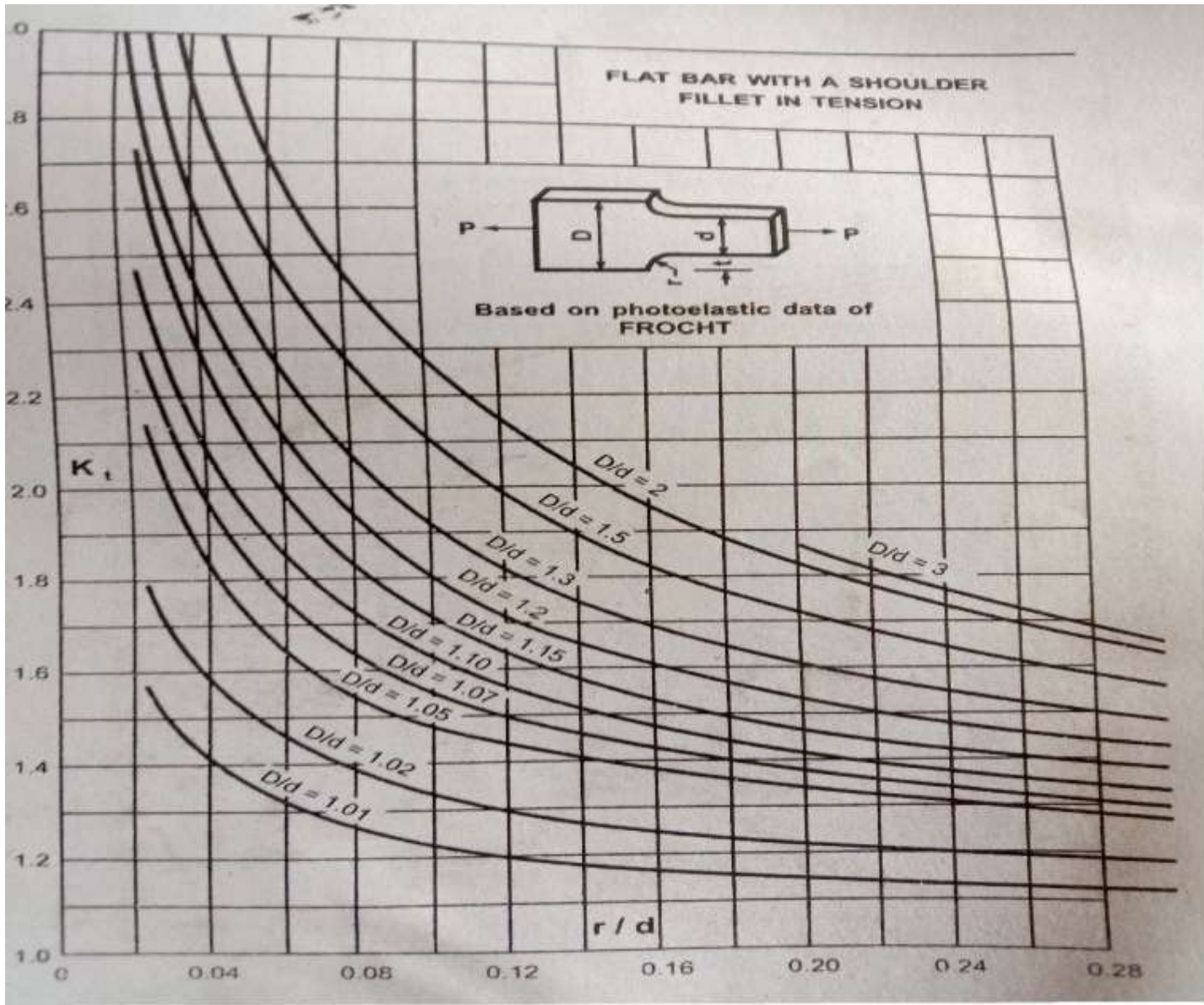
Tutorial 2 week2 20 marks credit



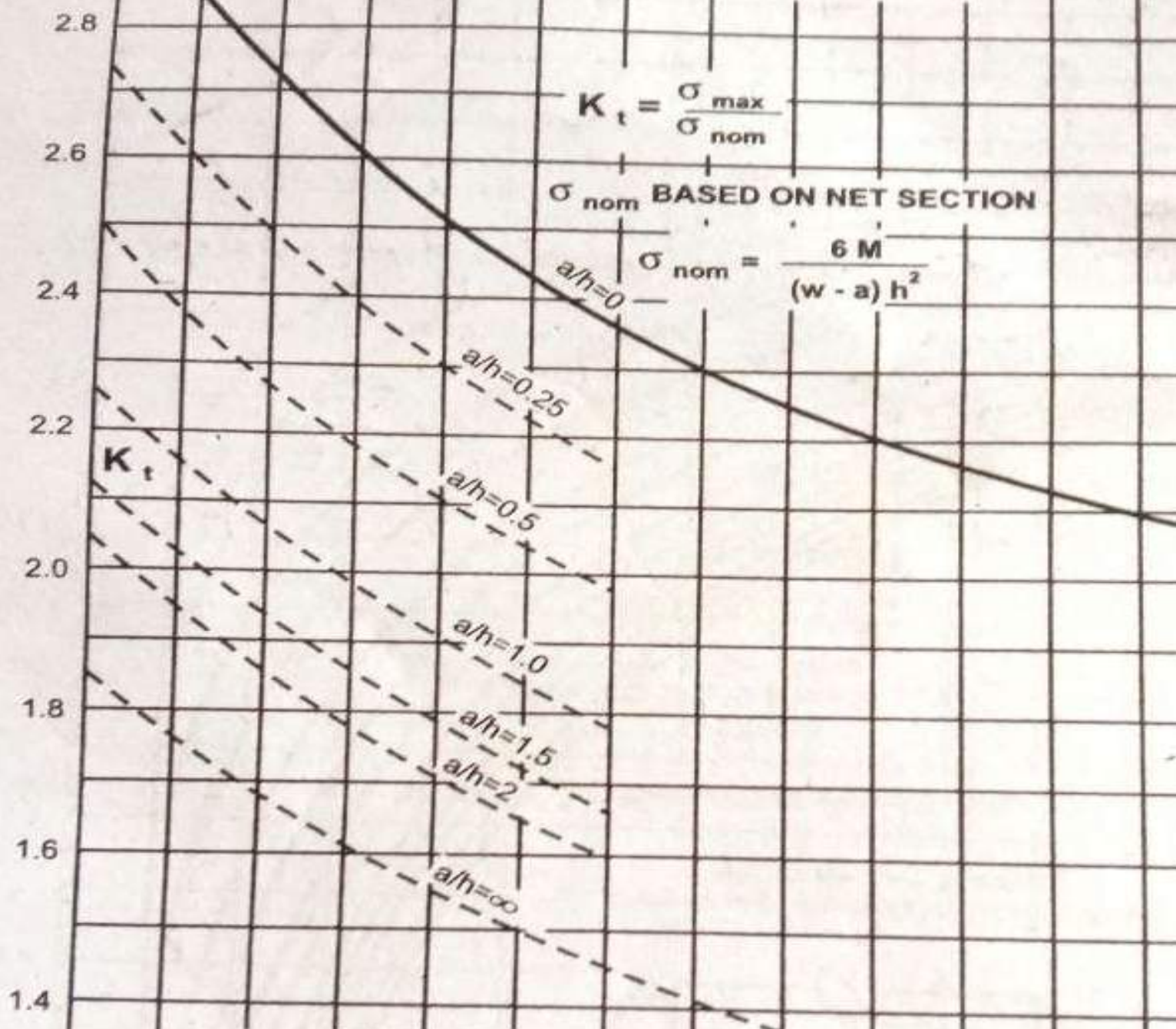
$M=150 \text{ N-mm}$

Width = 100, $a = 40$, $t = 20$ in mm





FINITE WIDTH PLATE WITH A TRANSVERSE HOLE IN BENDING



Principal stress			
$\sigma_1, \sigma_2 =$??		
$\sigma_x + \sigma_y$	$2.547E+05$	$*(1/d^3)N/sq.mm$	
$\sigma_x - \sigma_y$	$-2.547E+05$	$*(1/d^3)N/sq.mm$	
$(\sigma_x - \sigma_y)^2$	64870030360.2	$*(1/d^3)^2N/sq.mm$	
$(\sigma_x + \sigma_y)/2$	$1.273E+05$	$*(1/d^3)N/sq.mm$	
τ	101878.3827	$*(1/d^3)$	
τ^2	10379204857.6	$*(1/d^3)^2$	
$(\sqrt{(\sigma_x - \sigma_y)^2 + 4\tau^2})/2$	163084.985	$*(1/d^3)$	

Now			
$\sigma_1 =$	2.9043296E+05	$*(1/d^3)N/sq.mm$	
$\sigma_2 =$	-35737.007	$*(1/d^3)N/sq.mm$	

Apply the theories of failure

1. Max.Normal Stess Theory

Ref. Pg.7.3/DDB

$$\sigma_1 / \sigma_2 / \sigma_3 \leq \sigma_y / n$$

chose max value

$$\sigma_1 = \sigma_y / n$$



n	2.5	
Select material C45 ref. Pg.1.9/DDB (ASSUMED)		
$\sigma_y = 360$		N/mm ²
Use the equation 1		
$2.9043296E+05$	$360/2.5$	
$2.9043296E+05$	144	
$2.0168956E+03$,=	d ³
	$d ,=12.63\text{mm}$	

Now to check the design safe

Theory used is **Max .Normal Stress Theory**

Check for safe

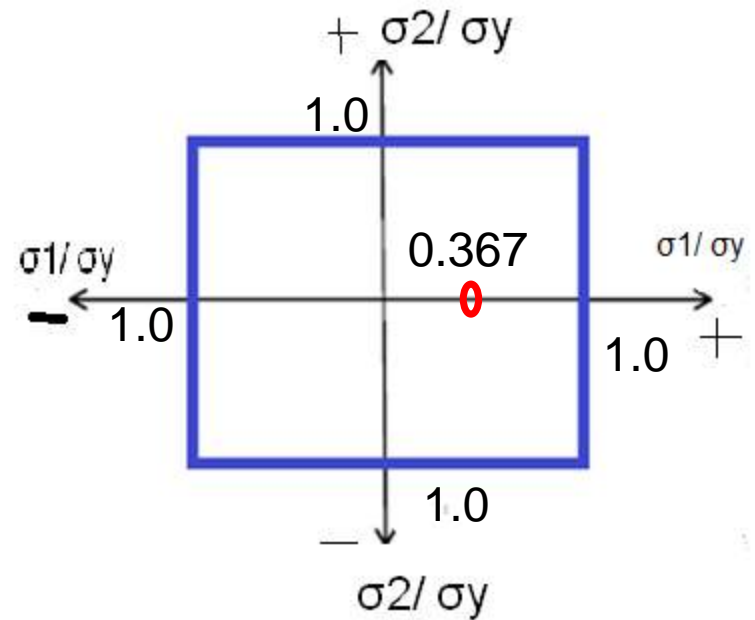
$$\sigma_1 = 2.9043296E+05 \quad *(1/d^3)N/sq.mm$$

substitute $d = 13 \text{ mm}$ (rounded off)

$$\sigma_1 = 1.3219525E+02 \text{ N /Sq.mm}$$

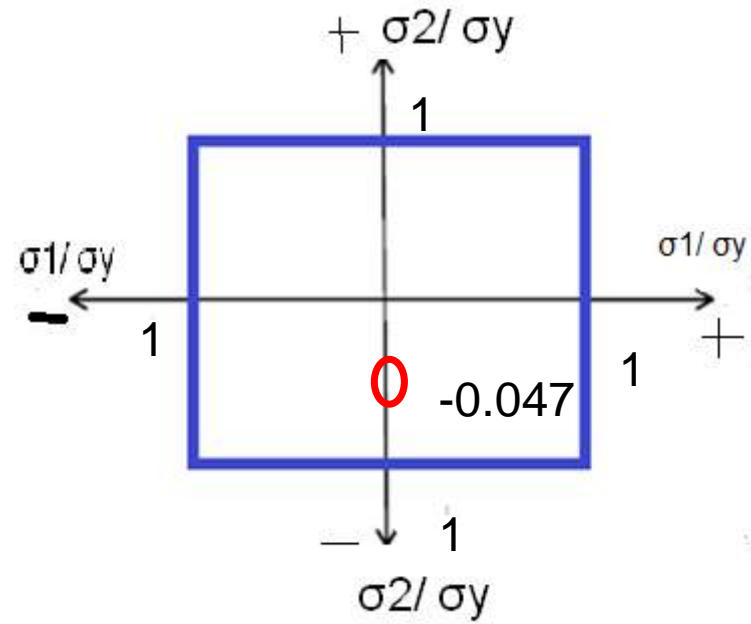
$$\sigma_y = 360 \text{ N/Sq.mm}$$

$$\sigma_1/\sigma_y = 3.7E-01 \quad (0.367)$$



As the ratio point lie in the region, **Design is safe**

$$\sigma_2/\sigma_y = -0.047$$



2Max Shear Stress Theory

Ref pg.7.3/DDB

$$\sigma_1 - \sigma_2 / \sigma_2 - \sigma_3 / \sigma_3 - \sigma_1$$

=

$$\sigma_y / 2n$$

$$\sigma_y = 360, n = 2.5$$

$$\sigma_1 - \sigma_2 = 3.2616997E+05 * (1/d^3)$$

$$\sigma_2 - \sigma_3 = -35737.007 * (1/d^3)$$

$$\sigma_3 - \sigma_1 = -2.9043296E+05 * (1/d^3)$$

Chose max, $\sigma_1 - \sigma_2$

$$3.261699x 10^5 / d^3 = 72$$

$$d^3 = 4.5301385E+03$$

$$d = 16.54 \text{ mm}$$

Similar way find the diameter by the other theories also check for the design safe

Dynamic Design

Fatigue & Endurance strength SN – Curve

Theories

Gerber equation

Goodman equation

Soderberg Equation

Solving the problem

To understand cyclic stress



Try to break it



1



2



Cyclic Stresses

General terms

ref.Pg.7.6/DDB

FATIGUE

Cyclic Stresses

Parameters used to characterize the fluctuating stress cycle:

1. Mean Stress (σ_m):

$$\sigma_m = \frac{\sigma_{\max} + \sigma_{\min}}{2}$$

4. Stress Ratio (R):

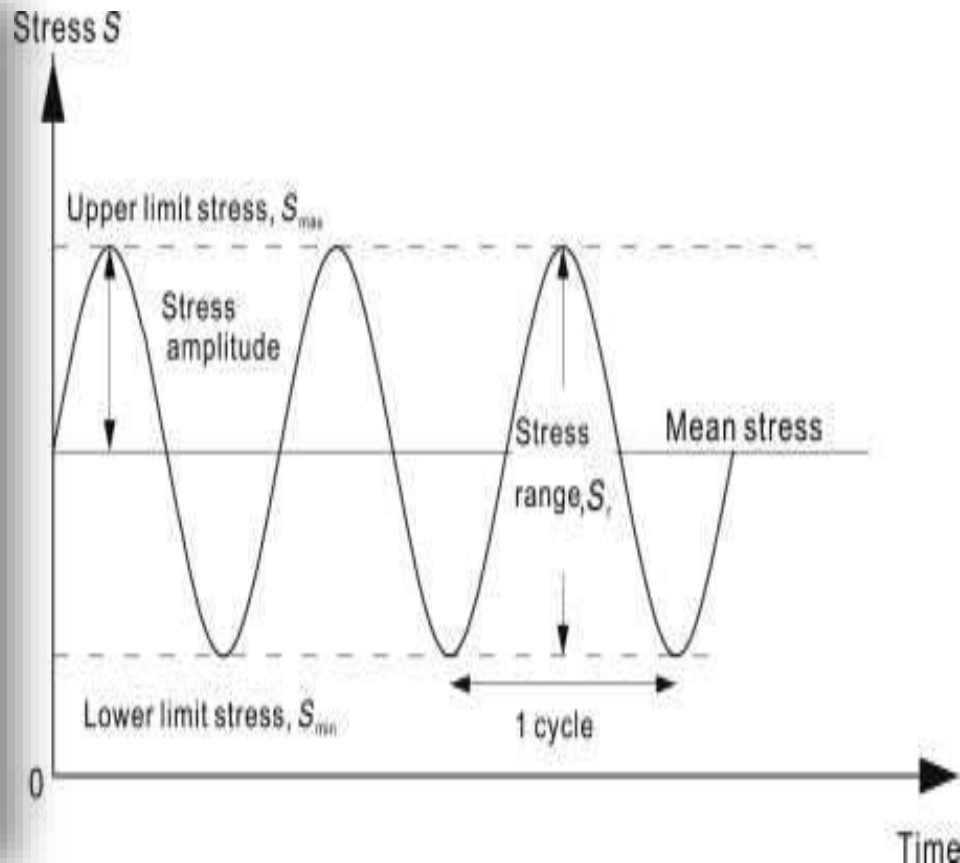
$$R = \frac{\sigma_{\min}}{\sigma_{\max}}$$

2. Range of Stress (σ_r):

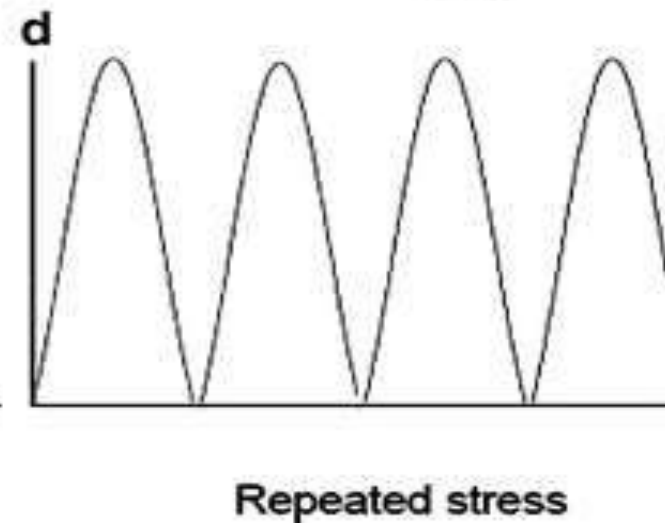
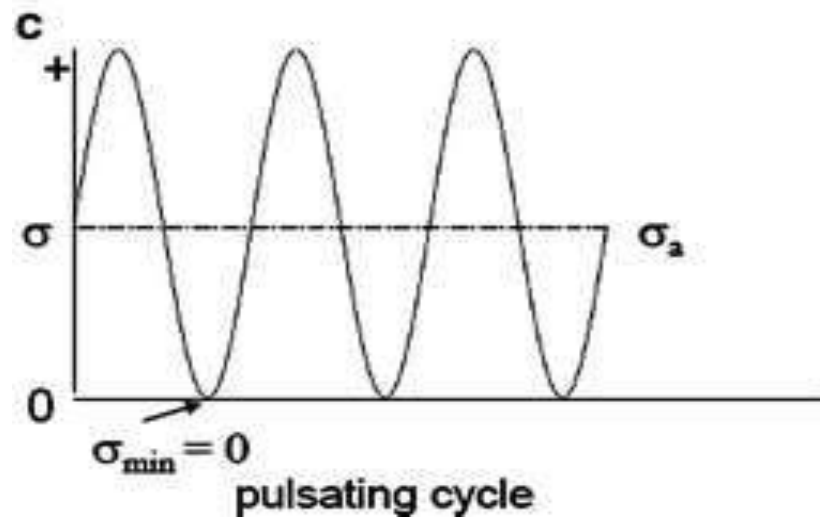
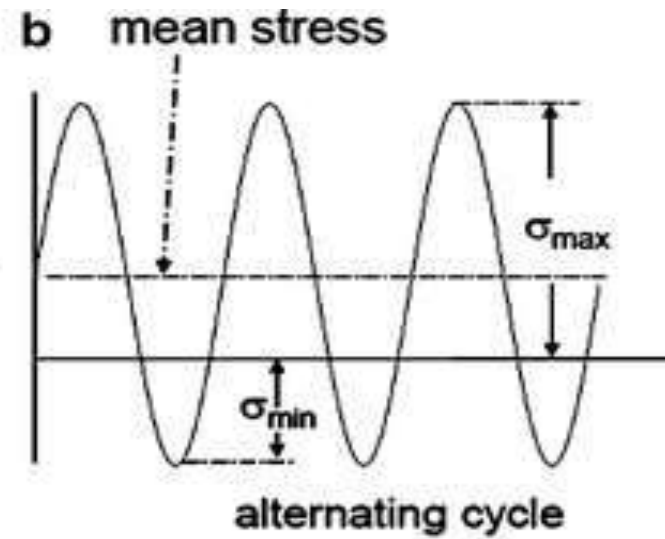
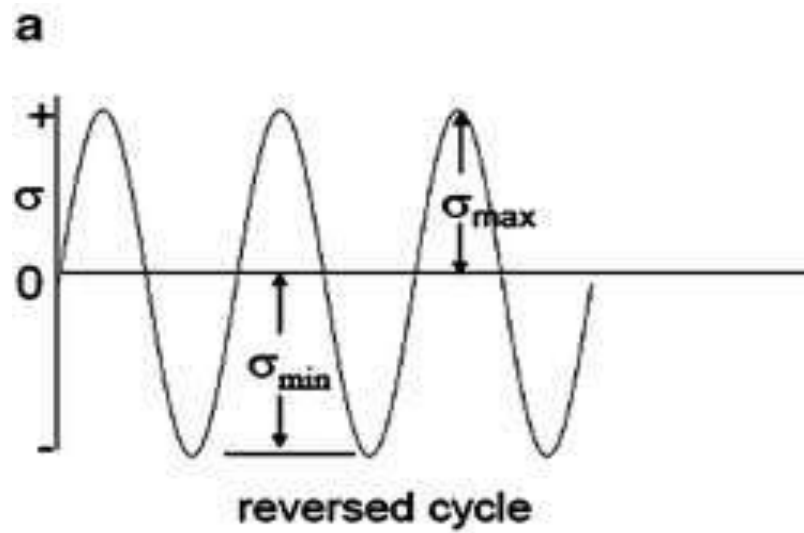
$$\sigma_r = \sigma_{\max} - \sigma_{\min}$$

3. Stress Amplitude (σ_a):

$$\sigma_a = \frac{\sigma_r}{2} = \frac{\sigma_{\max} - \sigma_{\min}}{2}$$



Cyclic stresses



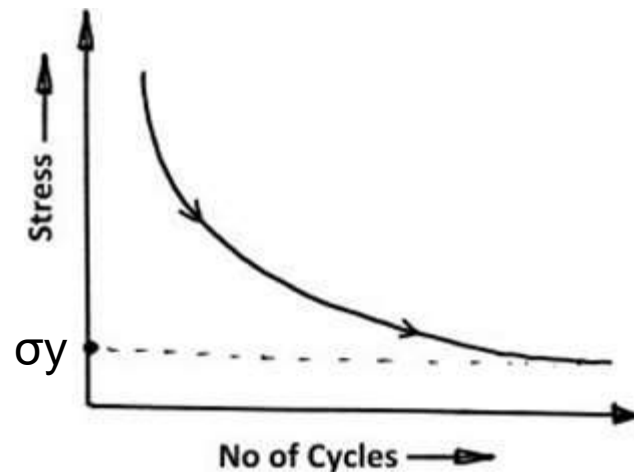
What is fatigue?

When a member subject to cyclic stresses , it fails below its yield stress value known as “fatigue”.

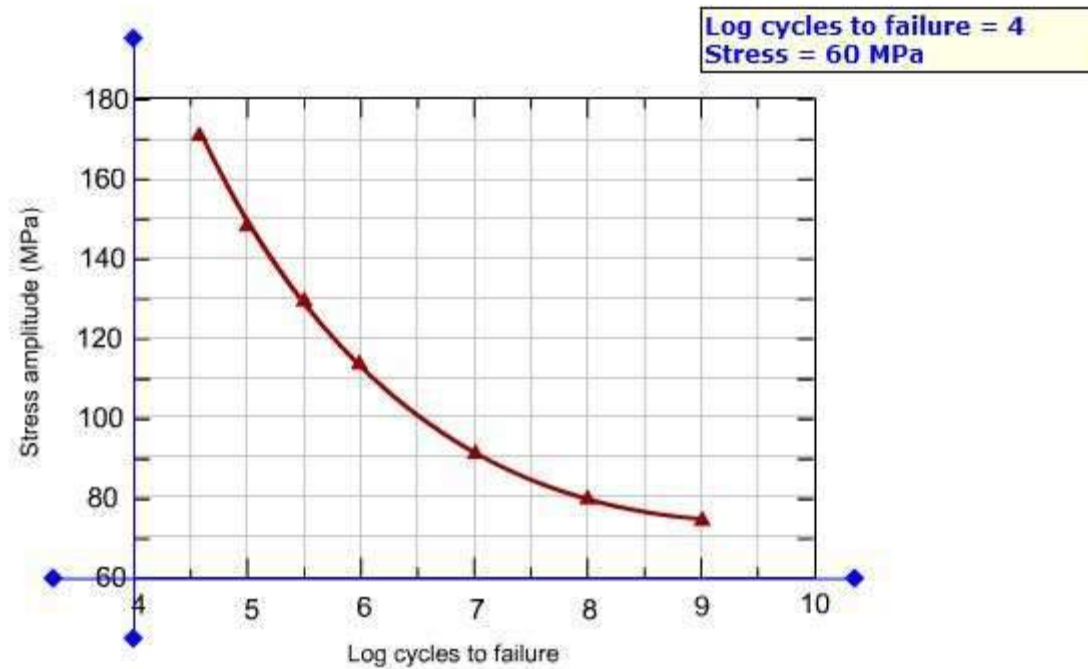
It is unpredictable, - its so dangerous. It occurs due to flaws in the member(under micro study), which initiates the crack formation, then propagation and finally fracture.

Endurance or fatigue limit?

A member will undergo n numbers of cycles. With out failure/fracture at a particular stress value. This stress is known as Endurance or fatigue limit.



S-N diagram – example to predict the stress values With respect to cycles



Fatigue stress concentration factor K_f

Def: Experimental definition only available

Fatigue stress concentration factor,

$$K_f = \frac{\text{Endurance limit without stress concentration}}{\text{Endurance limit with stress concentration}}$$

Notch sensitivity “q”

Def:

It is defined as the degree of attaining the K_t - theoretical stress concentration. q is estimated with some experimental curves. No extensive data available.

Ref. Pg.no:7.8/DDB

Notch sensitivity q index is defined by

$$q = \frac{K_f - 1}{K_t - 1} \quad \text{or} \quad q_{shear} = \frac{K_{fs} - 1}{K_{ts} - 1} \rightarrow K_f = 1 + q(k_t - 1)$$

STRESS CONCENTRATION FACTORS

Stress concentration factor } $K_t = \frac{\text{maximum stress}}{\text{nominal stress at the net section}}$

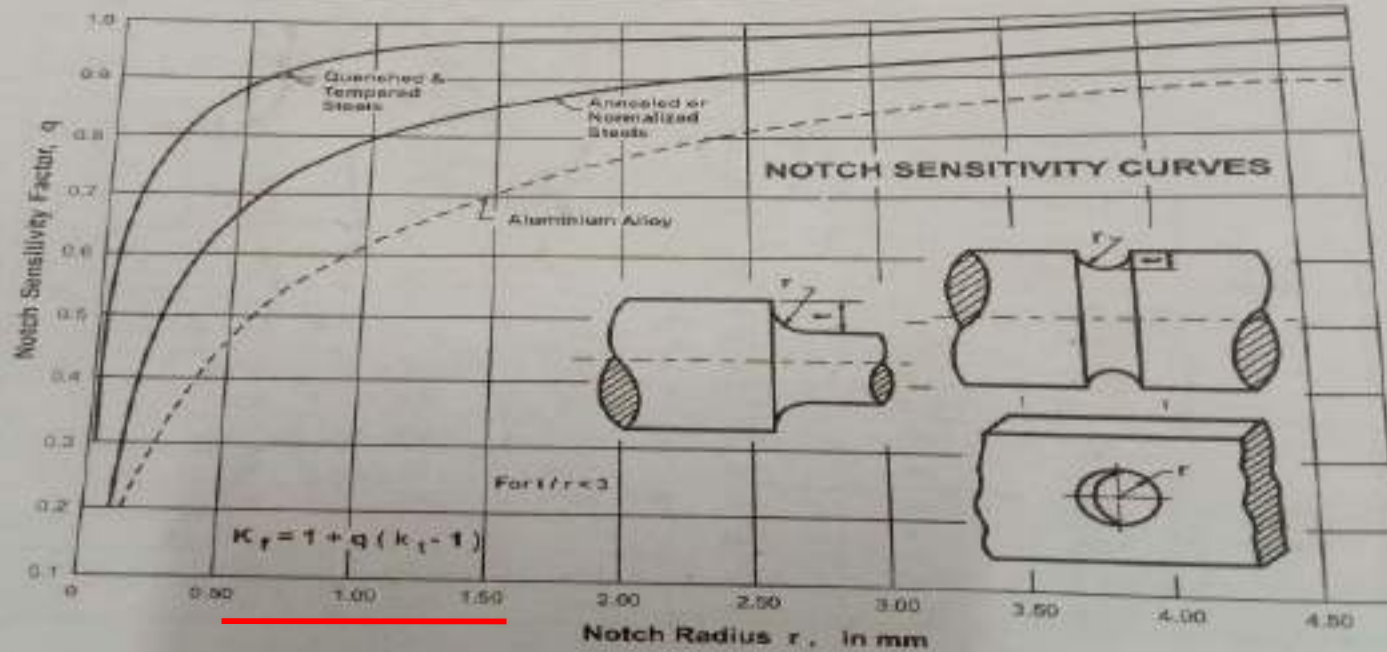
K_{ts} - combined stress concentration factor, taking into account biaxial or triaxial stresses and Von Mises - Hencky theory.

K_t is always greater than K_{ts} , therefore where information is lacking or doubtful, K_t should be used, to be on the safe side.

VALUES OF K_t FOR KEYWAYS



KIND OF KEYWAY	Annealed		Hardened	
	Bending	Torsion	Bending	Torsion
PROFILE	1.6	1.3	2.0	1.6
SLED RUNNER	1.3	1.3	1.6	1.6



Problem for finding “Kf”



Find the K_f of the member which is made of steel and annealed.

SOLUTION

The equation,

$$K_f = 1 + q(kt - 1)$$

Ref. Pg.no:7.8/DDB

1. $K_t = ?$ 2. $q = ?$

Ref. Pg.no:7.11/DDB

$$r/d = 0.2,$$

$$D/d = 2.33(2.0)$$

$$K_t = 1.5$$

Ref. Pg.no:7.8/DDB

$$r = 6 \text{ (4.5), material steel \& annealed}$$

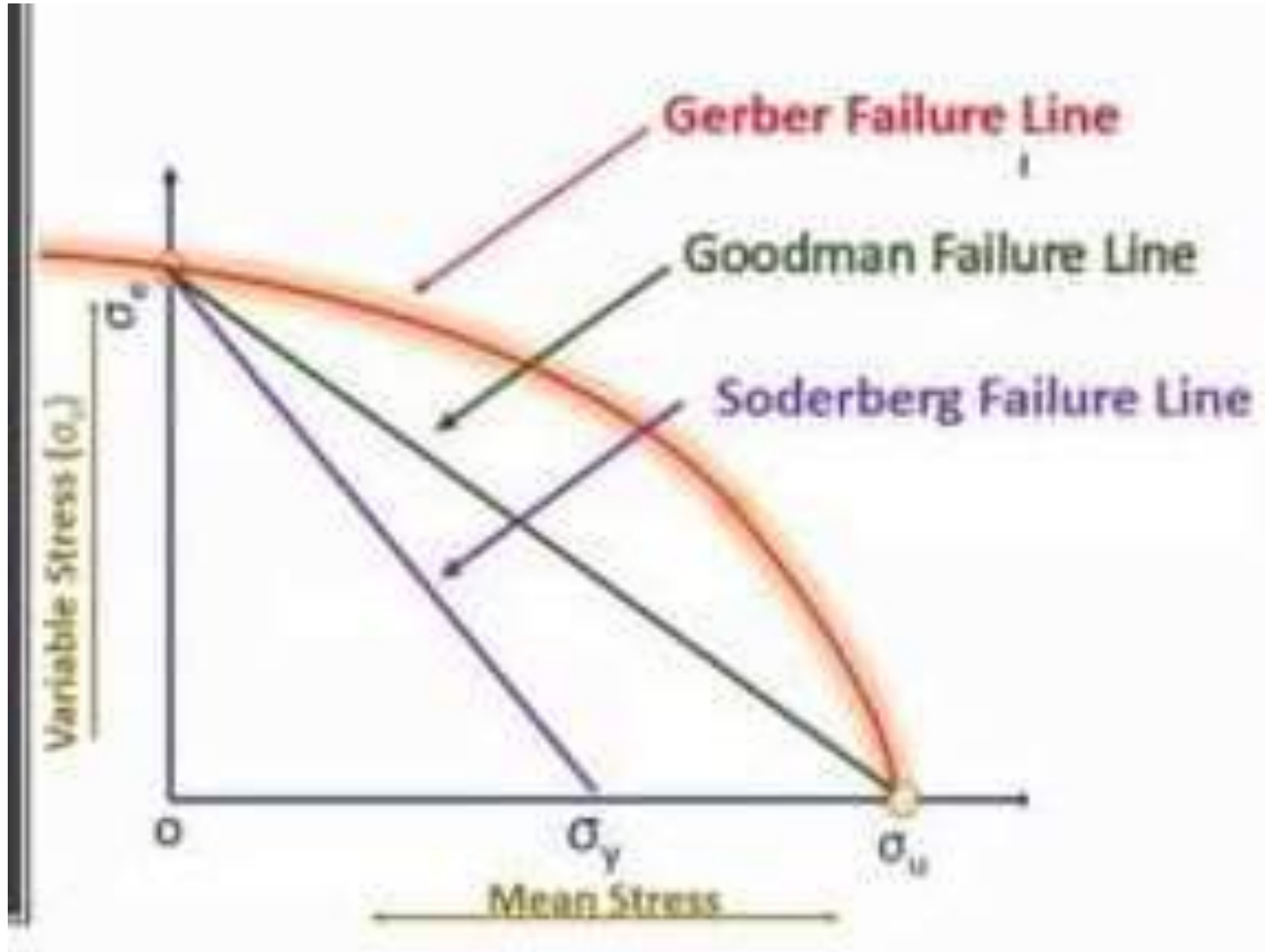
$$q = 0.95$$

Now

$$K_f = 1.475$$

Theories for Varying stresses

Refer pg.No.7.4 & 7.6 /DDB

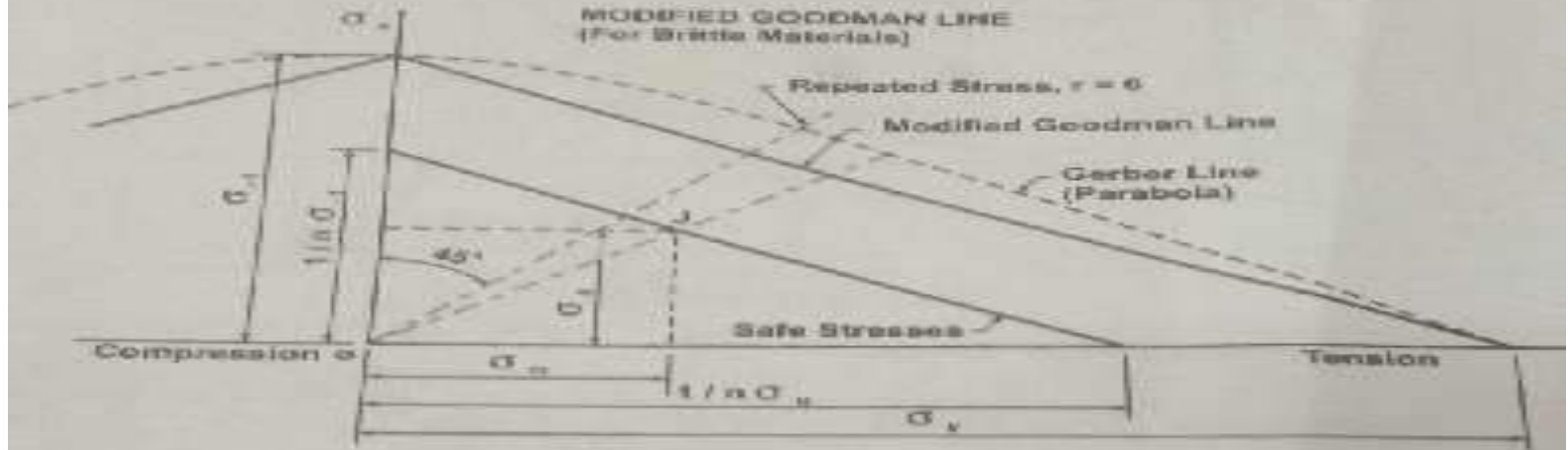


VARIOUS STRESSES

SODERBERG LINE (Recommended for ductile materials only)



MODIFIED GOODMAN LINE (For Brittle Materials)



Soderberg Equations

$\frac{1}{n} = \frac{\sigma_m}{\sigma_y} + \frac{\sigma_v}{\sigma_u}$	$\frac{1}{n} = \frac{\tau_m}{\tau_y} + \frac{\tau_v}{\tau_u}$
---	---

Goodman Equations

$\frac{1}{n} = \frac{\sigma_m}{\sigma_u} + \frac{\sigma_v}{\sigma_y}$	$\frac{1}{n} = \frac{\tau_m}{\tau_u} + \frac{\tau_v}{\tau_y}$
---	---

SODERBERG Equation

$$\frac{1}{n} = \frac{\sigma m}{\sigma y} + \frac{\sigma a}{\sigma - 1}$$

$$\frac{1}{n} = \frac{\tau m}{\tau y} + \frac{\tau a}{\sigma - 1}$$

Goodman equation

$$\frac{1}{n} = \frac{\sigma m}{\sigma u} + \frac{\sigma a}{\sigma - 1}$$

$$\frac{1}{n} = \frac{\tau m}{\tau u} + \frac{\tau a}{\sigma - 1}$$

Problem Objectives

- 1. Find factor of safety “n”**
- 2. Find size of the member**

Steps to solve the problem

Step1 Find the mean and amplitude loads

$$(W_{\max} + W_{\min})/2 = W_{\text{mean}} (W_m)$$

$$(W_{\max} - W_{\min})/2 = W_{\text{ampl}} \text{ or } W_a$$

(Amplitude or varying Load)

Step2 Find the mean and amplitude Stresses

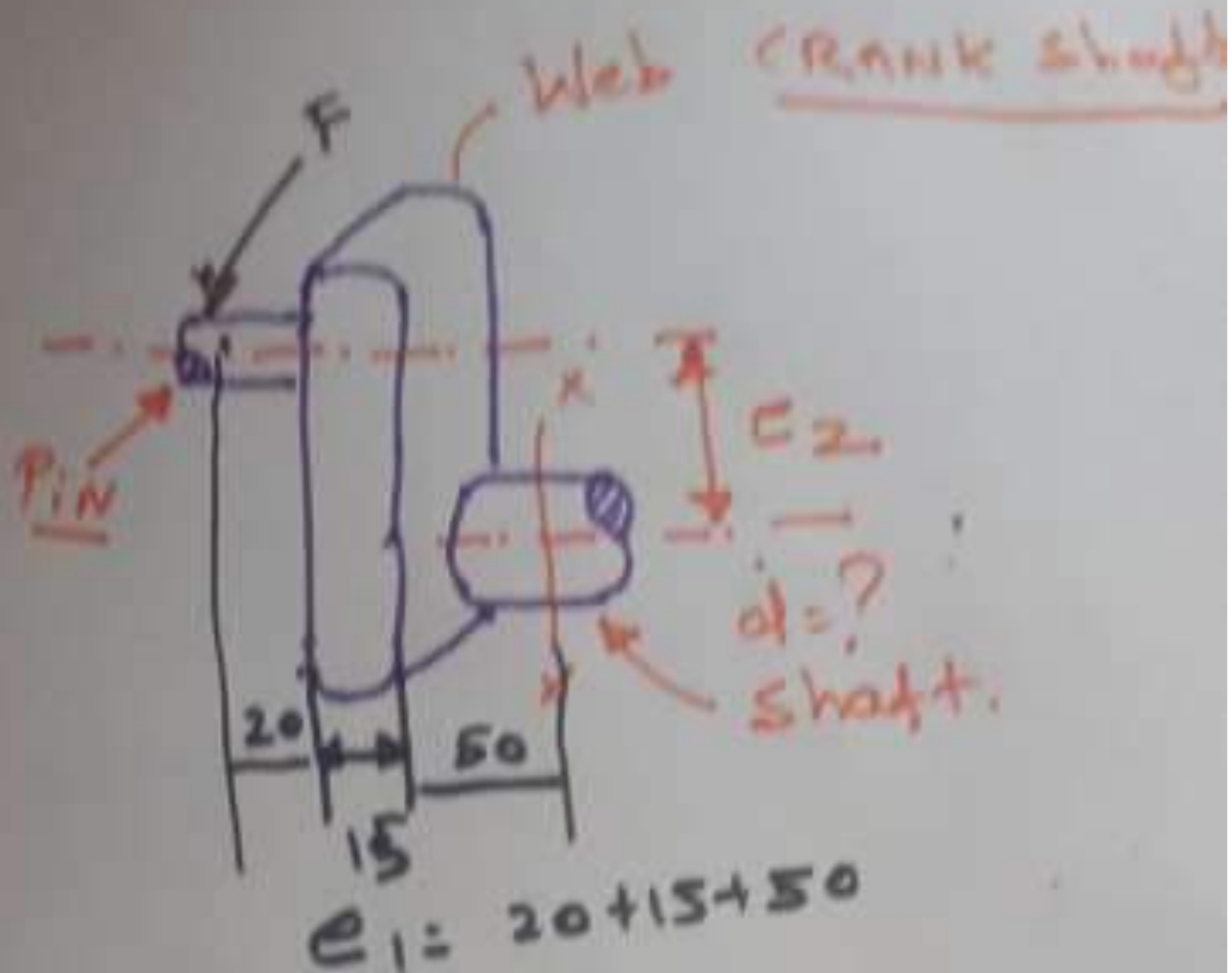
$$\sigma_{\text{mean}} = W_{\text{mean}} / \text{Area} ,$$

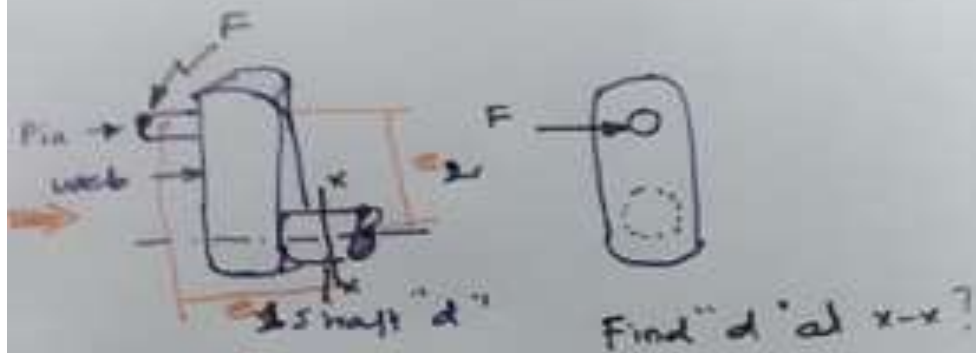
$$\sigma_a \text{ or } \sigma_v = W_a \text{ or } W_v / \text{Area}$$

Step3 Use the theories for varying stress

Find the **size** or **factor of safety**

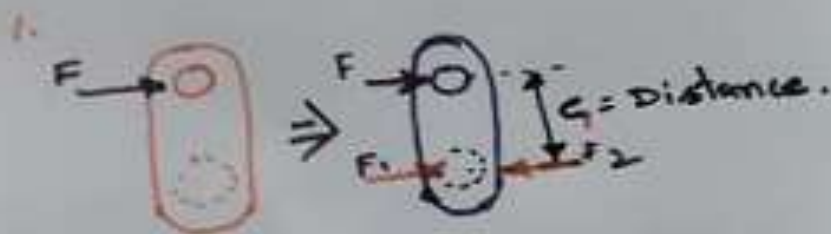
which ever is applicable with respect to the problem



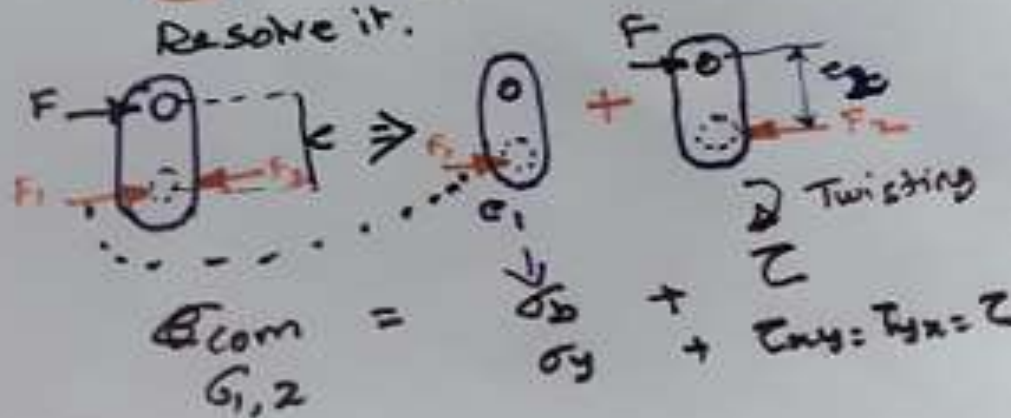


Solution

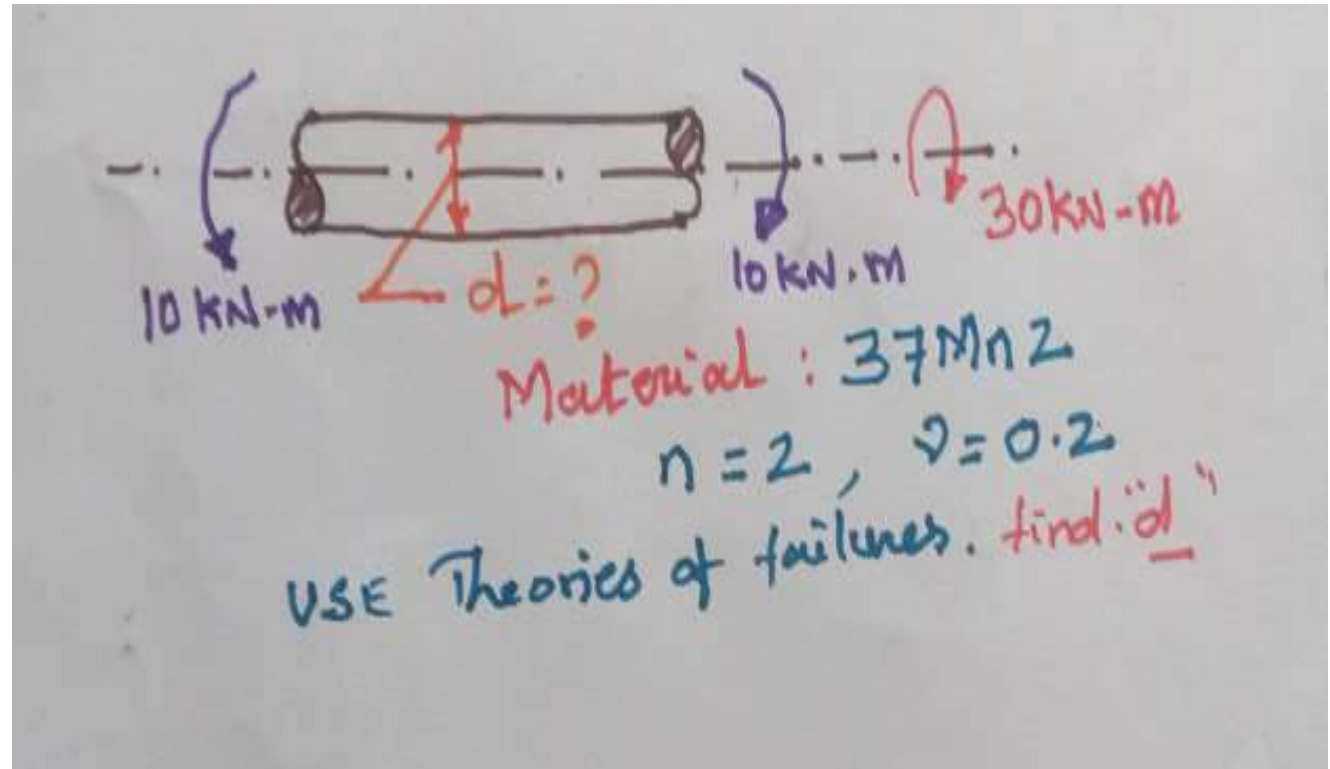
• Eccentric Load Concept.



Resolve it.



Tutorial 3 - Topic theories of failure

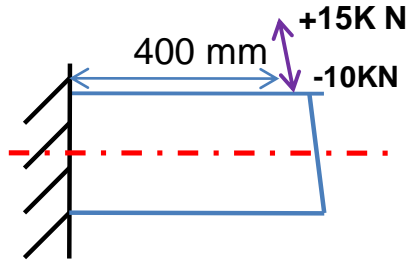


COMPOSITION AND RELATED MECHANICAL PROPERTIES OF ALLOY STEELS

Designation	% C	% Mn	% Ni	% Mo	% Cr	% Cu	Tensile strength kg/cm ²	Yield strength kg/cm ²	Elongation 1/4 in. gage length	Reduction of area 1/4 in. gage length
20 Mn 2 ^a	0.20-0.24	0.3-0.60	-	-	-	-	60-70	30	25	15
							Temp. (°C)			
27 Mn 2 ^a	0.20-0.27	0.3-0.70	-	-	-	-	60-70	30	25	15
							Temp. (°C)			
37 Mn 2 ^a	0.22-0.30	0.3-0.80	-	-	-	-	60-70	30	25	15
							Temp. (°C)			
40 Mn 2 30 Cr 2 ^a	0.3-0.4	0.3-0.80	-	-	1.7-1.8	-	60-70	30	25	15
							Temp. (°C)			
40 Mn 2 30 Cr 4 ^a	0.3-0.4	0.3-0.80	-	-	1.8-1.9	-	60-70	30	25	15
							Temp. (°C)			
40 Cr 2 ^a	0.20-0.30	0.3-0.80	-	-	-	-	60-70	30	25	15
							Temp. (°C)			

^a Values for Hardening and Tempering ^b Values for Normalizing ^c Values for Normalizing
^d A, based on 2.25% carbon ^e μ = 1070

Pg. 1.13 /DDB, alloyed steel



Determine the factor of safety for the member having diameter 50 mm which is subjected to fluctuating load from -10 to +15 kN. The ultimate and yield stress values are 60 N/mm² and 40 N/mm² respectively. Use the Goodman and Soderberg theories

Data:

$$n = ?$$

$$D = 50 \text{ mm}$$

$$\text{Max. load} = 15000 \text{ N}$$

$$\text{Min. load} = -10000 \text{ N}$$

$$\text{Ultimate} = \sigma_u = 60 \text{ N/mm}^2$$

$$\text{Yield} = \sigma_y = 40 \text{ N/mm}^2$$

Soderberg

$$\frac{1}{n} = \frac{\sigma m}{\sigma_y} + \frac{\sigma a}{\sigma - 1}$$

Goodman

$$\frac{1}{n} = \frac{\sigma m}{\sigma u} + \frac{\sigma a}{\sigma - 1}$$

Steps to solve the problem

Step1 Find the mean and amplitude loads

$$(W_{\max}+W_{\min})/2= W_{\text{mean}} (W_m)$$

$$(W_{\max}-W_{\min})/2= W_{\text{ampl}} \text{ or } W_a$$

(Amplitude or varying Load)

$$W_{\max} = 15000 \text{ N}$$

$$W_{\min} = -10000 \text{ N}$$

$$W_m \qquad 2500\text{N} \qquad (15000-10000)/2= W_{\text{mean}} (W_m)$$

$$W_a \qquad 12500\text{N} \qquad (15000+10000)/2= W_{\text{ampl}} \text{ or } W_a$$

Step2 Find the mean and amplitude Stresses

$$\sigma_{\text{mean}} = W_{\text{mean}} / \text{Area}$$

$$\text{Area} = 1963.1 \text{mm}^2$$

$$\sigma_m = 1.3 \text{N/mm}^2$$

$$\sigma_a \text{ or } \sigma_v = W_a \text{ or } W_v / \text{Area}$$

$$\sigma_v = 6.4 \text{N/mm}^2$$

Step3 Use the theories for varying stress

$$1/n = (1.27348/40) + (6.3674/\sigma_1)$$

Now to find σ_1

Refer pg.1.42/DDB

Take reversed cycle

Bending Load $\sigma_1 = 0.46\sigma_u$

$$\sigma_1 = 27.6\text{N/mm}^2$$

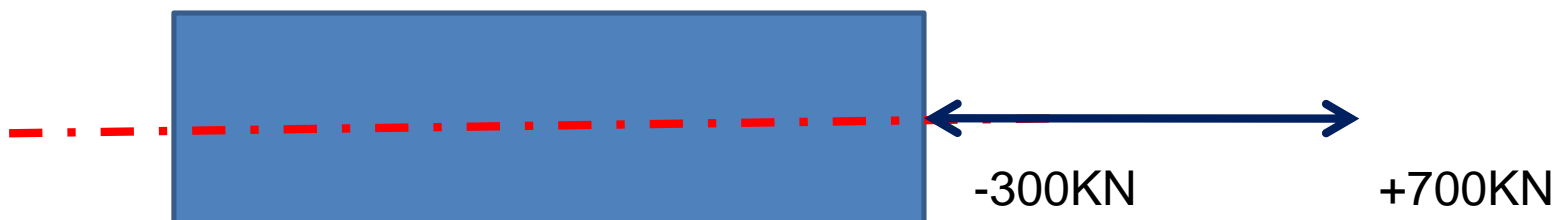
$$1/n = 0.031836995 + 0.230703$$

$$1/n = 0.262539854$$

$$n = 3.808945518$$

G&S Probm2

Determine the diameter of circular shaft subjected to axial load of range from - 300KN to 700 KN, Use factor of safety as 2.0.



NO yield, No Ultimate stress given, hence select suitable material
Then find the both stress values

Material : C50, refer pg.1.9, $\sigma_y = 380 \text{ N/mm}^2$ & $\sigma_u = 730 \text{ N/mm}^2$

G&S Prob2

DATA

$$n = 2$$

$$W_{\max} = 700000\text{N}$$

$$W_{\min} = -300000\text{N}$$

$$\sigma_u = 730\text{N/mm}^2$$

$$\sigma_y = 380\text{N/mm}^2$$

$$\sigma_1 = ?$$

Soderberg

$$\frac{1}{n} = \frac{\sigma_m}{\sigma_y} + \frac{\sigma_a}{\sigma - 1}$$

Goodman

$$\frac{1}{n} = \frac{\sigma_m}{\sigma_u} + \frac{\sigma_a}{\sigma - 1}$$

Step 1 Calculating **Wm**(mean) and amplitude **Wa**

Wm **200000N** $(W_{\max}+W_{\min})/2= W_{\text{mean}} (W_m)$

Wa **500000N** $(W_{\max}-W_{\min})/2= W_{\text{ampl}} \text{ or } W_a$

Step2 Find the mean and amplitude Stresses

$$\sigma_{\text{mean}} = W_{\text{mean}} / \text{Area}$$

$$\text{Area} = 0.78525 d^2 \text{ mm}^2$$

$$\sigma_m = 254695.96 / d^2 \text{ N/mm}^2$$

$$\sigma_a \text{ or } \sigma_v = W_a \text{ or } W_v / \text{Area}$$

$$\sigma_v = 636739.89 / d^2 \text{ N/mm}^2$$

Step3 Use the theories for varying stress

$$1/2.0 = [254695.96/(d^2*380) + (636739.89/(d^2*\sigma_1))]$$

Now to find σ_1

Refer pg.1.42/DDB

Take reversed cycle

Load: Tension- compression

$$\sigma_1 = 0.36\sigma_u$$

262.8

$$\sigma_1 = 335.8 \text{ N/mm}^2$$

$$1/2.0 = 670.2525176 \cdot (1/d^2) + 1896 \cdot (1/d^2)$$

$$1/2.0 = 2566.440403/d^2$$

$$d^2 = 5132.880805$$

$$d = 71.6 \text{ mm}$$

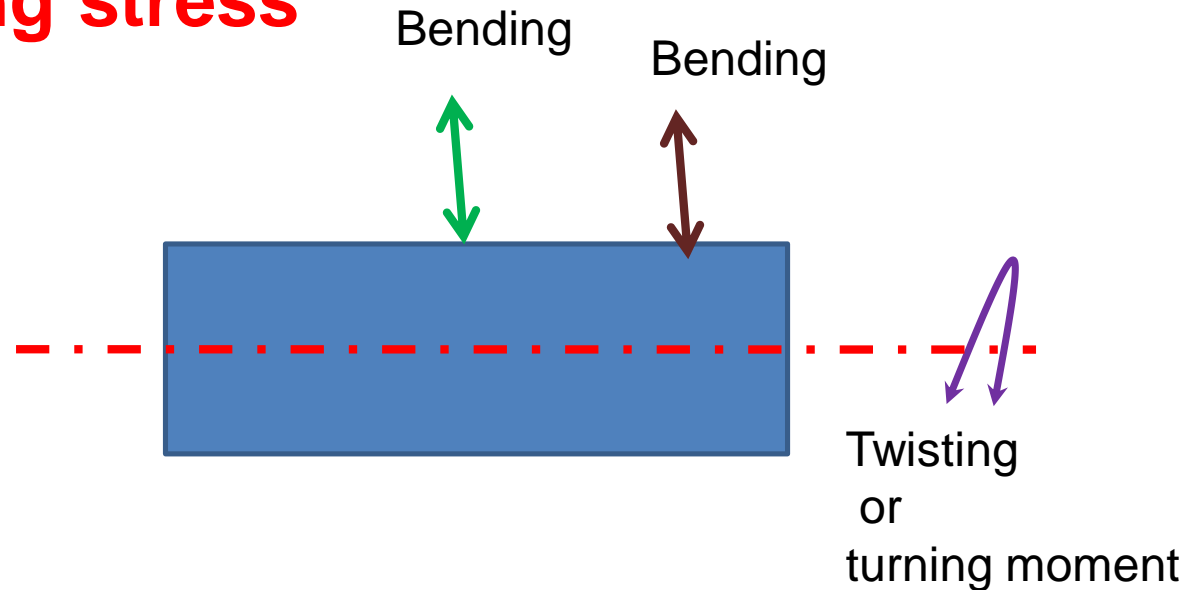
$$72 \text{ mm (78.65)}$$

mm

Right answer: 78.65 mm = 79 mm

Combined varying stress

Refer Pg.7.6/DDB



$$1/n = \left[\left(\frac{\sigma_{eq}}{\sigma_y} \right)^2 + \left(\frac{\tau_{eq}}{\tau_y} \right)^2 \right]^{1/2}$$

Required Task , to calculate σ_{eq} & τ_{eq} ,

Again refer Pg.7.6/DDB

Soderberg Equations

$$\frac{1}{n} = \frac{\sigma_m}{\sigma_y} + K_f \frac{\sigma_a}{\sigma_{-1}} \quad \left| \quad \frac{1}{n} = \frac{\tau_m}{\tau_y} + K_f \frac{\tau_a}{\tau_{-1}}\right.$$

For ductile materials

Goodman (modified) equation

$$\frac{1}{n} = K_f \left[\frac{\sigma_m}{\sigma_u} + \frac{\sigma_a}{\sigma_{-1}} \right] \quad \left| \quad \frac{1}{n} = K_f \left[\frac{\tau_m}{\tau_u} + \frac{\tau_a}{\tau_{-1}} \right]\right.$$

For brittle materials (notch sensitivity negligible)

Combined Stresses

$$\sigma_{eq} = \frac{\sigma_a}{n} = \sigma_m + K_f \frac{\sigma_a \sigma_f}{\sigma_{-1}} \quad \left| \quad \tau_{eq} = \frac{\tau_a}{n} = \tau_m + K_f \frac{\tau_a \tau_f}{\tau_{-1}}\right.$$

$$\frac{1}{n} = \left[\left(\frac{\sigma_{eq}}{\sigma_y} \right)^2 + \left(\frac{\tau_{eq}}{\tau_y} \right)^2 \right]^{1/2}$$

Max. shear theory : $\tau_y = \sigma_y/2$

Octahedral shear theory : $\tau_y = \sigma_y/\sqrt{3}$

Steps to solve the problem

Step1 Find the mean and amplitude Moments (bending & twisting)

$$(M_{\max}+M_{\min})/2= M_{\text{mean}} (M_m)$$

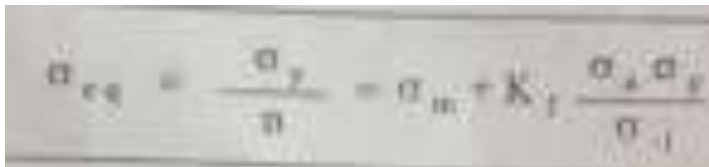
$$(M_{\max}-M_{\min})/2= M_{\text{ampl}} \text{ or } M_a \\ (\text{Amplitude or varying Load})$$

Step2 Find the mean and amplitude Stresses (bending & twisting)

$$\sigma_{b_{\text{mean}}} = M_{\text{mean}} / Z , \quad \tau_{\text{mean}} = 16 \times T_m / (\pi \cdot d^3)$$

$$\sigma_{b_a} \text{ or } \sigma_v = M_a \text{ or } M_v / Z \quad \tau_a = 16 \times T_a / (\pi \cdot d^3)$$

Step3 Calculating equivalent σ_e & τ_{eq} varying stress

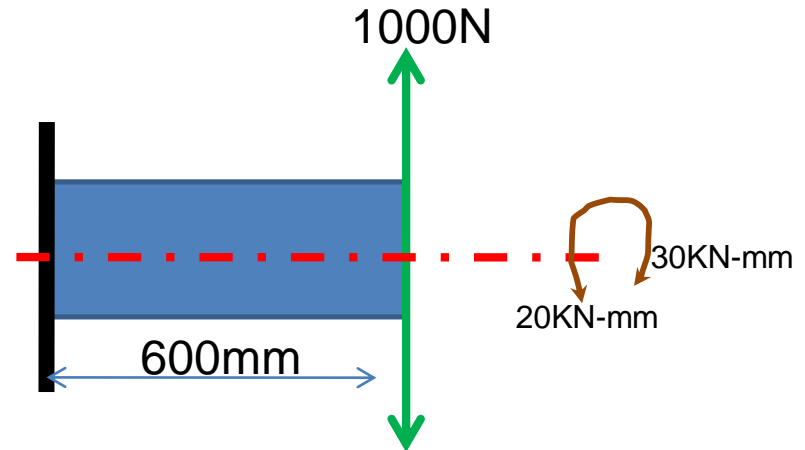

$$\sigma_{eq} = \frac{\sigma_v}{n} = \sigma_m + K_f \frac{\sigma_a \cdot \sigma_v}{\sigma_v}$$


$$\tau_{eq} = \frac{\tau_v}{n} = \tau_m + K_f \frac{\tau_a \cdot \tau_v}{\tau_v}$$

Step4 : Use of Soderberg equation, **Find the size or factor of safety**

$$\frac{1}{n} = \left[\left(\frac{\sigma_{eq}}{\sigma_y} \right)^2 + \left(\frac{\tau_{eq}}{\tau_y} \right)^2 \right]^{1/2}$$

Combined stress Problem1



Determine the diameter of the circular member which is subjected to loads as shown in figure. Take $n=2.0$

DATA

$d = ?$

$n=2.0$

Bending load = +1000N to - 700N

Twisting Moment = + 30KN-mm to -20KN-mm

Use combined varying stress equation to solve

Other works

Material Assumption
List yield & ultimate
Stresses

σ_1 & τ_1 – Ref.
DDB_Pg.1.42

APPROXIMATE VALUES OF ENDURANCE LIMIT FOR STEEL

Loading	Reversed cycle	Repeated cycle
Tension - compression	$\sigma_{-1t} = 0.36 \sigma_u$	$\sigma_{0t} = 0.5 \quad \sigma_u \leq \sigma_y$
Bending	$\sigma_{-1b} = 0.46 \sigma_u$	$\sigma_{0b} = 0.6 \quad \sigma_u \leq \sigma_y$
Torsion	$\tau_{-1} = 0.22 \sigma_u$	$\tau_0 = 0.3 \quad \sigma_u \leq \sigma_y$

APPROXIMATE RELATIONSHIP BETWEEN ENDURANCE LIMITS FOR DIFFERENT MATERIALS

Material	Relationship		
Steels (Generally)	$\sigma_{-1t} = (0.7 \text{ to } 0.8) \sigma_{-1b}$		$\tau_{-1} = (1.55 \text{ to } 0.58) \sigma_{-1b}$
Carbon Steels	$\sigma_{0b} = 1.5 \sigma_{-1b}$	$\sigma_{0t} = 1.6 \sigma_{-1b}$	$\tau_0 = (1.8 \text{ to } 2) \tau_{-1}$
Alloy Steels	$\sigma_{-1t} = 0.95 \sigma_{-1b}$	$\sigma_{0b} = 1.6 \sigma_{-1b}$ $\sigma_{0t} = (1.5 \text{ to } 1.6) \sigma_{-1t}$	$\tau_0 = (1.8 \text{ to } 2) \tau_{-1}$
Copper Alloys	$\tau_{-1} = 0.58 \sigma_{-1b}$	$\tau_0 = (1.4 \text{ to } 2) \tau_{-1}$	
Aluminum Alloys	$\sigma_{0b} = 1.8 \sigma_{-1b}$	$\sigma_{-1t} = 0.7 \sigma_{-1b}$ $\tau_0 = (1.4 \text{ to } 2) \tau_{-1}$	$\tau_{-1} = (0.55 \text{ to } 0.58) \sigma_{-1b}$
Grey Cast Iron	$\sigma_{-1t} = (0.6 \text{ to } 0.7) \sigma_{-1b}$ $\tau_{-1} = (0.75 \text{ to } 0.9) \sigma_{-1b}$	$\sigma_{0b} = (1.2 \text{ to } 1.5) \sigma_{-1b}$ $\tau_0 = (1.2 \text{ to } 1.3) \tau_{-1}$	$\frac{\sigma_{-1b}}{\sigma_u} \approx 0.38$
Endurance strength for finite life $\sigma'_{-1} = \sigma_{-1} \left(\frac{10^6}{N} \right)^{0.09}$ where N is the required life in cycles			

DATA

D **???**

twisting moment **30000N-mm**

twisting moment(-) **-20000N-mm**

Bending load **1000N**

Bending load **-700N**

Bending load Distance **600mm**

n- factor of safety **2**

Step1: Cal. Mean & amplitude for bending Moment & Twisting Moment

BM = F x Distance

Max.M 600000N-mm
Min.M -420000N-mm

MMean 90000
Ma 250000N-mm
510000
950000N-mm

Twisting Moment directly given

T mean 5000
Tamp--- 25000 20000N-mm
40000N-mm

Step2 Cal. σ_b and τ (mean and amplitude)

$$\sigma_b = M/Z$$

$$\sigma_{b\text{mean}} = M_{\text{mean}}/Z$$

$$Z = I/y$$

$$I = \pi * d^4 / 64$$

$$Y = d/2$$

$$I = 0.0491 d^4 \text{ mm}^4$$

$$Y = 0.5 d \text{ mm}$$

$$z = 0.09816 d^3 \text{ mm}^3$$

$$M_{\text{mean}} = \frac{90000}{390000} \text{ N-mm}$$

$$\sigma_{b\text{mean}} = 3973256.92 * (1/d^3) \text{ N/mm}^2$$

$$916496.945 \quad 5193482.688$$

$$\sigma_{b\text{ampl}} = 8252149 * (1/d^3) \text{ N/mm}^2$$

Now , find Shear Stress

τ Shear stress	$T \times 16 / (\pi \times d^3)$		N/mm ²		
τ mean	25469.59567 $\times (1/d^3)$		(Tmean)		
τ ampl	127347.9784 $\times (1/d^3)$		(Tamp)		

Step3: Cal. σ_{eq} and τ_{eq}

$$\sigma_{eq} = \frac{\sigma_1}{n} = \sigma_m + K_f \frac{\sigma_x \sigma_f}{\sigma_1}$$

$K_f=1$ (assumed)

$n=2$

$\sigma_1 = ?$

$\sigma_y = ?$

To find σ_y, σ_1

Material	C45	refer pg.1.9/DDB
σ_y	360 N/mm ²	
σ_u (630- 710)	670 N/mm ²	(assumed)

Now refer pg.1.42/DDB , Get σ_1

Load	Bending
reverse cycle	
	$\sigma_1 = 0.46\sigma_u$
$\sigma_1 = 308.2$	N/mm ²

$\sigma_{eq} = \sigma_m + k_f \cdot \sigma_a \cdot \sigma_y / \sigma_1$				
RHS	916496.945			
σ_m	3973256.925	$\cdot (1/d^3)$		
	1869653768			
$k_f \cdot \sigma_a \cdot \sigma_y$	2970773639	$\cdot (1/d^3)$		
$\sigma_1 =$	308.2			
$k_f \cdot \sigma_a \cdot \sigma_y / \sigma_1$	9639109.795	$\cdot (1/d^3)$	(LHS) of equation	
	6066365.243			
LHS	RHS			
$\sigma_{eq} =$	13612366.72	$\cdot (1/d^3)$		

6982862.188

Now , to find

$$\tau_{eq} = \tau_m + k_f \cdot \tau_a \cdot \tau_y / \tau_{_1}$$

$k_f = 1$ (Assumed)

$$\tau_y = \sigma_y / 2 \text{ (refer Pg. 7.6 /DDB)}$$

$$\tau_y = 180 \text{ N/mm}^2$$

To find $\tau_{_1}$

Refer Pg.1.42/DDB

Load

Torsion

Cycle: Reversed

$$\tau_{_1} = 0.22 \sigma_u$$

$$\tau_{_1} = 147.4 \text{ N/mm}^2$$

RHS of equation

$$\tau_m = 25469.59567 \cdot (1/d^3) \quad \text{N/mm}^2$$

$$k_f \cdot \tau_a \cdot \tau_y = 155513.135 \cdot (1/d^3) \quad \text{N/mm}^2$$

22922636.04

$$k_f \cdot \tau_a \cdot \tau_y / \tau_{_1} = 1055.041622 \cdot (1/d^3)$$
$$155513.135 \cdot (1/d^3)$$

Therefore

$$\tau_{eq} = 26524.63729 \cdot (1/d^3)$$

$$180982.7308$$

STEP 4: CALCULATION OF "d" USING COMBINED VARYING STRESS EQUATION (REFR Pg.7.6 /DDB)

LHS of equation

0.5

RHS of equation

$$(\sigma_{eq}/\sigma_y)^2 = 1429757158 * ((1/d^3))^2$$

376237379.1

$$(\tau_{eq}/\tau_y)^2 = 21714.7032 * ((1/d^3))^2$$

1010949.027

$$0.5 = 37812.4169 * (1/d^3)$$

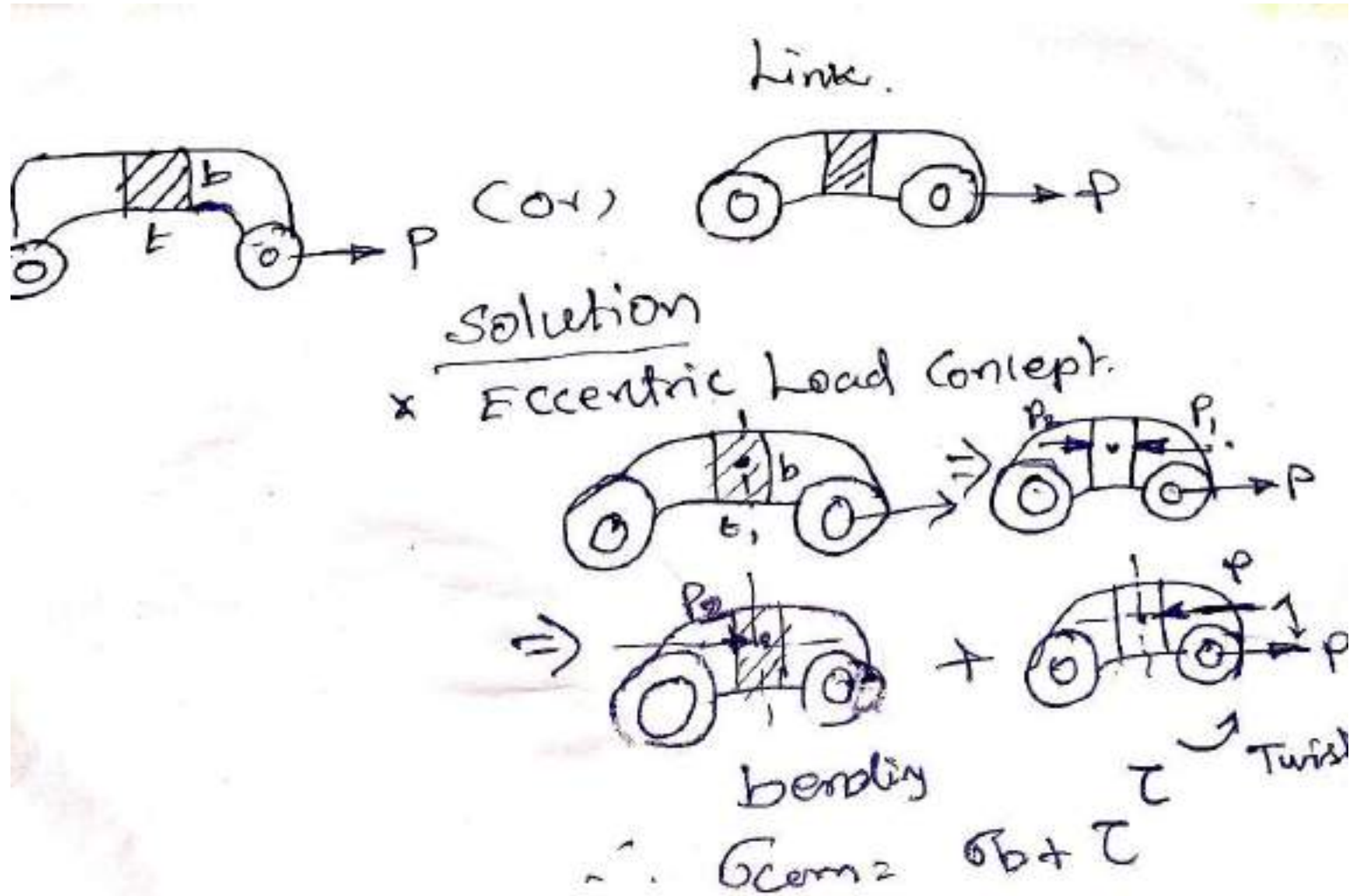
19422.882

$$d^3 = 75624.8338$$

38845.763

$$d = 42.130\text{mm} \quad 33.86$$

Some critical and important problem



Tutorial 4: Topic: combined varying stress

A hot rolled C60 steel shaft is subjected to torsion moment that varies from 330 N-m clockwise to 110N-m counter clockwise and an applied bending moment at a critical section varies from 440 N-m to -220N-m. The shaft is of uniform cross section and no keyway is present at the critical section. Determine the required diameter of shaft. Take factor of safety as 2, the size factor as 0.85 and surface finish factor as 0.62.

UNIT II

ME18503-Design Of Machine Elements

UNIT II DESIGN SHAFT, KEYS AND FITS AND TOLERANCE AND COUPLINGS 12

Preferred Numbers- Standardization Design of shafts under static and **fatigue loadings**, Keys – types of keys , design of keys. Design of Rigid coupling, and Flexible coupling. Fits- types of Fits and Tolerance- hole basis system Shaft basis problems.

Objective

•**This course will make acquainted design principles on shaft, fits and tolerances. and couplings.**

Outcome

Analysing and applying the design of solid, hollow shafts keys and couplings. Also Understanding knowledge of fits and tolerance and analysing it

ROAD MAP

SHAFT - DESIGN - static and fatigue loading

KEY Design

Couplings- Rigid coupling & Flexible coupling

Fits & Tolerance

Preferred Numbers & Standardization

SHAFT

What is shaft?

An element which is usually in circular or round bar to transmit the motion from one element to other

Types of Shaft

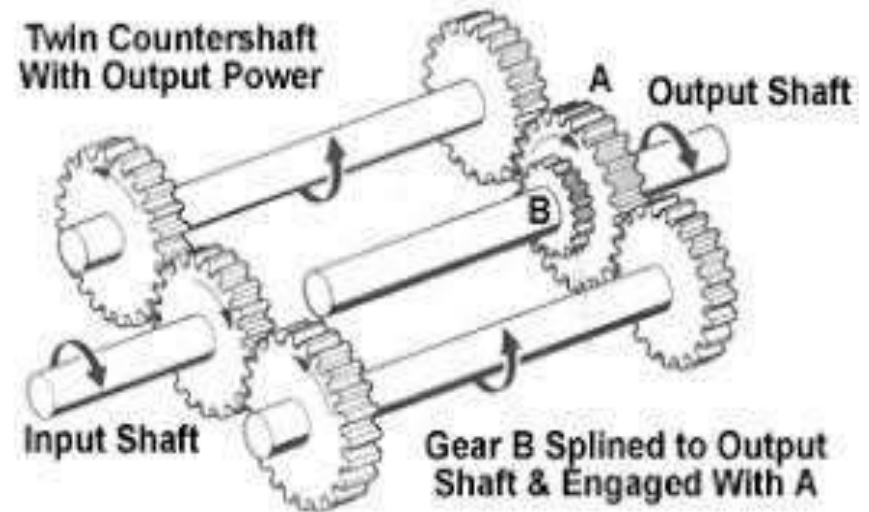
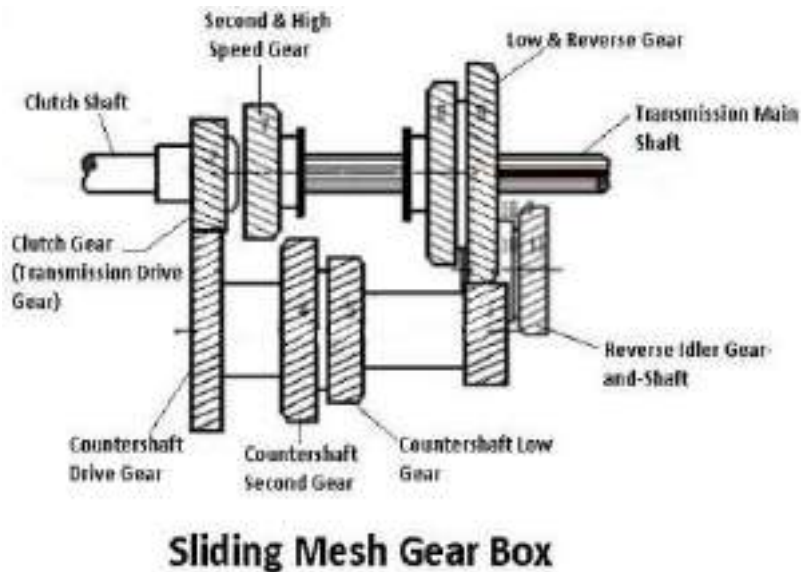
1. Shaft ---- cylindrical revolving member used in power transmission
2. Axle----- Non rotating element, acting as supporting for the rotational elements.
3. Spindle--- Short length shaft – machine tools

Line shaft

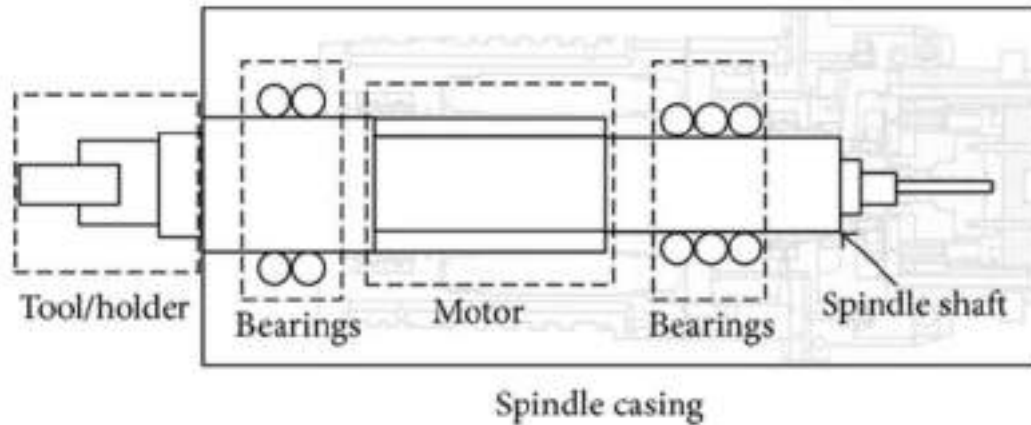
line shafting was used to distribute power from a large central power source to machinery



Counter shaft Or Lay shaft

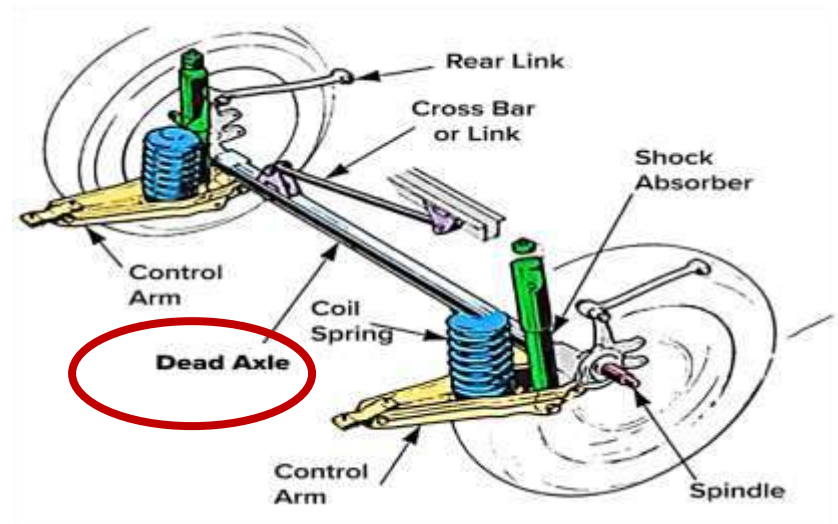


Spindle shaft



Axle

An **axle is a central shaft** for a rotating wheel or gear. On wheeled vehicles, the axle may be fixed to the wheels, rotating with them, or fixed to the vehicle, with the **wheels rotating around the axle**. ... Sometimes, especially on bicycles, bikes. bullock cart wheels on shaft



Shaft manufacturing : hot Rolled or cold working methods

Shaft standard Size: 5 – 7 meters.

Shaft Design

- 1. Strength Based :- theory of simple bending= $M/I = \sigma/Y$**
- 2. Rigidity /stiffness based:- Theory of Torsion= $T/J = \tau/y$**

Shaft generally subjected to

- 1. Torsion due to rotational action**
- 2. Bending due to pulleys mounted on it.**
- 3. Both**
- 4. axial loads in some specific application**

For remembrances

types of beam:

1. Cantilever
2. Simply supported
3. Over hanging (right, Left, or Both sides)

Shafts may be solid or hollow

hollow shaft is better than solid for the same power transmission.

due to saving of material.

Shaft design always prefers the Max Shear stress Theory

Refer the eqn. Pg.7.2/DDB. $\tau_{max} = \pm \sqrt{\sigma^2 + 4.\tau^2}$

SHAFT DESIGN STEPS

Objective = Diameter finding

Step 1: Identify the loads applied on shaft

Step 2: Select material (optional) List σ_u & σ_y

Step 3: calculate T torque transmission

$$P = (2\pi \times NT) / 60,$$

N- rpm, P = power, T = ?

Step 4: Find the maximum bending moment
(use of BMD- bending moment diagram)

Step 5 : Apply the equation to Find “ d “ of the shaft –
applying the loading both T & M

use the equation in Pg. 7.21/DDB

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{\left[K_b M_b + \alpha \frac{P d_o}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2}$$

SHAFTS

* Hollow shaft subjected to torsion only

$$\tau_{max} = \frac{16 M_t d_o}{\pi (d_o^4 - d_i^4)}$$

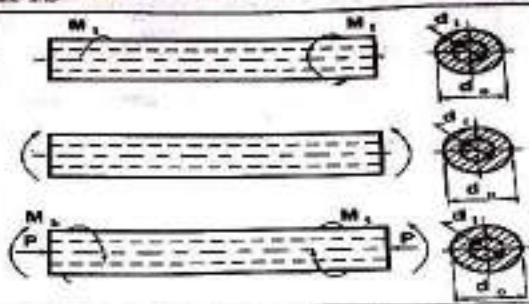
* Hollow shaft subjected to bending only

$$\sigma_{b,max} = \frac{32 M_b d_o}{\pi (d_o^4 - d_i^4)}$$

* Hollow shaft subjected to bending, torsion and axial load

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{ \left[K_b M_b + \alpha \frac{P d_o}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2 }$$

* For solid shaft put $d_i = 0$ Shaft sizes to be rounded off to R 20 series



Type	K_b	K_t
STATIONARY SHAFT		
Gradually applied load	1	1
Suddenly applied load	1.5 - 2	1.5 - 2
REVOLVING SHAFT		
Gradual loading	1.5	1
Minor shock loads	1.5 - 2	1 - 1.5
Heavy shock loads	2 - 3	1.5 - 3

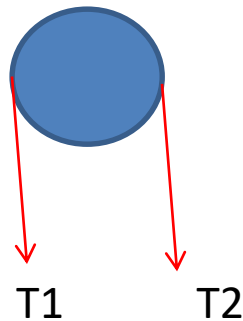
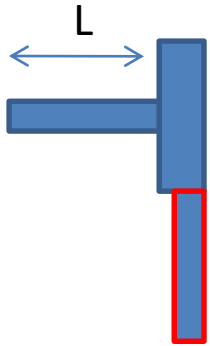
- P axial load
- M_t twisting moment
- M_b bending moment
- $[\tau]$ design shear stress
- K_b combined shock and fatigue factor applied to M_b
- K_t combined shock and fatigue factor applied to M_t
- τ shear stress
- σ_b bending stress
- α column action factor
- 1 for tensile load
- $= \frac{1}{1 - 0.0044 \left(\frac{l}{r} \right)}$ for $\frac{l}{r} < 115$
- $= \frac{\sigma_y}{\pi^2 n E} \left(\frac{l}{r} \right)^2$ for $\frac{l}{r} > 115$
- E young's modulus
- σ_y yield stress
- r radius of gyration
- l length of shaft under axial load
- n end condition coefficient

Standard Series

BASIC SERIES OF PREFERRED NUMBERS

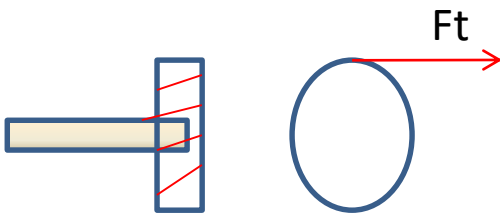
R 5 $\phi = 1.4$	R 10 $\phi = 1.25$	R 20 $\phi = 1.12$	R 40 $\phi = 1.06$
1.00	1.00	1.00	1.00
		1.12	1.10
		1.25	1.18
	1.25	1.25	1.25
		1.40	1.32
		1.60	1.40
1.60	1.60	1.60	1.50
		1.80	1.60
		2.00	1.70
	2.00	2.00	1.80
		2.24	1.90
		2.50	2.00
2.50	2.50	2.50	2.12
		2.80	2.24
		3.15	2.36
	3.15	3.15	2.50
		3.55	2.65
		4.00	2.80
4.00	4.00	4.00	3.00
		4.50	3.15
		5.00	3.35
	5.00	5.00	3.55
		5.60	3.75
		6.00	4.00
6.30	6.30	6.30	4.25
		7.10	4.50
		8.00	4.75
	8.00	8.00	5.00
		9.00	5.30
		10.00	5.60
10.00	10.00	10.00	6.00
			6.30
			6.70
			7.10
			7.50
			8.00
			8.50
			9.00
			9.50
10.00	10.00	10.00	10.00

Some task for finding the bending moment



Belt pulley
 $W = T_1 + T_2$

$$BM = W \times L$$



Gear drive / pulley

$$W_g = F_t / \cos \alpha$$

$$BM = W_g \times L/4$$

F_t = tangential force
 α = Pressure angle (20°)

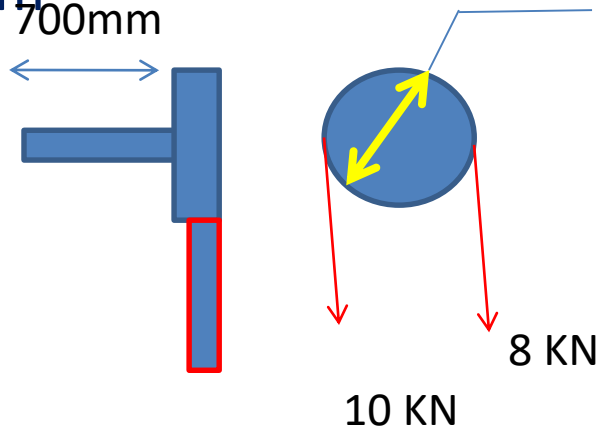
Pg.6.4/DDB

CANTILEVER BEAMS			
Bending Moment	Reaction	Deflection	Type of Loading
$M_x = -Px$ $M_{max} = -Pl$	$R_x = P$	$y = \frac{P}{6EI} [x^3 - 3L^2x + x^3]$ $y_{max} = \frac{Pl^3}{6EI}$ at $x=0$	
$M_x = -\frac{wx^2}{2}$ $M_{max} = -\frac{wL^2}{2}$	$R_x = wL$	$y = \frac{w}{24EI} [x^4 - 4L^3x + x^4]$ $y_{max} = \frac{wL^4}{8EI}$ at $x=0$	
$M_x = -P(x-a)$ $x > a$ $M_{max} = -P(l-a)$	$R_x = P$	$y = \frac{P}{6EI} [x^3 - 3L^2x + x^3]$ for $x < a$ $y = \frac{P}{6EI} [x^3 - 3L^2x + x^3]$ for $x > a$ $y_{max} = \frac{P(l-a)^3}{6EI}$ at $x=l$	
$M_x = -\frac{wx^2}{2}$ $M_{max} = -\frac{wL^2}{2}$	$R_x = \frac{wL}{2}$	$y = \frac{w}{120EI} [x^5 - 5L^4x + 4L^4]$ $y_{max} = \frac{wL^5}{120EI}$ at $x=0$	
$M_x = -\frac{wL}{6L} (L-x)^3$ $M_{max} = -\frac{wL^3}{6}$	$R_x = \frac{wL}{2}$	$y = \frac{w}{120EI} [x^5 - 5L^4x + 4L^4]$ $y_{max} = \left[\frac{11}{120EI} \right] wL^5$ at $x=0$	
$M_x = -mx$ constant	$R_x = 0$	$y = \frac{m}{2EI} [x^3 - 2Lx + x^3]$ $y_{max} = \frac{mL^3}{2EI}$ at $x=0$	

LAKSHON KATTA - PISA TUTOR

Simply Supported Beam		SIMPLY SUPPORTED BEAM	
Bending Moment	Reaction	Deflection	Shape of loading
$M_x = \frac{P}{2} x, 0 < x < \frac{L}{2}$ $M_{max} = \frac{PL}{4}$	$R_A = \frac{P}{2}$ $R_B = \frac{P}{2}$	$y = \frac{Px^2}{4EI} [3Lx - x^2]$ $y_{max} = \frac{PL^3}{48EI} \text{ at } x = \frac{L}{2}$	
$M_x = \frac{Pb}{L} x, 0 < x < a$ $M_{max} = \frac{Pab}{L}$	$R_A = \frac{Pb}{L}$ $R_B = \frac{Pa}{L}$	$y = \frac{Pbx}{6EI} [L^2 - x^2 - a^2] \text{ at } x = \frac{Pa}{L}$ $y_{max} = \frac{Pa^2b}{\sqrt{3}EI} [L^2 - 2a^2] \text{ at } x = \frac{Pa}{L}$	
$M_x = \frac{wLx}{2} - \frac{wx^2}{2}$ $M_{max} = \frac{wL^2}{8} \text{ at } x = \frac{L}{2}$	$R_A = \frac{wL}{2}$ $R_B = \frac{wL}{2}$	$y = \frac{wx^2}{24EI} [L^3 - 3Lx^2 + 2x^3]$ $y_{max} = \frac{5wL^4}{384EI} \text{ at } x = \frac{L}{2}$	
$M_x = Pa \sin \frac{\pi x}{L}$ $M_{max} = Pa$	$R_A = P$ $R_B = P$	$y = \frac{Pa}{24EI} [L^3 - 3Lx^2 + 2x^3]$ $y_{max} = \frac{Pa}{24EI} [L^3 - 3Lx^2 + 2x^3] \text{ at } x = \frac{L}{2}$	
$M_x = Wx \left[\frac{1}{2} - \frac{x}{L} + \frac{3}{2} \left(\frac{x}{L} \right)^2 \right]$ at $x = \frac{L}{2}$ $M_{max} = \frac{WL}{8} \text{ at } x = \frac{L}{2}$	$R_A = \frac{W}{2}$ $R_B = \frac{W}{2}$	$y = \frac{Wx^2}{48EI} \left[L^2 \frac{x}{2} - \frac{x^2}{3} + \frac{x^3}{6L} \right] \text{ at } x = \frac{L}{2}$ $y_{max} = \frac{WL^3}{96EI} \text{ at } x = \frac{L}{2}$	
$M_x = \frac{Wx}{3} \left[1 - \frac{x}{L} \right]^2$ $M_{max} = 0.128 WL$ at $x = 0.5774L$	$R_A = \frac{2W}{3}$	$y_{max} = 0.00768 \left[\frac{WL^3}{EI} \right] \text{ at } x = 0.5774L$	

SP1- Find the diameter of the cantilever shaft which carries a belt pulley at its end as shown in figure. Take allowable Shear Stress as 50 N/mm^2 $p=10\text{kW}$, at 800 rpm



DATA

$D = ?$

Loads = Only belt pulley - tensions

$T_1 = 10 \text{ kN}$,

$T_2 = 8 \text{ kN}$,

$[\tau] = 50 \text{ N/mm}^2$

Equation to be used for the combination Torsion and bending

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{\left[K_b M_b + \alpha \frac{P d_o}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2}$$

Type	K_b	K_t
STATIONARY SHAFT		
Gradually applied load	1	1
Suddenly applied load	1.5 - 2	1.5 - 2
REVOLVING SHAFT		
Gradual loading	1.5	1
Minor shock loads	1.5 - 2	1 - 1.5
Heavy shock loads	2 - 3	1.5 - 3

DATA

D ??

T1 10000N

T2 8000N

Diameter mm

[T] 50N/mm²

Step1 Loads identification

T1 & T2

Weight of belt pulley
(If not given in problem,)

W = T1+ T2

If weight of the pulley is given

W = Wp + T1 + T2

Now

W 18000N

(W = T1 + T2)

Step2: Material Selection, List σ_u & σ_y (optional)

Allowable shear stress is given, no need to select material

$$[\tau] = 50 \text{ N/mm}^2$$

Step3 Calculation of torque T

Step3: calculate T torque transmission

$$P = (2\pi \times NT) / 60,$$

N- 500 rpm, P = 8×10^3 , T = ?

$$T = Mt = 152.817 \text{ N-m},$$
$$= 152.817 \times 10^3 \text{ N-mm}$$

Step4 Calculation of Bending moment

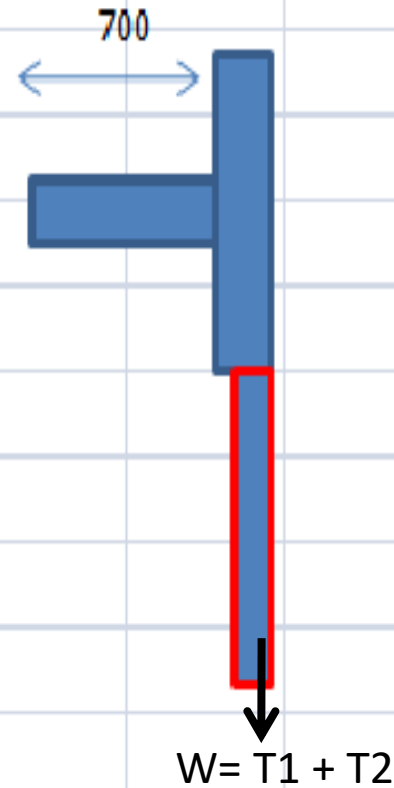
Beam cantilever

w T1+T2
18000 N

L 700 mm

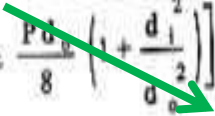
M = W X L

M = 12600000 N-mm



Step 5 CALCulation of Diameter

Refer Pg. 7.21/DDB

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{\left[K_b M_b + \alpha \frac{P d_o}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2}$$


$d_i = 0$, for solid shaft

[τ]	50
M_b	12600000
M_t	152817

Put $P = 0$ No axial load

K_b & K_t = ?

Refer Pg. 7.21/DDB

Take : revolving condition

Assumed Gradual Loading

K_b 1.5

K_t 1

Type	K _b	K _t
STATIONARY SHAFT		
Gradually applied load	1	1
Suddenly applied load	1.5 - 2	1.5 - 2
REVOLVING SHAFT		
Gradual loading	1.5	1
Minor shock loads	1.5 - 2	1 - 1.5
Heavy shock loads	2 - 3	1.5 - 3

$$D^3 = ??$$

$$K_b \times M_b \quad 1.9E+07 \quad 3.572E+14 (K_b \times M_b)^2$$

$$K_t \times M_t \quad 152817 \quad 2.335E+10 (K_t \times M_t)^2$$

$$\sqrt{K_b M_b^2 + K_t M_t^2} = 18900617.8$$

$$16 / (\pi \times [\tau]) = 0.101878383$$

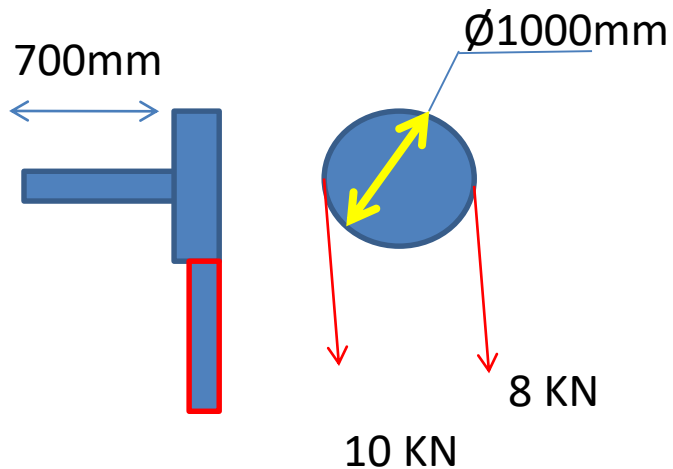
$$d^3 = 1925564.373$$

$$d = 123.8105999$$

Pg.7.20/DDB 124.409

STD R20 series 125mm

SP1a- Find the diameter of the cantilever shaft which carries a belt pulley at its end as shown in figure. Take allowable Shear Stress as 50



DATA

D= ?

Loads = Only belt pulley - tensions

T1= 10 kN,

T2= 8 kN,

$[\tau] = 50 \text{ N/mm}^2$

DATA

D ??

T1 10000N
T2 8000N
Diameter 1000mm
[T] 50N/mm²

Step1 Loads identification

T1 & T2
Weight of belt pulley
(If not given in prob,)

W = T1+ T2

If weight of the pulley is given

W = Wp + T1 + T2

Now

W 18000N

(W = T1 + T2)

**Step2: Material Selection, List σ_u & σ_y _
(optional)**

Allowable shear stress is given, no need to select material

$$\tau = 50 \text{ N/mm}^2$$

Calculation of torque

Step3 T

$$T = F \times R$$

R = radius of Pulley
F = net force (T1 - T2)

$$T = (T1 - T2) \times D / 2$$

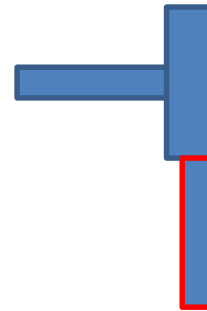
$$T = 1000000 \text{ N-mm}$$

Step4 Calculation of Bending moment

Beam cantilever
w T1+T2
18000 N
L 700mm

$$M = W \times L$$

$$M = 12600000 \text{ N-mm}$$



$$D^3 = ??$$

$K_b \times M_b$	$1.9E+07$	$3.572E+14 (K_b \times M_b)^2$
$K_t \times M_t$	1000000	$1.000E+12 (K_t \times M_t)^2$

$$\text{sqrt}(K_b M_b^2 + K_t M_t^2) = 18926437$$

$$16 / (\pi \times [\tau]) = 0.1018784$$

$$d^3 = 1928194.7$$

$$d = 123.86689$$

Refer Pg.7.20/DDB

STD R20 series 125mm

$$d_o^3 = \frac{16}{\pi [\tau] \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\}} \sqrt{\left[K_b M_b + \alpha \frac{P d_i}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2}$$

$D^3 =$??

$K_b \times M_b$ 18900000 $(K_b M_b)^2 = 3.572E+14$

$K_t \times M_t$ 2000000 $(K_t M_t)^2 = 4.000E+12$

$\text{sqrt}(K_b M_b^2 + K_t M_t^2) =$ 19005525.5

$16 / (\pi \times [\tau]) =$ 0.10187838

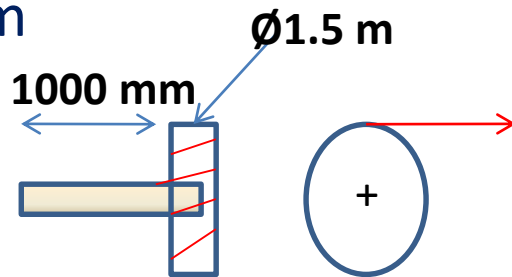
$d^3 =$ 1936252.2

$d =$ 124.039018

Pg.7.20/DDB

STD R20 series 125mm

SP2- Find the diameter of the cantilever shaft which carries a gear pulley with pressure angle 20° . at its end as shown in figure. Take allowable Shear Stress as 400 N/mm^2 . Power transmitted by the drive is 8 kW at 500 rpm



DATA


$D = ?$

Loads = Gear drive

$W_{\text{gear}} = Ft / \cos \alpha$

$\alpha = 20^\circ$

$[\tau] = 400 \text{ N/mm}^2$

DATA							
D	??						
P=	8 x 10 ³ watts						
L	1000 mm						
Diameter	1500 mm		N= 500 rpm				
[T]	400 N/mm ²						
Step1	Loads identification						
W =	Weifght of Gear Pulley						
W =	Ft/Cosα			T =	P x 60/(2x π x N)		
α =	20			T =	152.817574N-m		
Cos20	0.4081				152817.574N-mm		
T=Ft.R							
Ft =	T/R	203.7567654 N		(refer step3)			
Now							
Wgear=	499.3 N						

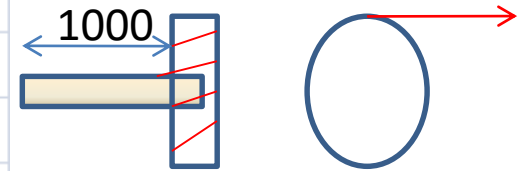
Step2: Material Selection, List σ_u & σ_y _ (optional)

Allowable shear stress is given, no need to select material

$$[\tau] = 400 \text{ N/mm}^2$$

Step3	Calculation of torque T			
	P =	$2 \times \pi \times N \times T / 60$		
	T =	$P \times 60 / (2 \times \pi \times N)$		
	N =	500		
	P	8000 W		
	π	3.141		
	T =	152.817574 N-m		
		152817.574 N-mm		

Step4		Calculation of Bending moment	
	Beam	cantilever	
	W _{gear} =	499.30341	
		499.30341 N	
	L =	1000 mm	
	M =	W X L	
	M =	499303.41 N-mm	



K_b & $K_t = ?$

Refer Pg. 7.20/DDB

Type	K_b	K_t
STATIONARY SHAFT		
Gradually applied load	1	1
Suddenly applied load	1.5 - 2	1.5 - 2
REVOLVING SHAFT		
Gradual loading	1.5	1
Minor shock loads	1.5 - 2	1 - 1.5
Heavy shock loads	2 - 3	1.5 - 3

Take : revolving condition

Assumed Gradual Loading

K_b 1.5

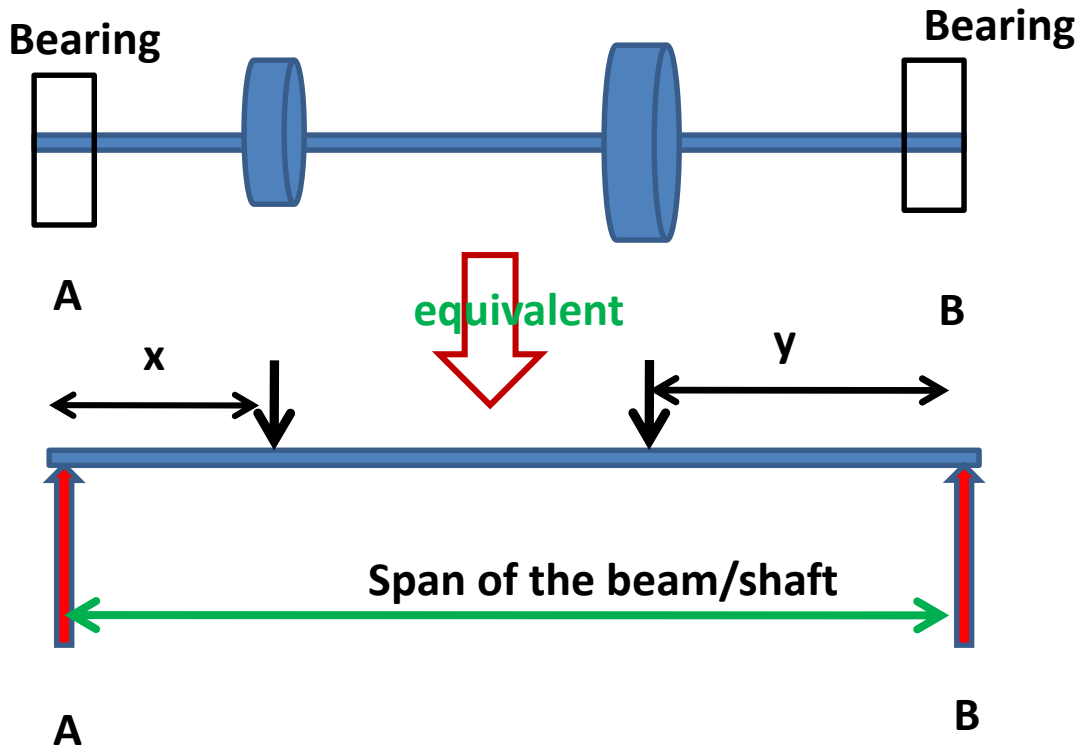
K_t 1

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{ \left(K_b M_b + a \frac{Pd}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right)^2 + (K_t M_t)^2 }$$

	$D^3 =$??			
	$K_b \times M_b$	748955.1162		$5.609E+11$	$= (K_b M_b)^2$
	$K_t \times M_t$	152817.574		$2.335E+10$	$= (K_t M_t)^2$
	$\text{sqrt}(K_b M_b^2 + K_t M_t^2) =$			764386.667	
	$16 / (\pi \times [\tau]) =$			0.0127348	
		$d^3 =$		9734.30968	
		$d =$		21.2865666	
	Pg.7.20/DDB				
		STD R20 series		22.4	mm

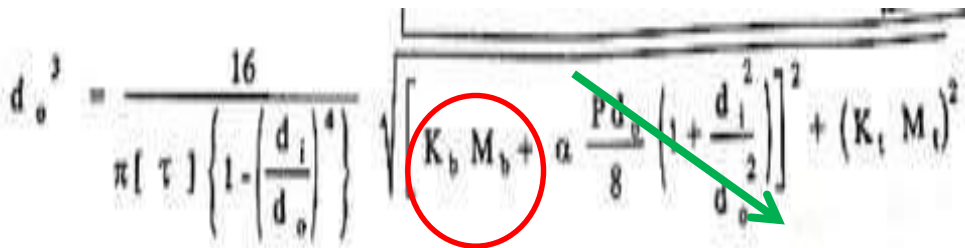
Various pulleys mounted on the shaft

Develop the shaft with respect the description in the problem



SP3. A shaft is having length of 2 meter and supported at its ends. A belt driven pulley of diameter 500 mm is mounted at distance of 0.5 meter from right end bearing. The maximum and minimum belt tensions are 12 KN and 8 KN respectively. An another gear pulley of 400 mm in diameter is mounted on the shaft at 400 mm from the left end support. The power transmitted by the system is 20 KW at 1200 rpm. Shaft is made of C45 steel. Determine the diameter of the shaft?

1. Develop the shaft arrangement

$$d_o^3 = \frac{16}{\pi [\tau]} \left\{ 1 - \left(\frac{d_i}{d_o} \right)^4 \right\} \sqrt{ \left[K_b M_b + a \frac{P d_1}{8} \left(1 + \frac{d_1^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2 }$$


DATA			
D	?? 20×10^{-3}		
P=	20000	watts	
N	1200	rpm	
Shaft L	2000	mm	
D_{belt pulley}	500	mm	
D_{gear pulley}	400	mm	
[T]	??	N/mm²	
T1	12 KN		12000 N
T2	8 KN		8000 N
Shaft Diagram			
<p>The diagram illustrates a shaft of total length 2000 mm. A gear is positioned 400 mm from the left end, exerting a downward force of 12000 N. A belt pulley is positioned 500 mm from the right end, exerting a downward force of 8000 N. The shaft is supported by bearings at both ends, shown as red vertical bars with upward-pointing arrows.</p>			

Step1	Torque calculation		
	P =	$2 \pi NT/60$	
	N=1200rpm		
	P= 20 x 10³		
	T =	159.1849729 N-m	
		159184.9729 N-mm	

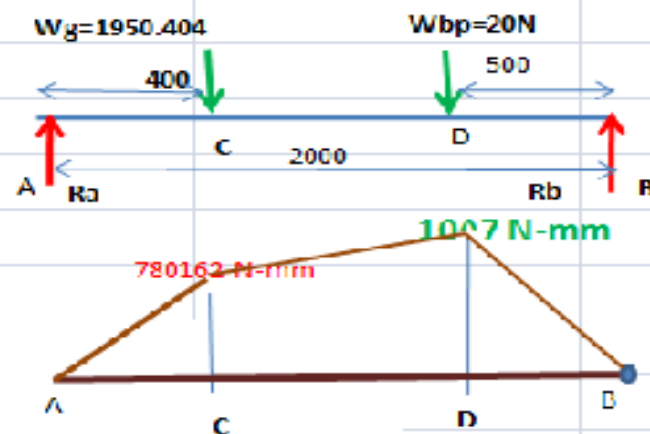
Step2	Loads identification		
<u>Gear pulley</u>			
W =	Weifght of Gear Pulley		
W =	Ft/Cosα		[α is not given]
			[α is taken 20^o]
$\alpha =$	20		
Cos20	0.408082		
T=Ft.R			
Ft =	T/R	795.9248647 N-m	
Now			
Wgear=	1950.404 N		
<u>Wbelt pulley</u>			
Wbp =	T1+T2		
	20	N	
	20 x 10³		

Step3: Material Selection, List σ_u & σ_y _ (optional)

	Material is given as C45		
	Allowable shear stress is not given		
	$[\tau] =$??	N/mm ²
	Refer Pg.1.9/DDB		
	C45		
	$\sigma_y =$	360	N/mm ²
	$\sigma_u =$		
	Refer Pg.7.6/DDB		
	Asper τ max theory,	$\tau = \sigma_y / 2$	
		$\tau =$	180 N/mm ²
		$[\tau] =$	180

Need to have line sketch of shaft

MaxBM= ???



$R_a + R_b =$ Total load

$R_a + R_b = 21950.4$ eqn1

Now bm at all points, A, C, D & B

BM at supports = 0 always

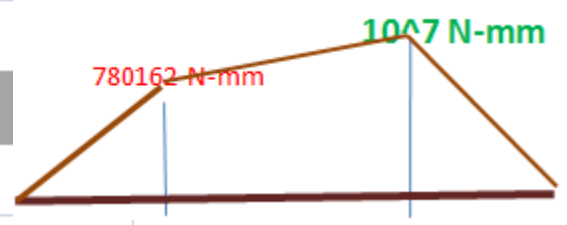
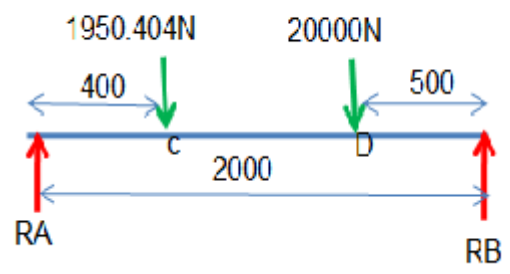
BM= F x L

Take fixed point A, moments calculated about it.

Bm at

A	0
C	780162 (400×1950.404)
D	$1E+07$ (500×20000)
B	0

Find Ra & Rb,			
take Moment about A,			
Clockwise= anti clock wise			
at C		at D	at B
$(400 \times 1950.404) + (1500 \times 20000) =$			$R_b \times 2000$
780162		30000000	
$3.1E+07$	=	$R_b \times 2000$	
Rb	=	15390.081 N	
Use Rb in eqn1, Ra=?			
	$R_a + R_b =$	21950.404	
	$R_a + 395.08 =$	21950.4	
	Ra =	6560.323 N	



Step 5

CALCulation of Diameter

Refer Pg. 7.21/DDB

$$d_s^3 = \frac{16}{\pi (1 + \alpha) \left[1 - \left(\frac{d_i}{d_o} \right)^4 \right]} \sqrt{\left[K_b M_b + \alpha \frac{P d_b}{8} \left(1 + \frac{d_i^2}{d_o^2} \right) \right]^2 + (K_t M_t)^2}$$

DI = 0, for solid shaft

[T] ??
Mb 10000000 N-mm
Mt 159184.9729

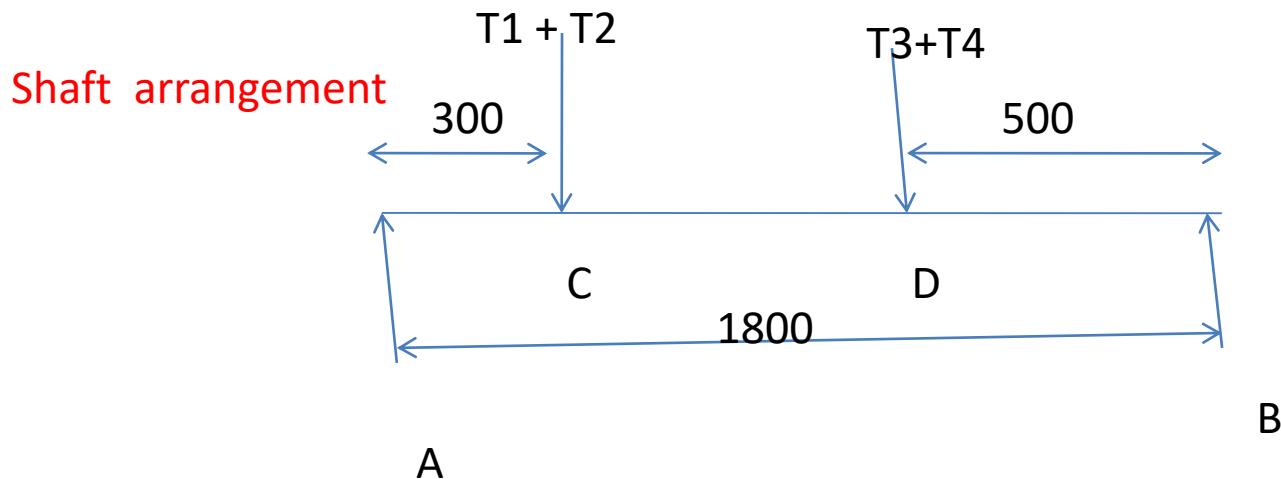
Put $\alpha = 0$ No axial load

Kb & Kt = ?
Refer Pg. 7.21/DDB

Type	K_b	K_t
FLAT SHAFT SHAFT		
Evenly applied load	1	1
Unidirectionally applied load	1.5-2	1.5-2
KEYWAY SHAFT		
Light loading	1.2	1
Medium shock loads	1.5-2	1.5-2
Heavy shock loads	2-3	1.5-2

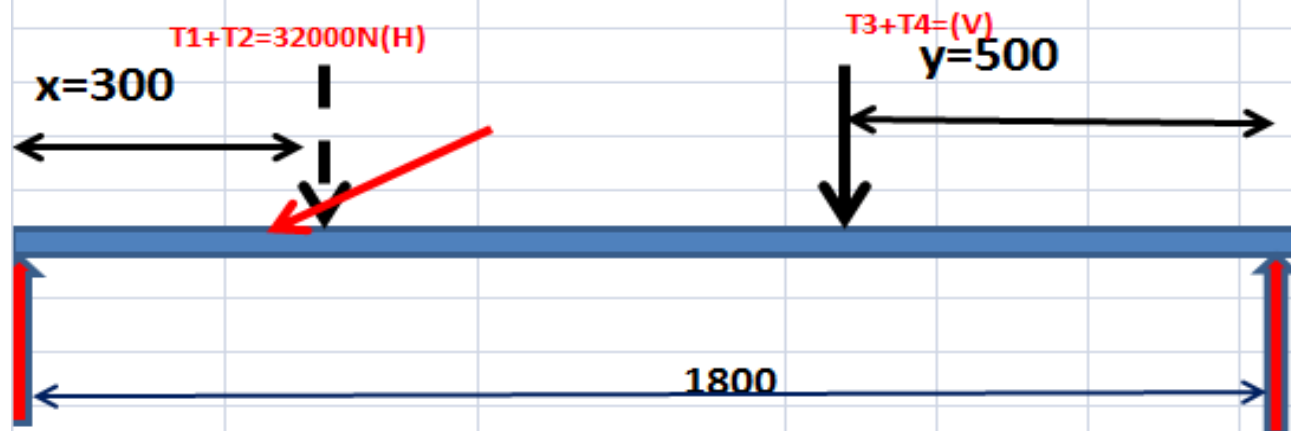
Take : revolving condition				
Assumed	Gradual Loading			
Kb	1.5			
Kt	1			
D³=	??			
Kb x Mb	15000000	2.250E+14	(Kb x Mb)²	
Kt x Mt	159184.9729	2.534E+10	(Kt x Mt)²	
sqrt(Kbmb²+KtMt²)=		15000845		
16/(π x [τ]) =		0.0282996		
	d³ =	424517.16		
	d =	74.832304		
Pg.7.20/DDB				
	STD R20 series	80	mm	

SP4. A shaft is supported at its two ends by two bearings A & B, the span between them is 1.8 meters. To the right of bearing A a belt pulley of diameter 600 mm is mounted at 300 mm takes the horizontal drive with the tensions 20KN and 12 KN. To the left of bearing B, another belt pulley of diameter of 400 mm, is located at 500mm. Also it transmits a vertical drive. Determine the diameter of the shaft when it takes minor shock loads. Take angle of contact 180deg.and $\mu=.25$



DATA		
D	??	
P=	?	
N	?	rpm
Shaft L	1800	mm
D1belt pulley	600	mm
D2belt pulley	400	mm
[T]	??	N/mm ²
T1	20000	N
T2	12000	N

Shaft Diagram



Step1	Torque calculation		
	P =	$2 \pi NT/60$	
	T= F X R		
	$T=(T1-T2) X D1/2$		
	T1-T2=	8000	
	D1/2 =	300	
	T =	2400000	N-mm

Step2 Loads identification

W1belt pulley(H)

$$W1bp = T1 + T2$$

$$32000 \quad N$$

$$W2Bp = T3 + T4 \quad ?$$

Assume the torque is same for The D2 belt Pulle

$$T = \quad T1/T2 = \quad e^{\mu\theta} = T3/T4$$

$$\mu = \quad 0.3$$

$$\theta = \quad 170 \quad 3$$

$$T1/T2 = \quad e^{\mu\theta}$$

$$T_1/T_2 = e^{\mu\theta}$$

$$2.1$$

$$T_3/T_4 = 2.1$$

Use Torque, find T3 & T4

$$T = (T_3 - T_4) \times D/2$$

$$D/2 = 200$$

$$T_3 = T_4 \times 2.0993$$

$$2400000 = T_4(2.0993 - 1) \times 200$$

$$T_4 = 10916.04 \text{ N}$$

$$T_3 = 22916.52 \text{ N}$$

$$W_2 B_p = 33832.6 \text{ N}$$

Step3: Material Selection, List σ_u & σ_y _ (optional)

Material is given as C45

Allowable shear stress is not given

$$[\tau] = 0 \text{ N/mm}^2$$

Refer Pg.1.9/DDB

C45

$$\sigma_y = 360 \text{ N/mm}^2$$

$$\sigma_u =$$

Refer Pg.7.6/DDB

Asper τ max theory,

$$\tau = \sigma_y / 2$$

$$\tau = 180 \text{ N/mm}^2$$

$$[\tau] = 180$$

Step4

Calculation of Bending moment

Need to have line sketch of shaft

MaxBM= ???

H & V

$R_a + R_b =$ Total load (only Horizontal)

$R_a + R_b = 32000$ eqn1

Now bm at all points, A, C, D & B

BM at supports = 0 always

$BM = F \times L$

Apply moment about A, to find R_a & R_b

$1800R_b = 300 \times 32000$

$R_b = 5333.33333$ N

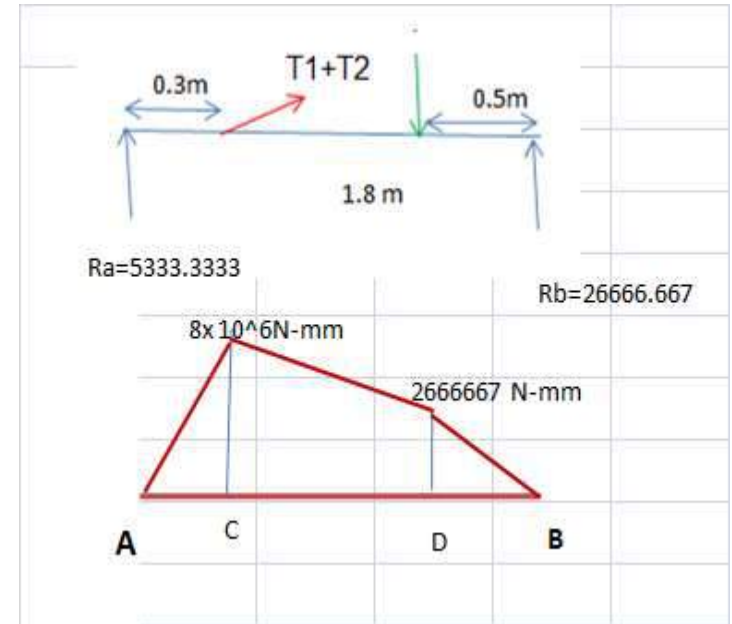
$R_a = 26666.6667$ N

Now find moment at a,b,c,d

At A & B = 0

At C $8000000 (R_a \times \text{dist from c})$

AT D $2666667 (R_b \times \text{dist from d})$



$R_a + R_b =$ Total load (only Vertical)

$R_a + R_b =$ 33832.56 eqn1

Now bm at all points, A, C, D & B

BM at supports = 0 always

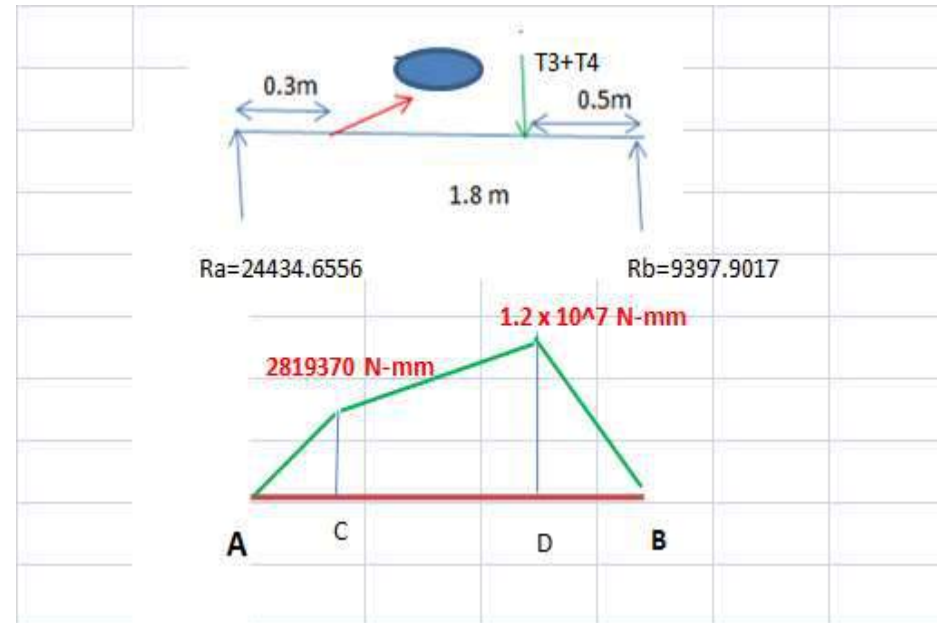
BM = $F \times L$

Apply moment about A, to find R_a & R_b

$$1800R_b = 1300 \times 33832.6$$

$$R_b = 24434.65556 \text{ N}$$

$$R_a = 9397.901655 \text{ N}$$



Now find moment at a,b,c,d

At A & B= 0

BM at Cv

$$2819370(Ra \times \text{dist from c})$$

BM at Dv

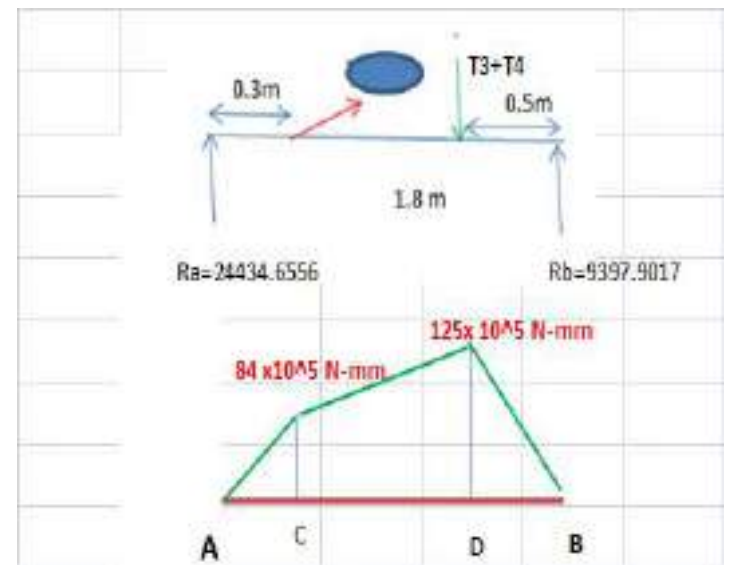
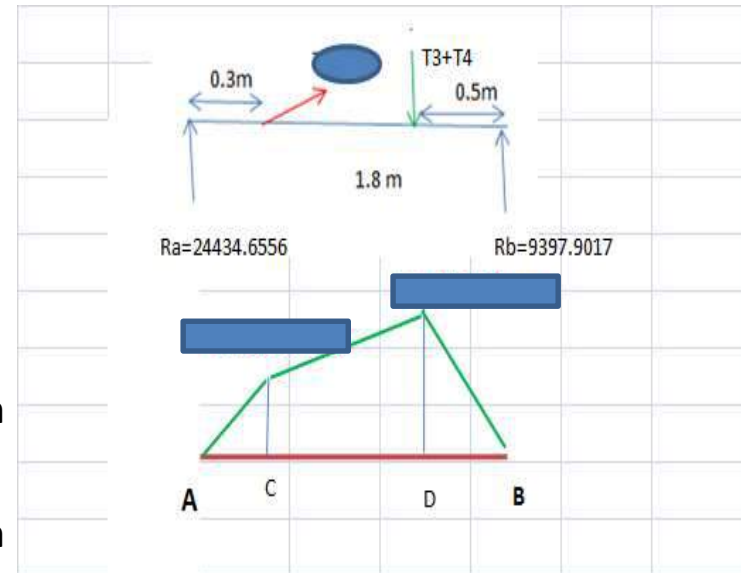
$$12217328(Rb \times \text{dist from d})$$

Now to find Resutant BM at C & D

$$\text{RBM.C} = \text{SQRT}[(\text{BMCh})^2 + (\text{BMCv})^2]$$
$$8482266.796 \quad \text{N-mm}$$

$$\text{RBM.D} = \text{SQRT}[(\text{BMDh})^2 + (\text{BMDv})^2]$$
$$12504967.38 \quad \text{N-mm}$$

Chose RBM at D as Max. BM



$D^3 =$??

Kb x Mb	22508941.28	$5.067E+14 (Kb \times Mb)^2$
Kt x Mt	3120000	$9.734E+12 (Kt \times Mt)^2$

$\sqrt{Kb \times Mb^2 + Kt \times Mt^2} =$ 22724146.6

$16 / (\pi \times [\tau]) =$ 0.02829955

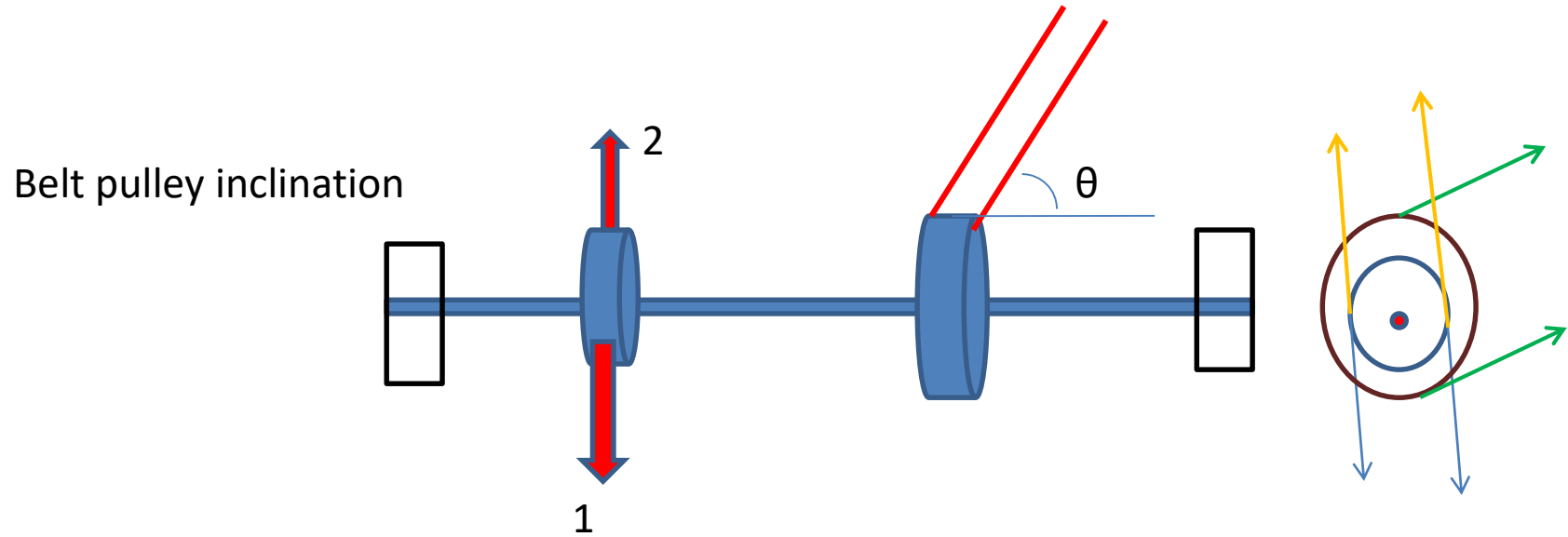
$d^3 =$ 643083.139

$d =$ 85.9316106

Pg.7.20/DDB

STD R20 series 90mm

Other important points to be considered

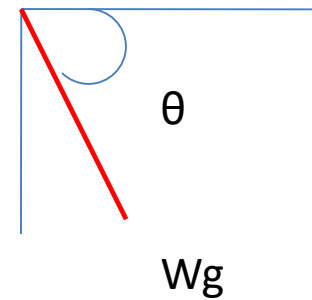
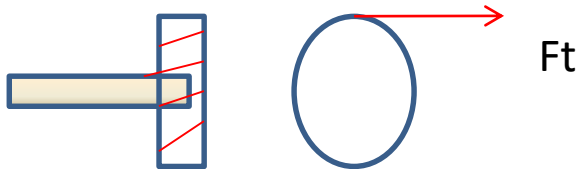


1. Vertical downward = easy to solve

2. Vertical upward = balance the force with the support.

3. Inclined = to be resolved into two components, horizontal $\cos\theta$ and $\sin\theta$

Other important points to be considered



Hence two point loads will come
1.ver. & 2. horizontal

This F_t acts downward or upward based on the rotation direction.

W_t of the gear component will act at given pressure angle to the vertical or horizontal

hence $W_g = F_t / \cos \alpha$ to be taken into two component $W_g \cos \alpha$ and $W_g \sin \alpha$

Shaft problems for Sketching

1. A solid steel shaft is supported on two bearings 1.8 m apart. A 20° involute gear D, is keyed to the shaft at a distance of 150 mm to the left of the right hand bearing. Two pulleys B and C are located on the shaft at distances of 600 mm and 1350 mm respectively to the right of left hand bearing. The drive B is vertically downward while from C the drive is downward at angle of 60° to the horizontal. Draw the arrangement.

2. A shaft carrying a pulley A and gear B and supported in two bearings C and D. The tangential force F_t on the gear acts vertically upwards. The pulley delivers the power vertically downward. B is located at 500 mm to right of C and A is located 400 mm to the left of D. The span between C and D is 2000 mm. Draw the arrangement on the shaft.

3. A horizontal shaft of 2.5 m AD supported in bearings at A and B and carrying the pulleys at C and D. The C pulley drives the power vertically downward and D pulley delivers the power horizontally. The span of AB is 1.8 m. C located at 600 mm to right of A bearing. D is located at 2500 mm to the left of A bearing. Draw the arrangement

Unit2 Tutorial 1

1. A horizontal nickel steel shaft rests on two bearings , A at the left and B at the right end and carries two gears at C and D located at distances of 250 mm and 400 mm respectively from the centre line of the left and right bearings. The pitch diameter of the gear C is 600 mm and the gear D is 200 mm. The distance between the centre line of the bearings is 2400mm. The shaft transmits 20 kW at 120 rpm. The power is delivered to the shaft at gear C is taken out at the gear D in such a manner that the tooth pressure F_{tC} of gear C and F_{tD} of the gear D act vertically downwards. Find diameter of the shaft. Take weight of the gears C and D 950 N and 350 N respectively. The combined shock and fatigue factors for bending and torsion may be taken as 1.5 and 1.2. The working stress is 100 MPa in tension and 56MPa in shear.

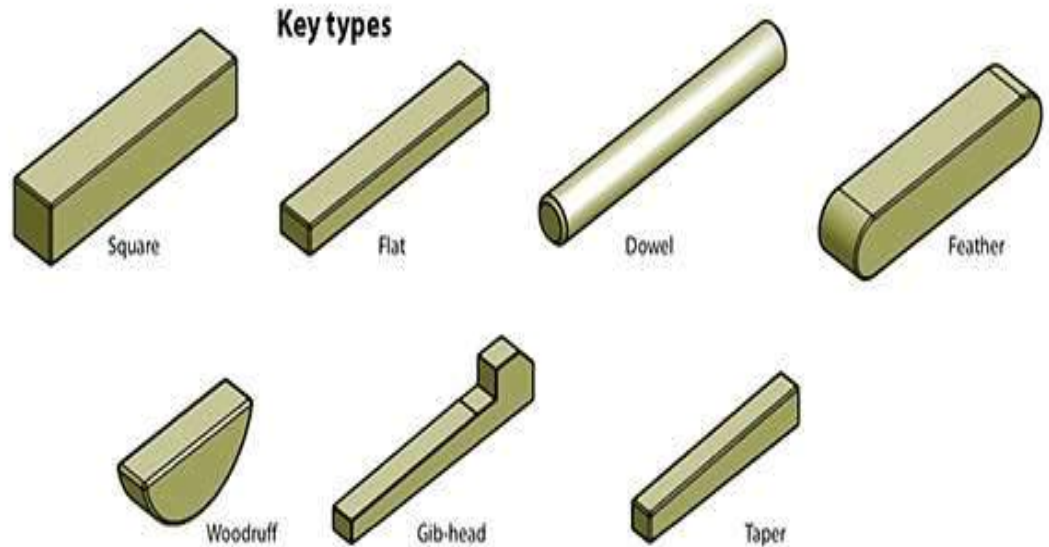
Key Design

What is key?

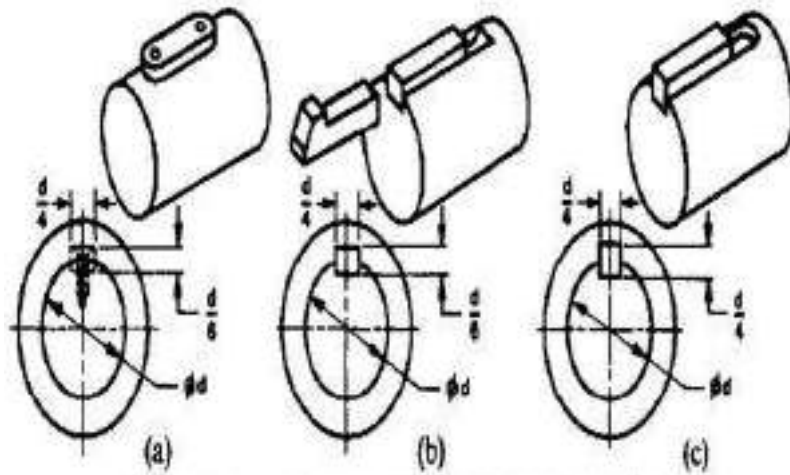
Key is a mild steel piece machine element inserted between the shaft and boss or hub of the pulley used to prevent the relative motion between the shaft and rotating element.

Types of keys

1. Sunk keys
2. Saddle keys
3. Tangent keys
4. Round keys
5. Splines

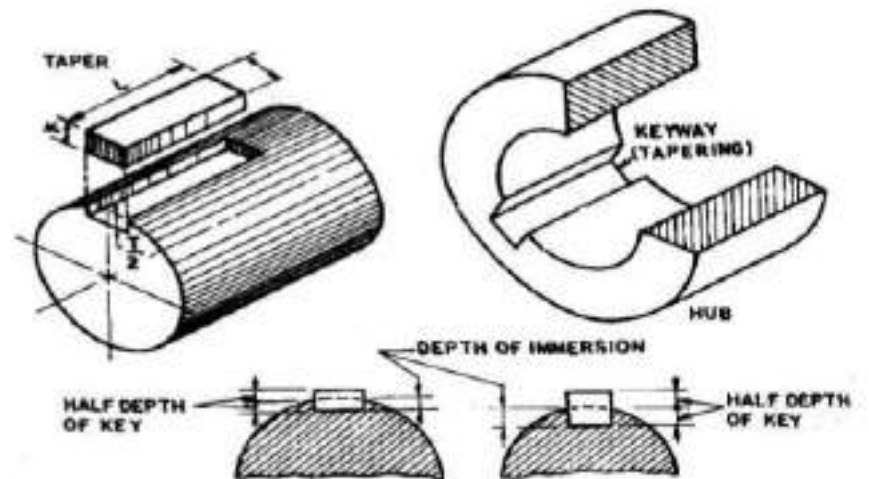


KEYS Design

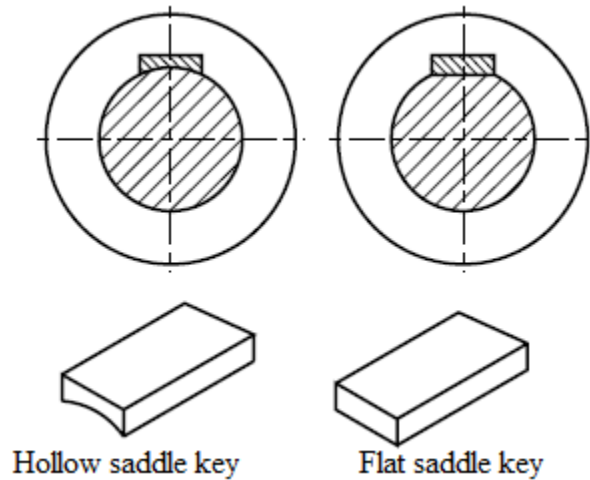
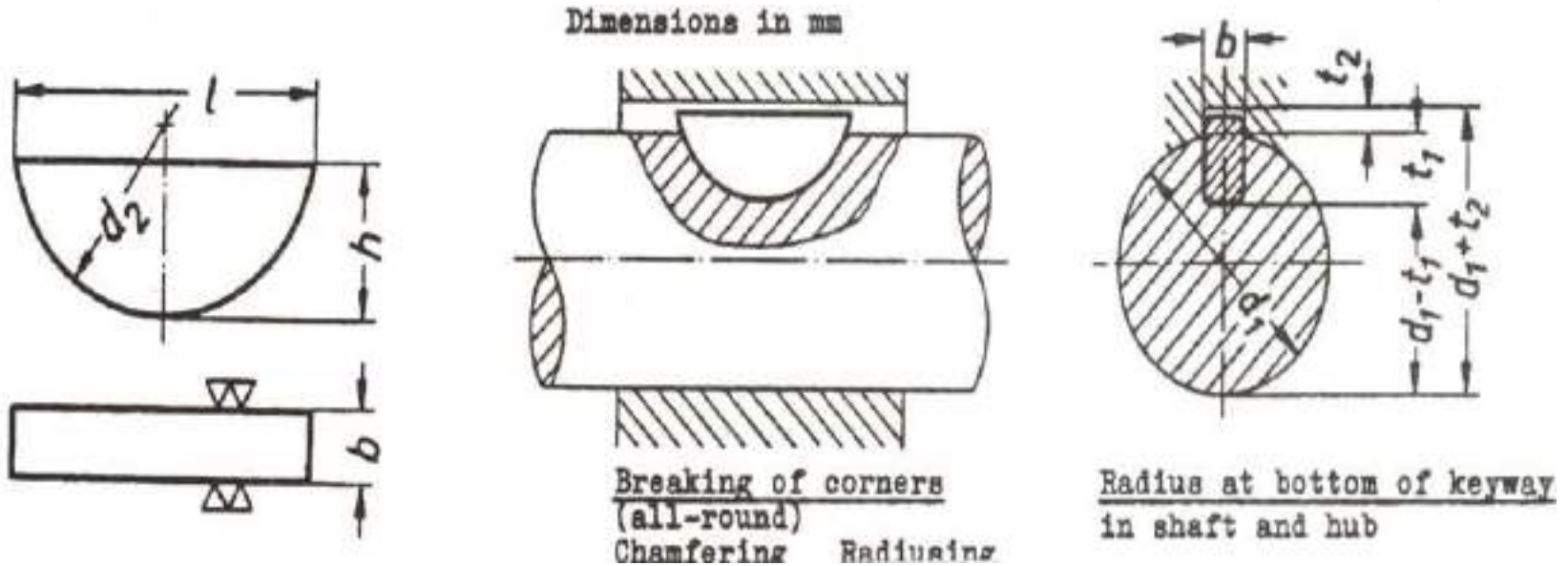


(a) Feather key, (b) rectangular key, and (c) square key

Sunk taper key



Woodruff key (used in automobile and machine tools, easy tilting in recess)

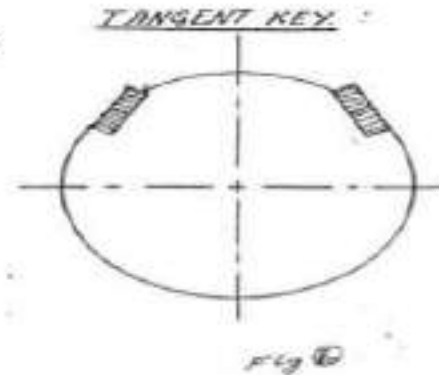


saddle keys

Used for light loads,
Key is fitted in the hub and flat is
rest on shaft surface

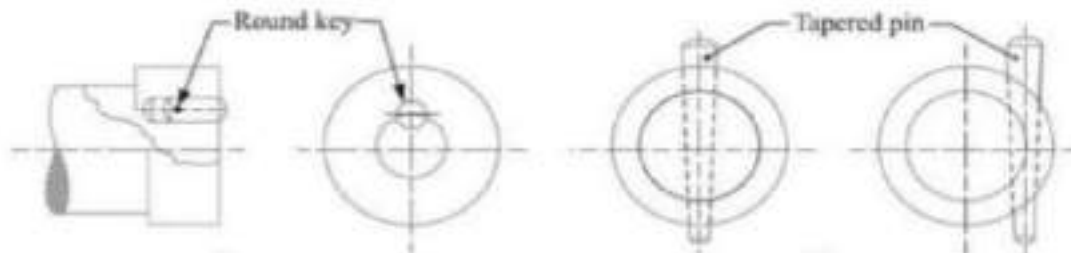
Tangent Keys

- The tangent key are fitted in a pair at right angles.
- Each key is to withstand torsion in one direction only.
- They are used in large heavy duty shaft.



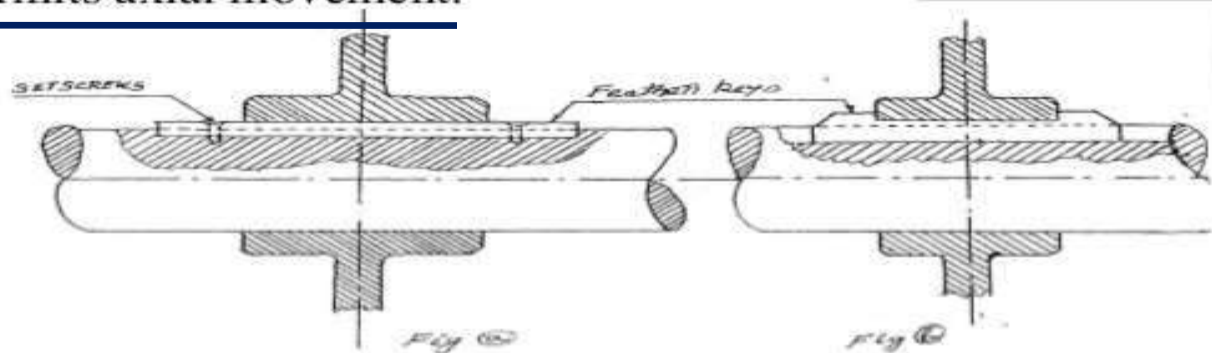
Round key

- The round keys, are circular in section and fit into holes drilled partly in the shaft and partly in the hub.
- They have advantage that their keyways may be drilled and reamed after the mating parts have been assembled.

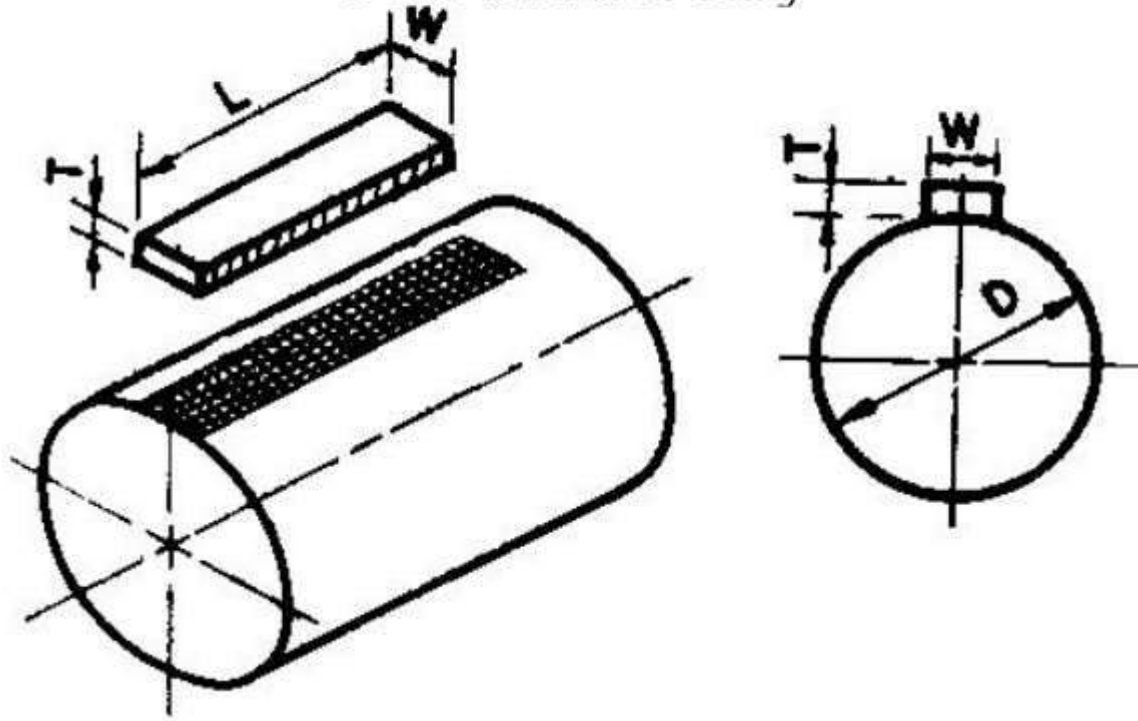


Feather key

- A key attached to one member of a pair and which permits relative axial movement is known as **feather key**.
- It is a special type of parallel key which transmits a turning movement and also permits axial movement.



Flat saddle key



Key design Procedure

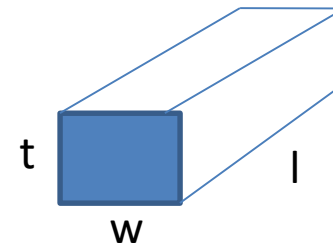
Step1: find torque. Using power eqn. $P=2\pi NT/60$

Step2: Calculate diameter of shaft “d” As per Max τ theory., $T = \pi/16 \times [\tau] \times d^3$

Step3: Based on d of shaft find key sizes – according to type of key selected.
w=?, t=? And L=?

Step4: Check for induced shear stress

Material of shaft = material of Key



$$T = F \cdot R$$

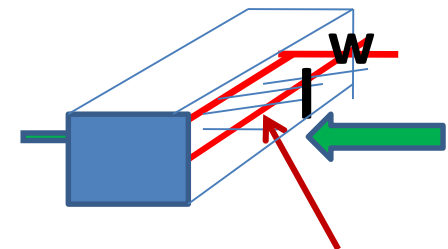
σ or $\tau = F/A$, Therefore $F = \tau \times A$

$$T = \tau \times A \times R$$

Where $R = d/2$, d- shaft dia,
 A for rectangular key, $A = l \times w$

$$T = \tau_{ind} \times l \times w \times d/2$$

$\tau_{ind} < [\tau]$ safe design



Shear Area is a plane

Kp1. Design a parallel key for the following data . Power= 20Kw at 1200 rpm. The shaft's allowable shear stress is 50 Mpa.

DATA					
P	20000 w				
N	1200 Rpm				
[τ]	50 N/mm²				
Key	parallel key				
Step1	find torque. Using power eqn. $P=2\pi NT/60$				
		T=	$60 \times P / (2 \times 3.141 \times N)$		
				159.18 N-m	
		T =	159185 N-mm		

Step3: Based on d of shaft find key sizes – according to type of key selected.

w=?, t=? And L=?

given key = parallel key

Refer page 5.16/DDB

to find sizes

shaft d= 28 mm

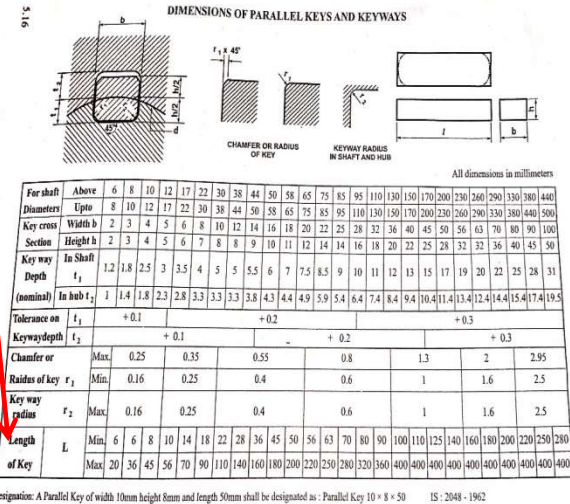
for d value , above 22 and upto 30 mm,

w= 8 mm

t /h= 7 mm

refer pg5.17/DDB

l = 45 mm



Step4: Check for induced shear stress		
Material of shaft = material of Key		
$[\tau]_{\text{shaft}} = [\tau]_{\text{key}}$ also		
T=	F x R	
F=	stress x A	
T=	stress x A x D/2	
A=	l x w	
T=	l x w x τ x d/2	
T=	45 x 8 x τ x 28/2	
τ induced: 31.6 N/mm² $[\tau]=50$		
31.584 < [50]		
design is safe		

Couplings

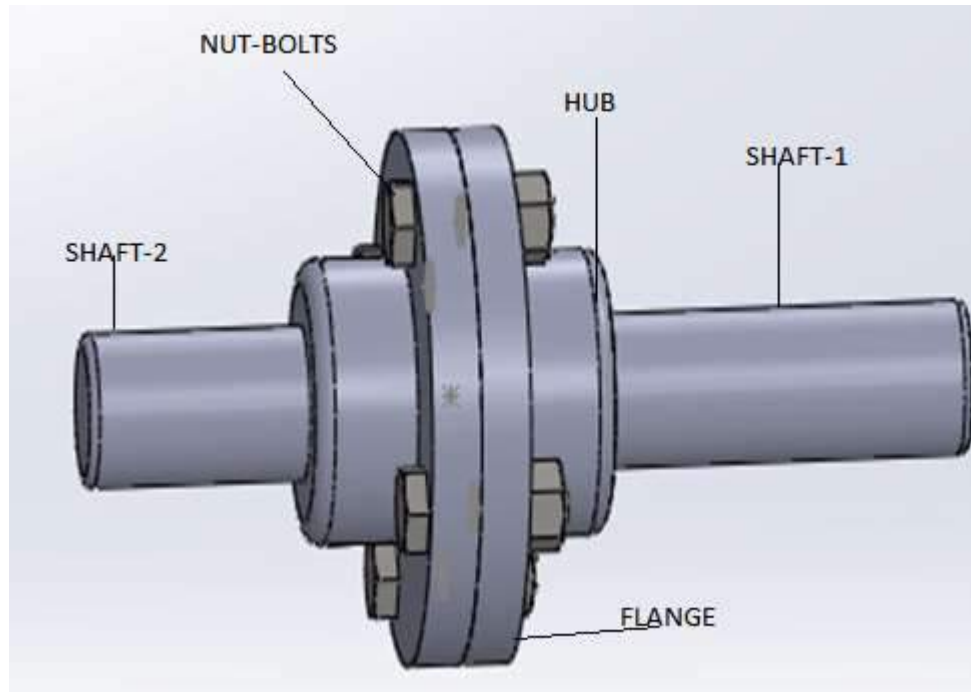
To transmit **motion** from driver to driven, act as connector between them

Types

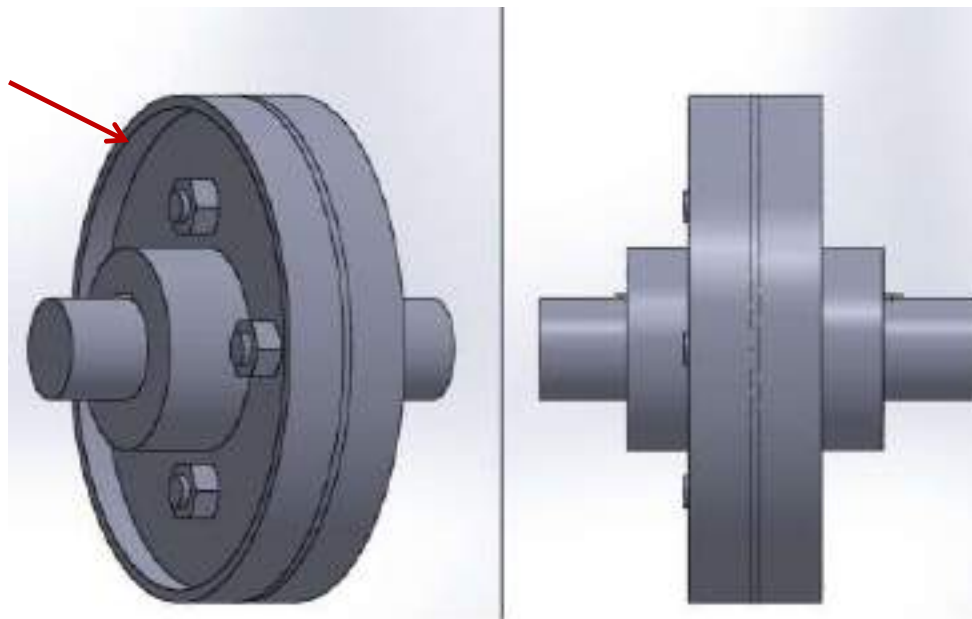
1. Rigid flange coupling ----a. Protective type
b. non protective
2. Flexible coupling– Bushed pin type

Clutch and coupling:

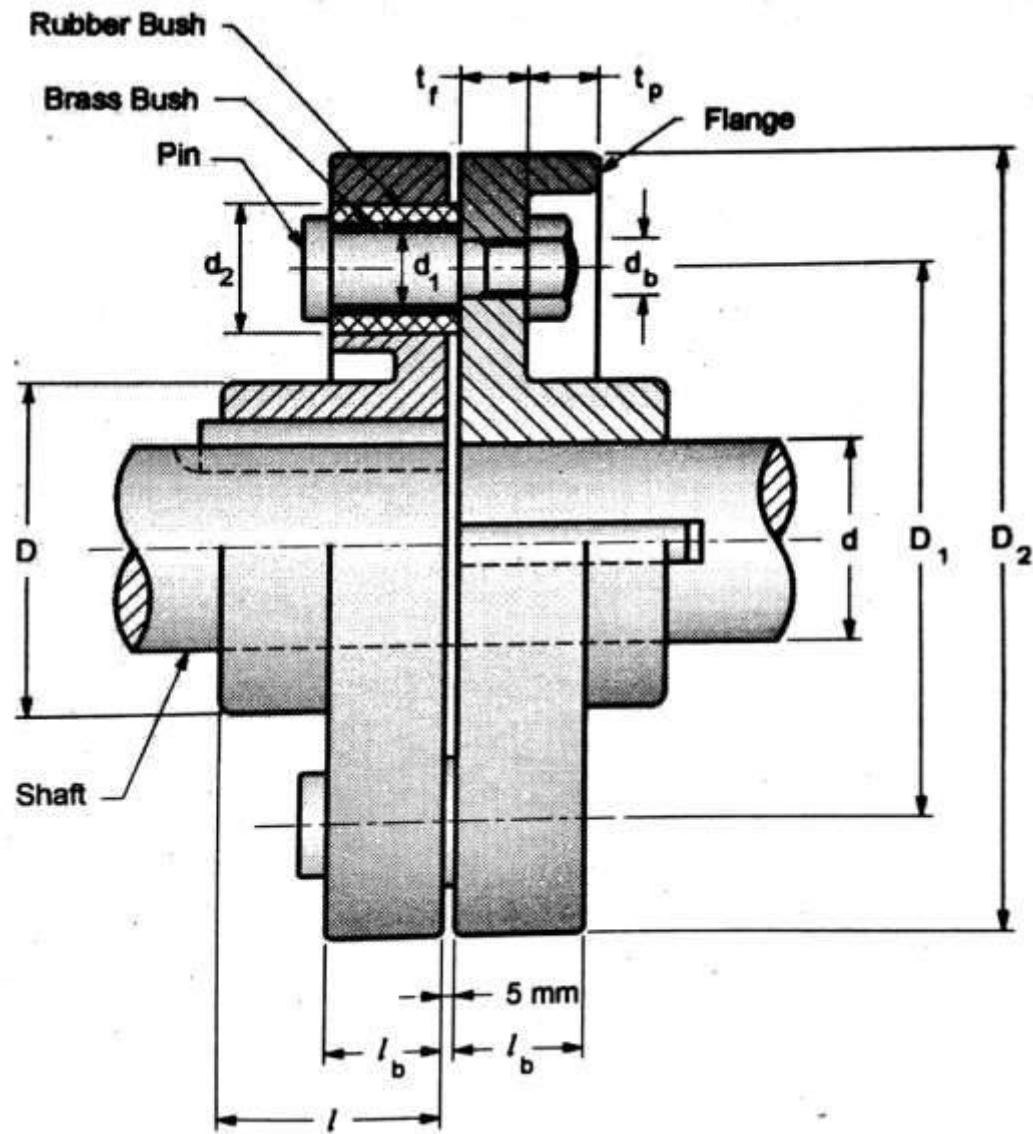
both functions are same , but clutch is distinguished by the transmission of motion can be intermittent . that is, when desired to stop or start is possible by disengage or engage the clutch.



Non protective flange



Protective flange



**Bushed Pin-
Flexible coupling**

Design procedure for Flange coupling (protective and non protective)

Step1: Find Torque using power eqn.

Step2: Find 'd' of the shaft using Max. T equation

Step3: List the basic sizes of the coupling using 'd' of the shaft
refer Pg. 7.134/DDB

Step4: design for hub(treating it as hollow shaft, $d_i = d$ of the shaft, $d_o = 2d$)

Step5: Design for key

Step6: Design for flange

Step7:Design for bolt

Step8: Draw the coupling with NTS free hand sketching with dimensions

CP1. Design and draw a rigid coupling for the following specifications.

Power 15 kW at 300 rpm. Allowable shear stress for Shaft and Key is 40 N/mm². Bolts working stress should not exceed 30 N/mm². Flange is made of Castiron and its limited shear stress is 14 N/mm². The torque transmission is 25% higher than the actual torque. The crushing stress for key is 2.5 times of its shear stress.

DATA			
P	15000	Watts	
N	300	rpm	
[τ] shfat& key =		40	N/mm²
bolts working stress=		30	N/mm²
crushing is thrice the Shear stress			
CI for FLANGE[τ] =		14	N/mm²
Tmax =	1.25 T		

Step1 : Torque finding

$$P = 2 \times 3.141 \times N \times T / 60$$

$$T = p \times 60 / (2 \times 3.141 \times N)$$

Some times service factor will be given,

$$T = \text{service factor} \times T_{\text{cal}}$$

P=	(2x3.141xN x T)/60
T=	477.5549188 Nm 477554.9188 N-mm
Tmax =	596943.6485 N-mm

$$(T_{\text{max}} = 1.25 \times T_{\text{cal}})$$

Step2: Find diameter of The shaft

$$T = 3.141/16 \times \tau \times d^3$$

Std the "d" to R20 series

	D ³ =	T x 16 / (3.141 x τ)
		76019.567
	d =	42.203469 mm
R20 series	d=	45 mm

Step3: List the basic proportions of coupling use d std

Refer Pg.7.134 /DDB

d = dia of the shaft

$D = 2d$ (d_0 = outter dia of the hub),

d_i = inner dia of the hub = dia of the shaft

$L = 1.5 d$ hub length

$D_1 = \text{PCD} = 3d$

D_2 = flange dia = $4d$

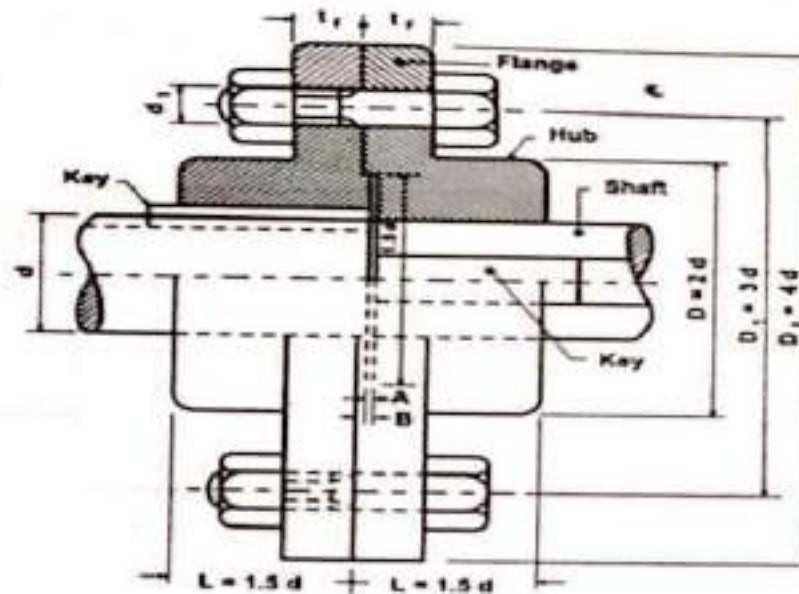
n = no.of bolts selecte according to d of the shaft

T_f = thickness of flange = $d/2$

Refer pg.7.134/DDB			
		shaft $d =$	45 mm
	HUB	$d_i =$	45 mm
		$d_o =$	90 mm
		L of Hub =	67.5 mm
		PCD for bolts =	135 mm
		Flange dia =	180 mm
		No.of bolts $n = (40 < d < 100)$	4 nos

FLANGE COUPLINGS

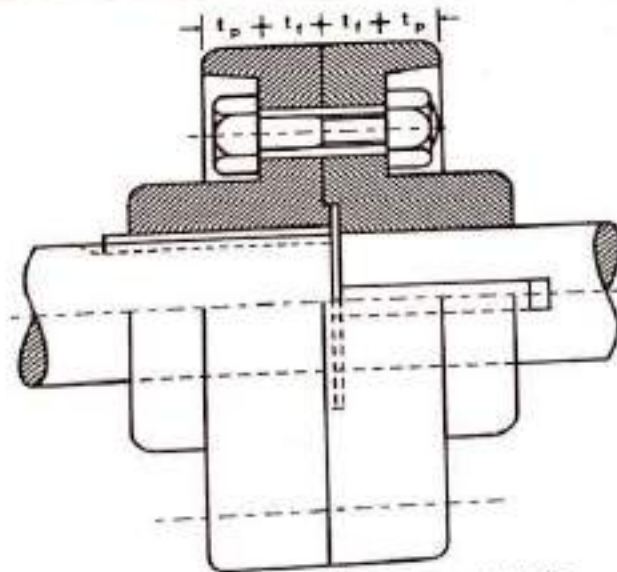
Equation	Nomenclature
$D = 2 d$	d Nominal diameter of bolts
$L = 1.5 d$	d Shaft diameter
$D_1 = 3 d$	D Outside diameter
$D_2 = 4 d$	D_1 Diameter of bolt circle
$t_f = \left(\frac{d}{2}\right)$	n number of bolts
$n = 3$ for d upto 40 mm $= 4$ for d upto 100 mm $= 6$ for d upto 180 mm	t_f Thickness of flange
$t_p = \left(\frac{d}{4}\right)$	t_p Thickness of protecting flange



UNPROTECTED TYPE FLANGE COUPLING

HUB

	Equation	Nomenclature
HUB	$T = \frac{\pi}{16} \tau_c \left[\frac{D^4 - d^4}{D} \right]$	τ_s Allowable shear stress for shaft
KEY	$T = l w \tau_k \left(\frac{d}{2} \right)$	τ_b Allowable shear stress for bolt
	$T = \left(\frac{t}{2} \right) \sigma_c \left(\frac{d}{2} \right)$	τ_k Allowable shear stress for key
FLANGE	$T = \pi \left(\frac{D^2}{2} \right) \tau_c t_r$	τ_c Allowable shear stress for flange
BOLTS	$T = n \left(\frac{\pi}{4} \right) d_1^2 \tau_k \left(\frac{D_1}{2} \right)$	σ_{cb} Allowable shear stress for crushing stress for bolt
	$T = n d_1 t_r \sigma_{cb} \left(\frac{D_1}{2} \right)$	σ_{ck} Allowable shear stress for crushing stress for key
		T Torque transmitted by the coupling

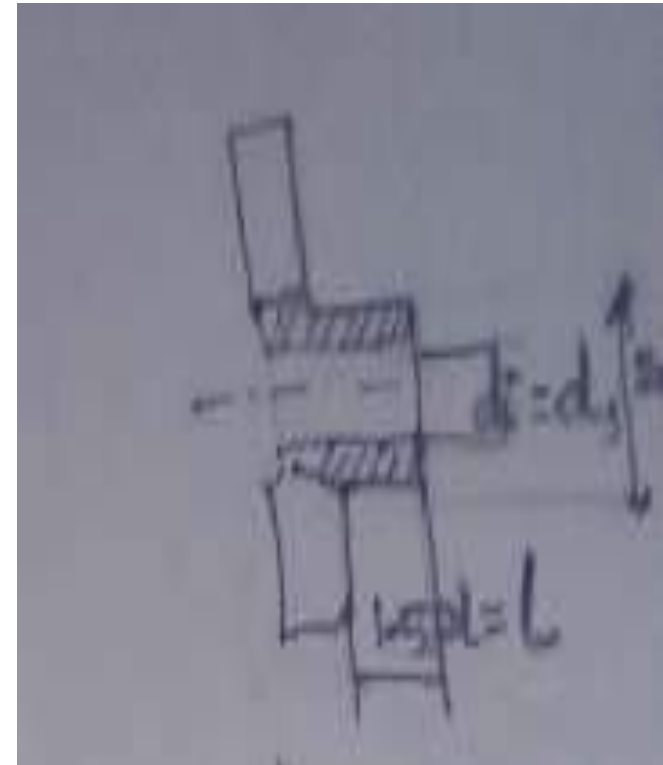


PROTECTIVE TYPE FLANGE COUPLING

Step 4: Hub design

$$T = 3.141/16 \times \tau \times [(d_o^4 - d_i^4)/d_o]$$

hollow shaft with torsion only	
$[\tau] =$	40 N/mm ²
$T =$	$(\pi/16) \times \tau_{ind} \times [(d_o^4 - d_i^4)/d_o]$
$(\pi/16)$	0.196313
$(d_o^4 - d_i^4)$	61509375
$(d_o^4 - d_i^4)/d_o$	683437.5
$T =$	596943.6
596943.6485	134167.3 τ_{ind}
$\tau_{ind} =$	4.449 N/mm ²
4.449 < [40]	
$\tau_{ind} < [\tau]$	safe design

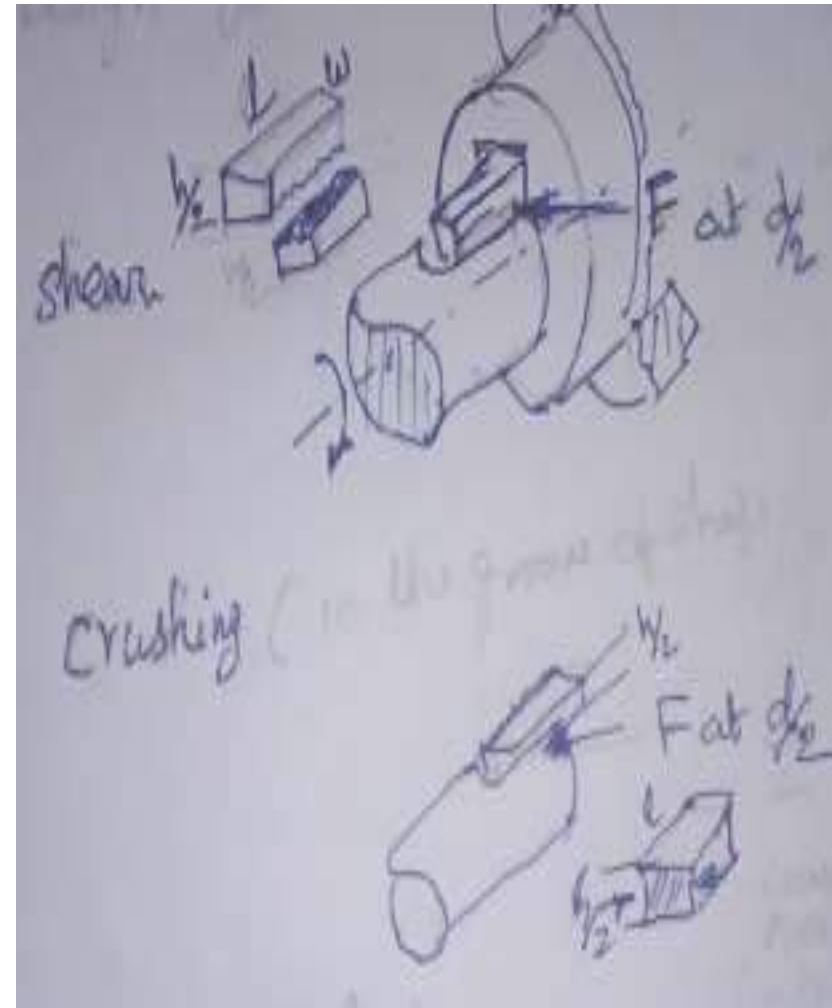


STEP.5 Key Design

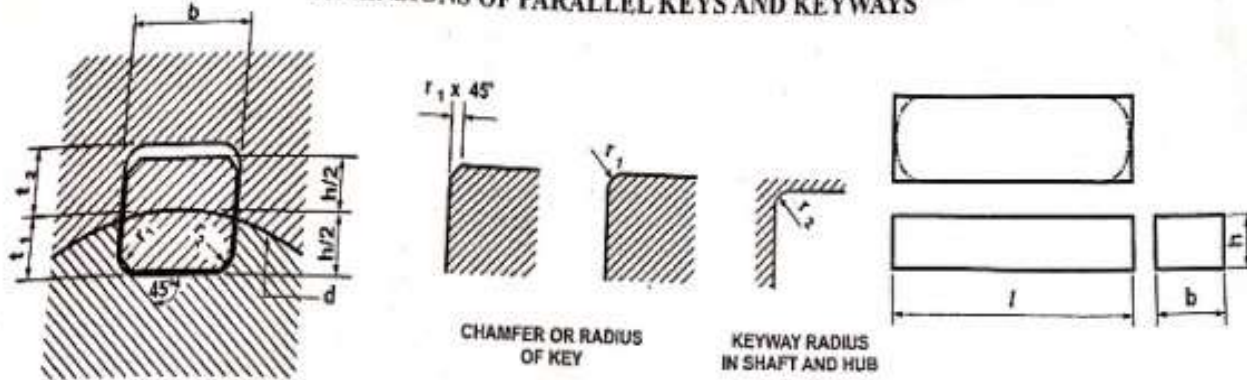
In shear

In crushing

Rectangular Key (Assumed) parallel		
Shaft d=	45 mm	
Refer Pg.5.16 & 5.17		
above	44 mm	
upto	50 mm	
b =	14 mm	
h =	9 mm	
l = hub length	67.5	
Pg.5.17/DDB preferred length	70 mm	



DIMENSIONS OF PARALLEL KEYS AND KEYWAYS



All dimensions in millimeters

For shaft Diameters	Above	6	8	10	12	17	22	30	38	44	50	58	65	75	85	95	110	130	150	170	200	230	260	290	330	380	440					
	Upto	8	10	12	17	22	30	38	44	50	58	65	75	85	95	110	130	150	170	200	230	260	290	330	380	440	500					
Key cross Section	Width <i>b</i>	2	3	4	5	6	8	10	12	14	16	18	20	22	25	28	32	36	40	45	50	56	63	70	80	90	100					
	Height <i>h</i>	2	3	4	5	6	7	8	8	9	10	11	12	14	14	16	18	20	22	25	28	32	32	36	40	45	50					
Key way Depth (nominal)	In Shaft <i>t</i> ₁	1.2	1.8	2.5	3	3.5	4	5	5	5.5	6	7	7.5	8.5	9	10	11	12	13	15	17	19	20	22	25	28	31					
	In hub <i>t</i> ₂	1	1.4	1.8	2.3	2.8	3.3	3.3	3.3	3.8	4.3	4.4	4.9	5.9	5.4	6.4	7.4	8.4	9.4	10.4	11.4	13.4	12.4	14.4	15.4	17.4	19.5					
Tolerance on Keyway depth	<i>t</i> ₁	+0.1					+0.2										+0.3															
	<i>t</i> ₂	+0.1										+0.2										+0.3										
Chamfer or Radius of key <i>r</i> ₁	Max.	0.25			0.35			0.55					0.8					1.3					2					2.95				
	Min.	0.16			0.25			0.4					0.6					1					1.6					2.5				
Key way radius <i>r</i> ₂	Max.	0.16			0.25			0.4					0.6					1					1.6					2.5				
	Min.	6	6	8	10	14	18	22	28	36	45	50	56	63	70	80	90	100	110	125	140	160	180	200	220	250	280					
Length of Key	Max.	20	36	45	56	70	90	110	140	160	180	200	220	250	280	320	360	400	400	400	400	400	400	400	400	400	400	400				

DESIGN DATA - PSG TECH

Designation: A Parallel Key of width 10mm height 8mm and length 50mm shall be designated as : Parallel Key 10 × 8 × 50 IS : 2048 - 1962

PREFERRED LENGTHS OF PARALLEL KEYS

Width (b), mm	Preferred Length (L), mm									
	2	3	4	5	6	8	10	12	14	16
6	6	-	-	-	-	-	-	-	-	-
8	8	8	-	-	-	-	-	-	-	-
10	10	10	10	-	-	-	-	-	-	-
12	12	12	12	-	-	-	-	-	-	-
14	14	14	14	14	-	-	-	-	-	-
16	16	16	16	16	-	-	-	-	-	-
18	18	18	18	18	18	-	-	-	-	-
20	20	20	20	20	20	-	-	-	-	-
22	22	22	22	22	22	22	-	-	-	-
25	25	25	25	25	25	-	-	-	-	-
28	28	28	28	28	28	28	-	-	-	-
32	32	32	32	32	32	32	-	-	-	-
36	36	36	36	36	36	36	36	-	-	-
40	40	40	40	40	40	40	40	-	-	-
45	45	45	45	45	45	45	45	-	-	-
50	50	50	50	50	50	50	50	50	-	-
56	56	56	56	56	56	56	56	56	56	-
63	63	63	63	63	63	63	63	63	63	63
70	70	70	70	70	70	70	70	70	70	70
80	80	80	80	80	80	80	80	80	80	80
90	90	90	90	90	90	90	90	90	90	90
100	100	100	100	100	100	100	100	100	100	100
110	110	110	110	110	110	110	110	110	110	110
125	125	125	125	125	125	125	125	125	125	125
140	140	140	140	140	140	140	140	140	140	140
160	160	160	160	160	160	160	160	160	160	160
180	180	180	180	180	180	180	180	180	180	180
200	200	200	200	200	200	200	200	200	200	200
220	220	220	220	220	220	220	220	220	220	220
250	250	250	250	250	250	250	250	250	250	250
280	280	280	280	280	280	280	280	280	280	280
320	320	320	320	320	320	320	320	320	320	320
360	360	360	360	360	360	360	360	360	360	360
400	400	400	400	400	400	400	400	400	400	400

Height
(h), mm

2 3 4 5 6 8 8 8 9 10 11 12 14 14 16 18 20 22 25 28 32 36 40 45 50 56 63 70 80 90 100



a.) SHEAR failure

$$T = F \times R$$

R= the tangential force acting at periphery of shaft because the key(half key) is located at shaft

$$\text{Stress} = F/A \text{ Therefore, } F = \text{stress} \times A$$

$$T = \text{Stress} \times \text{Area} \times R$$

$$A = l \times W$$

$$980 \text{ mm}^2$$

$$T =$$

$$596943.6 \text{ N-mm}$$

$$R =$$

$$22.5 \text{ mm}$$

$$596943.6485$$

$$22050 \text{ } \tau_{ind}$$

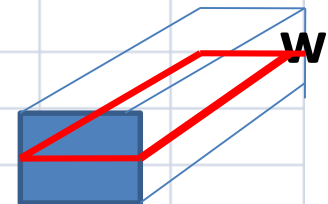
$$\tau_{ind} =$$

$$27.07227$$

$$27.0722 < [40]$$

$$\tau_{ind} < [\tau]$$

Safe design



b.) crushing failure

$$T = F \times R$$

R= the tangential force acting at periphery of shaft because the key(half key) is located at shaft

Stress= F/A Therefore, F= stress x A

A = $l \times h/2$ (crushing occurs in contained material zone)
 (half in shaft key way only to be considered)

A = 315 mm²
 T= 596943.6 N-mm
 R= 22.5 mm²

596943.6485 7087.5 σ

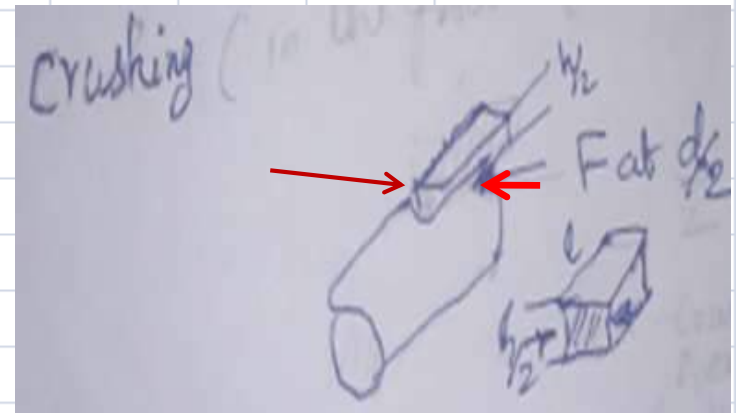
$\sigma = 84.22485$

To find [σ]

$[\sigma] = 2.5 \tau$
 $[\sigma] = 100 \text{ N/mm}^2$

$84.224 < [100]$

$\sigma_{ind} < [\sigma]$ Safe design



Step 5: Flange design

Flange is made of CI- cast Iron material

Flange failure occurs at the junction of HUB

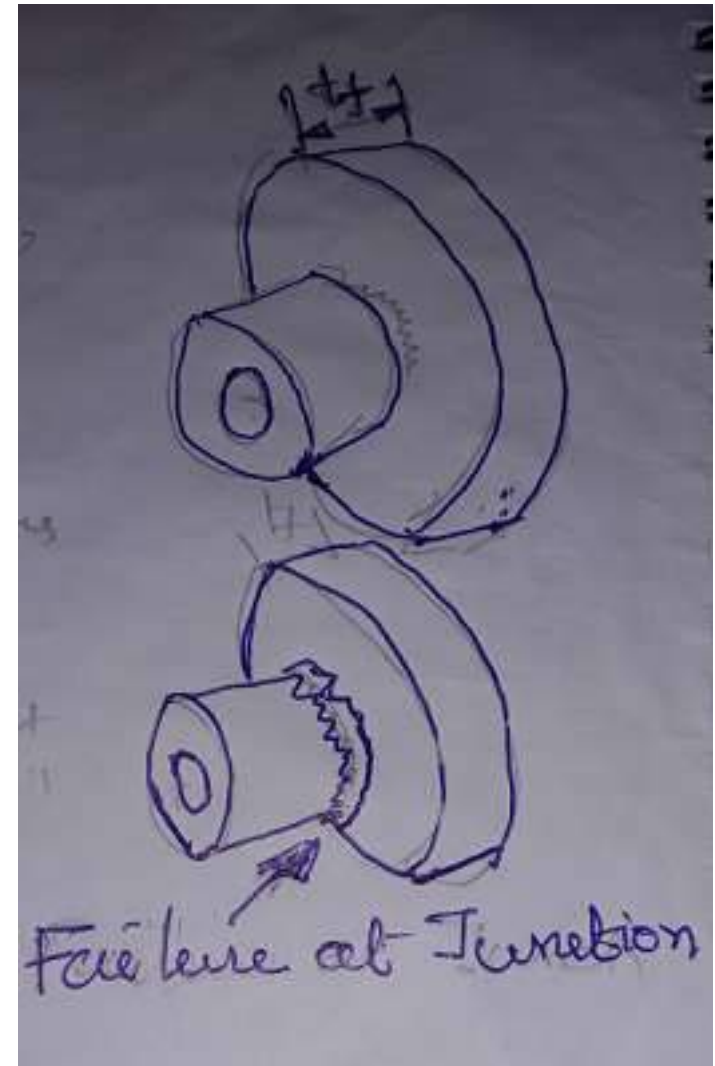
$$T = F \times R$$

$$T = \text{Stress} \times A \times R$$

for ,R, the hub diameter "do" to be selected
because The tangential force is taken at periphery of HUB

$$t_f = \text{Thck ness of Flage} = 0.5 d$$

$$t_f = 22.5 \text{ mm}$$

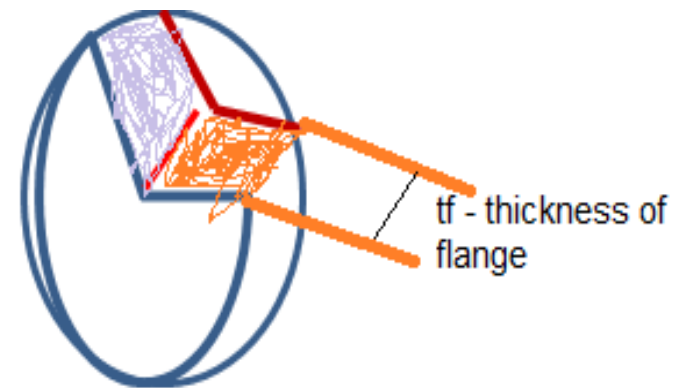
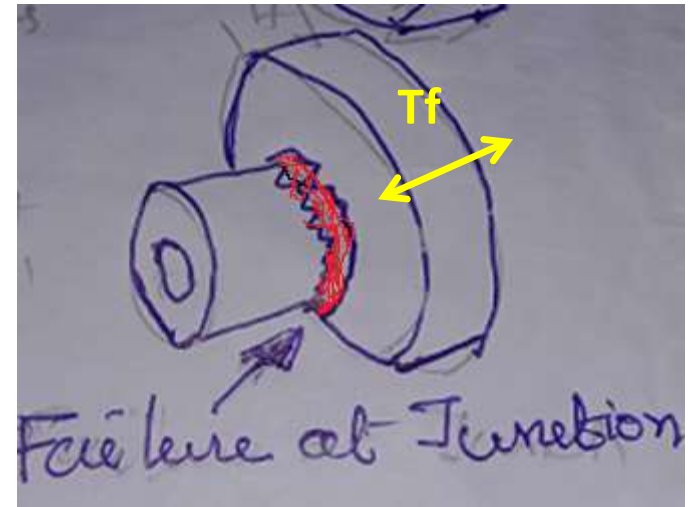


$A = \pi \times d_o \times t_f$	6360.525	mm ²
---------------------------------	----------	-----------------

Failure takes around its circumference, at the **junction** of Hub and Flange

Circumference = πD_o
 $A = \pi D_o t_f$

$A = \pi \times d_o \times t_f$	6360.525	mm ²
$T =$	596943.6	N-mm
$R =$	45	mm
$596943.6485 =$	286223.6	τ_{ind}
$\tau_{ind} = 2.085585 \text{ N/mm}^2$		
$2.08 < [14] \quad \text{safe design}$		



Design of bolts in shear and crushing

$$T = F \times R$$

(R, the radial distance of

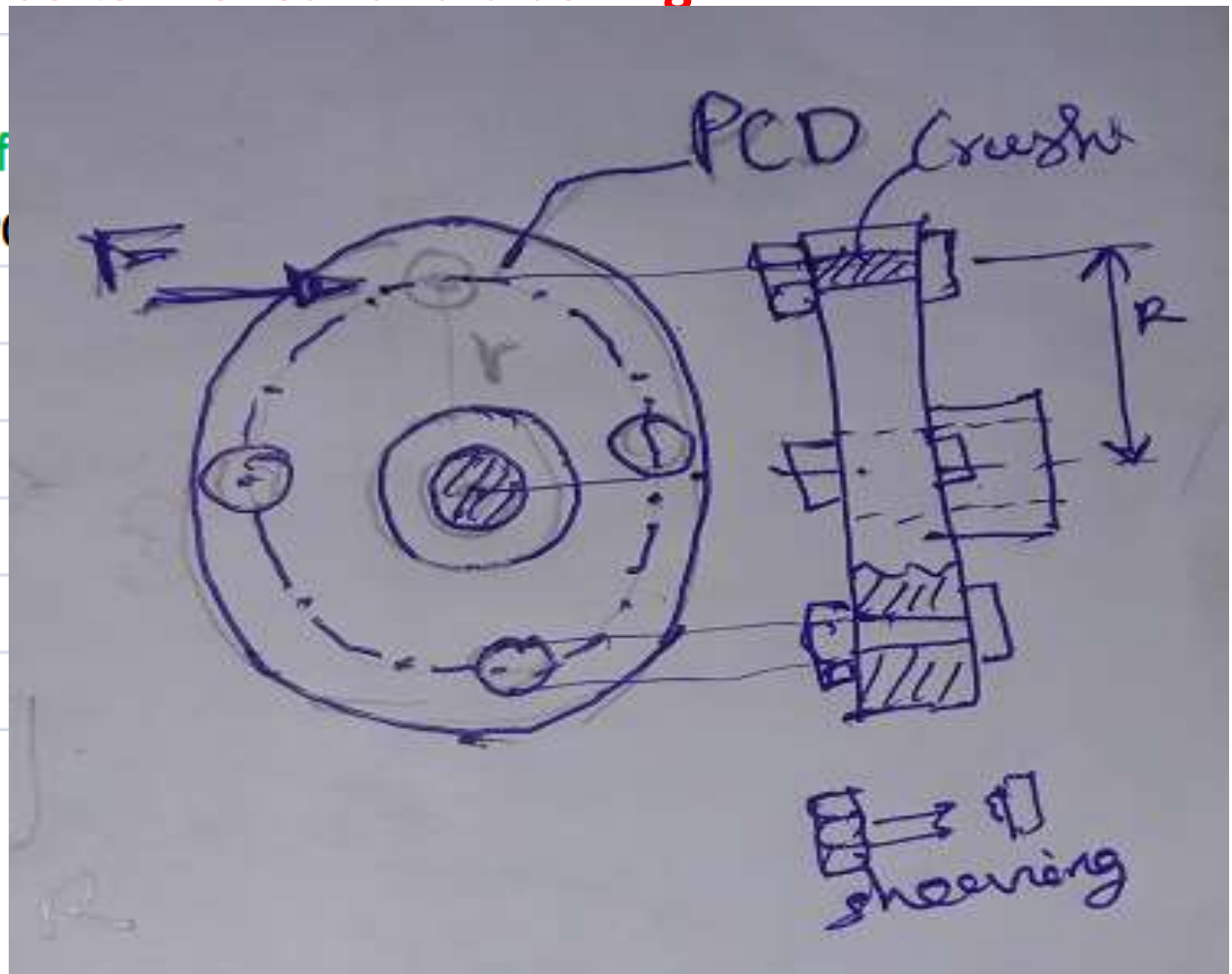
Tangential force acts at P

$$F = \text{stress} \times A$$

$$A = \frac{\pi}{4} \times d^2$$

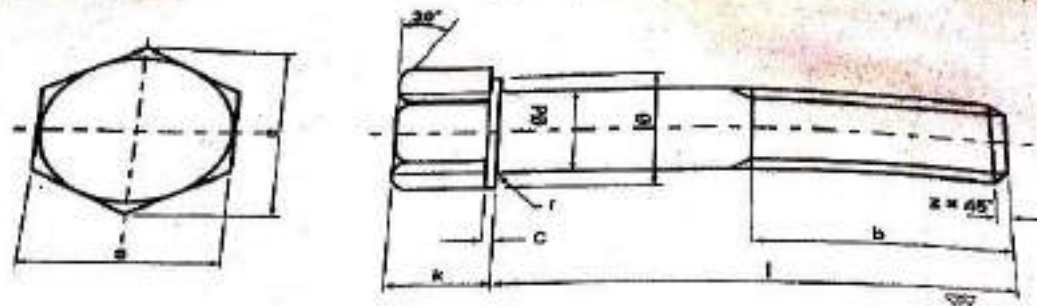
$$\text{no. of bolts} = 4$$

$$T = [\tau] \cdot \left(\frac{\pi}{4} \times d^2 \right) \cdot n \cdot R$$



T =	596943.6	N-mm	
[τ] =	30	N/mm²	
n	4	nos	
R = (PCD value)	67.5	mm	
d=	???		
596943.6485	=	6360.5	d²
d² =	93.85132		
d =	9.687689		
Ref.pg.5.49/DDB, std,			
d minimum= 9.78	to d max	M10 selected	
		d= 10 mm	

HEXAGONAL BOLT (contd...)
Precision (P) Grade
M5 - M30



All dimensions in mm

Size	M 5	M 6	M 8	M 10	M 12	M 16	M 20	M 24	M 30
d Max.	5	6	8	10	12	16	20	24	30
	4.82	5.82	7.78	9.78	11.73	15.73	19.67	23.67	29.67
s Max.	8	10	13	17	19	24	30	36	46
	7.85	9.78	12.73	16.73	18.67	23.67	29.67	35.38	45.38
k Max.	3.65	4.15	5.65	7.18	8.18	10.18	13.22	15.22	19.26
	3.35	3.85	5.35	6.82	7.82	9.82	12.78	14.78	18.74
b Min. for $l \leq 130$	16	18	22	26	30	38	46	54	66
for $130 < l < 300$	-	24	28	32	36	44	52	60	72
for $l > 300$	-	-	-	-	-	57	65	73	85
e Min.	7.2	9	11.7	15.3	17.1	21.6	27	32.4	41.4
c ~	0.2	0.3	0.4	0.4	0.4	0.4	0.4	0.5	0.5
e Min.	8.87	11.05	14.38	18.9	21.1	26.75	33.53	39.98	51.28
r Min.	0.2	0.25	0.4	0.4	0.6	0.6	0.8	0.8	1
z Max.	0.8	1	1.25	1.5	1.75	2	2.5	3	3.5
l	30 - 75	30 - 80	30 - 90	35 - 120	40 - 120	50 - 220	60 - 220	70 - 220	90 - 220

IS : 2389

IS : 1364

Crushing failure of Bolt

Crushing = straining of material in confined zone

$$\sigma_c = 2.5 \tau$$

$$\sigma_{c \text{ ind}} < [\sigma_c]$$

$$[\sigma_c] = 2.5 \tau$$

$$[\sigma_c] = 100 \text{ N/mm}^2$$

75

$$T = F \times R$$

$$T = \text{Stress} \times A \times R$$

(Here the R, , tangetial force acts at PCD of flange, |

n=	4 Nos		
T=	596944	N-mm	
A = n x d x tf			
d=	10 mm		
tf=	22.5 mm		
n =	4 nos		
R= pcd/2	67.5 mm		
A= 4 x 10 x 2	900	mm²	

T = n x dx tf x σc x pcd/2			
596943.65	=	60750	σc ind
σc ind	9.8262		
9.8 < [100]	Safe design		
[75]			

Flexible couplings

Types

1. Bushed Pin type
2. Oldham's couplings
3. Universal coupling

Why?

1. Abutting ends of shaft to be connected
2. Not exact alignment of shafts
3. Permits axial misalignment without power loss at the shaft

Design procedure for (Bushed pin flexible coupling)

Step1: Find Torque using power eqn.

Step2: Find 'd' of the shaft using Max. T equation

Step3: List the basic sizes of the coupling using 'd' of the shaft
refer Pg. 7.134/DDB

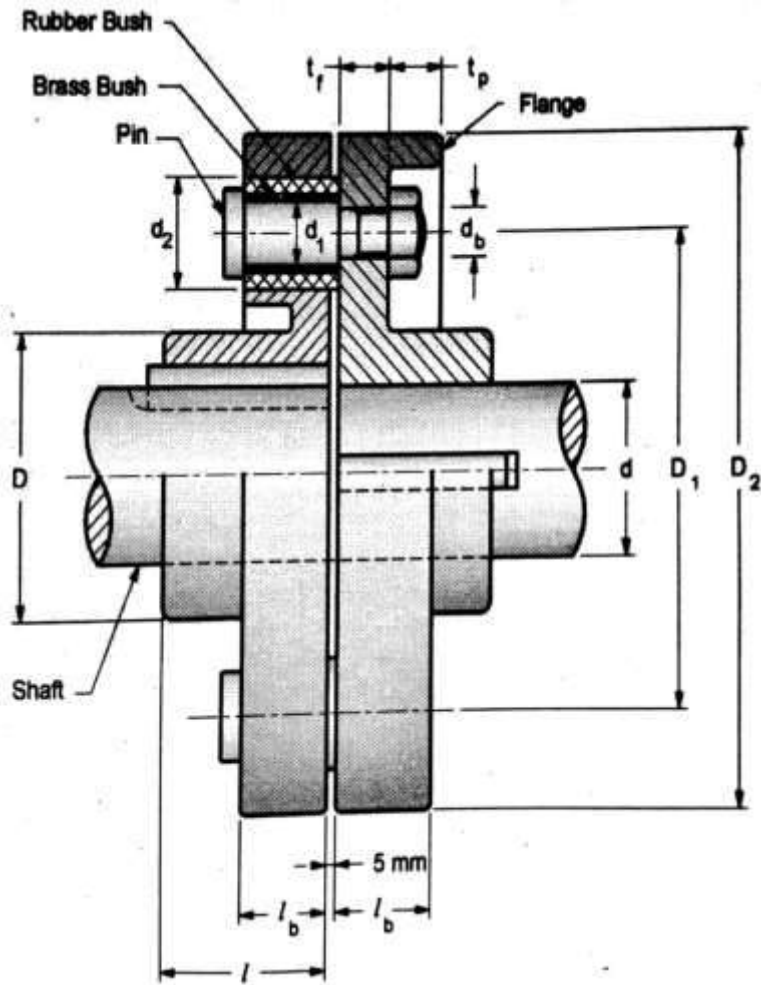
Step4: design for hub(treating it as hollow shaft, $d_i = d$ of the shaft, $d_o = 2d$)

Step5: Design for key

Step6: Design for flange

Step7:Design for PIN(instead of Bolt

Step8: Draw the coupling with NTS free hand sketching with dimensions



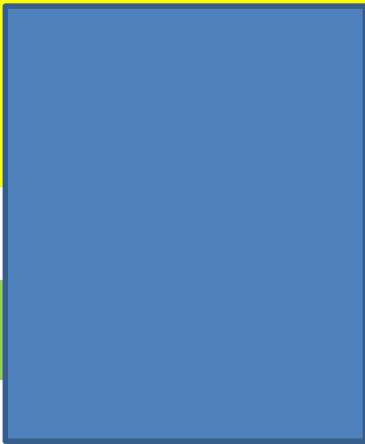
HUB

	Equation	Nomenclature
HUB	$T = \frac{\pi}{16} \tau_c \left[\frac{D^4 - d^4}{D} \right]$	τ_k Allowable shear stress for shaft
KEY	$T = l w \tau_k \left(\frac{d}{2} \right)$	τ_b Allowable shear stress for bolt
	$T = \left(\frac{t}{2} \right) \sigma_c \left(\frac{d}{2} \right)$	τ_k Allowable shear stress for key
FLANGE	$T = \pi \left(\frac{D^2}{2} \right) \tau_c t_f$	τ_c Allowable shear stress for flange
		σ_{cb} Allowable shear stress for crushing stress for bolt
BOLTS	$T = n \left(\frac{\pi}{4} \right) d_1^2 \tau_b \left(\frac{D_1}{2} \right)$	σ_{ck} Allowable shear stress for crushing stress for key
	$T = n d_1 t_f \sigma_{cb} \left(\frac{D_1}{2} \right)$	T Torque transmitted by the coupling

CP2. Design and draw a flexible coupling for the following specifications.

Power 32 kW at 960 rpm. Allowable shear stress for Shaft and Key is 40 N/mm². Bolts working stress should not exceed 30 N/mm². Flange is made of Castiron and its limited shear stress is 15 N/mm². The torque transmission is 20% higher than the actual torque. The crushing stress for key 80 N/mm².

DATA				
P	32000	Watts		
N	960	rpm		
[τ] shfat& key =		40	N/mm²	
bolts working stress=		30	N/mm²	
crushing is		80	N/mm²	
CI for FLANGE[τ] =		15	N/mm²	
Tmax =	1.20 T			

STEP1	Torquqe Finding		
	P=	$(2 \times 3.141 \times N \times T) / 60$	
	T=		Nm
			N-mm
	Tmax =		N-mm

STEP2 Shaft dia 'd' calculating

$$T = \frac{3.141}{16} \times \tau \times d^3$$

$$D^3 = \frac{T \times 16}{(3.141 \times \tau)}$$

d = mm

R20 serie d = mm

STEP 3 LIST the BASIC dimensions

Refer pg.7.134/DDB

		shaft d=		mm			
	HUB	di =		mm			
		do =		mm			
		L of Hub=		mm			
	PCD for bolts =			mm			
	Flange dia =			mm			
	No.of bolts n=	(40<d<100)		nos		but use pin	in case 6

STEP4 HUB design and check $\tau_{ind} < [\tau]$

hollow shaft with torsion only

$[\tau] = 40 \text{ N/mm}^2$

$T = (\pi/16) \times \tau_{ind} \times [(d_o^4 - d_i^4)/d_o]$

$(\pi/16)$
 $(d_o^4 - d_i^4)$
 $(d_o^4 - d_i^4)/d_o$
 $T =$



382043.9351

94230 τ_{ind}

$\tau_{ind} = 4 \text{ } ^2$



$\tau_{ind} < [\tau]$

safe design

STEP 4	KEY deisgn			
	Rectangular Key (Assumed)		parallel keys	
	Shaft d=	40 mm		
	Refer Pg.5.16 & 5.17			
	above	38 mm		
	upto	44 mm		
	b =	12 mm		
	h =	8 mm		
	l = hub length	60		
Pg.5.17	prefered length =	63 mm		

5.17

a.) SHEAR failure					
	$T = F \times R$				
		R= the tangential force acting at periphery of shaft because the key(half key) is located at shaft			
		Stress= F/A Therefore, F= stress x A			
	$T = \text{Stress} \times \text{Area} \times R$				
	$A = l \times W$		756 mm ²		
	T =		382043.94 N-mm		
	R=		20 mm		
		382043.9351	15120	τ_{ind}	
	$\tau_{ind} =$		25.267456		
	25.267				
	$27.0723 < [40]$				
	$\tau_{ind} < [\tau]$				
		Safe design			

b.) crushing failure

$$T = F \times R$$

R = the tangential force acting at periphery of shaft
because the key (half key) is located at shaft

$$\text{Stress} = F/A \quad \text{Therefore, } F = \text{stress} \times A$$

$$A = l \times h/2 \quad \text{(crushing occurs in contained material zone)}$$

(half in shaft key way only to be considered)

$$A = 252 \text{ mm}^2$$
$$T = 382043.94 \text{ N-mm}$$
$$R = 20 \text{ mm}$$

$$382043.9351 \quad 5040 \text{ } \sigma$$

$$\sigma = 75.802368$$

To find $[\sigma]$

$$[\sigma] = 80 \text{ N/mm}^2$$

$$75.802 < [80]$$

$\sigma_{ind} < [\sigma]$ Safe design



STEP5 Design of Flange

Flange is made of CI- cast Iron material

Flange failure occurs at the junction of HUB

$$T = F \times R$$

$$T = \text{Stress} \times A \times R$$

for ,R, the hub diameter "do" to be selected because The tangential force is taken at periphery of HUB

$$t_f = \text{Thck ness of Flage} = 0.5 d$$

$$t_f = 20 \text{ mm}$$

$$A = \pi \times d_o \times t_f$$

$$5025.6 \text{ mm}^2$$

$$T =$$

$$382043.94 \text{ N-mm}$$

$$R =$$

$$40 \text{ mm}$$

~~$$596943.6485 =$$~~

$$201024 \tau_{ind}$$

$$\tau_{ind} = 1.9004892 \text{ N/mm}^2$$

$$1.900$$

[15]

~~$$2.08 < [14]$$~~

safe design

STEP6 Design of (bolts) Pin

$$T = F \times R$$

(R, the radial distance of the tangential force, which acts at PCD of The flange)

Tangential force acts at PCD of The flange

$$F = \text{stress} \times A$$

$$A = \frac{\pi}{4} \times d^2$$

no.of .Pins= 6 always

Pin is Subjected to Torsion & Bending

Dia of the Pin =?

$$D_{pin} = 0.5 \times d / \sqrt{n}$$

Where d= shaft diameter

$$n = 6$$

Direct torsion

$$\tau = W/A$$

$$A = \pi/4 \times d^2$$

W= load on the pin - Pressure on the pin

$$p = 0.8 \text{ N/mm}^2$$

Bending Stress

Pin treated As UDL loaded shaft

$$\sigma = M/Z$$

$$M = (W \times l)/2$$

$$w = P \times d \times l \quad \leftarrow P = w/A$$

d- dia of Pin

l- length of the pin

Z- section modulus of circular Pin

$$Z = \pi/64 \times d^4/(d/2)$$

$$[z = I/y]$$

Apply principal stress equation

$\sigma_{1,2}=?$

$\sigma_1 < [\tau]$ prove it

T =	382043.9351 N-mm								
R =	120 mm								
P =	0.8 N/mm ²								
W =	Load on the pin due to pressure								
T =	$n \times W \times R$								
Due to Pressure									
W =	$P \times A$ (Projected area)								
A =	$D_{pin} \times l$								
		Pin dia =	8.16497	8.1650					
$D_{pin} = 2d_{pin}$		(secured in the left flange enlarged portion)							
$D_{pin} =$	32.33 mm	(2d _{pin})	(add rubber bush thick = 6 x 2 = 12 & brass bush thick = 2 x 2 = 4 in mm) Totally = 12 + 4 = 16						
$l = t_f$	in general								
$l =$	25 mm								

w=	646.5986324				
Wtotal=	3879.591794	(6 x W)			
Now					
Shear Stress calculation					
$\tau = W/A$					
W=	3879.591794	N			
A= $\pi/4 * D_{pin}^2$					
A=	820.7625217	mm ²			
$\tau =$	4.726814019	N/mm ²			

**Now
Bending stress calculation**

$$\sigma = M/Z$$

$$[z = I/y]$$

$$M = (W \times l)/2 \quad (\text{ udl \& simply supported beam })$$

$$I = \pi/64 \times D_{pin}^4$$

$$Y = D_{pin}/2$$

$$Z = 3316.899525 \text{ mm}^3$$

$$M = 48494.89743 \text{ N-mm}$$

$$\sigma_b = 14.62 \text{ N/mm}^2$$

Now Apply Principal stress equation, $\sigma_{1,2}$

$$\sigma_b = \sigma_x \quad \tau_{xy} = \tau$$

Ref.Pg.7.2 /DDB

$$\sigma_b = \sigma_y = 14.62 \text{ N/mm}^2$$

$$\tau_{xy} = \tau = 4.73 \text{ N/mm}^2$$

$$\sigma_{1,2} = (\sigma_x + \sigma_y)/2 \pm 1/2 \sqrt{(\sigma_x - \sigma_y)^2 + 4 \cdot \tau^2}$$

RHS

$$\sigma_y/2 = 7.310$$

$$\sigma_y^2/2 = 106.880$$

$$(4 \cdot \tau^2)/2 = 44.686$$

$$\text{Sqrt}(\sqrt{(\sigma_x - \sigma_y)^2 + 4 \cdot \tau^2}) = 17.410$$

8.707

Now

$$\frac{(\sigma_x + \sigma_y)}{2} \pm \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4 \tau^2} =$$

[redacted]

16.017 N/mm²

16.017 N/mm²

$\sigma_1 =$

[redacted]

N/mm²

16.017 < [40]

Safe design

[redacted]

But as per max. shear stress theory ,

$$\tau_{\max} = \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4 \tau^2}$$

$\tau_{\max} =$

8.707

[redacted]

N/mm²

8.707

< [40]

safe design

key design

1 Taper Key **5.21/5.22 DDB**

2 Gib-Head Key **5.19/5.20 DDB**

3 Tangential Key

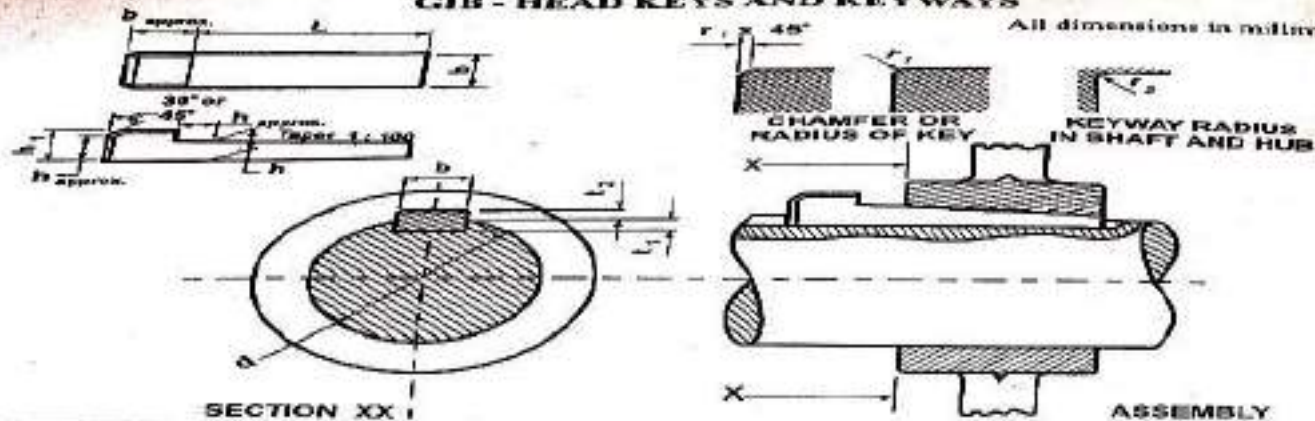
Power= 40000 Watts

N= 100 rpm

stress 20 N/mm²

GIB - HEAD KEYS AND KEYWAYS

All dimensions in millimetres.



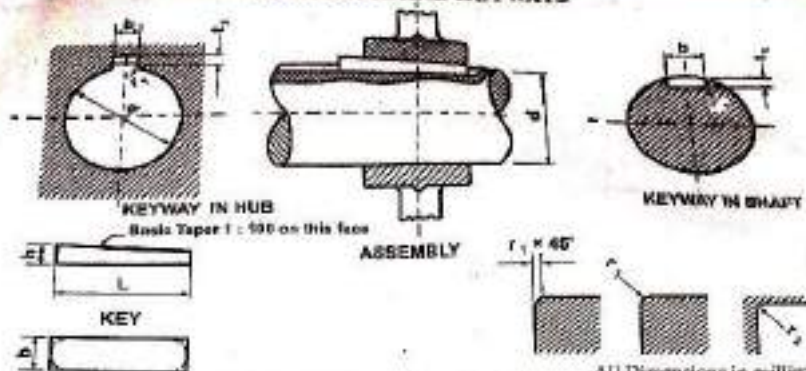
SHAFT DIAMETER, d		KEY					KEYWAY IN SHAFT AND HUB					
Above	Up to and including	Width b (h9)	Height, h (nominal)	Tolerance on h	Height of Gib-head h ₁	Chamfer or Radius r ₁ (Min.)	Width of Key way (D10)	Depth in Shaft, t ₁	Tol. on t ₁	Depth in Hub, t ₂	Tol. on t ₂	Radius at Bottom of Key way r ₂ (max)
10	12	4	4	+0.1	7	0.16	4	2.5	+0.1	1.2	+0.1	0.16
12	17	5	5		8	0.25	5	3		1.7		
17	22	6	6		10		6	3.5		2.1		
22	30	8	7		11		8	4		2.5		
30	38	10	8	+0.2	12	0.40	10	5	2.5	+0.15	0.40	
38	44	12	8		12		12	5	2.5			
44	50	14	9		14		14	5.5	2.0			
50	58	16	10		16	16	6	3.4				
58	65	18	11		18	18	7	3.3				
65	75	20	12		20	20	7.5	3.8				
75	85	22	14	+0.3	22	0.60	22	8.5	4.8	+0.15	0.60	
85	95	25	14		22		25	9	4.3			
95	110	28	16		25		28	10	5.3			
110	130	32	18		28	32	11	6.2				
130	150	36	20		32	36	12	7.2				
150	170	40	22		36	40	13	8.2				
170	200	45	23	+0.3	40	1.00	45	15	9.2	+0.3	1.00	
200	230	50	28		45		50	17	10.1			
230	260	56	32		50		56	19	11.1			
260	290	63	32		56	63	20	12.1				
290	330	70	36		63	70	22	13.1				
330	380	80	40		70	80	25	14.1				
380	440	90	45	+0.3	75	2.50	90	28	16.1	+0.3	2.50	
440	500	100	50		80		100	31	18.1			

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Height (h) mm	Preferred length (L), mm																			
	4	5	6	8	10	12	14	16	18	20	22	25	28	32	36	40	45	50	56	63
14	14	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
16	16	16	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
18	18	18	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
20	20	20	20	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
22	22	22	22	22	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
25	25	25	25	25	25	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
28	28	28	28	28	28	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
32	32	32	32	32	32	32	-	-	-	-	-	-	-	-	-	-	-	-	-	-
36	36	36	36	36	36	36	-	-	-	-	-	-	-	-	-	-	-	-	-	-
40	40	40	40	40	40	40	40	-	-	-	-	-	-	-	-	-	-	-	-	-
45	45	45	45	45	45	45	45	45	-	-	-	-	-	-	-	-	-	-	-	-
50	50	50	50	50	50	50	50	50	50	-	-	-	-	-	-	-	-	-	-	-
56	56	56	56	56	56	56	56	56	56	56	-	-	-	-	-	-	-	-	-	-
63	63	63	63	63	63	63	63	63	63	63	63	-	-	-	-	-	-	-	-	-
71	71	71	71	71	71	71	71	71	71	71	71	71	-	-	-	-	-	-	-	-
80	80	80	80	80	80	80	80	80	80	80	80	80	80	-	-	-	-	-	-	-
90	90	90	90	90	90	90	90	90	90	90	90	90	90	90	-	-	-	-	-	-
100	100	100	100	100	100	100	100	100	100	100	100	100	100	100	100	-	-	-	-	-
110	110	110	110	110	110	110	110	110	110	110	110	110	110	110	110	110	-	-	-	-
125	125	125	125	125	125	125	125	125	125	125	125	125	125	125	125	125	125	-	-	-
140	140	140	140	140	140	140	140	140	140	140	140	140	140	140	140	140	140	140	-	-
160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	160	-
180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180	180
200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200	200
220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220	220
250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250	250
280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280	280
315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315	315
355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355	355
400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400	400

DESIGN DATA - PSG TECH

TAPER KEYS AND KEYWAYS



All Dimensions in millimetres

SHAFT DIAMETER, d		KEY				KEYWAY IN SHAFT AND HUB				LENGTH		
Above	Upto and including	Cross Section		Chamfer or radius r_1 (Min)	Keyway width b (D10)	Depth in shaft, t_1	Tolerance on t_1	Depth in Hub, t_2	Tolerance on t_2	Radius, r_2 (Max)	Minimum	Maximum
		Width, b (h9)	Height h									
6	6	2	2	0.16	2	1.2	+0.05	0.5	+0.1	0.16	6	20
8	10	3	3	0.25	3	1.8		0.9			0.25	8
10	12	4	4		0.25	4	2.5	1.2	0.40	0.40		10
12	17	5	5	0.40		5	3.0	1.7			0.60	0.60
17	22	6	6		0.40	6	3.5	2.1	0.60	0.60		
22	30	8	7	0.40		8	4.0	2.5			0.60	0.60
30	38	10	8		0.40	10	5.0	2.5	0.60	0.60		
38	44	12	8	0.40		12	5.0	2.5			0.60	0.60
44	50	14	9		0.60	14	5.5	2.9	0.60	0.60		
50	58	16	10	0.60		16	6.0	3.4			0.60	0.60
58	65	18	11		0.60	18	7.0	3.3	0.60	0.60		
65	75	20	12	0.60		20	7.5	3.8			0.60	0.60
75	85	22	14		0.60	22	8.5	4.8	0.60	0.60		
85	95	25	14	0.60		25	9.0	4.3			0.60	0.60
95	110	28	16		1.00	28	10.0	5.3	0.60	0.60		
110	130	32	18	1.00		32	11.0	6.2			0.60	0.60
130	150	36	20		1.00	36	12.0	7.2	0.60	0.60		
150	170	40	22	1.00		40	13.0	8.2			0.60	0.60
170	200	45	25		1.00	45	15.0	9.2	0.60	0.60		
200	230	50	28	1.6		50	17.0	10.1			0.60	0.60
230	260	56	32		1.6	56	19.0	12.1	0.60	0.60		
260	290	63	32	1.6		63	20.0	11.1			0.60	0.60
290	330	70	36		1.6	70	22.0	13.1	0.60	0.60		
330	380	80	40	2.5		80	25.0	14.1			0.60	0.60
380	440	90	45		2.5	90	28.0	16.1	0.60	0.60		
440	500	100	50	2.5		100	31.0	18.1			0.60	0.60

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PREFERRED LENGTHS OF TAPER KEYS

Width (b) mm	Preferred Length (L), mm																			
	2	3	4	5	6	8	10	12	14	16	18	20	22	25	28	32	36	40	45	50
Height (h) mm	2	3	4	5	6	7	8	8	9	10	11	12	14	14	16	18	20	22	25	28
6	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
8	8	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
10	10	10	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
12	12	12	12	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
14	14	14	14	14	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
16	16	16	16	16	16	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
18	18	18	18	18	18	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
20	20	20	20	20	20	20	-	-	-	-	-	-	-	-	-	-	-	-	-	-
-	22	22	22	22	22	22	-	-	-	-	-	-	-	-	-	-	-	-	-	-
-	25	25	25	25	25	25	25	-	-	-	-	-	-	-	-	-	-	-	-	-
-	28	28	28	28	28	28	28	-	-	-	-	-	-	-	-	-	-	-	-	-
-	32	32	32	32	32	32	32	32	-	-	-	-	-	-	-	-	-	-	-	-
-	36	36	36	36	36	36	36	36	-	-	-	-	-	-	-	-	-	-	-	-
-	-	40	40	40	40	40	40	40	40	-	-	-	-	-	-	-	-	-	-	-
-	-	45	45	45	45	45	45	45	45	45	-	-	-	-	-	-	-	-	-	-
-	-	-	50	50	50	50	50	50	50	50	-	-	-	-	-	-	-	-	-	-
-	-	-	56	56	56	56	56	56	56	56	56	-	-	-	-	-	-	-	-	-
-	-	-	-	63	63	63	63	63	63	63	63	63	63	-	-	-	-	-	-	-
-	-	-	-	70	70	70	70	70	70	70	70	70	70	70	-	-	-	-	-	-
-	-	-	-	-	80	80	80	80	80	80	80	80	80	80	80	80	-	-	-	-
-	-	-	-	-	90	90	90	90	90	90	90	90	90	90	90	90	90	-	-	-
-	-	-	-	-	-	100	100	100	100	100	100	100	100	100	100	100	100	100	-	-
-	-	-	-	-	-	110	110	110	110	110	110	110	110	110	110	110	110	110	110	-
-	-	-	-	-	-	-	125	125	125	125	125	125	125	125	125	125	125	125	125	125
-	-	-	-	-	-	-	140	140	140	140	140	140	140	140	140	140	140	140	140	140
-	-	-	-	-	-	-	-	160	160	160	160	160	160	160	160	160	160	160	160	160
-	-	-	-	-	-	-	-	-	180	180	180	180	180	180	180	180	180	180	180	180
-	-	-	-	-	-	-	-	-	-	200	200	200	200	200	200	200	200	200	200	200
-	-	-	-	-	-	-	-	-	-	-	220	220	220	220	220	220	220	220	220	220
-	-	-	-	-	-	-	-	-	-	-	-	250	250	250	250	250	250	250	250	250
-	-	-	-	-	-	-	-	-	-	-	-	-	280	280	280	280	280	280	280	280
-	-	-	-	-	-	-	-	-	-	-	-	-	-	315	315	315	315	315	315	315
-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	355	355	355	355	355	355
-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	400	400	400	400	400

Fits and Tolerance

Its a manufacturing consideration for the sizes of the product

Its relation between the dimensions of the mating parts
Fit ?

The degree of tightness or looseness between the mating parts known as Fits

Three categories:

- 1.clearance,
- 2.location or transition, and
3. interference.

General example t

Hole and shaft assembly arrangement

1. Clearance Fit

Hole size $>$ Shaft Size, there is clearance between the shaft and hole.

Examples of clearance fit are door hinges, wheel, and axle, shaft and bearing

2. Interference fit

Hole size $<$ shaft size, then the assembly of parts made by means of forcing the shaft

Example : cotter pins in sleeve and cotter joint, Keys in coupling,

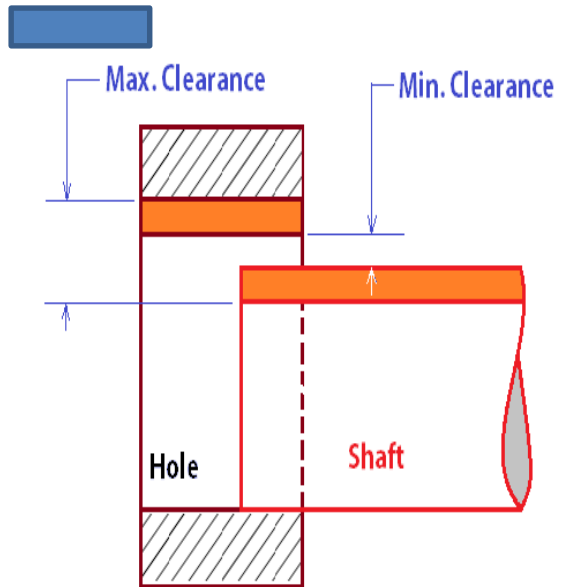
3. Transition Fit

Almost hole size and shaft size are closer or equal or their sizes overlap

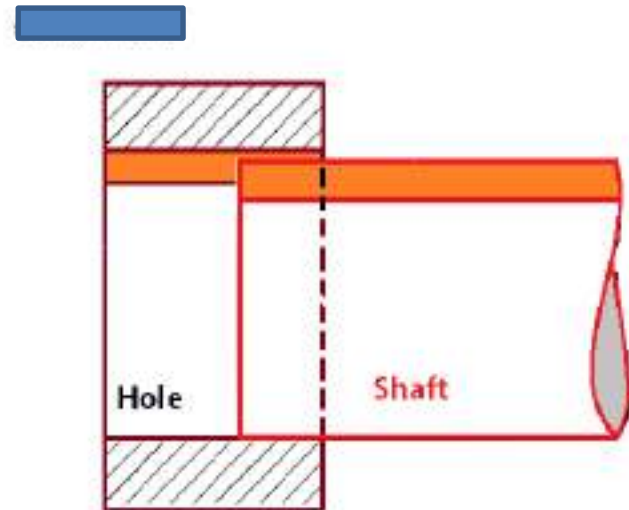
It is either clearance fit or interference fit

Example : coupling rings, Spigot mating

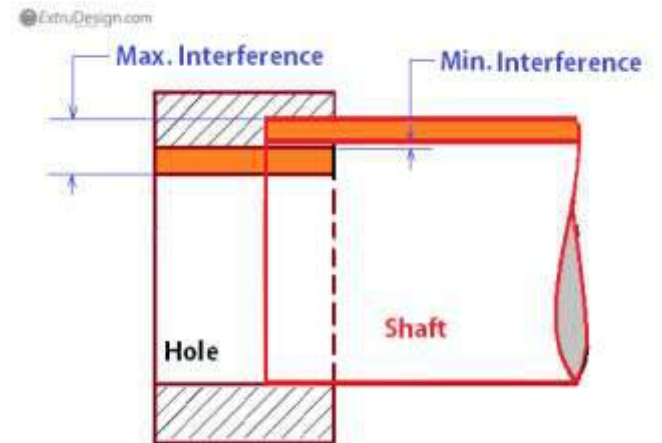
Clearance fit



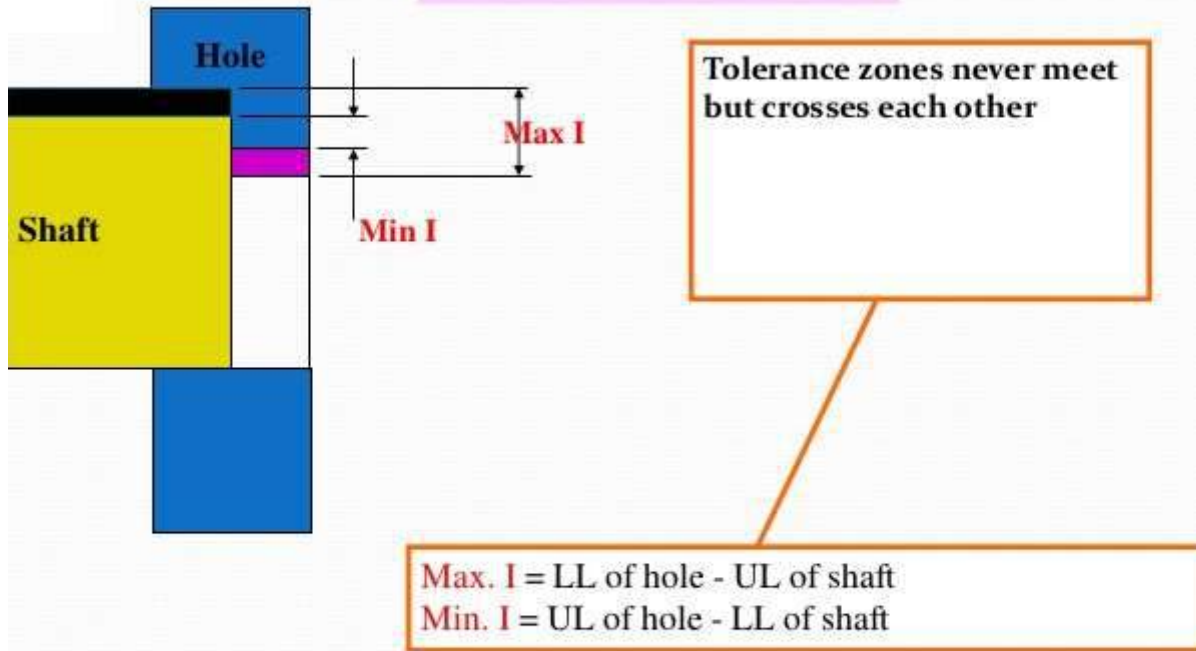
Transition fit



Interference fit



Interference Fits



The interference fits may be shrink fit, heavy drive fit and light drive fit.

Some basics

Nominal size= size provided in the drawing sheet

Basic size= production size, which might be nominal size

Example:

Dia 50 mm of shaft is required

Nominal or Basic size 50 mm

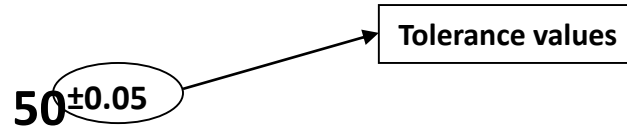
During machining , 50 will not be attained, the size may be lower or higher than the requirement, due to various errors. Eg. Machine error, operator error

50.05 mm- upper size or

49.95mm – lower size

Higher is upper limit and lower is the lower limit

Representation= $50^{+0.05}$ mm Limits: UL, LL



+0.05 mm is uL/uD--- denoted by **ES/es**

-0.05mm is lL/lD----- denoted by **EF/ef**

What is the tolerance?

The difference between upper and lower limits of dimensions

$$ES - EI = \text{Tolerance}$$

$$es - ei = \text{tolerance}$$

Types

1. Unilateral----- variation of basic size in one direction either + or -
2. Bilateral----- variation of basic size in both direction + & -

Limits systems

1. HOLE basis system

2. Shaft basis System

1. **HOLE basis system:** hole size kept as constant, shaft size will allow to vary

Denoted by Capital letters like H, G, L,,, etc

Always lower deviation is zero($EI=0$)

2. **Shaft basis System =** Shaft size kept as constant and hole size allowed to vary

Denoted by small letters like h,g,i,,, etc

Always upper deviation is zero($es=0$)

DATA BOOK usage; 3.1. 3.3 to 3.17 pages

3.3 to 3.10 more important for the calculation

Problems

Hole size: 25.00 and 25.02 mm, shaft size : 24.97 and 24.95mm

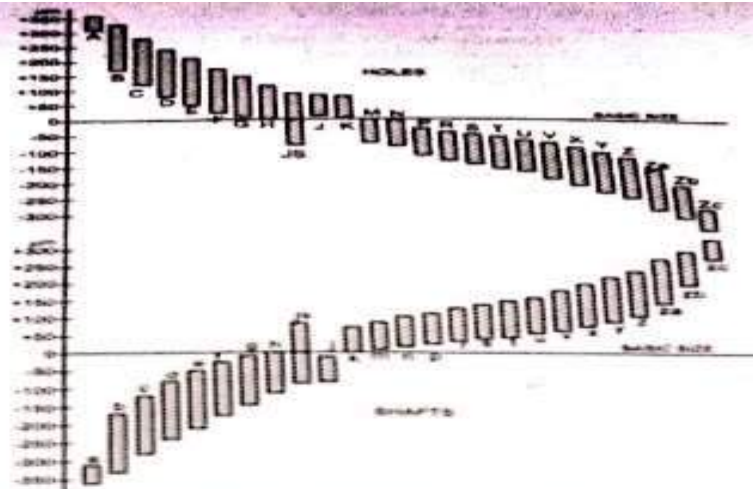
Tolerance calculation:

Hole: Upper limit- Lower limit of hole
=25.02-25.00
= 0.02 mm

Shaft: upper limit – lower limit
= 24.97-24.95
= 0.02 mm

Allowance: lower limit hole – upper limit shaft
=25.00-24.97
= 0.03 mm

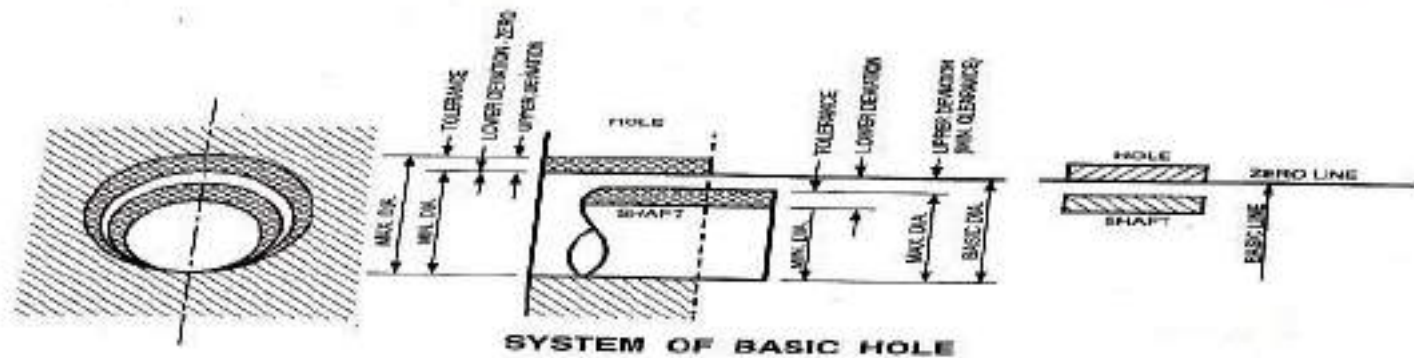
LETTER SYMBOLS FOR TOLERANCES



FUNDAMENTAL TOLERANCES OF GRADES 01, 0 AND 1 TO 16

Diameter Steps in mm	Values of tolerances in Microns																	
	Tolerance Grades																	
	01	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14*	15*	16*
To and inc 3	0.3	0.5	0.8	1.2	2	3	4	6	10	14	25	40	60	100	140	250	400	600
Over 3																		
To and inc 6	0.4	0.6	1	1.5	2.5	4	5	8	12	18	30	48	75	120	180	300	480	750
Over 6																		
To and inc 10	0.4	0.6	1	1.5	2.5	4	6	9	15	22	36	58	90	150	220	360	580	900
Over 10																		
To and inc 18	0.5	0.8	1.2	2	3	5	8	11	18	27	45	70	110	180	270	450	700	1100
Over 18																		
To and inc 30	0.6	1	1.5	2.5	4	6	9	13	21	33	52	84	130	210	330	520	840	1300
Over 30																		
To and inc 50	0.6	1	1.5	2.5	4	7	11	16	25	39	62	100	160	250	390	620	1000	1600
Over 50																		
To and inc 80	0.8	1.2	2	3	5	8	13	19	30	46	74	120	190	300	460	740	1200	1900
Over 80																		
To and inc 120	1	1.5	2.5	4	6	10	15	22	37	54	87	140	220	350	540	870	1400	2200
Over 120																		
To and inc 180	1.2	2	3.5	5	8	12	18	25	40	63	100	160	250	400	630	1000	1600	2500
Over 180																		
To and inc 250	2	3	4.5	7	10	14	20	29	46	72	115	185	290	460	720	1150	1850	2900
Over 250																		
To and inc 315	2.5	4	6	8	12	16	23	32	52	81	130	210	320	520	810	1300	2100	3200
Over 315																		
To and inc 400	3	5	7	9	13	18	27	36	57	89	140	230	360	570	890	1400	2300	3600
Over 400																		
To and inc 500	4	6	8	10	15	20	27	40	63	97	155	250	400	630	970	1550	2500	4000
Over 500																		

*Up to 1 mm, Grades 14 to 16 are not provided



RUNNING AND SLIDING FITS

Combination of Hole and shaft	Quality of fit	Typical uses
H 6 g 5 Fine H 7 g 6 Normal H 8 g 7 Coarse	Precision	Small clearance - used in precision equipment under very light load - Bearings for accurate link work and for piston and slide valves - Also used for spigot or location fits.
H 6 f 6 Fine H 7 f 7 Normal H 8 f 8 Coarse	Close running	Widely used as grease or oil lubricated bearings having low temperature differences - bearings for gear shafts, small electric motor shafts and pump shafts.
H 7 e 7 Fine H 8 e 8 Normal H 9 e 9 Coarse	Normal running	Used for properly lubricated bearings with appreciable clearance. Finer grades for high speeds and heavy loads. Turbo generator and large electric motor bearings.
H 8 d 8 Fine H 8 d 9 Normal H 9 d 9 Coarse	Loose running	For plumber block bearings and loose pulleys
H 8 c 8 } H 8 b 8 } Fine H 9 a 9 }	Slack running or positional fit	Large clearance - not widely used.
H 9 c 9 } H 11 c 11 } Normal		
H 11 c 9 } H 11 b 9 } Coarse H 11 a 9 }		

Over In	Blancher Steps, mm									
	1	2	3	4	5	6	7	8	9	10
65 G	-275	-270	-265	-260	-255	-250	-245	-240	-235	-230
66 G	-270	-265	-260	-255	-250	-245	-240	-235	-230	-225
67 G	-265	-260	-255	-250	-245	-240	-235	-230	-225	-220
68 G	-260	-255	-250	-245	-240	-235	-230	-225	-220	-215
69 G	-255	-250	-245	-240	-235	-230	-225	-220	-215	-210
70 G	-250	-245	-240	-235	-230	-225	-220	-215	-210	-205
71 G	-245	-240	-235	-230	-225	-220	-215	-210	-205	-200
72 G	-240	-235	-230	-225	-220	-215	-210	-205	-200	-195
73 G	-235	-230	-225	-220	-215	-210	-205	-200	-195	-190
74 G	-230	-225	-220	-215	-210	-205	-200	-195	-190	-185
75 G	-225	-220	-215	-210	-205	-200	-195	-190	-185	-180
76 G	-220	-215	-210	-205	-200	-195	-190	-185	-180	-175
77 G	-215	-210	-205	-200	-195	-190	-185	-180	-175	-170
78 G	-210	-205	-200	-195	-190	-185	-180	-175	-170	-165
79 G	-205	-200	-195	-190	-185	-180	-175	-170	-165	-160
80 G	-200	-195	-190	-185	-180	-175	-170	-165	-160	-155
81 G	-195	-190	-185	-180	-175	-170	-165	-160	-155	-150
82 G	-190	-185	-180	-175	-170	-165	-160	-155	-150	-145
83 G	-185	-180	-175	-170	-165	-160	-155	-150	-145	-140
84 G	-180	-175	-170	-165	-160	-155	-150	-145	-140	-135
85 G	-175	-170	-165	-160	-155	-150	-145	-140	-135	-130
86 G	-170	-165	-160	-155	-150	-145	-140	-135	-130	-125
87 G	-165	-160	-155	-150	-145	-140	-135	-130	-125	-120
88 G	-160	-155	-150	-145	-140	-135	-130	-125	-120	-115
89 G	-155	-150	-145	-140	-135	-130	-125	-120	-115	-110
90 G	-150	-145	-140	-135	-130	-125	-120	-115	-110	-105
91 G	-145	-140	-135	-130	-125	-120	-115	-110	-105	-100
92 G	-140	-135	-130	-125	-120	-115	-110	-105	-100	-95
93 G	-135	-130	-125	-120	-115	-110	-105	-100	-95	-90
94 G	-130	-125	-120	-115	-110	-105	-100	-95	-90	-85
95 G	-125	-120	-115	-110	-105	-100	-95	-90	-85	-80
96 G	-120	-115	-110	-105	-100	-95	-90	-85	-80	-75
97 G	-115	-110	-105	-100	-95	-90	-85	-80	-75	-70
98 G	-110	-105	-100	-95	-90	-85	-80	-75	-70	-65
99 G	-105	-100	-95	-90	-85	-80	-75	-70	-65	-60
100 G	-100	-95	-90	-85	-80	-75	-70	-65	-60	-55

Note: G - Co; N - No pp; Tolerances in Microns 1 Micron = 0.001 mm = 1 x 10⁻³ in.

TABLE OF TOLERANCES

A 9	Over To	Diameter Steps, mm																				
		3	6	10	18	24	30	40	50	65	80	100	120	140	160	180	200	250	300	350	400	
N	G	295	300	314	333	353	372	392	414	436	457	481	500	518	538	558	575	595	615	635	655	675
N	G	165	170	184	193	212	232	242	264	274	287	307	327	346	366	385	404	424	444	464	484	504
N	G	85	90	105	118	142	162	182	192	214	224	237	257	267	289	308	328	348	368	388	408	428
N	G	28	28	47	50	73	88	108	114	148	158	170	186	198	218	238	258	275	295	315	335	355
N	G	16	22	28	34	44	58	68	80	126	72	85	111	144	185	231	281	331	381	431	481	531
N	G	5	18	15	16	10	25	35	50	71	83	95	108	128	158	198	248	298	348	398	448	498
N	G	2	4	5	6	7	9	11	13	15	17	19	21	24	27	31	35	40	45	50	55	60
N	G	4	5	6	8	9	11	13	15	17	19	21	24	27	31	35	40	45	50	55	60	65
N	G	8	8	9	11	13	16	19	21	24	27	31	36	41	46	51	56	61	66	71	76	81
N	G	8	8	9	11	13	16	19	21	24	27	31	36	41	46	51	56	61	66	71	76	81
N	G	10	12	15	18	21	26	31	36	41	46	51	56	61	66	71	76	81	86	91	96	101
N	G	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	14	18	22	27	33	40	46	54	63	71	80	89	98	107	116	125	134	143	152	161	170
N	G	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	25	30	36	43	52	62	74	87	100	115	130	145	160	175	190	205	220	235	250	265	280
N	G	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	48	48	58	70	84	100	120	148	180	215	255	295	340	385	435	485	535	585	635	685	735
N	G	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	69	75	90	110	138	160	188	220	258	298	345	395	445	495	545	595	645	695	745	795	845
N	G	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	6	6	8	10	12	14	18	22	26	31	36	41	46	51	56	61	66	71	76	81	86
N	G	-6	-6	-7	-8	-9	-11	-12	-15	-18	-21	-24	-28	-31	-34	-38	-41	-45	-49	-53	-57	-61
N	G	8	1	2	2	2	3	4	4	4	4	4	5	5	5	5	5	5	5	5	5	5
N	G	-6	-6	-7	-9	-10	-13	-15	-18	-21	-24	-28	-31	-34	-38	-41	-45	-49	-53	-57	-61	-65
N	G	8	3	5	6	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22
N	G	-18	-9	-10	-11	-15	-18	-21	-25	-28	-32	-36	-40	-44	-48	-52	-56	-60	-64	-68	-72	-76
N	G	-2	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
N	G	-11	-12	-15	-18	-21	-25	-29	-33	-37	-41	-45	-49	-53	-57	-61	-65	-69	-73	-77	-81	-85
N	G	-4	-4	-4	-5	-7	-8	-9	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24
N	G	-14	-16	-19	-22	-28	-35	-39	-45	-51	-57	-63	-69	-75	-81	-87	-93	-99	-105	-111	-117	-123
N	G	-4	-8	-8	-11	-14	-17	-21	-24	-28	-32	-36	-40	-44	-48	-52	-56	-60	-64	-68	-72	-76
N	G	-14	-16	-20	-25	-31	-38	-47	-53	-61	-69	-77	-85	-93	-101	-109	-117	-125	-133	-141	-149	-157
N	G	-29	-24	-29	-36	-44	-54	-64	-72	-82	-92	-102	-112	-122	-132	-142	-152	-162	-172	-182	-192	-202
N	G	-14	-15	-17	-21	-27	-34	-42	-49	-58	-67	-77	-87	-97	-107	-117	-127	-137	-147	-157	-167	-177
N	G	-24	-27	-32	-39	-48	-59	-72	-84	-98	-112	-127	-142	-157	-172	-187	-202	-217	-232	-247	-262	-277

Note : G - Go; N - No go; Tolerances in Microns 1 Micron = 0.001 mm = 1 x 10⁻³ m

Designation : Number- (hole)capital letter .IT gradeNo./((shaft)small letter. It grade No

↓
Diameter

Shaft designation : 40 H8/f7

Geometric mean diameter= $\text{Sqrt}(D1 \times D2)$

D1 and D2 are shaft range

$D1 < D_{\text{given}} < D2$

Tolerance calculation

$$i = 0.45 \sqrt[3]{D} + 0.0001 D_{GM}$$

pg.3.6/ddb.

UNIT III

- ME18503-Design Of Machine Elements

UNIT III DESIGN FOR SPRINGS

Design of Close coil helical springs under varying load condition. Design of Leaf spring, Disc Spring and Torsion spring

OBJECTIVE

This course will familiarize the design principles of springs under dynamic and static conditions

COURSE OUT COME 3 - CO3

- 1.Examining the close coil helical springs under variable loading .
- 2.Acquiring the knowledge of leaf, disc and torsion springs.

What is spring?

a steel wire wound around an imaginary cylinder- Helical springs

a steel wire wound around an imaginary cone - conical spring




Elastic member absorbs energy when it is loaded and releases the energy when it is unloaded

It will distort when loaded and recover its original shape when unloaded

General - coil spring

Springs classifications

1. Open coil and closed coil

helix angle $> 10^\circ$  helix angle $< 10^\circ$ 
Clo 

1. Helical springs

compression and tension

2. Conical & volute

3. Torsion springs & spiral springs- bi-cycle hand lever brake, writing pads, toys, clocks

4. Leaf or laminated springs; pre stressed plates of different lengths held together by means of central bolt and clamps (U- clamps)

5. Disc or Belleville springs.- conical discs held together by central bolt
Needed: high spring rate in compact unit.

6. Special purpose springs

rubber , air or liquid, ring springs

Spring diagrams



Helical Compression Spring



Helical Extension Spring



Conical Spring



Torsion Spring



Laminated or Leaf Spring



Disc or Belleville Spring



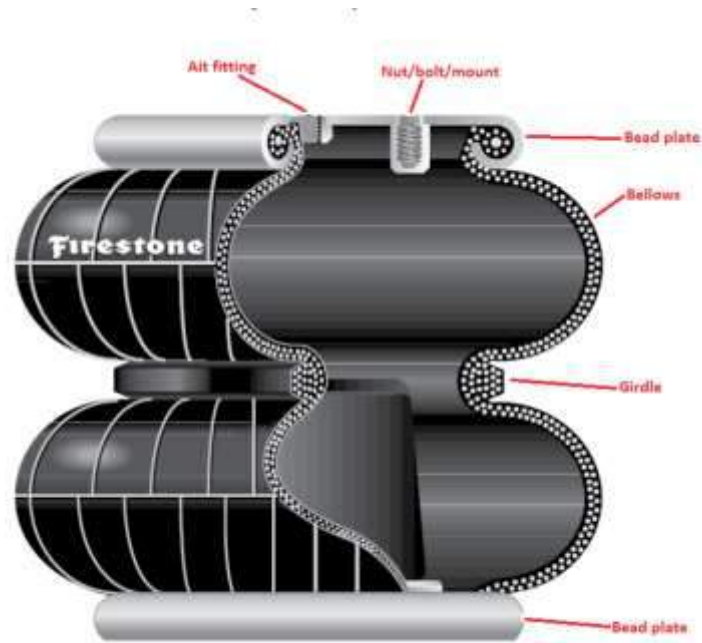
Spiral spring



**Disc springs--- industrial uses, brakes, clutches
piping- shock mounting**



Rubber spring



Air spring

Uses/functions/applications

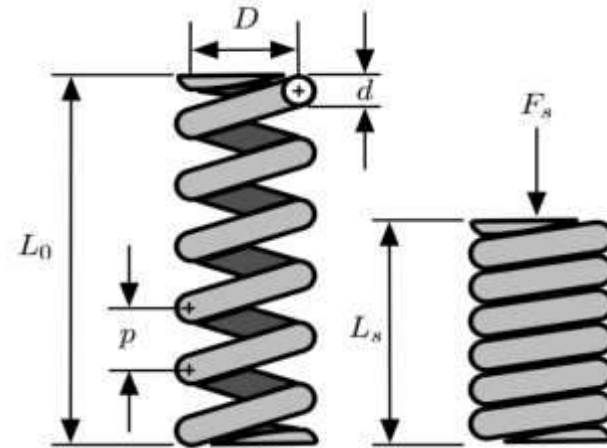
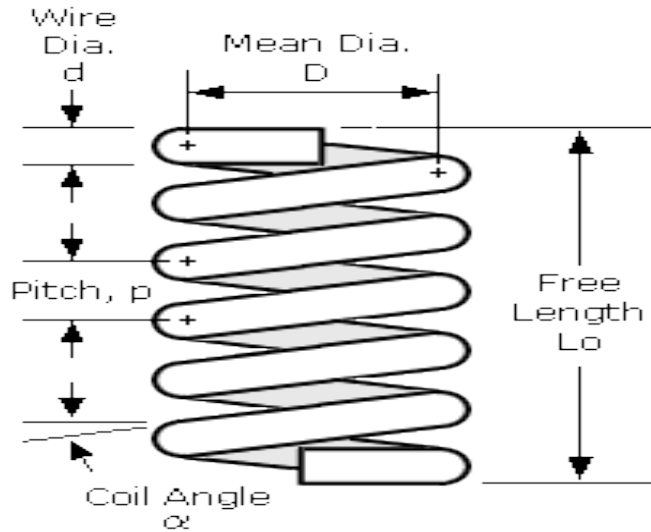
To absorb the shock or vibration as in-car springs, railway buffers, etc.

To measure the forces in a spring balance.

Apply forces in brakes and clutches to stop the vehicles.

Spring is also used to store the **energy** as in clocks, toys, etc.

Spring Terminologies



Spring stiffness(k): Load required to produce Unit deflection

$$k = w/\delta \text{ N/mm, } w = \text{load, } \delta = \text{deflection}$$

Spring Index (C): $C = Dm/d$ - No unit

Where Dm = mean diameter of coil

d = wire diameter

Note: the clearance between the two adjacent turn is 1 mm always.

Pitch of the coil

$$p = \frac{\text{Free length}}{n' - 1}$$

$$p = \frac{L_F - L_S}{n'} + d$$

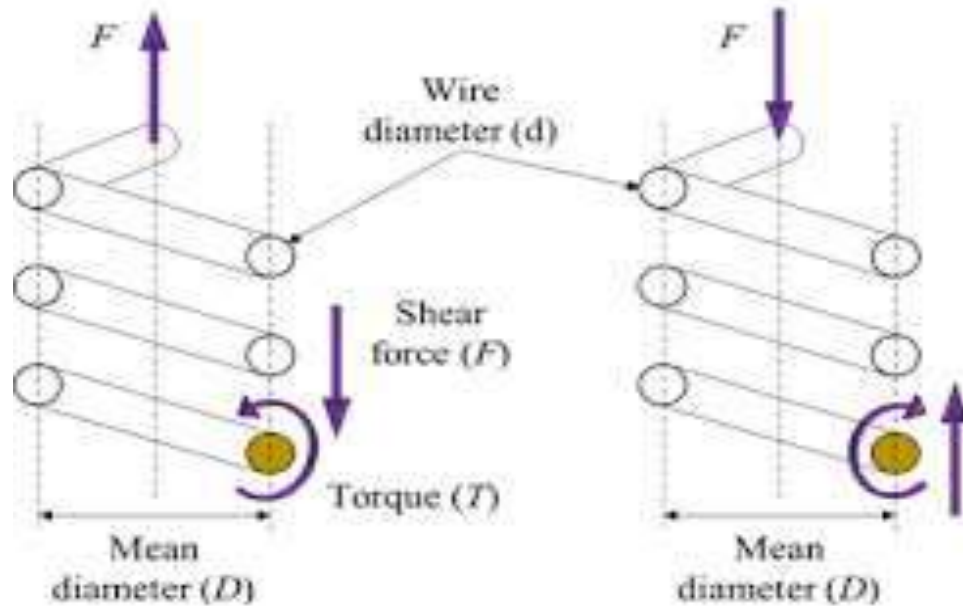
where LF = Free length of the spring.

LS = Solid length of the spring,

n' = Total number of coils, and

d = Diameter of the wire.

Stresses



1. Shear stress by external load
2. Torsion at wire curvature- $T = F \times R$

Radius of curvature is neglected in static loading

to consider this radius of curvature, **Wahl's factor used. K**

DESIGN PROCEDURE spring compression spring

Step1: Find d , D_m , D_o , & D_i for the spring using Shear stress(pg.7.100/DDB)

Step2: Find the deflection of spring ' y ' (Pg. 7.100/DDB) (either y or n based on The available either y or n)

Step3: Find the stiffness of the spring ' q ' (Pg. 7.100/DDB)

Step4: Find L_f - free length of coil

$$L_f = L_s + y + 15\% \text{ of } y$$

$$L_s = nxd \quad n = \text{no. of turns, } d = \text{wire diameter.}$$

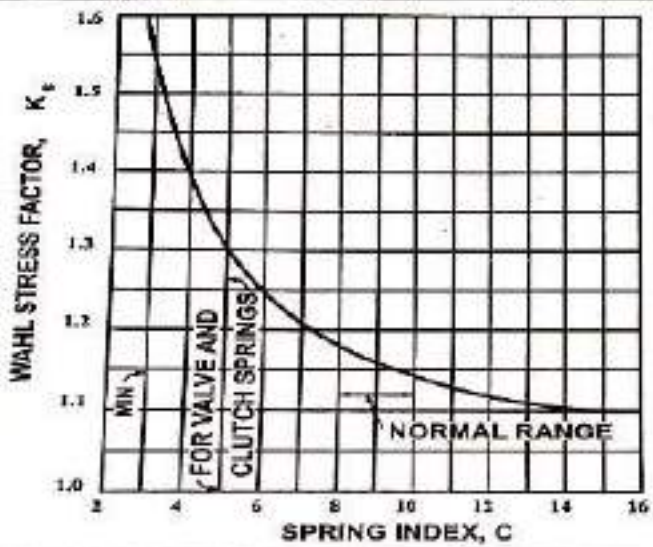
Step 5: Find pitch ' p '

Step 6 ; check for buckling Pg.7.101/DDB

Step 7: Check for surging (optional step) pg.7.101/DDB

SPRINGS

Helical Springs	
Circular Section	Rectangular Section
$\tau = K_s \frac{8PD}{\pi d^3} = K_s \frac{8WFC}{\pi d^3}$	$\tau = \frac{Q_2 PD}{2tb^2}$
$y = \frac{8PD^3n}{Gd^4} = \frac{8PC^3b}{Gd}$	$y = \frac{Q_1 \pi P D^3 n}{4 G t^3 b}$
$q = \frac{Gd^4}{8D^3n} = \frac{Gd}{8C^3n}$	$q = \frac{4 G t^3 b}{Q_1 \pi D^3 n}$

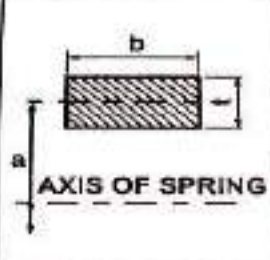


$$U = \frac{Py}{2}$$

Wahl stress factor

$$K_s = \frac{4C-1}{4C-4} + \frac{0.615}{C}$$

(may be obtained from the graph also)




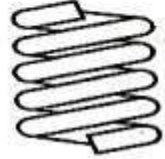
Factors for rectangular wire section


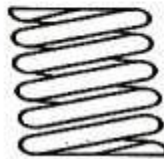
b / t	1	1.5	2	3	4	6	8	10	α
Q ₁	7.09	5.10	4.36	3.80	3.56	3.36	3.26	3.21	3
Q ₂	4.79	4.35	4.05	3.71	3.52	3.35	3.25	3.2	3

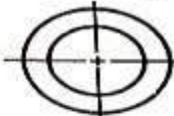
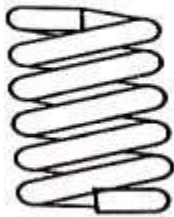
- P axial load, kgf
- D mean diameter of spring, cm
- d diameter of wire, cm
- C spring index, D / d
- b breadth of wire, cm
- t thickness of wire, cm
- n number of active coils
- τ shear stress, kgf / cm²
- G modulus of rigidity, kgf / cm²
- y deflection of spring, cm
- q spring rate or stiffness, kgf / cm
- K_s Wahl stress factor
- Q₁, Q₂ factors for springs of rectangular section
- U resilience, kgf cm
- f lowest natural frequency for circular coil helical springs, cycles per second
- γ specific weight of spring material, kgf / cm³
- g gravitational constant, 981 cm / s²
- L_f free length of spring, cm
- L_s solid length, cm
- p pitch of coils, cm
- α helix angle, < 12°
- λ solid deflection, cm
- Δ clearance between concentric springs


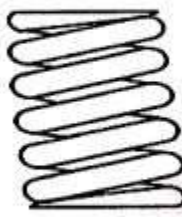
SPRINGS (contd...)

lowest natural frequency $f = \frac{(q/m)^{1/2}}{2} \frac{d}{\pi D^2 n} \sqrt{\frac{G g}{8 \gamma}}$				m mass of the active coil in the spring f _i frequency of the applied load τ _s shear stress in spring when compressed solid
Check for surging $f \geq 12 f_i$				
Check for solid stress $\tau_s < 0.5 \sigma_u$ for hard drawn carbon steels $< 0.5 \sigma_u$ for alloy steels				
To avoid Buckling $\frac{L_f}{D} < 3$ for $\frac{L_f}{D} > 3$ the spring must be suitably guided				
Coaxial springs (suffixes 1, 2 refer to outer and inner springs respectively)				
$\tau_1 = \tau_2 < [\tau]$				
$\frac{D_1}{d_1} = \frac{D_2}{d_2} = C$				
$\frac{P_1}{P_2} = \left(\frac{C}{C-2}\right)^2$				
$\Delta = \frac{(D_1 - D_2) - (d_1 + d_2)}{2} \approx \frac{d_1 - d_2}{2}$				
$d_1 \leq \frac{D_1 - D_2}{2}$				
The winding of the springs should be of opposite hands				
End conditions and length of springs				
Type of end	Total coils	Free length, L _f	Solid length, L _s	
Plain	n	pn + d	dn + d	
Plain and Ground	n	pn	dn	
Squared	n + 2	pn + 3d	dn + 3d	
Squared and Ground	n + 2	pn + 2d	dn + 2d	



Plain Ends



Plain and Ground Ends



Squared Ends



Squared and Ground Ends

S1 Design a helical spring subjected to a load of 1000 N for a deflection of 25 mm and the spring index is 5. The allowable shear stress is 420Mpa. Take modulus of Rigidity as 84 kN/mm².

DATA

Load W or P = 1000 N

C = D/d= 5

Y = 25 mm

[T]= 420 N/mm²

G= 84 x 1000 N/mm²

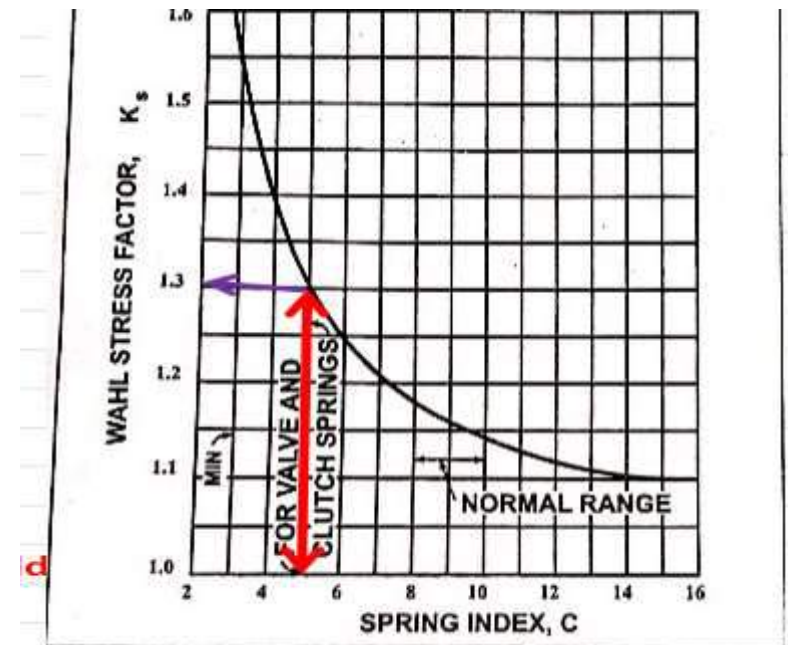
Step1: find d, Dm, etc refer 7.100/DDB

$$\tau = K_s \frac{8PD}{\pi d^3} = K_s \frac{8WC}{\pi d^2}$$

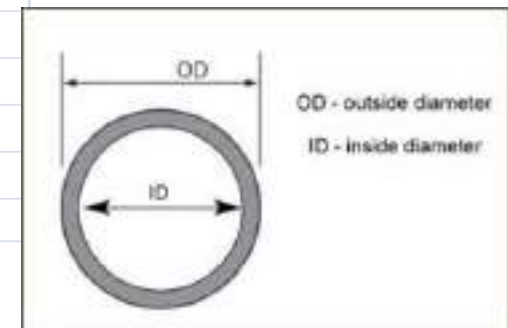
P=	1000
Pi=	3.14
[τ]	420
c=	5

To find Ks: refer pg.7.100/dd

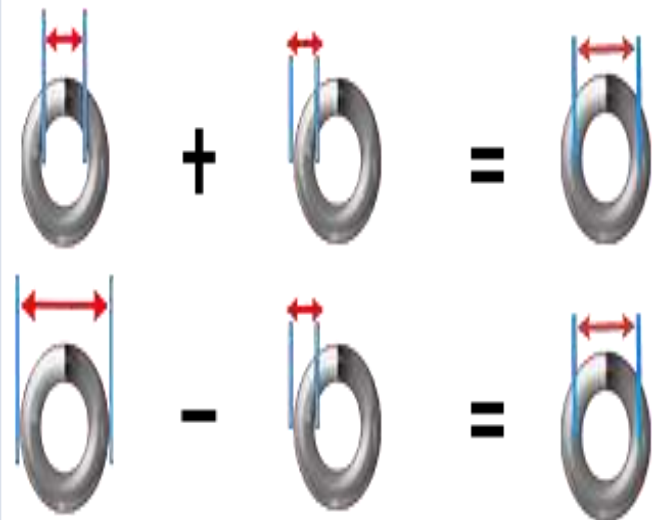
For c = ks = 1.3



LHS =	420		
RHS =	[1.3 x (8 x 1000 x 5)]/(3 .141 x d^2)		
	16555.24 /d^2		
LHS	=	RHS	
420	=	16555.2 /d^2	
d^2 =		39.4172	
d =		6.28	



Mean D=?		
C= D/d		
D=	31.39 mm	
Inner dia=? Di		
Di =	Dm-2r	
	Dm-d	[2r=d]
Di =	25.11 mm	
Outter diameter Do?		
Do=	Dm+2r	
	Dm+ d	
Do=	37.67 mm	



Step2:

Find the deflection of spring ' y ' (Pg. 7.100/DDB) (either y or n based on

The available either y or n

$$y = \frac{8PD^3n}{Gd^4} = \frac{8PC^3n}{Gd}$$

LHS =	25				
RHS =	8x1000 x31.39xn/(84x10 ³ x6.28)				
			Nr	247473770 n	
			Dr	130512323	
25	1.8961717 n			1.8961717 n	
n=	13.18446				
n=	13	turns			

Step3 : Finding stiffness of the spring ' q '

ref. Pg.7.100/DDB

$$q = \frac{G d^4}{8 D^3 n} = \frac{G d}{8 C^3 n}$$

D= 31.39 mm

D= 6.28 mm

n= 13 nos

G= 84000 N/mm²

q = $[984 \times 10^3] \times (6.28)^4 / [8 \times (31.39)^3 \times$

q= 40.56757 N/mm

Step4: Find L_f - free length of coil

$$L_f = L_s + y + 15\% \text{ of } y$$

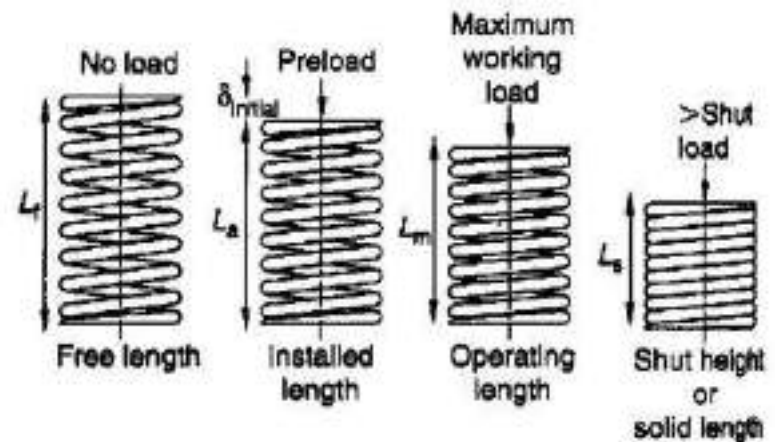
$$L_s = n \times d \quad n = \text{no. of turns, } d = \text{wire diameter}$$

$$\begin{aligned} L_s &= n \times d \\ &= 13 \times 6.28 \\ &= 81.64 \text{ mm} \end{aligned}$$

$$Y = 25 \text{ mm}$$

$$\begin{aligned} 15\% \text{ of } Y &= 0.15 \times 25 \\ &= 3.75 \text{ mm} \end{aligned}$$

$$\begin{aligned} L_f &= 81.64 + 25 + 3.75 \\ &= 110.39 \text{ mm} \end{aligned}$$



STEP 4: Calculation of Lf - free length of the coil.

Refer Pg.7.101/DDB

applying End conditions

Already Assume d

Plain End

$l_s = (dxn) + d$ solid length

$L_f = L_s + y + 15\% \text{ of } y$

n=	13		
d=	6.28		
LS=	81.64 mm	[Ls = 13 x 6.28]	
y=	25 mm		
15% of Y =	3.75 mm		

Now			
Lf=	81.64 + 25 + 3.75		
Lf =	110.39 mm		

STEP 5: Find pitch 'p'

$$p = \frac{\text{Free length}}{n' - 1}$$

Lf= 110.39

n= 13

p= 110.39/(13-1)

P= 9.199 mm



STEP 6 Check for Buckling			
REFER pg. 7.101/DDB			
LF/D > 3			Lf/d < 3 = no buckling
Lf=	110.39		
D=	31.39		
Therefore			
Lf/D =	3.516549		This shows buckling is there
Lf/D > 3,			
<u>"Guidance support is must for the spring"</u>			

Step 7	Check for solid stress		
Material selected: steel wire unalloyed cold drawn			
for d= 7.00 mm			(Pg.7.105/DDB)
$\sigma_u = \text{kgf/mm}^2$			
$\sigma_u = 111$	1110		
allowable shear stress is =[420]			
420 < 0.5 x σ_u			
420 < 555		safe design/ material selection is correct.	

S2: design and draw a valve spring of a petrol engine for the following specifications

Spring load when the valve is open=400 N

Spring load when the valve is closed = 250 N

Maximum inside diameter of the spring= 25 mm

Length of the spring when the valve is open=40 mm

Length of the spring when the valve is closed = 50 mm

Maximum permissible stress = 400 Mpa.

DATA

Basic needs C, y , Load

C= ?

Y = ?

D_m =?

But “ D_i ” is given in the

Y = can be calculated by comparing length of spring when valve is opened and closed

Closed length- open length

$Y = 50 - 40$

= 10 mm

***Load is max: 400 N for shear stress**

***Load for Y: max- min:**

400-250= 150 N

DATA				
W_o=		400 N	opened	
W_c=		250 N	closed	
C = D/d	???			
Y =	???	mm		
[τ] =		400 N/mm²		
G =		84000 N/mm²		
length of spring (opened)Lo			40 mm	
length of closed coil Lc			50 mm	
Inside dia. Of coil (Di)			25 mm	

1. $y =$	Lc-Lo		
	10 mm		
2. $C = D/d$	(to be assumed from the Ks graph)		
C=	5 (initial assumption)		
	Ks=	1.3	
3. Load			
Max=	400	(for shear stress)	
Max- Min	150	(for y equation)	

Step1

Findin d, Dm etc

Refer Pg.7.100 /DDB

$$\tau = K_s \frac{8PD}{\pi d^3}$$

=

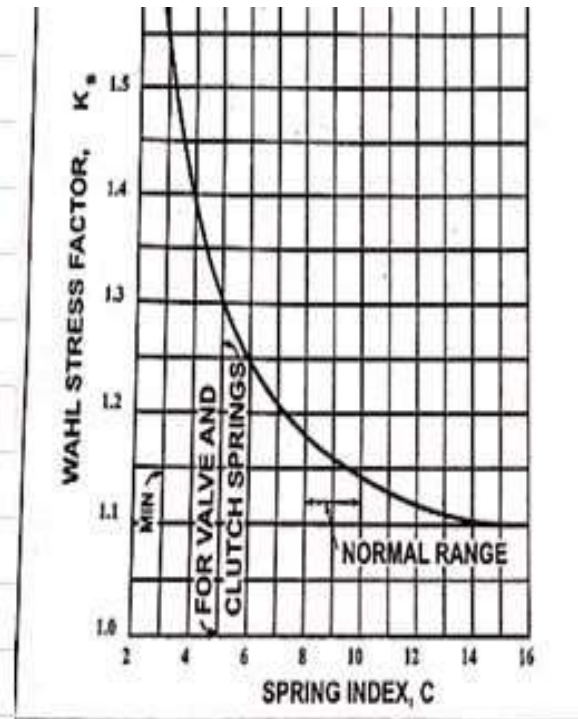
$$K_s \frac{8WC}{\pi d^3}$$

P= 400

Pi= 3.141

[T] = 400

c= 5 (Assumed)



LHS = 400

RHS = [1.3 x (8 x 400 x 5)]/(3.141 x d^2)

= 6622.09487 /d^2

LHS = RHS

LHS	=	RHS
400	=	6622.09 / d ²
d ² =		16.5552372
d	=	4.06881275 mm
d=		4.5 SWG pg.7.105/dbb

Now to find Dm

$$D_m = D_i + d$$

$$D_o = D_m + d$$

$$C = D/d = 5$$

$$\text{Corrected } C = 6.556$$

$$C = 6.6$$

$$D_m = 29.5 \text{ mm}$$

$$D_o = 34 \text{ mm}$$

PROPERTIES OF SPRING STEELS

Diameter of wire mm	Average values of Tensile strength, kgf / mm ² , σ _w									
	Steel wire unalloyed cold drawn				Spring steel oil hardened and tempered		Spring steels for moderately elevated temperatures			
	Gr 1	Gr 2	Gr 3	Gr 4	†SW	⊕VW	**1S	*1D	**2S	*2D
0.1	-	-	258	-	-	-	-	-	-	-
0.5	170	205	244	263	-	-	-	-	210	210
1.0	160	194	228	245	190	180	-	-	200	200
1.5	152	184	214	231	180	165	190	180	200	200
2.0	145	175	203	220	172	160	190	180	195	195
2.5	140	167	193	209	166	155	180	170	195	195
3.0	135	160	187	202	162	151	180	170	190	190
3.6	130	154	178	193	158	148	170	160	190	190
4.0	128	151	173	188	158	148	170	160	190	190
4.5	125	147	169	184	154	145	162	155	185	185
5.0	121	142	163	178	154	145	162	155	185	185
5.6	117	138	158	172	150	142	155	150	180	180
6.0	115	135	156	170	150	142	155	150	180	180
6.5	113	132	151	165	146	138	150	145	175	175
7.0	111	129	149	164	146	138	150	145	175	175
7.5	109	127	146	160	146	138	-	-	170	170
8.0	107	124	143	157	140	-	-	-	170	170
8.5	104	122	140	153	140	-	-	-	-	-
9.0	102	120	138	151	140	-	-	-	-	-
9.5	101	117	134	-	140	-	-	-	-	-
10.0	100	115	132	-	135	-	-	-	-	-
10.5	-	112	-	-	135	-	-	-	-	-
11.0	0	110	-	-	135	-	-	-	-	-
12.0	-	106	-	-	135	-	-	-	-	-
12.5	-	105	-	-	135	-	-	-	-	-
13.0	-	104	-	-	135	-	-	-	-	-

† SW intended for general purposes

⊕ VW intended for valve springs subjected to high dynamic stresses

* 1D and 2D intended for springs subjected to dynamic loads

** 1S and 2S intended for springs subjected to static loads

IS : 4454 - 1967

DESIGN DATA - PSG TECH

7.105

Step2:

Find the deflection of spring ' y ' (Pg. 7.100/DDB) (either y or n based on

The available either y or

n

Y= 10

LHS = 10

RHS = $8 \times 150 \times (5)^{3.5} \times n / (84 \times 10^3 \times 4.5)$ 1

10	0.8943715n	Nr	150000n
		Dr	378000
			0.3968254n

n= 11.181036

n= 12 turns

Apply end conditions: plain end (Assumed)

n= n

STEP3: Finding stiffness of the spring ' q '

ref. Pg.7.100/DDB

D= 29.50mm
d= 4.50mm
n= 12nos
G= 84000N/mm²

$$q = \frac{[(84 \times 10^3) \times (4.5)^4]}{[8 \times (29.5)^3 \times 12]}$$

$$q = 13.976295\text{N/mm}$$

STEP 4: Calculation of Lf - free length of the coil.

Refer Pg.7.101/DDB

applying End conditions

Already Assume d

Plain End

$L_s = (dxn) + d$ solid length

$L_f = L_s + y + 15\% \text{ of } y$

n= 12

d= 4.5

LS= 54mm

[$L_s = 12 \times 4.5$]

y= 10mm

15% of Y = 1.5mm

Now

Lf= 65.5mm

Lf = 65.5mm

Note:

To find L_f , the Y max to be obtained = (Y /difference ofload) x max.load

$$L_f = 54 + (10/150 * 400) + 0.15 * (10/150 * 400)$$

$$= 84.6666667$$

$$0.06666667$$

$$26.6666667$$

4

STEP 5: Find pitch 'p'

$$p = \frac{\text{Free length}}{n' - 1}$$

Lf= 65.5

n= 12

p= 65.5/(12-1)

P= 5.955mm

6mm

STEP 6 Check for Buckling

REFER pg. 7.101/DDB

$$\text{Lf/D} < 3$$

$$\text{Lf} = 65.5$$

$$\text{D} = 29.50$$

Therefore

$$\text{Lf/D} = 2.22033898$$

No guidance required, buckling is zero

Surging of spring

it must be prevented or it will cause failure of spring



Time interval of applied load/force= time taken by the wave propagation to and fro between Support and load taking end.

Resonance will occur

lowest natural frequency

$$f = \frac{(q/m)^{1/2}}{2} \frac{d}{\pi D^2 n} \sqrt{\frac{G}{8 \gamma}}$$

S3: A safety valve of 60 mm diameter is to blow off at a pressure of 1.2 N/mm². It is held in its seat by a close coil helical spring. The maximum lift of valve is 10 mm. Design a suitable compression spring of spring index 5 and providing an initial compression of 35 mm. The maximum shear stress in the material of the wire is limited to 500 Mpa. $G=80\text{kN/mm}^2$.

DATA

Valve seat= 60mm

Pressure=1.2 N/mm²

Y= 10mm

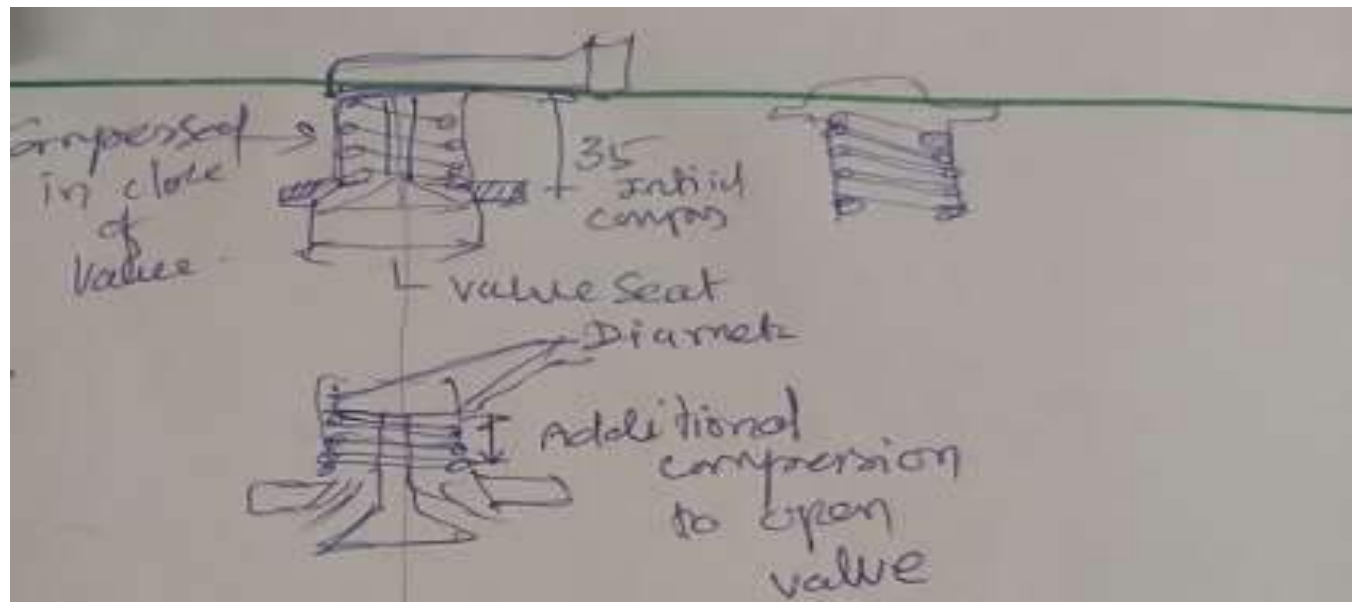
C=5

Allowable shear stress=500 N/mm²

Initial compression = 35 mm

$G= 80 \times 10^3 \text{ N/mm}^2$.





Pressure is the force or load

$$P = \frac{F}{A} \text{ or } \frac{W}{A}$$

$$\therefore W = P \cdot A \cdot \left[A = \frac{\pi}{4} d_{\text{valve}}^2 \right]$$

$$Y_{\text{max}} = \text{Initial compression} + \text{Additional compression (to open valve)}$$

$$Y_{\text{total}} = 35 + 10 = 45 \text{ mm.}$$

SPRINGS UNDER VARYING LOADS

Repeated Loading

$$\tau_s = \frac{8k_s P_s D}{\pi d^3} = \frac{8k_s P_s C}{\pi d^3}$$

$$\tau_m = \frac{8k_{s,d} P_m D}{\pi d^3} = \frac{8k_{s,d} P_m C}{\pi d^3}$$

$$\frac{1}{n} = \frac{\tau_m - \tau_s}{\tau_s} = \frac{2\tau_s}{\tau_s}$$

Curvature Factor, k_s

C	3	4	6	7	8	9	10
k_s	1.35	1.25	1.15	1.12	1.11	1.1	1.09

Static Approach to Varying Loads

No. of cycles	Classification	* Recommended Design stress [τ]
$\geq 10^6$	Severe Service	$= 0.263 \sigma_u$
$\geq 10^4$ but $< 10^6$	Average Service	$= 0.324 \sigma_u$
$< 10^4$	Light Service	$= 0.405 \sigma_u$

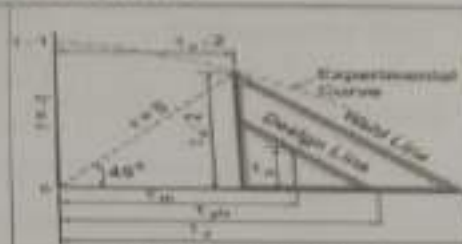
* For extension springs 0.8 times the values recommended for compression springs may be used.

EXTENSION SPRINGS

$$P_i = \frac{\pi \tau_i d^3}{8 D} = \frac{\pi \tau_i d^3}{8 C}$$

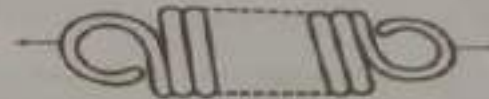
$$P = P_i + y q$$

The calculations for extension springs are done as per the compression springs based on the total load P.



VARIABLE STRESS IN SPRINGS

- P_s amplitude load, kgf
- P_m mean load, kgf
- τ_m mean shear stress, kgf/cm^2
- τ_s amplitude shear stress, kgf/cm^2
- τ_d endurance shear stress for repeated loading
- k_s Wahl stress factor, $k_{s,d}$, $k_{s,c}$
- $k_{s,d}$ direct shear factor
- k_s curvature factor
- n factor of safety
- P total load on the spring, kgf
- P_i initial load to separate the coils, kgf
- τ_i maximum value of initial stress, kgf/cm^2



Approximate maximum stresses at initial tension

C	3	4	5	6	7	8	9	10	11	12	13
τ_i kgf/cm^2	1700	1600	1430	1300	1160	1000	930	830	760	690	590



Shot on Y15
Vivo AI camera

2020-09-23 09:54
DESIGN DATA - P&I II

DESIGN PROCEDURE for compression spring –varying load

Step1: find d, using soderberg eqn. for spring.7.102/DDB)

Find Mean load, amplitude load, τ_m & τ_a , finally find 'd'

Step2: Find the deflection of spring 'y' (Pg. 7.100/DDB) (either y or n based on The available either y or n

Step3: Find the stiffness of the spring 'q' (Pg. 7.100/DDB)

Step4: Find L_f - free length of coil

$$L_f = L_s + y + 15\% \text{ of } y$$

$$L_s = nxd \quad n = \text{no. of turns, } d = \text{wire diameter.}$$

Step 5: Find pitch 'p'

Step 6 ; check for buckling Pg.7.101/DDB

Step 7: Check for surging (optional step) pg.7.101/DDB

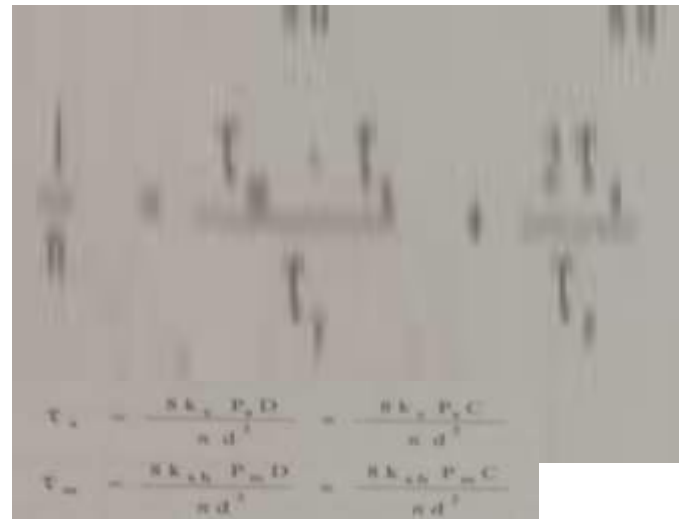
S3. A helical compression spring made of oil tempered carbon steel, is subjected to a Load which varies from 600N to 1600 N. The spring index is 6 and the factor of safety is 1.5 . The yield stress in shear is 700MPa. The endurance in shear is 350 N/mm². the compression at the maximum load is 40 mm. Take 80GPa. Design and draw the spring.

DATA

Max load= 1600N
 Min load= 600N
 C= 6
 [T] = 700N/mm²
 G= 80Gpa
 80 x 10⁹ N/m²
 80 x 10³ N/mm²

[T₁] = in shear 350N/mm²
 n= 1.5
 Y= 40 mm

Refer Pg.7.102/BBB



Step1: Find 'd' Dm, Di, do using soderberg eqn. 7.102/DDB

$\tau_m = ?$

$\tau_a = ?$

Mean Load $P_m = (P_{max} + P_{min}) / 2$
1100 N

amplitude Load $(P_{max} - P_{min}) / 2$

$P_a = 500$ N

Handwritten equations showing the relationship between shear stress and load:

$$\tau_m = \frac{8k_s P_m D}{\pi d^3} = \frac{8k_s P_m C}{\pi d^3}$$

$$\tau_a = \frac{8k_s P_a D}{\pi d^3} = \frac{8k_s P_a C}{\pi d^3}$$

$\tau_m = ?$ $(8 \times K_s \times P_m \times C) / (3.141 \times d^2)$

= 21012.4 (1/d²)

$\tau_a = ?$ $(8 \times K_s \times P_a \times C) / (3.141 \times d^2)$

= 9551.1 (1/d²)

Use soderberg relation 7.102			
$\frac{1}{n} = \frac{\sigma_m - \sigma_s}{\sigma_y} + \frac{2\sigma_s}{\sigma_y}$			

LHS=	1/1.5	0.66667	
	1 term	+	2 term
RHS=	16.3733	+	54.5777
	d^2		d^2
	0.66667 =	70.951	(1/d^2)
	d^2 =	106.427	
	d =	10.3163	
Std. 7.105/d_{db}	d=	10.5	mm

Now			
Dm=	C x d :	63	mm
Do=	Dm+d :	73.5	mm
Di=	Dm-d =	52.5	mm

$$N=13 + 2=15$$



$$L_f=211 \text{ mm}$$



$$\text{Pitch}=15\text{mm}$$



Leaf Spring Design

What is ?

A number of curved or cambered plates held together by by means of Centre bolt or shrank at its middle. Also known as semi – elliptic laminated lea spring. – (carriage spring)

The leaves are two types **1. Full length leaves or master leaf.**
2. graduated leaves

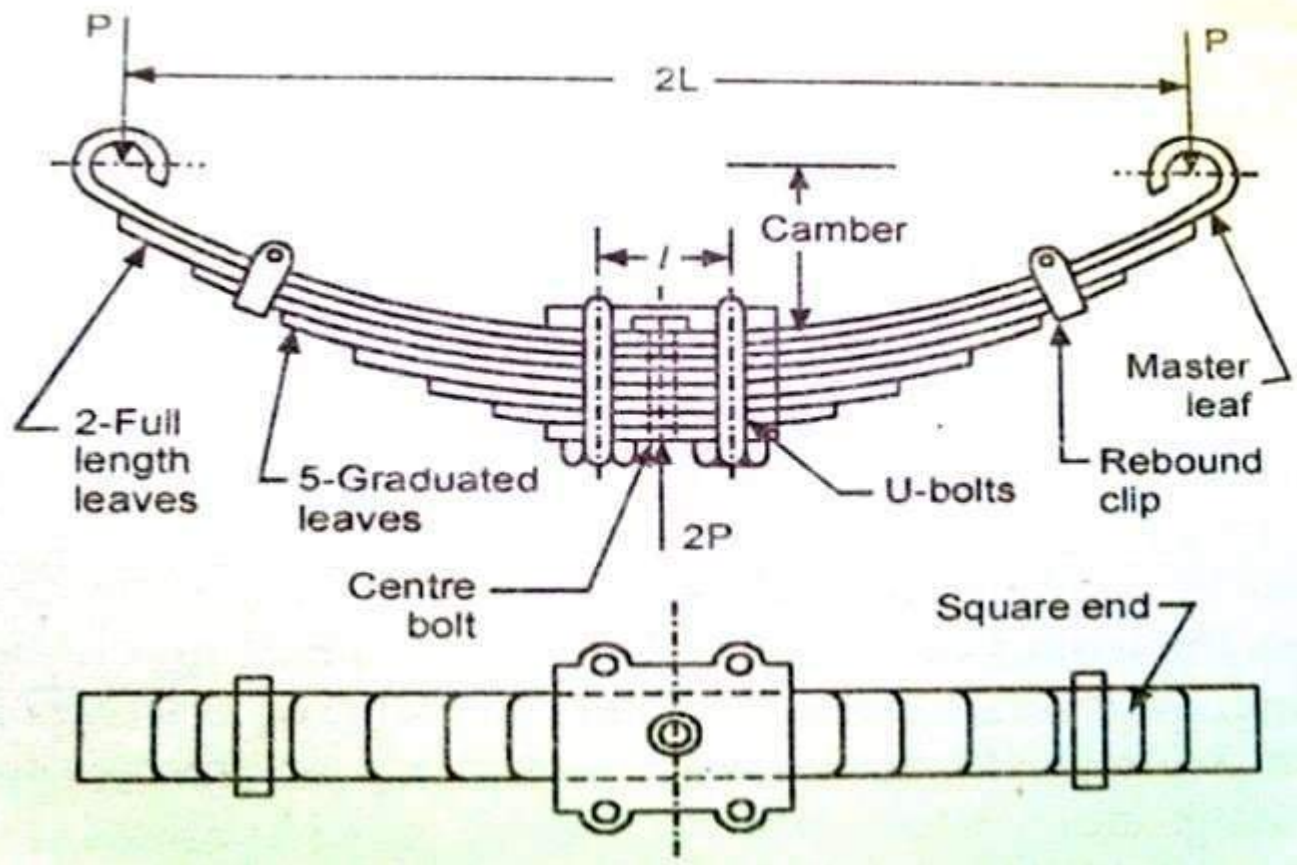
To avoid digging action between the leaves, the graduated leaves **end are trimmed**

To have curved or triangular shapes.

Master leaf has eyes at its ends. Through which , hanger or shackle pin are inserted

And supports the structure.

Widely used in automobile suspension systems



Design procedure for laminated spring

Leaf spring is treated as cantilever beam

Springs with full length leaves

Refer Pg.7.104/DDB

Objectives

1. If the sizes b and t of the leaves are given find stress induced and Y deflection, n ip
2. Stress will be given, b & t to be calculated

Step1. Cal. Of σ_b pg.7.104/DDB

Step2. cal of y deflection

Step3. cal. Of Nip h

When you have extra full length and graduated leaves, the same method to be followed, one step to be added as follows

Step-4 : find the all the leaves length

Treating the spring as a cantilever beam of uniform strength

$$\sigma_b = \frac{6PL}{nbt^2} ; y = \frac{6PL^3}{Enbt^3}$$

$$Nip, h = \frac{2PL^3}{nEbt^3}$$

Load on the clip bolts required to close the gap, $P_b = \frac{2n_e n_g P}{n(2n_g + 3n_e)}$

Spring with extra full length leaves

$$\sigma_{bg} = \frac{12PL}{bt^2(3n_e + 2n_g)}$$

$$\sigma_{be} = \frac{18PL}{bt^2(3n_e + 2n_g)}$$

$$y = \frac{12PL^3}{bt^3(3n_e + 2n_g)}$$

BELLEVILLE SPRINGS

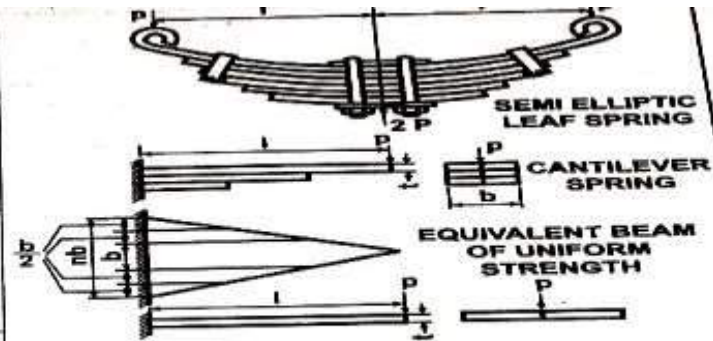
$$P = \frac{Ey}{(1-\nu^2)M\left(\frac{d_o}{2}\right)^2} \left[\left(h - \frac{y}{2}\right)(h-y)t + t^3 \right]$$

$$\sigma = \frac{Ey}{(1-\nu^2)M\left(\frac{d_o}{2}\right)^2} \left[C_1 \left(h - \frac{y}{2}\right) + C_2 t \right]$$

$$M = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)} \left[\frac{\left(\frac{d_o}{d_i} - 1\right)}{\frac{d_o}{d_i}} \right]^2$$

$$C_1 = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)} \left[\frac{\frac{d_o}{d_i} - 1}{\log_e \left(\frac{d_o}{d_i}\right)} - 1 \right]$$

$$C_2 = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)}$$



- 2 P load on the spring, kgf
 y deflection of the spring, cm
 L Length of the cantilever beam (spring) cm
 b width of the leaf, cm
 t thickness of the leaf, cm
 n number of leaves in the spring
 n_g number of graduated leaves
 n_e number of extra full length leaves
 σ_b bending stress, kgf / cm²
 σ_{bg} bending stress in graduated leaves, kgf / cm²
 σ_{be} bending stress in extra full length leaves, kgf / cm²
 E modulus of elasticity, kgf / cm²



- P axial load on the spring, kgf
 d_i internal diameter, cm
 d_o outside diameter, cm
 y deflection, cm
 t thickness of the disc, cm
 h free height minus thickness, cm
 E modulus of elasticity, kgf / cm²
 ν Poisson's ratio
 σ stress at the inside circumference, kgf / cm²
 M, C_1 , C_2 constants

LS1: A truck spring has 10 leaves and is supported at a span of length 100cm with a central band of 80 mm wide. A load of 6 kN is applied at the centre of the spring whose permissible stress is 300 Mpa. The spring has the ratio of total depth to width of about 2.5. Design the spring.

DATA

Load $2P = 6000\text{N}$
 $P = 3000\text{N}$

Central band

$a = 80\text{mm}$
 $[\sigma] = 300\text{N/mm}^2$
span ($2L$) 1000mm
 $L = 500\text{mm}$

No. of leaves (n) 10

assume $n_e = 2$ (at least 1 and maximum 3)

assume $n_g = 8$

$n = n_e + n_g$

Effective length = $2L - a$
 920mm

now new $l = 460\text{mm}$

ratio= Total depth of spring/ width



total depth= n x t

breadth= b

therefore , $nt/b=2.5$

$$b=nt/2.5$$

$$b=4t \quad [n=10]$$

objective to find b & t

objective to find b & t

Step1: find d & t using extra full length leaves eqn.

refer pg.7.104/DDB

$$\sigma_{bc} = \frac{18 PL}{bt^2 (3n_e + 2n_g)}$$

P= 3000

L= 460

b= 4t

ne= 2

ng= 8

[σ]= 300

LHS =					
		300			
RHS=					
	Nr=	18 x 3000 x 460 =		24840000	
	DR=	4t.t²[(3.2)+(2.8)]		88 t³	
					282272.7 (1/t³)
	=	282273 (1/t³)			
RHS	=	LHS			
	300	=	282273 (1/t³)		
t³	=	940.91			
t	=	9.7767			
t	=	10 mm			
now, b	=	4 x t			
	=	40 mm			

Step2: Finding defelection y

$$y = \frac{12PL^3}{bt^3(3n_e + 2n_g)}$$

Refer pg.7.104/DDB

E= to be introduced

Y= Px L/AXE

p= 3000

L= 460

ne= 2

ng= 8

b= 40

t= 10

E= 2.0 x 10⁵ N/mm²

Step3: Find Nip h

$$h = \frac{2PL^3}{nEbt^3}$$

h= 7.3002 mm

step4: Load excreted on band clips

$$P_b = \frac{2 n_e n_g P}{n (2 n_g + 3 n_e)}$$

Pg.7.104/DDB

P_b 436.364 N

Step5. Find the lengths of the leaves

First start with small leaf as l1

nth leaf length= effective length/(n-1) x n1 + Ineffective length

Effective length= 920

n= 10

Ineffective length= a= 80

n= 2

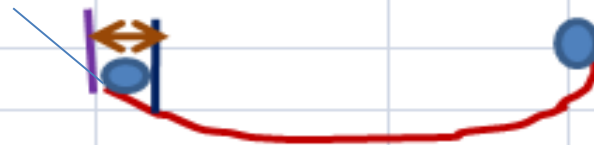
nth leaf

nth leaf					
1	1st leaf=	182.22	mm		$Ln=(920/(10-1) \times 1) + 80$
2	2nd leaf=	284.44	mm		
3	3rd leaf=	386.67	mm		
4	4th leaf=	488.89	mm		
5	5th leaf=	591.11	mm		
6	6th leaf=	693.33	mm		
7	7th leaf=	795.56	mm		
8	8th leaf=	897.78	mm		
9	9th leaf=	1000.00	mm		
10	10th leaf=	1102.22	mm		

But Masterleaf



But Masterleaf



actual span is = $2L$

Now ,
eye length to be added

since its a circular shape
llinear of circular = $\pi(d+t)$

d = inside dia of eye

if not given, assume as 20mm

No.of Eyes = 2

Therefore

Length of master leaf as follows

$L_m = 2l + 2$ eyes circumference

= $2l + 2(3.141 \times 20 + 20)$

$L_m = 1188.46$ mm¹⁰

Some special kind problems

① Safety Valve (Springs)

Important points

- ① Initial compression given, deflection given

$$\boxed{Y_T = Y_i + Y} \rightarrow \text{used to find "n"}$$

- ② Valve Seat Diameter: given.

$$\therefore W = P \cdot A, \quad A = \frac{\pi (d_v)^2}{4}$$

P: operating pressure

This "W_{op}" \Rightarrow used in both "Z & Y" equation

- ③ Max. load calculation (or)

Y_i = initial deflection given

Y = lift or deflection determined

$$Y_T = Y_i + Y$$

$$\boxed{\therefore W_{max} = \frac{W_0}{Y_i} \times Y_T = W_{max}}$$

- ④ Using Energy Eqn.

$$U = \frac{1}{2} P \cdot Y$$

P - load
Y - deflection

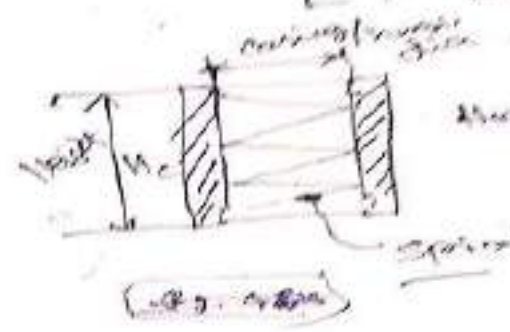
$\Rightarrow KE = \frac{1}{2} m v^2$ - Weegon (rule way)

\Rightarrow check for no. of springs. 2 (or) 3

$$U = \frac{1}{2} P \cdot Y \cdot (n) \quad n \rightarrow \text{no. of springs}$$

(b) Parabolic Profile

" D_o " - outer coil diameter is fixed
 $C \rightarrow$ spring index is determined by material
 \rightarrow outer most
 $L =$ length of the spring wire



Manufacturing constraints / casting tolerance
 D_o - given.
 $\therefore D_o = C \times D_i$

$H_c = L_{spring}$

$L \geq H_c$ - This condition also to be satisfied. (4.25)

Spring's conventional representation



Load for Spring arrangement

Springs in series
 in parallel

Note: 1. Load Range - "static only"

2. Varying load / fluctuating load -
 \rightarrow Dynamic

Sp2.

Design a spring for spring loaded safety valve for the following condition

Operating pressure=1 Mpa

Diameter of the valve seat =110mm

Design shear stress or the spring =360MPa.

G= 82GPa.

The spring is kept in the casing of 130 mm inner diameter and 400 mm long, The spring should be at maximum lift of 6 mm when the pressure is 1.08 MPa.

Answers

C – to be assumed 4/5/6

Load: operating load and maxi load to lift the spring

Load 1= N

Load2= N

Load = P x area of valve seat,
d=110mm

Load1-load2 = N

Y=6 mm

1. Find Dm, Do is = Di= Do

Dm= mm

2. Find n , n=

W= L1-L2

$L_s = \square \text{ nm}$

$L_f = \square \text{ nm}$

$P = \square \text{ mm}$

$L_f/D = \square$

Springs in Series

- Consider two springs with force constants k_1 and k_2 connected in series supporting a load $F = mg$.
- Let the force constant of the combination be represented by k
- For the combination, supporting the load $F = mg$:

$$F = kx \quad (\text{where } x = \text{the total stretch})$$

$$\text{and } x = \frac{F}{k}$$

- For each spring
 - the bottom supports $mg = F$ and stretches by x_1

$$F = k_1 x_1 \quad \text{or} \quad x_1 = \frac{F}{k_1}$$

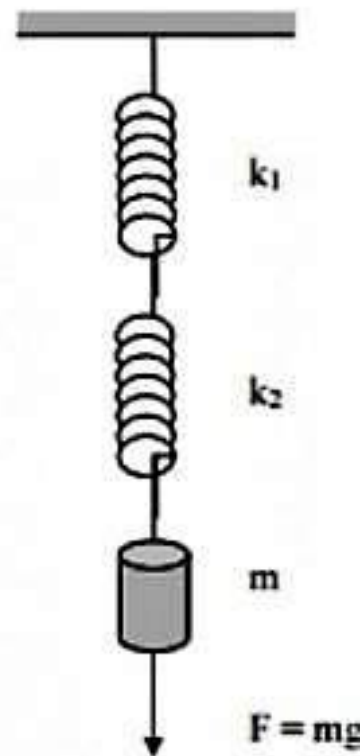
- the top spring support mg plus the weight of the bottom spring (which is negligible - Thus F is the stretching force for both springs)

$$F = k_2 x_2 \quad \text{or} \quad x_2 = \frac{F}{k_2}$$

- The total stretch

$$x = x_1 + x_2 \quad \text{or} \quad \frac{F}{k} = \frac{F}{k_1} + \frac{F}{k_2}$$

$$\text{and } \boxed{\frac{1}{k} = \frac{1}{k_1} + \frac{1}{k_2}}$$



Springs in Parallel

- Consider two springs with force constants k_1 and k_2 connected in parallel supporting a load $F = mg$.
- Let the force constant of the combination be represented by k
- For the combination supporting the load $F = mg$:
 $F = kx$ (where $x =$ the total stretch)

- The two individual springs both stretch by x but share the load ($F = F_1 + F_2$) and

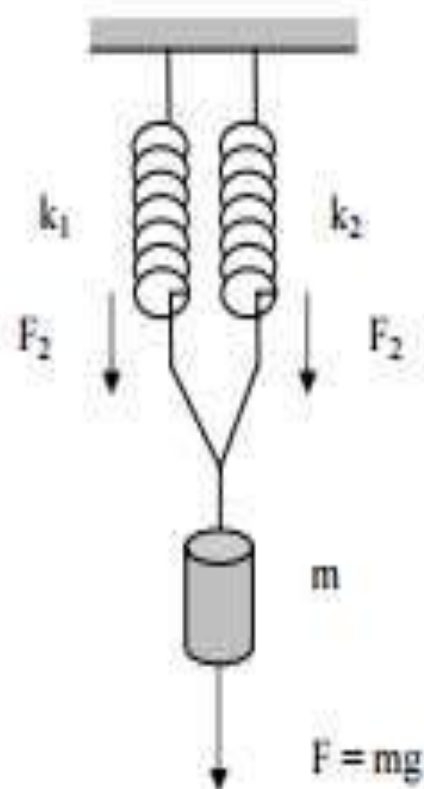
$$F_1 = k_1x \quad \text{while} \quad F_2 = k_2x$$

- Thus the total force is

$$F = F_1 + F_2 \quad \text{or} \quad kx = k_1x + k_2x$$

and

$$\boxed{k = k_1 + k_2}$$



Design procedure for disc or Belleville spring

Used: Less or compact space need high stiffness

Step1 Find load 'P' or 'Y' on the spring

Refer Pg. 7.104/DDB,

P=? Based on data , if 'p' given, find y or vice versa

Step2. Find stress " σ " on the spring

Refer Pg. 7.104/DDB,

$\sigma=?$

Prior to the above steps

TO be calculated M, C1 & C2.

Treating the spring as a cantilever beam of uniform strength

$$\sigma_b = \frac{6PL}{nbt^2} ; y = \frac{6PL^3}{Enbt^3}$$

$$Nip, h = \frac{2PL^3}{nEbt^3}$$

Load on the clip bolts required to close the gap, $P_b = \frac{2n_e n_g P}{n(2n_g + 3n_e)}$

Spring with extra full length leaves

$$\sigma_{bg} = \frac{12PL}{bt^2(3n_e + 2n_g)}$$

$$\sigma_{be} = \frac{18PL}{bt^2(3n_e + 2n_g)}$$

$$y = \frac{12PL^3}{bt^3(3n_e + 2n_g)}$$

BELLEVILLE SPRINGS

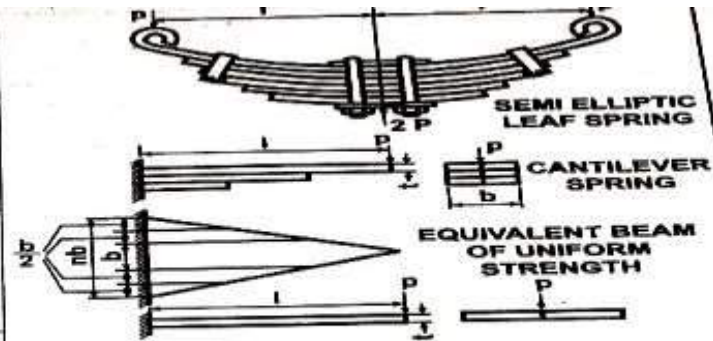
$$P = \frac{Ey}{(1-\nu^2)M\left(\frac{d_o}{2}\right)^2} \left[\left(h - \frac{y}{2}\right)(h-y)t + t^3 \right]$$

$$\sigma = \frac{Ey}{(1-\nu^2)M\left(\frac{d_o}{2}\right)^2} \left[C_1 \left(h - \frac{y}{2}\right) + C_2 t \right]$$

$$M = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)} \left[\frac{\left(\frac{d_o}{d_i} - 1\right)}{\frac{d_o}{d_i}} \right]^2$$

$$C_1 = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)} \left[\frac{\frac{d_o}{d_i} - 1}{\log_e \left(\frac{d_o}{d_i}\right)} - 1 \right]$$

$$C_2 = \frac{6}{\pi \log_e \left(\frac{d_o}{d_i}\right)}$$



- 2 P load on the spring, kgf
 y deflection of the spring, cm
 L Length of the cantilever beam (spring) cm
 b width of the leaf, cm
 t thickness of the leaf, cm
 n number of leaves in the spring
 n_g number of graduated leaves
 n_e number of extra full length leaves
 σ_b bending stress, kgf / cm²
 σ_{bg} bending stress in graduated leaves, kgf / cm²
 σ_{be} bending stress in extra full length leaves, kgf / cm²
 E modulus of elasticity, kgf / cm²



- P axial load on the spring, kgf
 d_i internal diameter, cm
 d_o outside diameter, cm
 y deflection, cm
 t thickness of the disc, cm
 h free height minus thickness, cm
 E modulus of elasticity, kgf / cm²
 ν Poisson's ratio
 σ stress at the inside circumference, kgf / cm²
 M, C₁, C₂ constants

SP3 Design a disc spring or the following specifications
Spring is made of 4 mm steel sheet has 120 mm outer diameter and 60 mm inner diameter . It is dished by 5 mm . Calculate when deflection of the spring is 2.5 mm due to an axial load P. Also calculate the stress induced in the spring take $E=200\text{kN/mm}^2$, Poisson's ratio as 0.3

**$t=4$, $d_o=120$, $d_i=60$ mm, $h=5$,
 $y=2.5$ mm, $E=200 \times 10^3$ N/mm²**

Answers:

$$M = \square$$

$$C1 = \square$$

$$C2 = \square$$

$$Y = \square$$

$$P = \square$$

$$\text{Stress} = \square$$

UNIT IV

- ME18503-Design Of Machine Elements

UNIT IV DESIGN FOR RIVETED AND WELDING JOINTS, FASTNERS

Rivet – Types of rivet joints, Caulking and caulking, Design of riveted joints for structural and pressure vessels. Eccentrically loaded rivet joint, Welding – Welding symbols, Design of welded joints under eccentrically load. Geometry of thread forms, Terminology of screw threads. Design of screws and bolts.

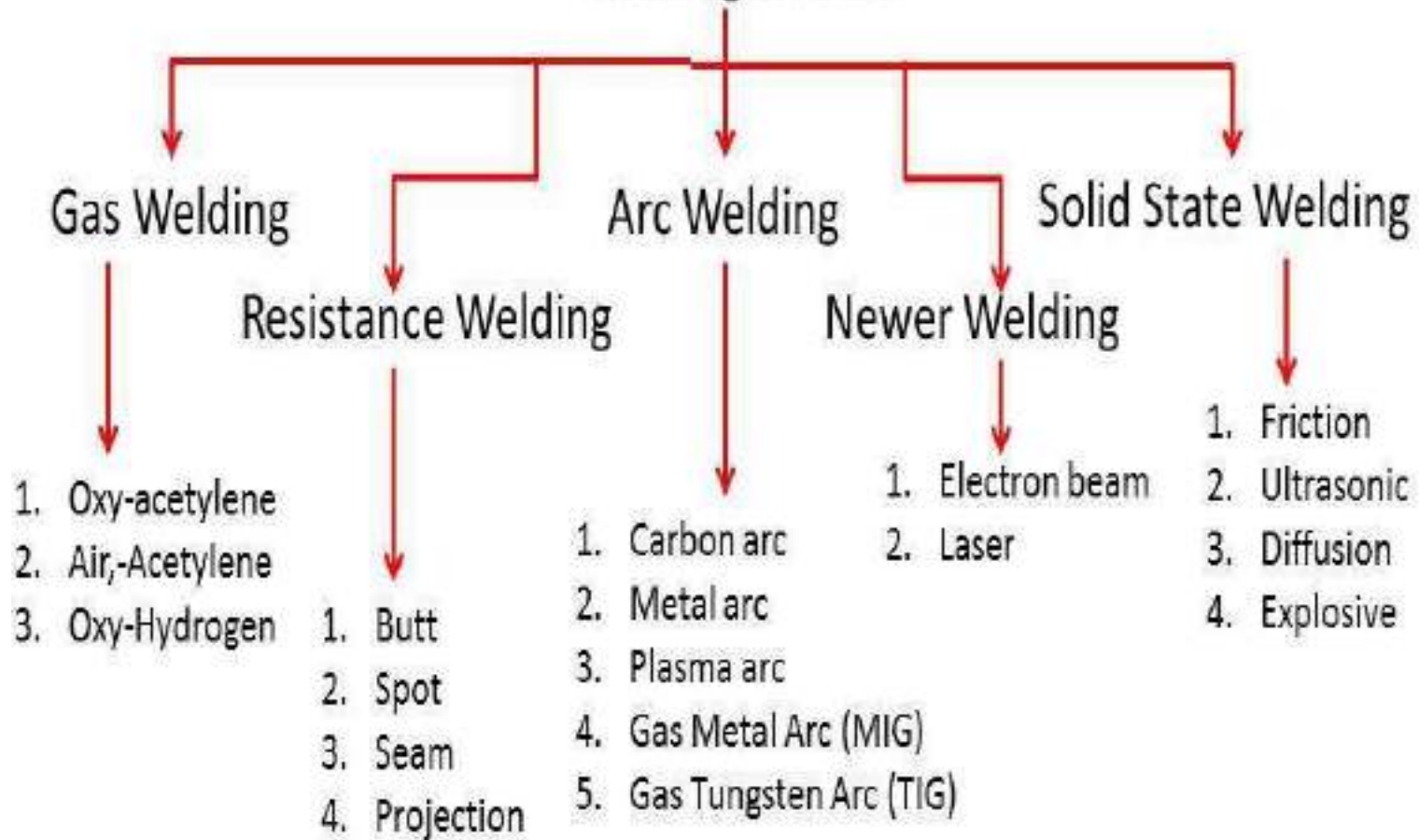
OBJECTIVE

- **This course will enable to check strength of fasteners kind of both rivet and welding.**

COURSE OUT COME 4 - CO4

Proficient in Design of riveted joint and welding joints under eccentric loading

Welding Process



Designs

- 1. Rivet design under axial load and eccentric loading**
- 2. Welding design under axial load and eccentric loading**

Welding

What ?

joining metal by heat and with or without application of pressure

Why?

Large size parts can not be manufactured by casting process

Ex. Ship building

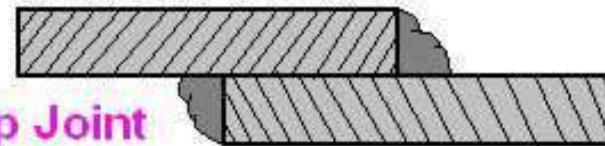
Types:

Lap joint butt joint joining method

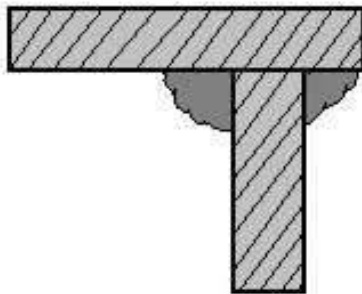
other types are there



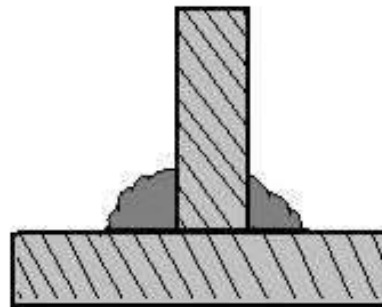
Butt Joint



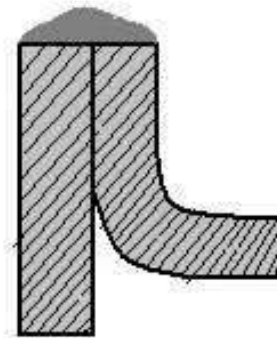
Lap Joint



Corner
Joint



Tee Joint



Edge Joint

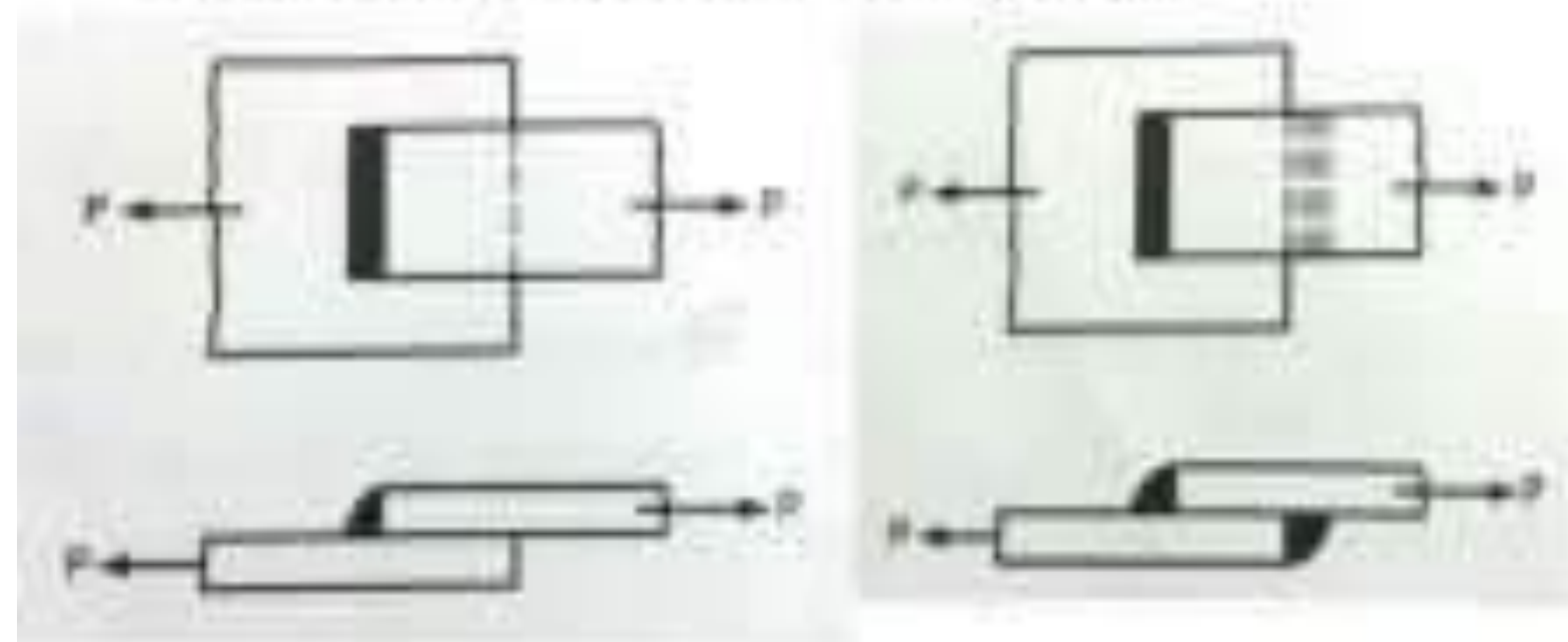
□ Types of Welded Joints:

a. Lap joint or fillet joints

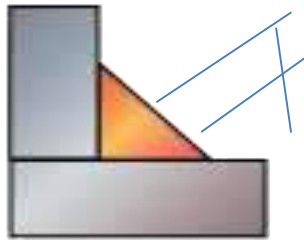
b. Butt joint

(a) Types of fillet joint:

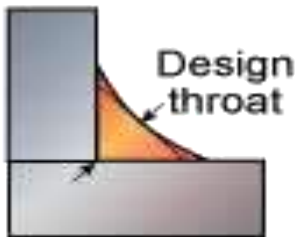
1) Single transverse 2) Double transverse



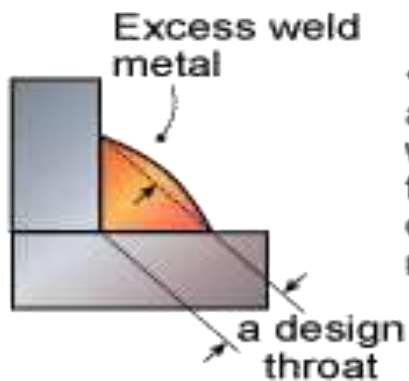
Throat



- Mitre:-
shown as a right angled
triangle with a design throat
thickness 'a' $a = 0.7072$



- Concave:-
as above, but with a reduced
design throat thickness.



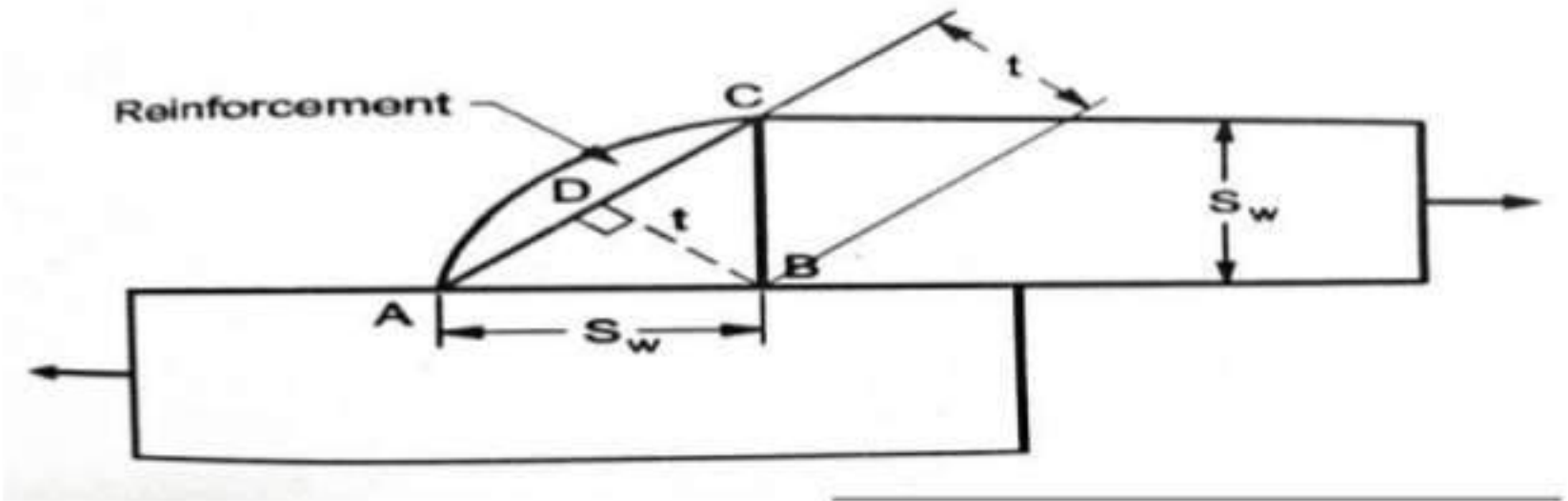
- Convex:-
as for the mitre shape but
with an increased design
throat thickness and
consequential excess weld
metal

Understanding of fillet weld



Strength of Transverse Fillet Welded Joint:-

- Consider a single transverse fillet weld.



ΔABC is a right angle isosceles triangle

Let, $t=BD$ =Throat thick. In mm

$S_w=AB=BC$ =size of weld

L_w =Length of weld in mm

$\angle BAC=\angle BCA=45^\circ$

$$T = \sin 45^\circ \times h$$

Objectives of the weld design

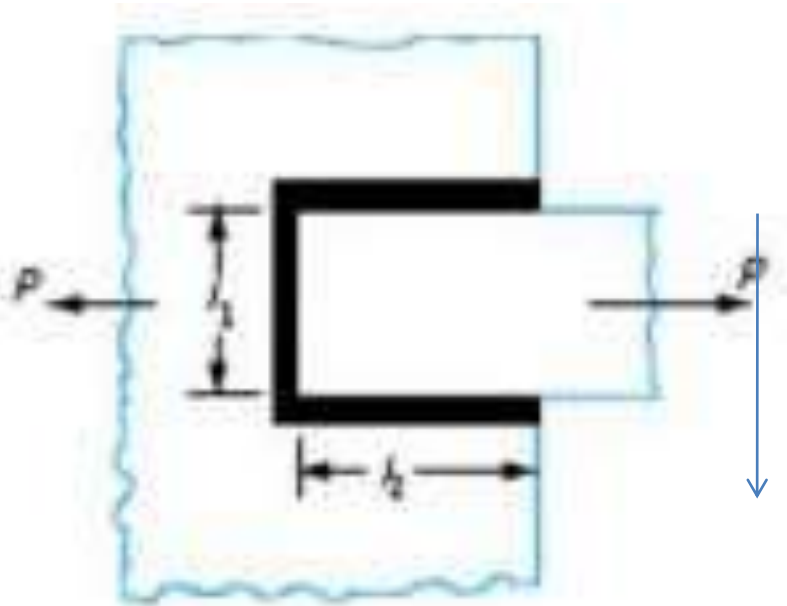
Weld size as “ h” or “ length”
(both axial load and eccentric load)

$$\text{Load} = \text{area} \times \text{stress}$$

Area of throat of weld bead or run or fillet

Stress= tensile and shear are major in consideration

bending will also be considered based on the structure and load application

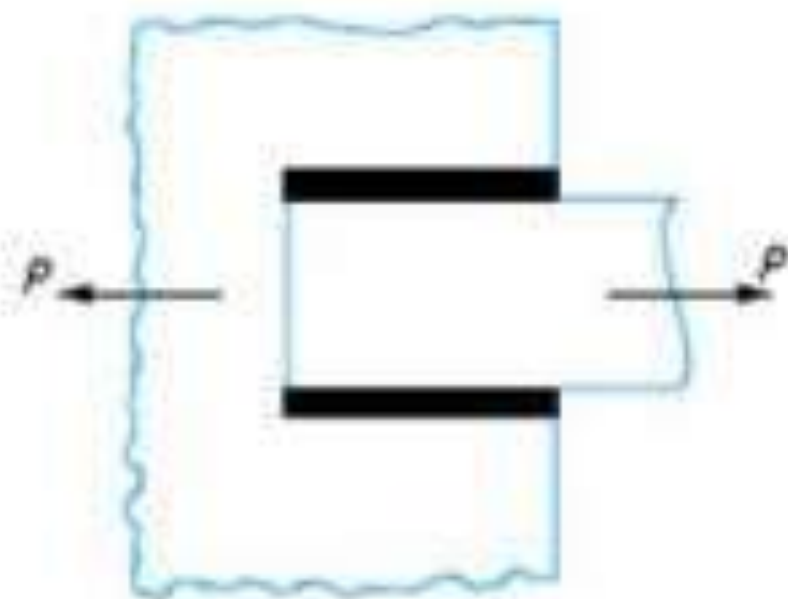


(B) Combination of transverse and parallel fillet weld.

3 fillets

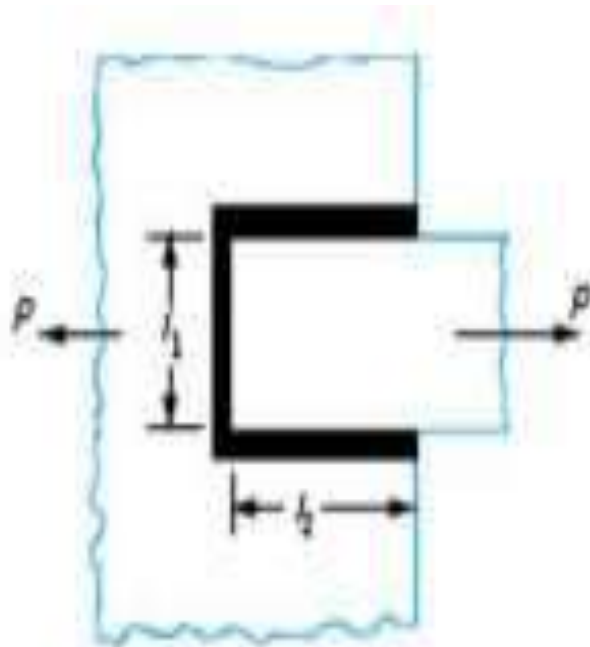
1- set parallel fillet or double parallel Fillet

1. Transverse i fillet



(a) Double parallel fillet weld.

WP1 A plate 75 mm wide and 12.5 mm thick is joined with another plate as shown in figure. The maximum tensile and shear stresses are 70 MPa and 56 MPa respectively.



(a) Combination of transverse and parallel fillet weld.

Data

Width plate = 75 mm

$H = 12.5$ weld size

$L_1 =$ Width of plate = 75 mm

$L_2 = ?$

Total load by the weld joint (Tensile Load)

Load by L1 by P1 single

$$P_1 = \text{Throat area} \times \text{Allowable shear stress} = 0.707 s \times l \times \tau$$

1

Load by L2 by P2 double

$$P_2 = \text{Throat area} \times \text{Allowable shear stress} = 0.707 s \times l \times \tau$$

2

Step1: Cal. Of total load

P= stress x area

Stress= 70 Mpa

Area= 75 x 12.5

P= 65 625 N

Step2: cal. Of Load P1

P1= 0.707 x h x l1 x σ
= 0.707 x h x 75 x 70/56

h= plate thickness= 12.5

P1= 46397 N

~~38 664 N~~

Step3: cal of L2

$$\text{Use } P_2 = 1.414 h \times L_2 \times \tau \quad (2 \text{ Single fillet} = 2 \times 0.707 h)$$

$$= 1.414 \times h \times 56$$

$$= 990 L_2 \text{ N}$$

$$\text{Now } P = P_1 + P_2$$

$$65\,625 = 46\,397 \text{ } ~~38\,664~~ + 990 L_2$$

$$L_2 = 19.43 \text{ } ~~27.2~~ + \text{weld size}$$

$$\text{Weld size} = 12.5 \quad L_2 = 39.7 \text{ } ~~31.93~~ \text{ mm}$$

Important pages to be used from DDB

11.3

11.4

11.5

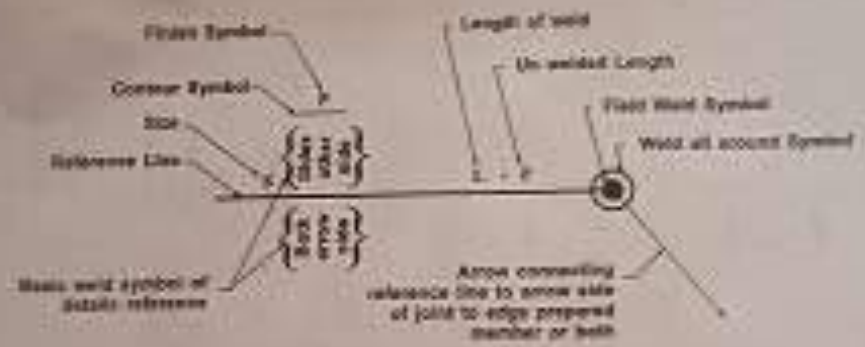
11.6

General information

11.1 & 11.2

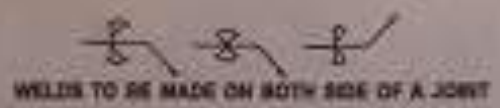
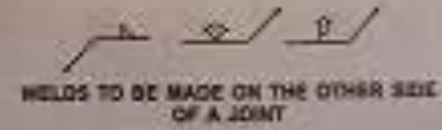
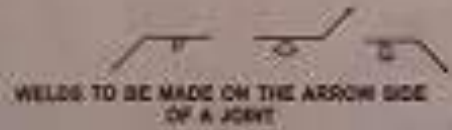
WELDED JOINTS
TYPE OF WELDS AND SYMBOLS

Form of weld	Sectional representation	Approximate symbol	Form of weld	Sectional representation	Approximate symbol
Flare			Flare joint		
Open butt			Butting strap		
Single V butt					
Double V butt			Butt		
Single U butt					
Double U butt			Matted butt		
Single bevel butt					
Double bevel butt			Matted butt		
Single J butt					
Double J butt			Projection		
Flare					
Flare (edge or wall)			Flare		
Butting strap					
Butting strap			Butt resistance or Pressure (Spot)		



STANDARD LOCATION OF ELEMENTS OF A WELDING SYMBOL

Special Instructions	Drawing Representation	Symbol
Weld all round		
File weld (Fillet weld)		
Flare contour		
Concave contour		
Convex contour		
Grinding finish		
Machining finish		
Dressing finish		



TECHNICAL - VIVO CAMERA

Shot on Y15
Vivo AI camera

 $\sigma = \frac{P}{A}$	 $\sigma = \frac{P}{(t_1 + t_2)l}$	 $\sigma = \frac{P}{A}$	 $\sigma = \frac{6M}{t^3}$	 $\tau = \frac{P}{tA}$
 $\sigma = \frac{6M}{t^3}$	 $\sigma = \frac{3M}{l^2(2t_1^2 - 6t_1t_2 + 4t_2^2)}$	 $\sigma = \frac{6M}{(t_1 + t_2)l^2}$	 $\sigma = \frac{3M}{l^2(2t^2 - 6t_1t_2 + 4t_2^2)}$	 $\tau = \frac{3PL}{l^2(2t^2 - 6t_1t_2 + 4t_2^2)}$
 $\sigma = \frac{0.707 P}{A}$	 Stress in weld A equals stress in weld B $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$	 $\sigma = \frac{0.707 P}{A}$	 $\sigma = \frac{0.707 P}{A}$	 Weld A $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$ Weld B $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$
 $\sigma = \frac{0.36 P}{A}$	 $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$	 $\sigma = \frac{0.707 P}{A}$	 Both plates same thickness $\sigma = \frac{0.707 P}{A}$	 Weld B $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$
 $\sigma = \frac{0.36 P}{A}$	 $\sigma = \frac{1.414 P}{(t_1 + t_2)l}$	 BUTT WELD (V) $\sigma = \frac{2.83 M}{h^2}$	 BUTT WELD (V) $\sigma = \frac{5.66 M}{h^2}$	 BUTT WELD (V) $\sigma = \frac{4.24 M}{h^2}$
 $\sigma = \frac{0.707 P}{A}$	 $\sigma = \frac{1.414 M}{h^2(D+t)}$	 Avg $\sigma = \frac{0.707 P}{A}$ Max $\sigma = \frac{1.414 P}{h^2(D+t)}$	 $\sigma = \frac{4.24 M}{h^2}$	 Avg $\sigma = \frac{0.707 P}{A}$ Max $\sigma = \frac{4.24 PL}{h^2}$
 $\sigma = \frac{6M}{h^2}$	 $\sigma = \frac{6PL}{h^2}$	 $\tau = \frac{M(3l + 1.5h)}{h^2}$	 $\sigma = \frac{3M}{h^2}$	 $\sigma = \frac{3PL}{h^2}$
 $\tau = \frac{3PL}{2(t_1 + t_2) \cdot P}$	 Butt Weld $\sigma = \frac{1.414 P}{2(t_1 + t_2) \cdot P}$	<p>σ = Normal Stress τ = Shear Stress M_x = Bending moment M_t = Twisting moment</p> <p>P = External Load L = Linear Distance h = Size of weld l = Length of weld</p>		

WELD STRESS FORMULAE

PLATE THICKNESS AND WELD SIZE

Plate thickness, mm	3 to 5	6 to 8	10 to 16	18 to 24	26 to 35	over 38
Weld size, mm	3	5	6	10	14	20

* Permissible static load per cm length : (Mild Steel Fillet welds)

Weld size mm	Permissible static load, kgf / cm			
	Bare Electrode		Covered Electrode	
	in tension	in shear	in tension	in shear
3	170	135	210	170
4	225	180	280	225
5	280	225	350	280
6	335	270	420	335
8	450	360	560	450
10	560	450	700	560
12	670	540	840	670
14	785	630	980	785
16	900	720	1120	900
20	1120	900	1400	1120

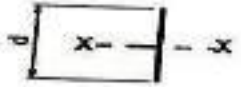
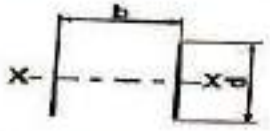

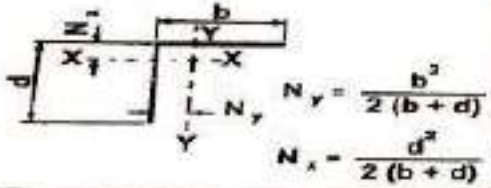
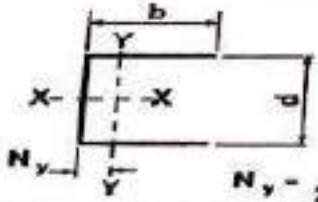

*For machine welding increase the permissible load by 25 to 30 %

DESIGN STRESSES FOR WELDED JOINTS, kgf / cm²

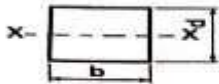
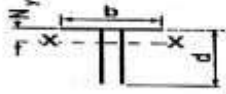
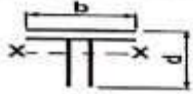
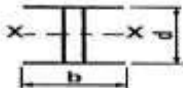
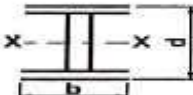

Kind of weld and stresses	Bare Electrodes		Covered Electrodes		Fatigue Design Stresses			
	Steady load	Reversed load	Steady load	Reversed load	Structural Steels		Alloy Steels	
					2 × 10 ⁴ cycles	10 ⁵ cycles	2 × 10 ⁴ cycles	10 ⁵ cycles
<i>Butt Welds</i>								
Tension	900	350	1100	550	$\frac{1100}{1 - 0.8 r}$	$\frac{1250}{1 - 0.5 r}$	$\frac{1150}{1 - 0.8 r}$	$\frac{2300}{1 - 0.6 r}$
Compression	1000	350	1250	550	$\frac{1250}{1 - 0.5 r}$	$\frac{1250}{1 - 0.5 r}$	-	-
Shear	550	210	700	350	$\frac{625}{1 - 0.5 r}$	$\frac{900}{1 - 0.5 r}$	-	-
<i>Fillet Welds</i>								
All	790	210	950	350	-	-	-	-

$$r = \frac{F_{min}}{F_{max}}$$

PROPERTIES OF WELD TREATED AS A LINE

Outline of welded joint	BENDING about horizontal axis XX, Z_w	TWISTING about centroidal axis, J_w
	$\frac{d^2}{6}$	$\frac{d^3}{12}$
	$\frac{d^2}{3}$	$\frac{d(3b^2 + d^2)}{6}$
	bd	$\frac{b^3 + 3bd^2}{6}$
 <p> $N_y = \frac{b^3}{2(b+d)}$ $N_x = \frac{d^3}{2(b+d)}$ </p>	$\frac{4bd + d^2}{6}$, top $\frac{d^2(4bd + d)}{6(2b + d)}$, bottom	$\frac{(b+d)^4 - 6b^2d^2}{12(b+d)}$
 <p> $N_y = \frac{b^3}{2b+d}$ </p>	$bd + \frac{d^2}{6}$	$\frac{(2b+d)^3}{12} - \frac{b^2(b+d)^2}{2b+d}$
 <p> $N_x = \frac{d^3}{3}$ </p>	$\frac{2bd + d^2}{3}$, top $\frac{d^2(2b+d)}{3(b+d)}$, bottom	$\frac{(b+2d)^3}{12} - \frac{d^2(b+d)^2}{b+2d}$

PROPERTIES OF WELD TREATED AS A LINE (contd...)

Outline of welded joint	BENDING about horizontal axis XX, Z_w	TWISTING about centroidal axis, J_w
	$bd + \frac{d^2}{3}$	$\frac{(b+d)^3}{6}$
 $N_y = \frac{d^2}{b+2d}$	$\frac{2bd + d^2}{3}, \text{ top}$ $\frac{d^2(2b+d)}{3(b+d)}, \text{ bottom}$	$\frac{(b+2d)^3}{12} - \frac{d^2(b+d)^2}{(b+2d)}$
 $N_y = \frac{d^2}{2(b+d)}$	$\frac{4bd + d^2}{3}, \text{ top}$ $\frac{4bd^2 + d^3}{6b + 3d}, \text{ bottom}$	$\frac{d^3(4b+d)}{6(b+d)} + \frac{b^3}{6}$
	$bd + \frac{d^2}{3}$	$\frac{b^2 + 3bd^2 + d^3}{6}$
	$\frac{2bd + d^2}{3}$	$\frac{2b^3 + 6bd^2 + d^3}{6}$
	$\frac{\pi d^2}{4}$ $\frac{\pi d^2}{2} + \pi D^2$	$\frac{\pi d^3}{4}$

Points to identify

- 1. How many fillets are there**
- 2. Static load**
- 3. Fatigue load --- K_t to be used to find the weld size.**
- 4. Eccentric loaded – weld bead are treated with line**
- 5. In pg 11.5 & 11.6 , the given formula to be “x “ by ‘t’**

Design Procedure

[Welding \rightarrow Eccentric Load]

Step 1: Find & locate "G" - Centre of Gravity.

Step 2: Introduce P_1 & P_2 at failure Point "A"

Step 3: Find τ_1 - direct shear stress $\tau_1 = ?$ $\tau_1 = \frac{P}{A}$

Step 4: Find $\tau_2 \rightarrow$ secondary shear stress

$$\tau_2 = ?$$

$$\tau_2 = \frac{P \cdot r_e \cdot r_2}{J}$$

$$r_2 = r_0$$

$$\cos \theta = \frac{r_1}{r_2}$$

Step 5: $\tau_2 = ?$

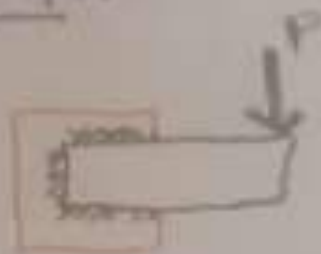
$$\tau_x = [\tau], \tau_x = \sqrt{\tau_1^2 + \tau_2^2} \quad \tau_x = \frac{P}{A} \sqrt{1 + \frac{r_2^2}{r_0^2}}$$



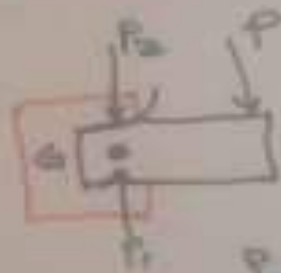
Eccentric Load Concept:

1. Find $G - CG$.

2. Introduce $P_1 + P_2$.



3.

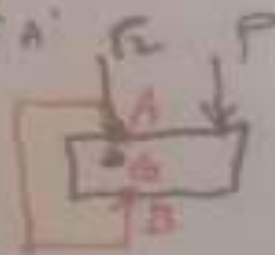


$$P_1 = P_2 = P$$

3. Locate the failure Pt: A

a - direct stress

$$\sigma = \frac{P}{A}$$



4. Secondary Shear stress τ_2

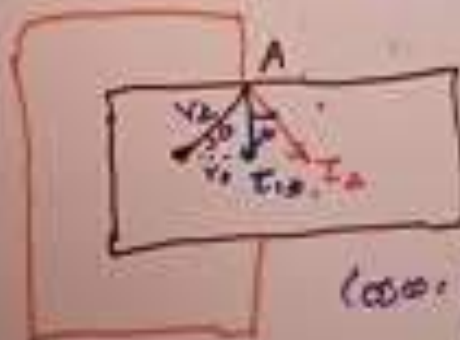


Secondary τ

which makes angle α (acute)

5.
$$\tau_2 = \frac{T \cdot r_2}{J}$$

J = Polar M.I.
 r_2 = Radius between G & A.



$\cos \alpha = \frac{r_2}{r}$
 $\alpha = \cos^{-1} \frac{r_2}{r}$

5. Resultant τ . [1/known] [cm]

$$\tau_R = \sqrt{(\tau_1)^2 + (\tau_2)^2 + 2\tau_1 \cdot \tau_2 \cdot \cos\theta}$$

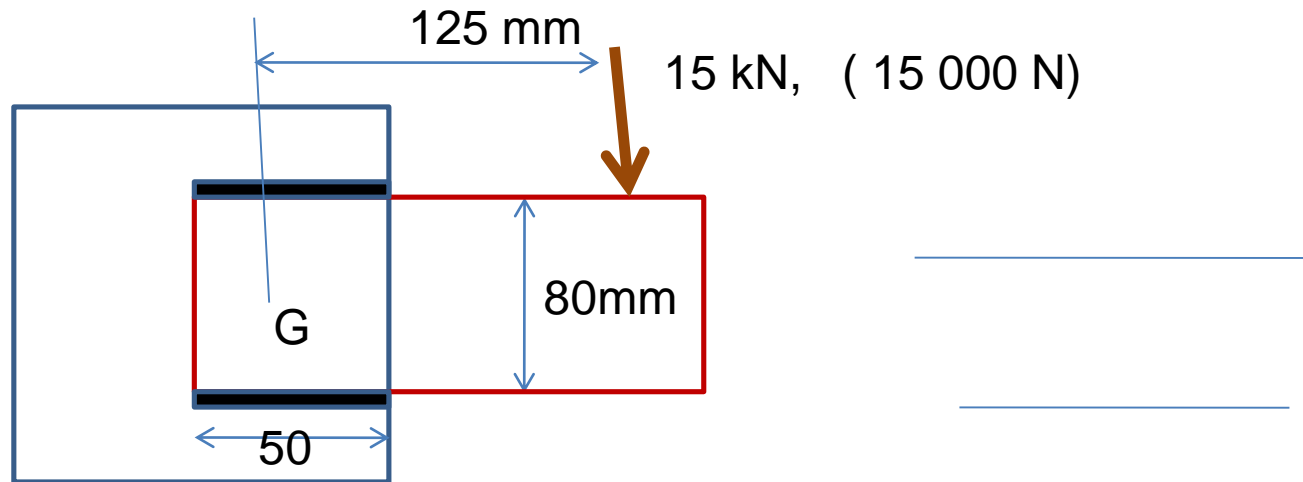
$$\cos\theta = \frac{\tau_1}{\tau_2}$$

find "h" of weld has

$$\tau_R = [\tau]$$

WP2.A bracket carrying a load of 15 kN is to be welded as shown in figure

Find the size of the weld required if the allowable shear stress is 80 Mpa.



Data :

Load = 15000 N

plate width= 80 mm

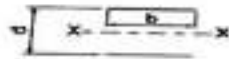
e- eccentric distance = 125 mm

$[\tau] = 80 \text{ Mpa} = 80 \text{ N/mm}^2$

DATA			
Load- P		15000 N	
[T]		80 N/mm ²	
width plate -d		80 mm	
eccentric. Dist.		125 mm	
length of weld- l		50 mm	
sin45		0.707	

Step1 Find and locate "G" of the given weld arrangement

Refer Pg 11.5/DDB



b= 50 mm
d= 80 mm

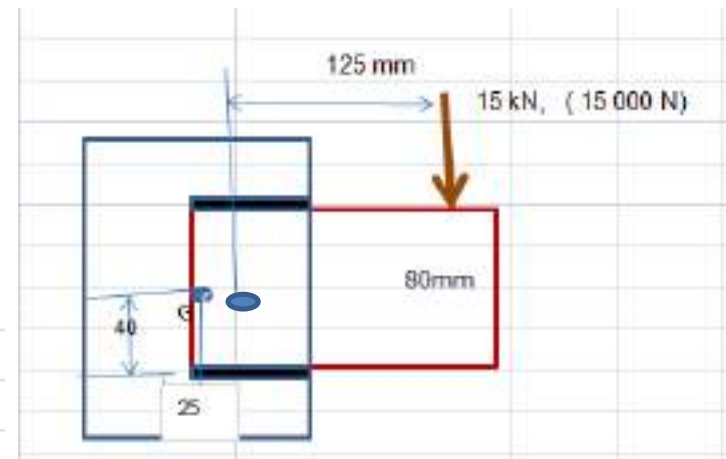
G=?

the structure is symmetrical about X-X

(x,y)

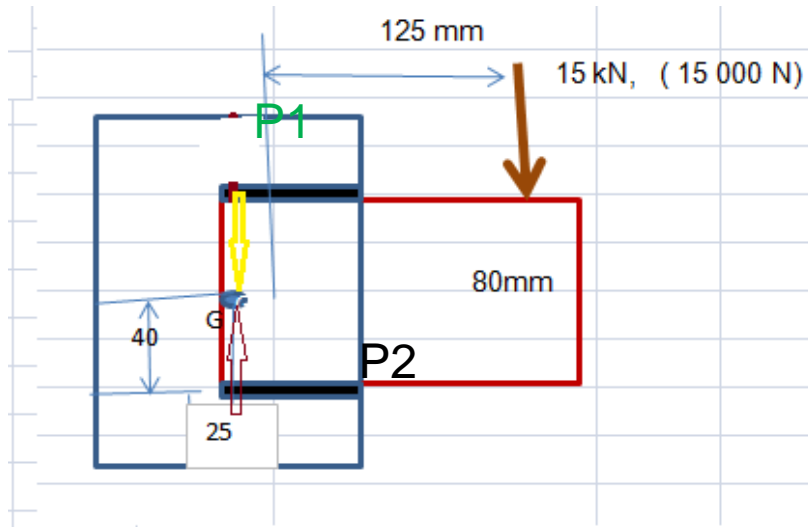
G(x,y)	x=	b/2	25
	y=	d/2	40

center of gravity G is (25 ,40)

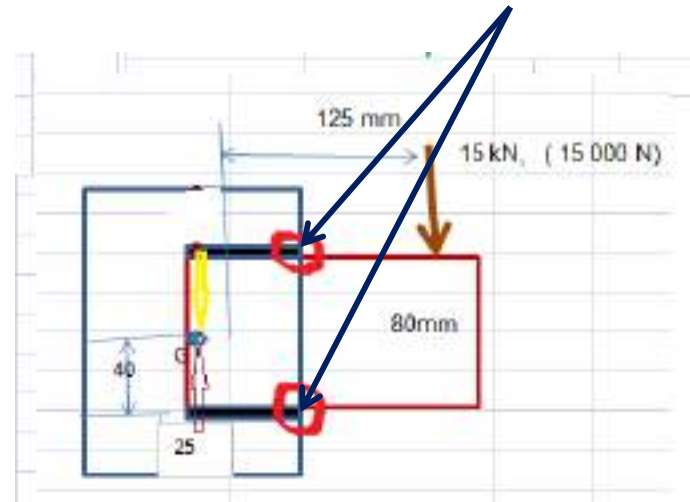


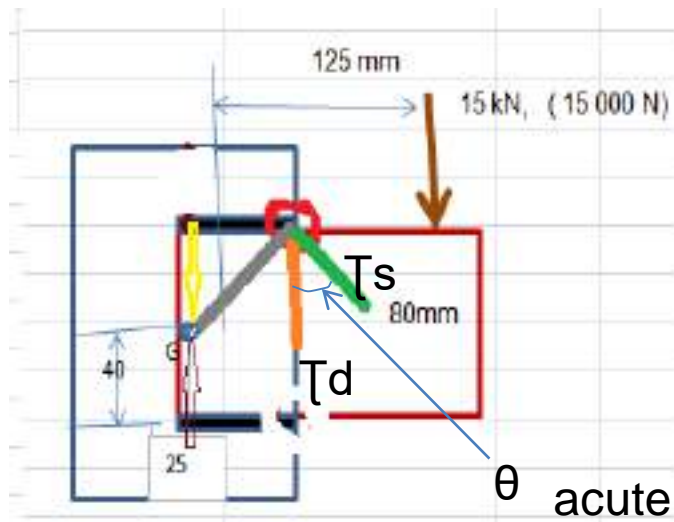
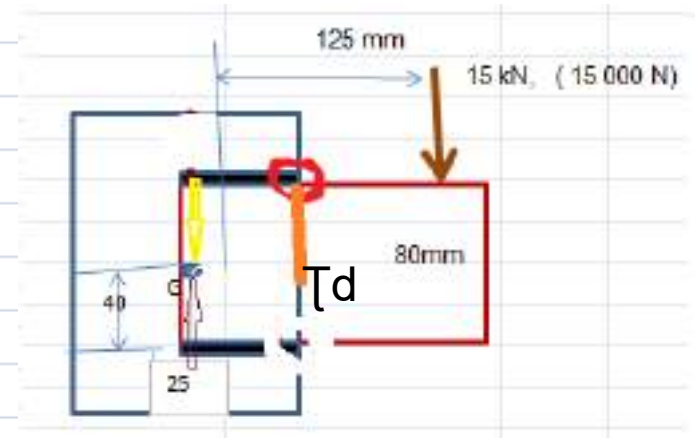
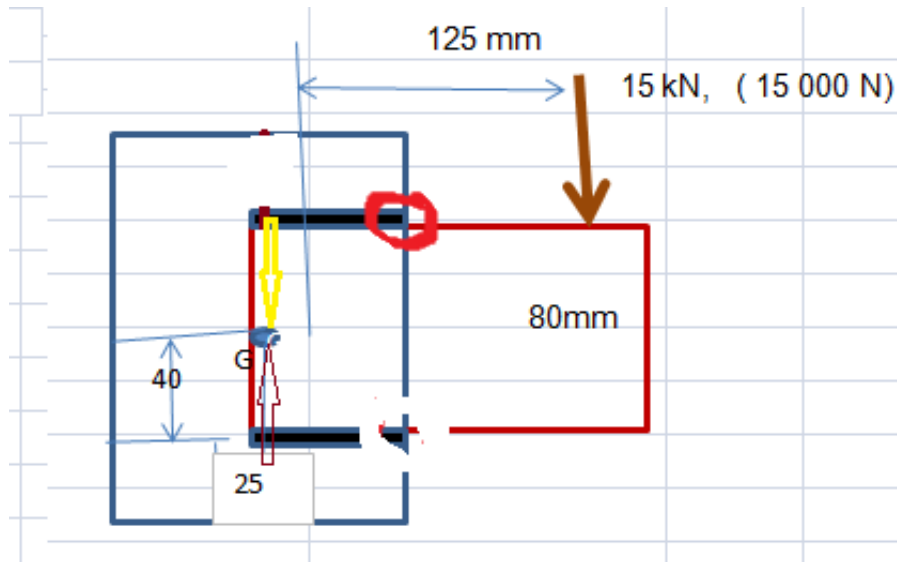
Step2: Introduce p1 & P2 and prepare the stresses induced

Two stresses will be **1. direct shear stress**
2. Secondary shear stress



Failure points





Angle between the two forces
Is acute, hence Resultant is given
Parallelogram law of forces

step3. Cal. Direct stress

stress = load / A

Load= 15000 N

A= lt for (single fillet)

A=2 x l x hx sin45 (for 2 fillet)

A= 70.7 h

$\tau_d = \text{load/Area}$

212.1640736 (1/h)

Step3: cal. Of secondary shearstress " τ_s "

Use of torsion theory refer pg 7.1 DDB

$$T/J = \tau_s/r$$

J= polar M.o.I

T= twisting moment = P x e

r= r2 from the digram

$$\tau_s = \frac{T \times r}{J}$$

From Pg. 11.5, J=?

$$\frac{b^3 + 3 b d^2}{6}$$

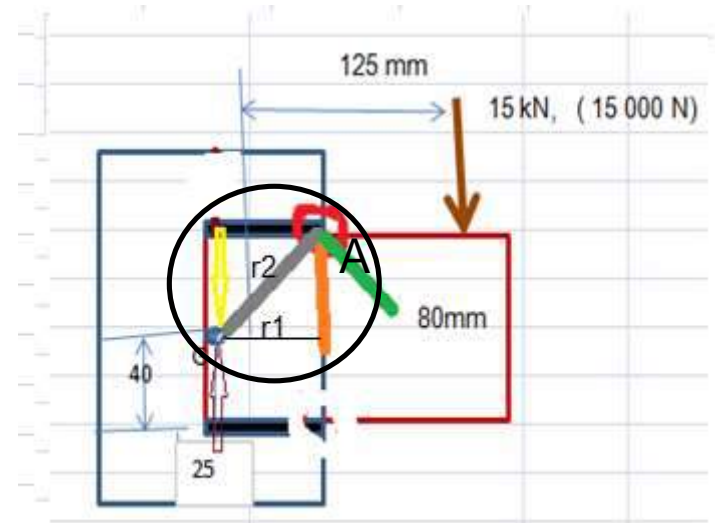
x t

$$t = 0.707h$$

$T =$	1875000 N-mm
$J =$	127849.1667 h
$r_2 =$	47.16990566 mm use of trigonometry-Phythagras theorem

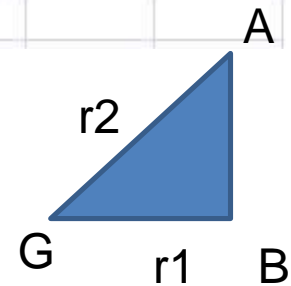
now

$$T_s = 691.8 (1/h)$$



$$r_1 = 25 = GB$$

$$AB = 40$$



$$r_2 = \sqrt{(AB)^2 + (GB)^2}$$

Step5 Cal.of " h" (τ_r by parallelogram law of forces)

$$\tau R = \sqrt{\tau d^2 + \tau s^2 + 2 \cdot \tau d \cdot \tau s \cdot \cos\theta}$$

$$\tau d^2 = 45014 \text{ h}^2$$

$$\tau s^2 = 5E+05 (1/h^2)$$

$$478587.24 * (1/h^2)$$

$$\cos\theta = r1/r2 = 0.532$$

where

$$r1 = 25$$

$$r2 = 47 \quad [\text{from triangle}]$$

$$2 \cdot \tau d \cdot \tau s \cdot \cos\theta = 2E+05 (1/h^2)$$

$$\tau R = 824.5 (1/h)$$

$$80 = 824.5 (1/h)$$

$$h = 10.31 \text{ mm}$$

$$11.299 \text{ mm}$$

Design procedure for closed system of welds

Step1 Find weld areas

Step2: Find direct shear stress

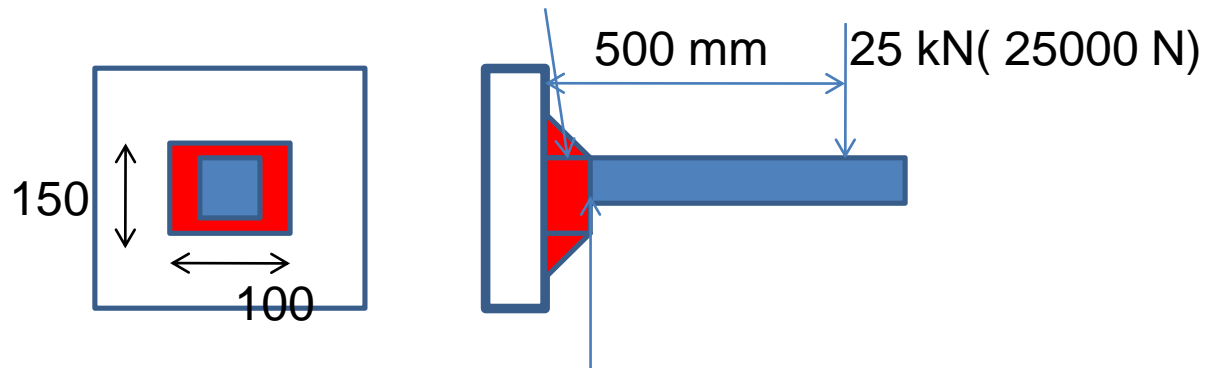
Step3: Find either bending or tensile stress according to the weld structure.

Step4: Find h

- 1. apply max.normal stress theory – (When tensile failure is available)**
- 2 Apply max shear stress theory- (when shear stress is available)**

wP3

A shaft of rectangular cross section is welded to a support by means of fillet welds as shown in figure below. Determine the size of weld, if the permissible stress in the weld is limited to 75 MPa



Data

$$P = 25\,000\text{ N}$$

$$e = 500\text{ mm}$$

$$[\tau] = 75\text{ N/mm}^2$$

$$h = ?$$

This is a closed system of weld

1. Max normal stress theory (case1)
2. max. Shear stress theory (case 2)

Since , shear stress only given work for “2” case

Step1 Total weld area

4 –sides are there
weld length $l = 2(b+d)$

$$A = t \times l$$

$$A = 0.707h \times l$$

$$A = 0.707 h \times [2 b + 2d]$$

$$\text{Area} = 353.5 h$$



$$d = 150$$

$$b = 100$$

Step2: find direct shear stress

$$\tau = P/A$$

$$= 25000/(353.5h)$$

$$= 70.72/h$$

Step 3 Find bending stress

$$\sigma_b = M/Z$$

$$M = p \times e$$

$$= 25\,000 \times 500$$

$$= 125 \times 10^5 \text{ N-mm}$$

Now to find Z Ref. Pg.11.6/DDB

$$Z = [bd + (d^2/3)] \times t$$

$$= [bd + (d^2/3)] \times 0.707 h$$

$$Z = 15907.5h$$

$$\sigma_b = 785.8/h$$

Step 4: find h

here , as shear stress value only given

choose, max shear stress theory

$$\tau_{\max} = [\tau] = \frac{1}{2} \sqrt{(\sigma_b)^2 + 4 \tau^2}$$

Take [tensile] = [120] N/mm²

$$[75] = \frac{1}{2} \sqrt{(785.8/h)^2 + 4 \cdot (70.72/h)^2}$$

$$h = 399.2/75$$

$$h = 5.32 \text{ mm}$$

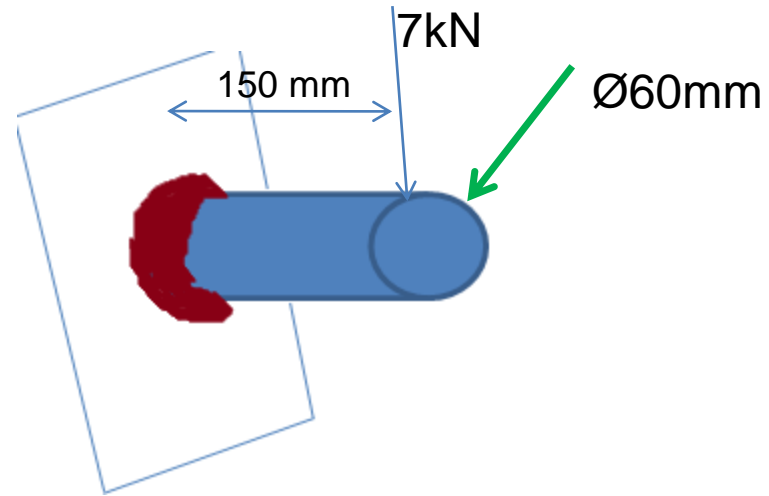
Closed system concept

1. Normal stress theory (if allowable tensile is given)
2. max shear stress theory (if allowable shear stress alone given)

Steps

1. Find direct stress
2. Find bending or shear stress
3. Find weld size “h”

Apply **max shear stress theory/ Normal stress theory**



WP4

A circular shaft of dia 60 mm is welded to a support by means of a fillet weld as shown in figure. Determine the size of the weld if the permissible shear stress is limited to 85 MPa

Step1

direct stress Shear = load/A

$$\begin{aligned} A &= \pi \times d \times t \\ &= 3.141 \times 0.707h \\ &= 133.27h \end{aligned}$$

$$P = 7000 \text{ N}$$

$$\tau_d = 52.52/h$$

Step2 bending stress

$$\sigma_b = M/Z$$

$$\begin{aligned} M &= 7000 \times 150 \\ &= 105 \times 10^4 \text{ N.mm} \end{aligned}$$

$$Z = \text{pg.no.11.6/DDB}$$

$$\begin{aligned} Z &= [(\pi \times d^4)/4] \times 0.707 h \\ &= 1999h \end{aligned}$$

$$\sigma_b = M/Z = 525.26/h$$

Step4 :

$H=3.15$ mm

Unit V

Rolling contact bearing- ball bearings takes both radial and axial load

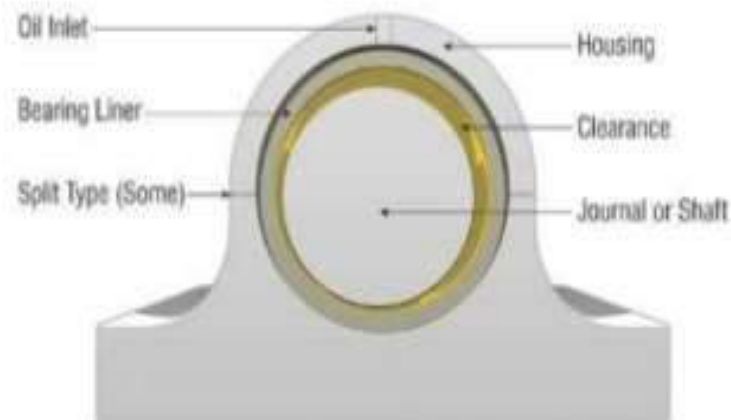
Sliding contact bearing - Journal bearings- sleeve and shaft – radial loads

Journal bearing

*** shaft and sleeve arrangement***

INTRODUCTION

In journal bearing sliding action is along the circumference of circle or an arc of circle and carrying radial loads. [1]



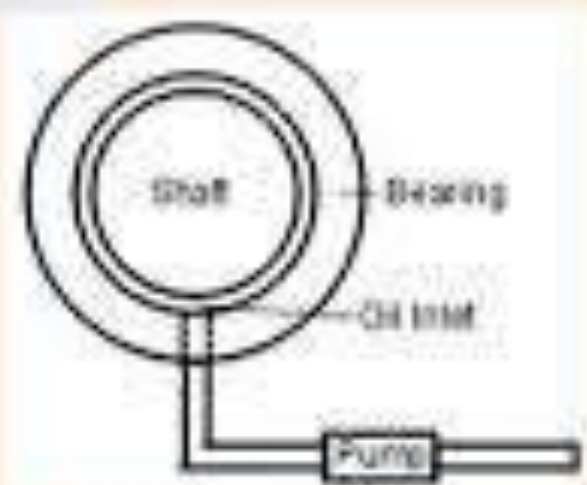
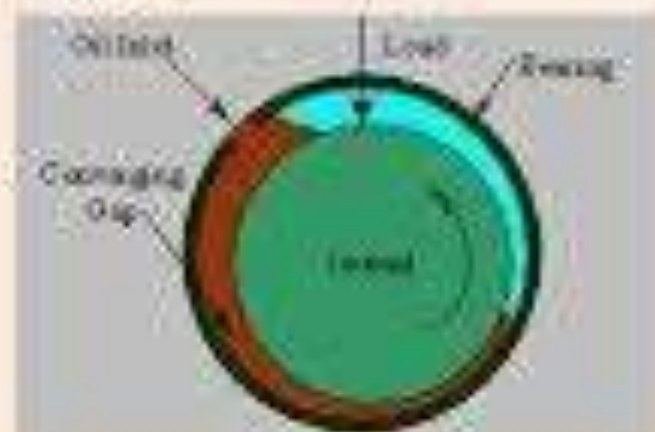
Common Journal Bearing Components

- Housing
- Bearing liner
- Segment (split type)
- Oil inlet
- Drain
- Journal

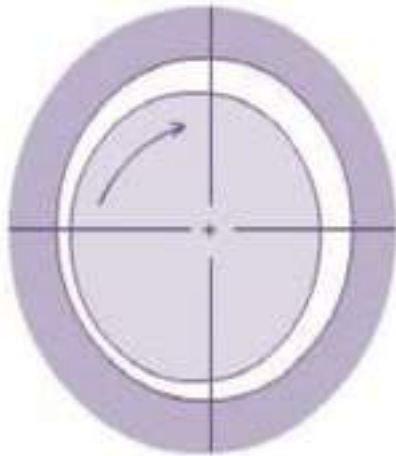
Figure 2. Plain Bearings (Journal Bearings)

Journal bearings

- Dry
- Hydrodynamic
- Hydrostatic
- Squeeze Film

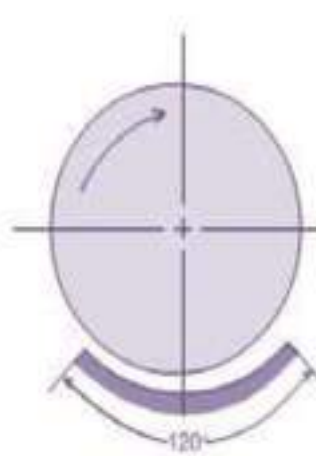


Bearing Classification



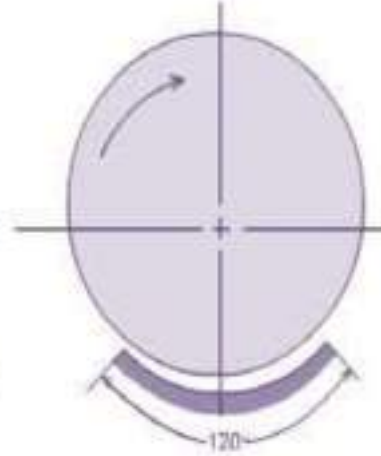
(a) Full

When the angle of contact of the bearing with the journal is 360° as shown in (a), then the bearing is called a full journal bearing.



(b) Partial

When the angle of contact of the bearing with the journal is 120° , as shown in Fig (b), then the bearing is said to be partial journal bearing.

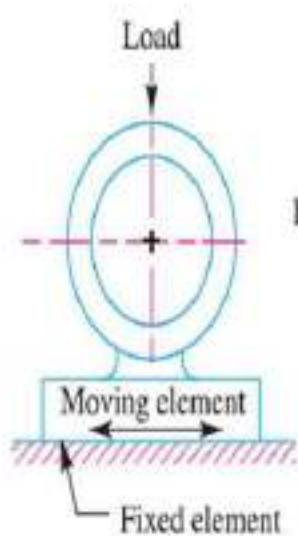


(c) Fitted

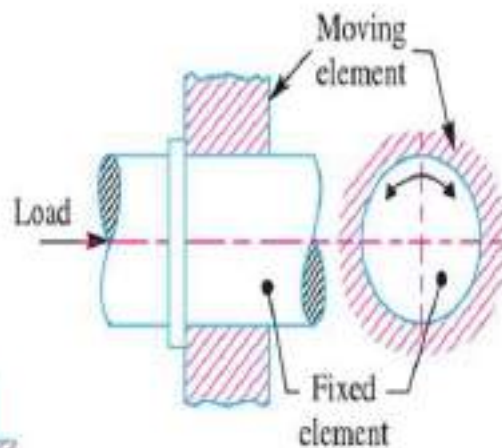
the diameters of the journal and bearing are equal, then the bearing is called a fitted bearing, as shown in Fig. (c).

Classification of Bearings

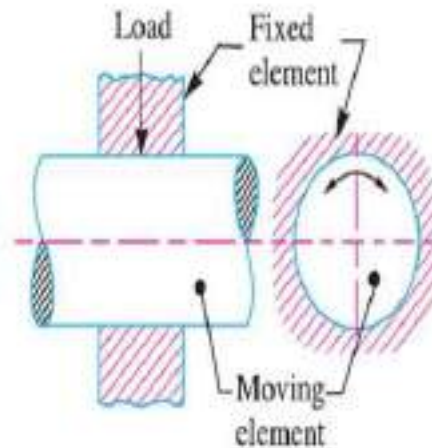
1. Depending upon the direction of load to be supported.
 - a) Radial bearings, and
 - (b) Thrust bearings.



(a) Radial bearing.



(b) Radial bearing.



(c) Thrust bearing.

Design procedure for sliding contact bearing- Journal bearing

Basic data needed

Load= W

Journal Diameter or Length= D or L

RPM = speed

Step1: find the application chose L/D ration ref.Page N0 7.31/DDB

Step2: Find L & D of the journal bearing & List the values/propertes

Step3: Check for bearing pressure < [pressure] Pg.731
 $P = W / (L \times D)$

Step4: selection of SAE oil refer. Pg.No.7.31 & 7.41

Step 5: Find coefficient of friction using Mc kees eqn. Pg.No.7.34

Step6: Find heat generation Hg? Pg. 7.34

Step 7 : Find Hd heat dissipation pg. 7.34

Step8: check for Artificial cooling required or not $H_g < H_d$ or $H_g > H_d$

DESIGN PRACTICES - JOURNAL BEARING

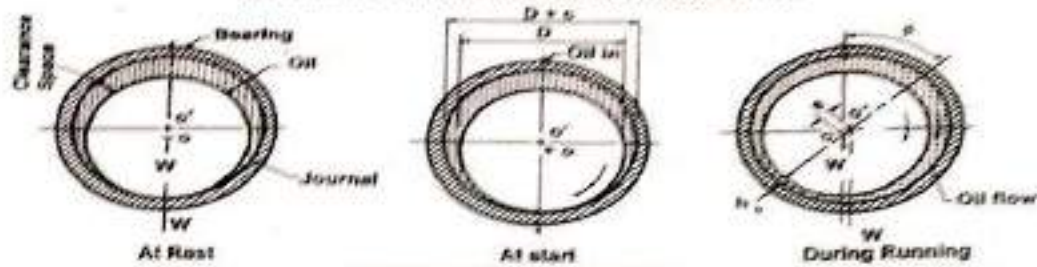
MACHINERY	BEARING	L / D	BEARING PRESSURES ALLOWABLE	LUBRICANT	
			kgf / cm ²	Z	Z _n / P min
Stationary High speed steam Engines	Main	1.5 - 3.0	17.50	15	355.6
	Crank pin	0.9 - 1.5	42.00	30	85.3
	Wrist pin	1.3 - 1.7	126.00	25	71.1
Gas and Oil Engines (Four Stroke)	Main	0.6 - 2.0	49 - 84	20 - 65	284.5
	Crank pin	0.6 - 1.5	108 - 126		142.2
	Wrist pin	1.5 - 2.0	125 - 154		71.1
Gas and Oil Engines (Two Stroke)	Main	0.6 - 2.0	35 - 125	20 - 65	355.6
	Crank pin	0.6 - 1.5	70 - 105		170.7
	Wrist pin	1.5 - 2.2	84 - 125		142.2
Aircraft & Automobile Engine	Main	0.8 - 1.8	56 - 119	8	213.3
	Crank pin	0.7 - 1.4	105 - 245		142.2
	Wrist pin	1.5 - 2.2	161 - 350		113.8
Reciprocating Compressors and Pumps	Main	1.0 - 2.2	17.5	30 - 80	426.7
	Crank pin	0.9 - 1.7	42		284.5
	Wrist pin	1.5 - 2.0	70		142.2
Centrifugal Pump, Motors and Generators	Rotor	1.0 - 2.0	7 - 14	25	2844.5
Machine Tools	Main	1.0 - 4.0	21	40	14.2
Steam Turbines	Main	1.0 - 2.0	7 - 20	2-16	1422.3
Railway Cars	Axle	1.9	35	100	711.2
Marine Steam Engines	Main	0.7 - 1.5	35	30	284.5
	Crank pin	0.7 - 1.2	42	40	213.3
	Wrist pin	1.2 - 1.7	105	30	142.2
Transmissions	Light, Fixed	2.0 - 3.0	1.8	25	1422.3
Gyroscopes	Rotor	-	60	30	782.3
Shafting	Self Aligning	2.5 - 4.0	11	60	426.7
	Heavy	2.0 - 3.0	11	60	426.7
Cotton Mills	Spindle	-	0.07	2	142231
Punching and Shearing Machines	Main	1.0 - 2.0	280	100	-
	Crank Pin	1.0 - 2.0	560	100	-
Rolling Mills	Main	1.0 - 1.3	210	50	142.2

Z, absolute viscosity, centipoises

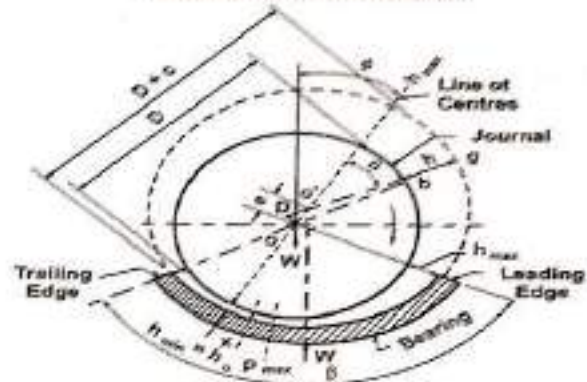
n, speed, rpm

P, pressure, kgf / cm²

HYDRODYNAMIC JOURNAL BEARINGS



MECHANISM OF LUBRICATION

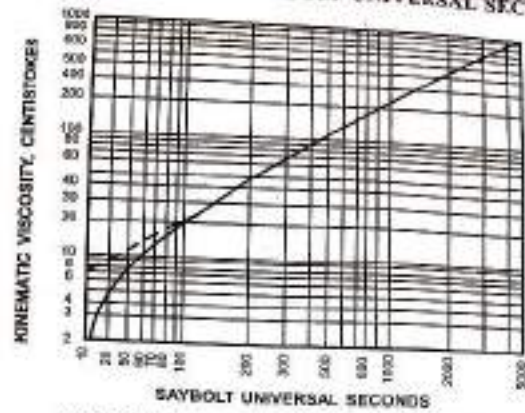


GEOMETRIC RELATION FOR ANY JOURNAL BEARING
(Shown here partial clearance bearing)

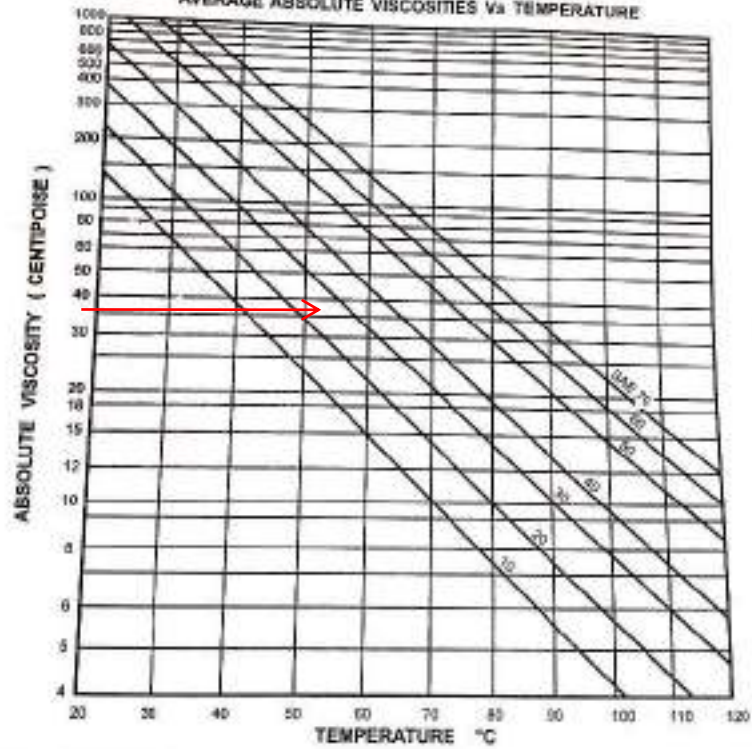
GEOMETRIC RELATIONS FOR A CLEARANCE BEARING

Clearance ratio, C/D	D journal diameter, cm
Eccentricity, $e = \frac{C}{2} - h_0$	C diametral clearance, cm
Eccentricity factor or Attitude,	$e = \frac{2e}{C} = 1 - \frac{2h_0}{C}$
Film thickness at any angle θ ,	$h = \frac{C}{2} (\epsilon \cos \theta + 1)$
Minimum film thickness, $h_0 = \frac{C}{2} (1 - \epsilon)$	h_0 minimum film thickness, cm
	ϵ attitude, dimensionless
	ϕ attitude, angle
	L Length of bearing, cm
	W Load, kgf
	P bearing pressure on projected area, kgf/cm ²
	e eccentricity, cm
	n speed of journal, rpm
	n' speed of journal, rps
	Z absolute viscosity of the oil, centipoises

KINEMATIC VISCOSITY - SAYBOLT UNIVERSAL SECONDS



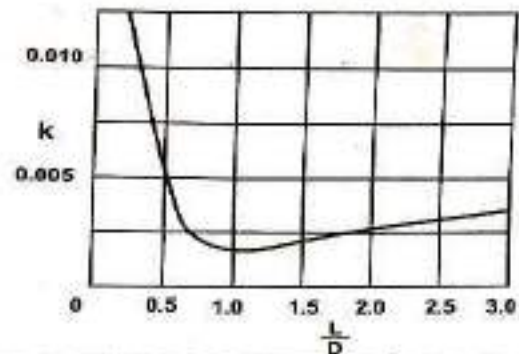
AVERAGE ABSOLUTE VISCOSITIES vs TEMPERATURE



Pg.7.41

DIMENSIONLESS PERFORMANCE PARAMETERS	
Coefficient of friction Variable, $\mu \frac{D}{C}$	
Flow variable, $4q / D C \rho' L$	
Flow ratio, q_2 / q	
Pressure ratio, P / p_{max}	
Temperature rise variable, $\frac{\rho C' \Delta t_o}{P}$	
FLOW THROUGH THE BEARING	
Axial flow in a 360° bearing, pressure fed, through a cylindrical hole at the centre	
$q_{sl} = \frac{C^3 p_1}{24 Z'} \left(\tan^{-1} \frac{\pi D}{L} \right) (1 + 1.5 \epsilon^2)$	
Flow is 2 to 3 times greater when the feed hole is located in a longitudinal groove	
With the feed hole in a central circumferential groove	
$q_{sl} = \frac{\pi D C^3 p_1}{24 Z' L} (1 + 1.5 \epsilon^2)$	
Energy increase of the oil = $q \rho C' \Delta t_o$	
AVERAGE TEMPERATURE RISE	
Oil ring bearings (still air), $\Delta t_s = 2 \Delta t$	
Oil bath bearings (still air), $\Delta t_s = 1.3 \Delta t$	
Waste packed bearings (still air), $\Delta t_s = 2.5 \Delta t$	
Design oil film temperature, 60 - 95 °C	
FRICTIONAL LOSS IN THE CAP	
Average film thickness in the cap	
$h_{av} = \frac{C}{2} (1 + 0.74 \epsilon^2)$	
$F = \frac{Z' A v}{0.6 h_{av}}$	
$U_f = F v$	
	<p>Δt increase in bearing surface temperature from ambient temperature, °C</p> <p>k constant for McKees equation</p> <p>v surface speed of Journal m/min</p> <p>K constant for heat dissipation, 437 for heavy construction, well ventilated. 775 for light construction in still air</p> <p>h_{av} average film thickness, cm</p> <p>h film thickness at any angle, θ, cm</p> <p>F frictional force on the cap, kgf</p> <p>U_f frictional loss in the cap, kgf m/min</p> <p>A sliding area of the cap, cm²</p> <p>Δt_s difference in temperature of oil and ambient temperature, °C</p>

Sommerfeld number, $S = \left(\frac{Z'n'}{P}\right) \left(\frac{D}{C}\right)^2$ $S = (2 + \epsilon^2) \sqrt{1 - \epsilon^2} / 12 \pi^2 \nu$ for $L/D = \infty$	ν Kinematic viscosity centistokes Z absolute viscosity, $\text{kgf sec} / \text{cm}^2 = \frac{Z}{9.81 \times 10^7}$ P hydro dynamic pressure developed at any angles θ , kgf / cm^2 P_1 inlet pressure, kgf / cm^2 P_{max} maximum pressure, kgf / cm^2 ρ density of the oil, kgf / cm^3 P_0 Pressure at $\theta = 0^\circ$
Coefficient of friction	
McKEES EQUATION $\mu = \frac{33.25}{10^{10}} \left(\frac{Z_0}{P}\right) \left(\frac{D}{C}\right) + k$ PETROFF'S EQUATION $\mu = 2 \pi^2 \left(\frac{Z'n'}{P}\right) \left(\frac{D}{C}\right)$	for lightly loaded bearings $S > 0.15$



$\mu = \frac{C}{D} \frac{(1 + 2\epsilon^2)}{3\epsilon}$ for medium or heavily loaded bearings $S < 0.15$	S Sommerfeld number, dimensionless μ coefficient of friction q oil flow through the bearing, cm^3 / sec q_a axial flow of oil, cm^3 / sec q_{s1} axial flow of oil due to inlet pressure cm^3 / sec C' specific heat of oil $17100 \text{ kgf} / \text{cm}^3 \text{ } ^\circ\text{C}$ Δt_o temperature increase of the oil $^\circ\text{C}$
Reynolds' equation for pressure distribution at any angle θ $P - P_0 = \frac{20 Z' \nu D}{C^2} \left[\frac{\epsilon (2 + \epsilon \cos \theta) \sin \theta}{(2 + \epsilon^2) (1 + \epsilon \cos \theta)^2} \right]$	
Load capacity $W = 20 Z' \nu L \left(\frac{D}{C}\right)^2 \left[\frac{\pi \epsilon}{(2 + \epsilon^2) \sqrt{1 - \epsilon^2}} \right]$	
Heat generated, $H_g = \mu W \nu$ $\text{kgf m} / \text{min}$	
Heat dissipated, $H_d = \frac{(\Delta t + 18)^2 L D}{K}$ $\text{kgf m} / \text{min}$	

JBP1

**Design a full journal bearing for a railway car, running at 600 rpm.
Diameter of the journal is 200 mm and load on the bearing is 230 KN.**

Data

W= load= 230 000 N

D= diameter= 200 mm

Speed= n= 600 rpm

Step1.find the application chose L/D ratio ref.Page N0 7.31/DDB

Application: **railway car**

From Pg.7.31

chose L/D ratio, **L/D=1.9**

[pb]= 35 kgf/cm² ---- 35 x 10 N/cm²----35 x 10/10² N/mm²

[Pb]= [3.5] N/mm²

Z= 100 centipoise

Zn/P= 711.2

Step2: Find L of the journal bearing

$$L/D=1.9$$

$$L= 1.9 \times D$$

$$L= 1.9 \times 200 = 380 \text{ mm}$$

Step3; Check for $P_{ind} < [P_b]$

$$P= w/(L \times D)$$

$$= 230 \times 10^3 / (380 \times 200)$$

$$= 3.026 \text{ N/mm}^2$$

$$3.026 < [3.5]$$

$P_{ind} < [P_b]$ design safe.

Step4: find SAE oil Pg. 7.41

Assume Operating temperature : 40° - 120° if not given in problem

Assume 70° C

Z= in cp?

from step 1,

$$Z_n/P = 711.2$$

$$Z = ? \quad 711.2 \times 30.26 / (600)$$

$$Z = 35.87 \text{ cP.}$$

Now use :

$$z = 35.87 \text{ cP, } t_o = 70^{\circ} \text{ C}$$

SAE50 oil selected

Refer. Pg.No:7.41/ddb from chart,

To=70° C and Z= 35.87 --- 45 cP

oil selected is **SAE 50**

Step5: find μ

Refer Pg.7.34/DDB

$$\mu = 33.25/10^{10} \times (zn/P) \times (D/C) + k$$

$$z=45/36 \text{ cP}$$

$$n= 600 \text{ rpm}$$

$$P_{ind}= 30.26 \text{ kgf/cm}^2$$

$$D/C= 1000 \text{ (in general clearance ratio } C/D= 1/1000)$$

To find k
Pg.no. 7.34/DDB
for L/D= 1.9,

$$k= 0.0025$$

$$\mu= \mathbf{0.00546}$$

$$= 0.0026 \text{ for } z=36\text{cP}$$

Step 5' find Hg?

$$Hg = \mu \times W \times v$$

$$\mu = 0.00546$$

$$W = 23000 \text{ kgf}$$

$$V = \pi \times d \times n / 1000$$

$$d = 200 \text{ mm}, \quad n = 600 \text{ rpm},$$

$$V = 376.99 \text{ m/min}$$

Now

$$Hg = 47342.5 \text{ kgf m/min}$$

Step6 Find Hd=?

Pg. No. 7.34/DDB

$$Hd = (\Delta t + 18)^2 \times L \times d / K$$

$2\Delta t = \Delta t_a$ Pg.No. 7.35/ddB

$$\Delta t = 1/2 \times \Delta t_a$$

$$= 1/2 \times (t_o - t_a)$$

$$= 1/2 \times (70 - 30)$$

$$= 20$$

$$L = 380 \text{ mm} = 38 \text{ cm}$$

$$D = 200 \text{ mm} = 20 \text{ cm}$$

$$k = 437 \text{ or } 775 \text{ (pg.no: 7.35/ddB)}$$

Assume $k = 437$ - heavy construction

Now **$Hd = 2511.3 \text{ kgf m/min}$**

Step7

$$H_g > H_d$$

Artificial cooling arrangement
is required

A3-3 Design a journal bearing for the following data

Diameter of journal=75 mm

Load on journal=3500N

Length of journal =75 mm

Speed=400 rpm

Minimum film thickness= 0.02 mm

Take operating temperature 60° C

Rolling Contact Bearing

Structure of ball bearing

Point of contact

Also called ‘antifriction bearing’

Loads

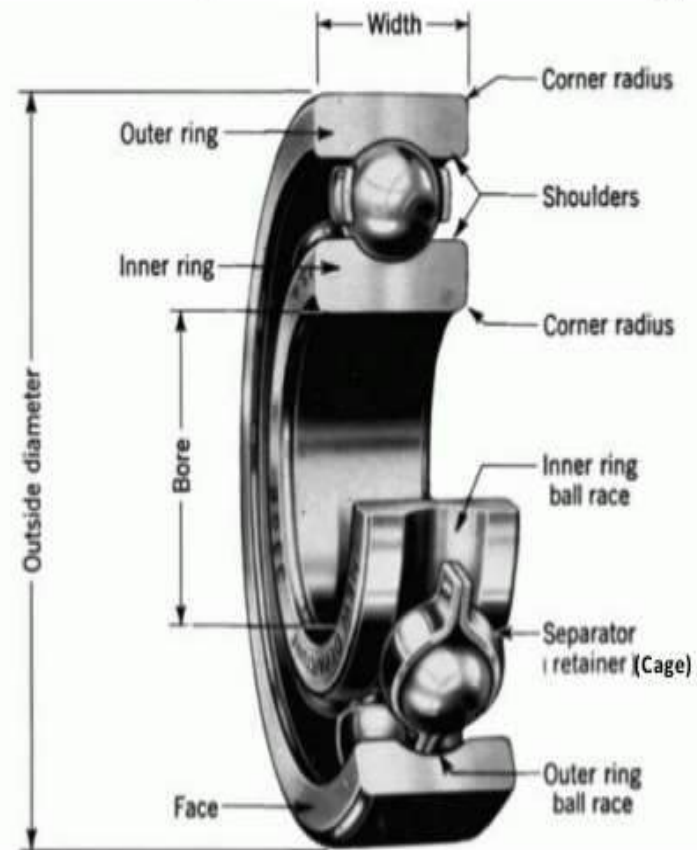
1. Axial
2. Radial load

Needs

Static capacity

Dynamic capacity

Rolling Contact Bearing



Advantages and Disadvantages of Rolling Contact Bearings Over Sliding Contact Bearings

Advantages

1. Low starting and running friction except at very high speeds.
2. Ability to withstand momentary shock loads.
3. Accuracy of shaft alignment.
4. Low cost of maintenance, as no lubrication is required while in service.
5. Small overall dimensions.
6. Reliability of service.
7. Easy to mount and erect.
8. Cleanliness.

Disadvantages

1. More noisy at very high speeds.
2. Low resistance to shock loading.
3. More initial cost.
4. Design of bearing housing complicated.

Types of Rolling Contact Bearings

Following are the two types of rolling contact bearings:

1. Ball bearings; and 2. Roller bearings.



Why need of rolling contact bearings

Available in many sizes and cross sections

Minimal lubricant supply needed

Low driving torque required

High load carrying capacity

Accurate positional capability

Wide temperature operating range (with solid lubricants)

Wide speed range capability

Many analytical programs available

DDB Pgs

4.1

4.2

4.4

4.8

4.9

Selection of bearings

4.12 to 4.36

DESIGN PROCEDURE

- 1 FIND $F_a/F_r=e$, select factors X & Y Pg.4.14/DDB
- 2 Find equivalent load P Pg.4.2/DDB
- 3 Find dynamic capacity “C” of Bearing Pg.4.2/DDB
- 4 Select suitable rolling contact bearing based on “C”
state bearing designation
5. Rated life of the bearing at 90%, 95% or given percentage